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RESEARCH MEMORANDUM

EFFECTS OF SPOILER AILERONS ON THE AERODYNAMIC LOAD

DISTRIBUTION OVER A 45° SWEEPBACK WING AT

MACH NUMBERS FROM 0.60 TO 1.03

By Joseph M. Hallissy, Jr., F. E. West, Jr.,
and George Liner

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**NATIONAL ADVISORY COMMITTEE
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WASHINGTON
May 14, 1954

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EFFECTS OF SPOILER AILERONS ON THE AERODYNAMIC LOAD

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SUMMARY

An investigation was conducted with 73-percent-semispan inboard spoiler ailerons, projecting 4 percent of the local chord from the wing surface, and located at the 70-percent-chord line of a 45° sweptback-wing-fuselage combination. The model consisted of a wing with an aspect ratio of 3.98, taper ratio of 0.61, and NACA 65A006 airfoil sections parallel to the plane of symmetry in combination with a fuselage of fineness ratio 10. Pressure data were measured on the wing and spoiler at several spanwise stations at Mach numbers from 0.60 (Reynolds number 5.1×10^6) to 1.03 (Reynolds number 6.2×10^6) for angles of attack that usually extended to 20° or more.

Upper-surface spoilers resulted in normal-force decrements which were largest in the 0.6 to 0.8 semispan area of the wing for low and moderate angles of attack. Most of this decrement was associated with increased upper-surface pressures ahead of the spoiler, but some of the decrement resulted from decreased lower-surface pressures. The addition of a gap through the wing behind the spoiler relieved the low pressure behind the spoiler on the upper surface and thus increased the rolling effectiveness. Lower-surface spoilers give reversed rolling-moment effectiveness at angles of attack higher than about 10° primarily because there is a large decrease in pressure behind the spoiler on the lower wing surface at the higher angles of attack.

Spoiler loads were highest at the inboard end. For upper-surface spoilers, the loads on the spoiler decreased rapidly with increasing angle of attack, but for lower-surface spoilers they increased to the highest test angles of attack.

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INTRODUCTION

Recently there has been considerable interest in spoiler ailerons, primarily because they maintain rolling effectiveness through the transonic speed range (ref. 1) and because they provide high reversal speeds for thin flexible wings. In addition, they can be designed to have very low hinge moments. However, very few pressure data for spoiler configurations have been available at high subsonic and transonic speeds (refs. 2 and 3) for load calculations or for studying the effects of spoilers on the flow about a wing.

Hence, a systematic test program has been initiated in the Langley 16-foot transonic tunnel to provide pressure data for various spoiler configurations in the transonic speed range at moderately high Reynolds numbers for a large angle-of-attack range. The initial investigation of this program was conducted on a 6-percent-thick 45° sweptback-wing-fuselage combination at 0° yaw and Mach numbers from 0.60 to 1.03. The spoilers investigated on this wing were of the retractable type, and they extended along the 70-percent-chord line from the fuselage (14 percent of the wing semispan) to 87 percent of the wing semispan and had a projection from the wing surface of 4 percent of the local wing chord.

This paper presents results for some of the configurations that were tested during the initial investigation. The effects of upper- and lower-surface spoilers and of a wing gap behind the spoilers are shown on the wing normal-force characteristics, chordwise pressure distributions, span-load distributions, and centers of load. Also included are tabulated wing static-pressure coefficients, spoiler pressure distributions, and spoiler span-load distributions. A limited amount of summary data from this investigation has already been published (ref. 4), and six-component force data obtained simultaneously with the pressure data are presented in reference 5.

SYMBOLS

b	wing span
c	section wing chord
\bar{c}	average wing chord
c'	wing mean aerodynamic chord

$c_m \frac{c^2}{c'}$ wing-section pitching-moment parameter with moment about the 25-percent position of the mean aerodynamic chord,

$$\frac{c^2}{c' c'} \int_0^1 (P_L - P_U) \left(\frac{x_1}{c} - \frac{x}{c} \right) d\frac{x}{c}$$

$c_n \frac{c}{c'}$ wing-section normal-load parameter,

$$\frac{c}{c'} \int_0^1 (P_L - P_U) d\frac{x}{c}$$

$c_{ns} \frac{h}{h}$ spoiler-section load parameter which acts parallel to plane of symmetry and perpendicular to h at a given spanwise station,

$$\frac{h}{h} \int_0^1 (P_F - P_R) d\frac{z}{h}$$

C_N wing-panel normal-force coefficient,

$$\int_{0.135}^{1.0} c_n \frac{c}{c'} d\frac{y}{b/2}$$

C_l rolling-moment coefficient

h length of chord of exposed front face of spoiler at any spanwise station

\bar{h} average h , $\frac{(h \text{ at } 0.14b/2) + (h \text{ at } 0.87b/2)}{2}$

M free-stream Mach number

P pressure coefficient, $\frac{P - P_0}{q}$

p local static pressure

p_0 free-stream static pressure

q free-stream dynamic pressure

x distance from wing leading edge at a given spanwise station, positive downstream

x_l distance from wing leading edge at a given spanwise station to line perpendicular to plane of symmetry and passing through 25-percent position of mean aerodynamic chord, positive downstream

$\frac{x_{cp}}{c}$ longitudinal location of wing-section center of pressure,

$$0.25 - \frac{\int_0^1 (P_L - P_U) \left(0.25 - \frac{x}{c}\right) d\frac{x}{c}}{c_n},$$
 fraction of section wing chord

$\frac{x_{cp}}{c'}$ longitudinal location of wing-panel center of pressure,

$$0.25 - \frac{\int_{0.135}^{1.0} \left(c_m \frac{c^2}{cc'}\right) d\frac{y}{b/2}}{C_N},$$
 fraction of mean aerodynamic chord

y spanwise distance from the plane of symmetry

$\frac{y_{cp}}{b/2}$ lateral location of wing-panel center of pressure,

$$\frac{\int_{0.135}^{1.0} \left(c_n \frac{c}{c'}\right) \left(\frac{y}{b/2}\right) d\frac{y}{b/2}}{C_N},$$
 fraction of semispan

z distance measured from wing surface along h at a given spanwise station (not perpendicular to x - and y -axes)

α angle of attack of fuselage center line relative to test-section center line

ΔP change in P across spoiler at a given spanwise station (P at $0.65c$ - P at $0.75c$)

Subscripts:

F forward surface of spoiler

L lower surface of wing

R rear surface of spoiler

U upper surface of wing

APPARATUS AND TESTS

Tunnel.- The investigation was conducted in the Langley 16-foot transonic tunnel, which is a single-return wind tunnel having a slotted throat of octagonal cross section. The maximum variation of average Mach number was about ± 0.002 along the test-section center line in the vicinity of the model. Additional details of the test-section configuration and of the calibration of the tunnel are given in reference 6.

Model.- Figures 1 and 2 show details of the basic model and the spoiler configurations included in this investigation. The steel wing had NACA 65A006 airfoil sections parallel to the plane of symmetry, quarter-chord line sweep of 45° , taper ratio of 0.61, and aspect ratio of 3.98. The wing was designed to have no incidence, dihedral, or twist and was mounted in a midwing position on the fuselage. The fuselage, constructed of steel, had a fineness ratio of 10, and the quarter chord of the wing mean aerodynamic chord was located at the longitudinal position of the maximum fuselage diameter.

The spoilers for these tests (fig. 1) simulated retractable spoiler-aileron configurations pivoted about the 50-percent-chord line. These spoilers were located along the 70-percent-chord line of the wing and were projected four percent of the local wing chord from the wing surface. They extended from the fuselage (14 percent of the wing semispan) to the 87-percent wing semispan and had a sweep angle of 41.6° . Spoilers were tested without and with a gap in the wing behind the spoiler. The gap, when used, extended outboard from the 15-percent to the 87-percent wing semispan station. The lower-surface spoiler with gap configuration was obtained by inverting the model with upper-surface spoiler and wing gap. The oppositely deflected spoiler configuration had one spoiler mounted on the upper surface of the left wing and one on the lower surface of the right wing with no gap behind the spoilers.

Figure 2 shows location of the wing static-pressure orifices which were distributed over the left wing at seven spanwise stations. The orifices at the inboard station (average $\frac{y}{b/2} = 0.135$) were actually located on the fuselage 0.1 inch from the wing surface. Pressure orifices were also located on the front and rear surfaces of the left wing spoiler and in the wing gap behind this spoiler at five spanwise stations as shown in figure 2. The pressures were transmitted by means of small tubing through the model support system to mercury manometer boards.

Model support system.- A cantilever strut, described in reference 7, supported the sting-mounted model. The model was near the tunnel center line at all angles of attack. A straight coupling between the sting and

the model permitted variations in the angle of attack from -4° to 15° ; a 10° coupling extended the range.

Tests.- Tests were generally made for all configurations at angles of attack of -2° to 26° for Mach numbers from 0.60 to 0.90. At Mach numbers from 0.94 to 1.03, the maximum angle of attack of these tests was limited by sting-support stresses or available tunnel power.

The Reynolds number variation over the Mach number range of the tests is shown in figure 3.

DATA REDUCTION

Data reduction methods.- Extensive use of a punched-card system greatly facilitated the reduction of data. Pressure data recorded with manometer board cameras were first transferred to cards by the use of a commercially available manual film reading device coupled to a card-punch machine. The data were then processed on electronic computing machines to obtain individual pressure coefficients as well as section normal-force and pitching-moment coefficients (using a rectangular step integration). The data cards were then fed to an automatic plotting device for the preparation of the chordwise pressure plots of this paper and were also used to prepare the tables of pressure data.

Corrections.- The angles of attack presented include an adjustment for an incremental angle determined from static calibration of the model angular deflection as a function of pitching moment and normal-force loads. Based on the repeatability of deflection measurements made during the static calibrations, the estimated maximum error of the angle-of-attack measurements is $\pm 0.1^{\circ}$. No corrections were made for tunnel-flow angularity. The cumulative effect of model asymmetry and tunnel flow angularity is shown to be small by the basic model normal-force curves in figure 5. No corrections were made for sting interference. Sting interference was not considered of importance for these tests because all lateral-control configuration changes were made on the wing, which was relatively remote from the sting. The data have not been corrected for tunnel boundary-interference effects since the results of reference 8 indicate that these effects would be small.

In calculating wing section and panel coefficients, the effect of the forces on the spoiler was neglected. The magnitude of the error thus introduced was checked and found to be within the accuracy of the data.

RESULTS AND DISCUSSION

Static-pressure measurements are given in coefficient form for the basic wing and the several spoiler configurations as follows:

Table	Static-pressure coefficients for -	Page
I	Basic wing	19 to 43
II	Wing with upper-surface spoiler (no gap)	44 to 63
III	Wing with upper-surface spoiler (with gap)	64 to 88
IV	Wing with lower-surface spoiler (with gap)	89 to 104

Each table shows the pressure coefficients at seven spanwise wing stations for various Mach numbers and angles of attack. Some of the high-angle-of-attack data for the upper-surface spoiler without gap and for the lower-surface spoiler with gap have been omitted since these configurations are of less interest because of loss of effectiveness or reversal at high angles of attack. No data have been tabulated for the oppositely deflected spoiler configuration, since, as will be shown, these were only slightly different from the data presented in table II.

In discussing the test results, some of the more important effects of spoiler operation on panel and section characteristics are first noted. Subsequent discussion makes use of chordwise pressure distributions to illustrate the manner in which the various spoiler configurations affect the air flow and the load distribution over the swept wing. Finally, the effect of a lower-surface spoiler on the opposite wing loading is considered and loads on the spoilers are discussed.

In figures which show comparisons at one angle of attack, the angle of attack given is an average for the compared configurations. This average does not differ more than $\pm 0.15^\circ$ from the extreme value for any of the compared configurations.

Effect of Spoiler on Wing-Panel Loading

Since the rolling-moment coefficients produced by the spoiler configurations are given in reference 5, there was no need to integrate the present pressure data for rolling moment. In order to be certain,

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however, that such integrations would indicate the same rolling moments as were measured with the strain-gage balance, integrations have been carried out for a few pressure-data points. A comparison with the balance measurements is made in figure 4. Even though the chordwise pressure force rolling-moment component (about the stability roll axis) has been neglected, as have crossover effects from the left wing to the right wing, agreement between the two sets of data is generally excellent. It is concluded, therefore, that an analysis of the pressure measurements made in these tests will help to provide a correct understanding of the functioning of a spoiler as a lateral control.

Figure 5 shows the variation of integrated wing normal-force coefficient with angle of attack at all test Mach numbers for the basic wing and three spoiler configurations. As in reference 5, the upper-surface spoilers caused decreases in normal-force coefficient which were largest in the region of 4° to 6° and which became small at the higher angles of attack. The presence of a gap through the wing behind the spoiler provided an additional decrement which was more or less constant throughout the angle-of-attack range. The beneficial effect of such a wing gap at transonic speeds has been previously shown. (For example, see ref. 9.) A lower-surface spoiler provided an increase in wing normal force at the lower angles of attack but a reversal was indicated for angles of attack above about 10° as was also shown in reference 5.

The increments in normal-force coefficient in figure 5 caused by operation of the spoiler were about twice the magnitude of the lift-coefficient increments shown on a similar figure in reference 5 for the same configurations. The reason for this effect was that the lift coefficient in reference 5 was based on total wing area, whereas the normal-force coefficient in this report was based on the semispan wing area.

Figures 6 and 7 show, respectively, the longitudinal and lateral locations of wing-panel center of pressure for the basic wing and three spoiler configurations at all test Mach numbers. These curves show discontinuities at some low angle of attack in both longitudinal and lateral center-of-pressure locations. These discontinuities indicate that, for an upper-surface spoiler, there was extensive positive loading inboard and negative loading outboard for the condition of zero panel lift.

Changes in center-of-pressure location caused by operation of the upper-surface spoiler were generally less for the spoiler with no gap than for the spoiler with a gap. At moderately high angles of attack, the center-of-pressure locations were affected only slightly by the spoilers, although the spoilers were still effective in reducing normal force at these angles. (See fig. 5.)

Effect of Spoiler on Wing Section Loadings

Figure 8 presents the semispan load distributions for the basic wing and the three spoiler configurations at all test Mach numbers. For the lower angles of attack, when the spoilers were most effective, the important loading changes caused by the spoiler occurred outboard of 0.3 semi-span, the largest decrements being in the region between 0.6 and 0.8 semi-span. This large influence over the outboard regions of the wing was the cause of the inboard position of the panel center of pressure for upper-surface spoilers (or outboard for lower-surface spoilers) at low positive load conditions. The loss in effectiveness for the upper-surface spoilers as the angle of attack was increased beyond 6° is evident as a reduction in the load decrement beginning near the tip.

Addition of a gap through the wing behind the spoiler produced an added decrement in section normal-load parameter which extended across most of the semispan.

Figure 9 shows the wing-section center-of-pressure locations across the span of the wing at all test Mach numbers. For a swept wing the line of section centers of pressure tends to be somewhat less swept than the wing itself as shown by the basic wing data in this figure. The effect of adding the upper-surface spoilers was to exaggerate this tendency, the local center of pressure being farther rearward inboard and farther forward outboard. This effect was most noticeable at low angles of attack ($\alpha = 4^\circ$ and 6°) but rapidly decreased at higher angles for two reasons: The spoiler decrements were smaller in absolute magnitude at high angles of attack and were a smaller proportionate amount of the total normal force.

The lower-surface spoiler at low angles of attack had the opposite effect, that is, the center of pressure was more forward inboard and more rearward outboard at low angles of attack ($\alpha = 4^\circ$) than it was for the basic wing.

Figure 9 also shows that the section center-of-pressure locations moved ahead of the leading edge at low angles of attack for the upper-surface-spoiler configurations. This movement occurred whenever the section normal force approached zero while a finite section moment remained.

Chordwise Pressure Distributions

In the discussion which follows, selected chordwise pressure distributions for the basic wing and the three spoiler configurations are used in conjunction with unpublished tuft photographs obtained on one spoiler configuration (upper-surface spoiler - no gap) to permit the study of the effect of the spoilers on the flow over the wing.

Upper-surface spoiler without a gap.- Figure 10 shows the chordwise pressure distributions obtained at seven spanwise stations at several Mach numbers. For a Mach number of 0.60 (fig. 10(a)) and low angles of attack, the presence of the spoiler resulted in increased pressures on the upper surface ahead of the spoiler. The highest pressures were immediately ahead of the spoiler, the peak being in the corner ahead of the spoiler-fuselage juncture. This peak was sharp, but elsewhere along the spoiler face a relatively flat peak about 0.1 chord in width indicated a separated flow region. Tuft studies confirmed this result and showed that the flow on the wing surface in this region was toward the tip and parallel to the spoiler.

Behind the spoiler on the upper surface the pressures (fig. 10(a)) and tuft studies also indicated a separated flow condition. Immediately adjacent to the fuselage, the separation extended only a few percent of the chord behind the spoiler and was followed by complete recovery of the flow. Since the expansion or turn made by the air over the top of the spoiler was very abrupt, the pressures reached at this point were quite low, being about the same as those reached at the leading edge at high angles of attack. Somewhat farther out on the wing (at 0.25 semispan), the extent of the separation was greater, recovery being made at about the trailing edge. Because of the increased extent of the separated region, the air turned down toward the surface less abruptly than at the more inward locations, and the pressures reached were not as low. Outboard of 0.25 semispan, complete separation of the flow behind the spoiler was indicated by the flat pressure distributions. The pressure in this region steadily increased toward the wing tip but was always less than for the basic wing.

Apparently the flow about the spoiler was somewhat similar to that described in reference 10 for a two-dimensional supersonic case and as illustrated in figure 11 but combined with the circulatory flow patterns in the boundary layer was the outboard movement of the boundary layer in the regions ahead and behind the spoiler. The pressure distributions were somewhat similar to the two-dimensional case, and the circulation in the separated areas also appeared to be present. Ahead of the spoiler this circulation was evidenced by the spoiler face pressure distributions to be discussed later, whereas behind the spoiler the direction of the tufts indicated that the air on the wing surface was moving toward the spoiler (in addition to moving outboard). Since this circulation was combined with a spanwise movement, the flow may be described as two vortex-type motions, one ahead of the spoiler and the other behind the spoiler.

The lower-surface pressures on the wing, as shown in figure 10(a), were not changed in the inboard areas because of the presence of a spoiler on the upper surface. At about 0.30 semispan, however, the separated area behind the spoiler on the upper surface reached the trailing edge and from

this point on the semispan to the tip the pressures on the lower surface were reduced because of the influence of the upper-surface flow. Outboard of 0.55 semispan the reduction in lower-surface pressure extended all the way to the leading edge.

As the angle of attack was increased, it was apparent that the effect of the spoiler on the wing did not change appreciably until angles of attack were reached where the flow separation on the basic wing began to progress inboard from the tip. (See fig. 10(a).) At these angles of attack, the spoiler effectiveness was reduced, as would be expected, since raising the spoiler into a separated flow region where the air is already moving parallel to the spoiler should not have any effect. (A discussion pertaining to the flow over the basic wing may be found in ref. 11.)

As the Mach number was increased from 0.6 at zero angle of attack (fig. 10(a)), there was no change apparent in the way which the spoiler affected the flow until about Mach number 1.0 (fig. 10(e)). Beginning at this speed the lower-surface influence was less extensive, being generally confined to the area behind the 0.70-chord line. At higher angles of attack the upper-surface influence ahead of the spoilers was also less extensive for the higher speeds. This effect was mainly noticeable over the inboard stations and was probably caused by the presence of a shock wave associated with the separation point ahead of the spoiler. The presence of this shock wave would have opposed the transmission of the pressure increase ahead of this point except outboard where the boundary layer was considerably thickened.

The effects of these pressure changes on the rolling moment were as follows: The pressures on the upper surface ahead of the spoiler and on all the lower surface changed in a direction to decrease the lift and thus contribute to the rolling moment, whereas the pressure decreases behind the spoiler on the upper surface were adverse. Inboard this adverse contribution was enough to completely offset the small favorable contribution which occurred at these stations. Thus, the span load distributions (fig. 8) show the largest normal-force decrements to have occurred outboard (between 0.6 and 0.8 semispan) and that the contribution of that part of the wing inboard of 0.3 semispan was generally unfavorable. These pressure changes also indicate the reasons for the discontinuities in the panel center-of-pressure locations shown in figures 6 and 7. Zero panel lift occurred at a small positive angle of attack and for each angle of attack the unaffected inboard sections were positively loaded, whereas the outboard sections, where the spoiler was most effective, were negatively loaded.

The ineffectiveness of the spoiler inboard of 0.3 semispan is not too important since the roll moment arm is small, but the large pressure difference across the spoiler near the fuselage is undesirable since it

contributes heavily to the drag. These adverse roll and drag effects inboard may indicate that an improvement could be obtained by removing the inboard part of the spoiler, but there is a likelihood that the region of adverse effects would then move outboard. In other words, the trouble may be associated with the inner end of the spoiler more than with the inboard area of the wing.

Effect of a gap through the wing behind an upper-surface spoiler.- Figure 12 compares chordwise pressure distributions of the basic wing with those of the spoiler-with-gap configuration throughout the Mach number and angle-of-attack range, and figure 13 compares the configurations with and without gap at a few points.

It appeared that the gap served to relieve the pressure difference between the upper- and lower-surface trailing-edge regions but its presence did not affect the extent of flow separation behind the spoiler. Evidently, the quantity of flow through the gap was too small to affect the extent of flow separation or the flow was not directed through the gap properly to decrease the extent of flow separation.

At all angles of attack and Mach numbers the gap was effective in increasing the upper-surface pressures behind the spoiler. The gap was also effective in increasing upper-surface pressures ahead of the spoiler, but this effect was very small at low angles of attack. Both of these effects increased the rolling effectiveness. On the lower surface, effects of the gap were limited to localized pressure changes.

Lower-surface spoiler with a gap.- The effect of a lower-surface spoiler ahead of a wing gap on the wing pressures is shown in figure 14. At the lower angles of attack, the lower-surface spoiler was equivalent in its effect on the flow to the upper-surface spoiler, the basic configuration being completely symmetrical about the chord plane. The appropriate part of the discussion on upper-surface spoilers therefore applies. As the angle of attack was increased, however, the lower-surface spoiler became ineffective and above about 10° produced losses in lift rather than increases. The changes leading to these losses in lift were as follows:

- (1) The region of increased pressure ahead of the spoiler on the lower surface became less extensive.
- (2) Behind the spoiler, the lower-surface pressures became much less than corresponding basic wing pressures and hence there was a large reduction in normal force over the trailing-edge region.
- (3) When the angle of attack was reached where separation existed on the upper surface, the influence of the lower-surface spoiler in reducing upper-surface pressures vanished.

Thus, the two favorable influences noted for upper-surface spoilers both became less for the lower-surface spoiler as the angle of attack was increased, whereas the unfavorable influence became greater. This increase with angle of attack of the unfavorable trailing-edge loading did not occur with the spoiler on the upper surface.

Oppositely Deflected Spoilers

A configuration was tested in which the left wing had an upper-surface spoiler and the right wing had a lower-surface spoiler. There was no gap through the wing behind either spoiler.

Comparisons showing the effect of the lower-surface spoiler on span-load distributions for the wing having the upper-surface spoiler are presented in figure 15 for three representative Mach numbers. Study of this figure and unpublished comparisons at other Mach numbers showed that the presence of the right-wing lower-surface spoiler reduced the effectiveness of the left-wing upper-surface spoiler at an angle of attack of 0° for Mach numbers above 0.90. For all other conditions the reverse was true, although the differences appeared to be small in every case.

At about 11° there were inconsistencies in the span loadings throughout the speed range, possibly because at this angle of attack the extent of the separated flow on the wing was not always consistent. Small differences in surface conditions on the wing could probably have varied the extent of separation at this angle of attack enough to account for the differences shown.

In order to determine the magnitude of the rolling moment represented by the differences shown in the span-load plots of figure 15, the differences were plotted to a larger scale and integrated for moment about the plane of symmetry. The rolling-moment-coefficient increments so obtained in general were less than ± 0.0016 , with no consistent Mach number effects discernible. Thus, the carry-over effects between wings were quite small, probably because of the inability of the spoiler to affect the inboard wing section loads to any great extent. This indication should lend credence to the use of reflection-plane test techniques for obtaining data on this type of control on swept wings.

Chordwise pressure differences have been compared at one speed only (fig. 16) to show the effect of adding the lower-surface spoiler to the opposite wing. The generally small differences were typical of all speeds.

Spoiler Loadings

Pressures at each spanwise station on the spoilers were integrated in such a manner as to obtain the pressure force coefficient normal to the spoiler chord line as indicated in the list of symbols. Figure 17 shows distributions across the wing span of these spoiler section normal-force coefficients for the three spoiler configurations weighted for spoiler height.

For upper-surface spoilers, the highest loads occurred at zero angle of attack. As the angle of attack was increased, the spoiler loads, especially near the tip, fell rather rapidly. At about an angle of attack of 11° , when flow separation existed over a considerable portion of the outboard wing area (see figs. 10 and 12), the spoilers were unloaded outboard of 0.40 semispan for all Mach numbers. This loss of spoiler load contrasted with the wing span-load distributions (fig. 8) which show that lift decrements caused by the spoilers extended much farther outboard at this angle of attack. Apparently, the inboard portions of the spoiler had an influence on the separated flow over the outboard wing areas. At higher angles of attack, where flow separation extended over more of the wing, this influence diminished.

Addition of a gap through the wing behind the spoiler resulted in a reduction of the loads on the spoiler for nearly all conditions. Figure 18, which makes a few comparisons of pressures on the upper-surface spoiler with and without gap, shows that this reduction resulted from increases in the low pressure on the back of the spoiler. These increases were probably due to the flow of high pressure air from the lower surface through the gap. The largest relief, as might be expected, was inboard where the pressures behind the spoiler were lowest.

Figure 18 shows that, at 0.25 semispan for angles of attack up to 16° and at other stations for low angles, the front face pressures on the spoiler were somewhat lower in the center of the spoiler than at the top and bottom. This type of pressure distribution is similar to that shown in reference 12 and probably indicates that there was a circulation of the separated air ahead of the spoiler as illustrated in figure 11, since such a circulation would result in higher velocities (and lower pressures) near the center than at the top or bottom.

For a lower-surface spoiler, figure 17 indicates that the loads continued to increase to the highest angles of attack of these tests. Figure 19, which compares the pressures on upper- and lower-surface spoilers, shows that the load on the lower-surface spoiler increased with angle of attack because the pressure coefficient on the front face of the spoiler remained at an approximately constant positive value through the angle-of-attack range, whereas the rear face pressure coefficient became more negative with increasing angle of attack.

Inasmuch as the spoiler pressure distributions were quite rectangular for most conditions, it would seem that a satisfactory measure of the spoiler loads could be obtained from a pair of orifices at each station located in the wing at the base of the spoiler, one orifice being immediately ahead of the spoiler and the other immediately behind. In order to test this supposition, the loads on the spoiler have been determined by this method at two speeds, Mach number 0.60 and 0.98, by using the closest available wing orifices, which were at 0.65- and 0.75-chord locations. The pressure at the 0.65-chord location was assumed to be the same as that on the front face of the spoiler and the pressure at 0.75-chord location was assumed to be the same as that on the rear face. The results are shown by the symbols in figures 17(a) and 17(f). Agreement was good for most conditions but would probably have been better if orifices closer to the spoiler had been available. This is especially true at the inboard stations where the extent of the separated flow ahead and behind the spoiler was small. It is therefore believed that a pair of properly located orifices on the wing at each station would give loads on this type spoiler as good as those obtained in the present tests with seven orifices on the spoiler at each station.

CONCLUSIONS

An investigation was conducted with 73-percent-semispan inboard-spoiler ailerons having heights of 4 percent of the local chord and located on the 70-percent-chord line of a 45° sweptback-wing—fuselage combination. Pressure data were measured on the wing and spoiler at several spanwise stations at Mach numbers from 0.60 (Reynolds number 5.1×10^6) to 1.03 (Reynolds number 6.2×10^6) for angles of attack that usually extended to 20° or more. The results of the investigation indicate the following conclusions:

(1) Operation of upper-surface spoilers at low and moderate angles of attack produces normal-force decrements which are largest in the 0.6 to 0.8 semispan area of the wing. Most of the normal-force decrement is associated with increases in the upper-surface pressures ahead of the spoiler due to a deceleration of the air approaching the spoiler. An additional contributing factor is a decrease in lower-surface pressures resulting from transmission of upper-surface pressure changes around the trailing edge.

(2) Rolling-moment effectiveness is reduced for upper-surface spoilers at high angles of attack because they do not have much effect on the separated flow which occurs on the basic swept wing at these angles of attack.

(3) A gap through the wing behind the spoiler is effective in increasing the rolling-moment effectiveness because it permits a relief of the pressure difference between the upper- and lower-wing surfaces.

(4) Lower-surface spoilers give reversed rolling-moment effectiveness at angles of attack higher than about 10° primarily because there is a large decrease in pressure behind the spoiler on the lower-wing surface at the higher angles of attack.

(5) A lower-surface spoiler has only a small effect on the rolling-moment effectiveness of a spoiler located on the upper surface of the opposite wing.

(6) Spoiler section loadings are highest at the inboard end. For upper-surface spoilers, the spoiler loading decreases rapidly with increasing angle of attack, especially outboard. For a lower-surface spoiler, however, the spoiler loading increases with angle of attack.

(7) Spoiler section load at any point along a spoiler of this type can be determined by measurement at the wing surface of the pressure drop across the spoiler.

Langley Aeronautical Laboratory,
National Advisory Committee for Aeronautics,
Langley Field, Va., March 8, 1954.

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TABLE 1

BASIC WING

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:							
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
		M = 0.60 α = 0.0°								
UPPER SURFACE		.00	.019	.485	.442	.684	.449	.467	.443	
		1.25	.137	.180	.144	.093	.044	.091	.088	
		2.50	.063	.097	.180	.098	.103	.138	.071	
		5.00	.002	.093	.112	.098	.069	.071	.096	
		7.50	.028	.105	.094	.091	.102	.090	.111	
		10.00	.038	.116	.189	.107	.110	.101	.123	
		15.00	.069	.124	.138	.125	.114	.124	.104	
		20.00	.081	.132	.146	.134	.126	.125	.122	
		25.00	.101	.146	.156	.139	.127	.126	.112	
		30.00	.119	.154	.164	.147	.133	.134	.112	
		35.00	.118	.159	.169	.150	.138	.141	.114	
		40.00	.130	.169	.169	.154	.140	.138	.106	
		45.00	.145	.170	.169	.155	.136	.137	.103	
		50.00	.159	.174	.169	.150	.135	.130	.095	
		55.00	.137	.169	.155	.149	.128	.125	.083	
LOWER SURFACE		60.00	.149	.164	.145	.100	.115	.104	.059	
		65.00	.157	.146	.180	.099	.099	.082	.048	
		70.00	.117						.040	
		75.00	.106	.115	.085	.074	.061	.046	.017	
		80.00	.139	.083	.085	.044	.038	.017	.010	
		85.00	.114	.054	.005	.020	.000	.011	.037	
		90.00	.099	.030	.002	.009	.022	.039	.052	
		95.00	.064	.005	.032	.017	.048	.066	.070	
		1.25	.134	.092	.125	.134	.124		.140	
		2.50	.069	.069	.098	.098	.091	.074	.109	
		5.00	.021	.071	.098	.047	.101	.107	.109	
		7.50	.001	.087	.106	.080	.114	.114	.137	
		10.00	.028	.099	.115	.106	.116	.103	.136	
		15.00	.053	.121	.128	.129	.122	.111	.140	
		20.00	.090	.130	.144	.139	.132	.128	.136	
	25.00	.065	.142	.150	.144	.142	.137	.118		
UPPER SURFACE		30.00	.117	.153	.154	.143	.144	.138	.118	
		35.00	.124	.163	.147	.147	.143	.139	.115	
		40.00	.126	.162	.163	.147	.146	.136	.111	
		45.00	.141	.170	.163	.148	.147	.139	.108	
		50.00	.163	.178	.163	.148	.137	.131	.099	
		55.00	.159	.170	.157	.143	.128	.126	.092	
		60.00	.158	.155	.136	.129	.110	.110	.073	
		65.00	.161	.138	.129	.113	.086	.084	.050	
		70.00	.143						.039	
		75.00	.131	.108	.084	.069	.048	.042	.029	
		80.00	.103	.078	.057	.051	.031	.015	.011	
		85.00	.123	.058	.032	.010	.008	.022	.035	
		90.00	.096	.031	.007	.008	.014	.020	.031	
		95.00	.102	.003	.032	.045	.044	.050	.057	
			M = 0.60 α = 4.0°							
UPPER SURFACE		.00	.013	.241	.471	.349	.905	.969	.673	
		1.25	.170	1.092	1.322	1.238	1.471	1.344	1.376	
		2.50	.243	.588	.718	.766	.844	.901	.679	
		5.00	.275	.465	.546	.561	.541	.555	.512	
		7.50	.277	.400	.458	.479	.493	.472	.440	
		10.00	.264	.373	.429	.448	.446	.424	.374	
		15.00	.266	.341	.394	.393	.370	.370	.275	
		20.00	.261	.322	.355	.356	.343	.319	.251	
		25.00	.264	.312	.338	.331	.315	.287	.219	
		30.00	.270	.307	.326	.317	.296	.268	.200	
		35.00	.256	.300	.310	.302	.281	.243	.191	
		40.00	.260	.297	.301	.286	.261	.222	.172	
		45.00	.264	.291	.285	.265	.244	.201	.161	
		50.00	.276	.283	.270	.256	.234	.188	.147	
		55.00	.239	.273	.244	.241	.205	.166	.135	
LOWER SURFACE		60.00	.247	.252	.217	.175	.177	.135	.101	
		65.00	.251	.224	.168	.172	.149	.105	.085	
		70.00	.198						.075	
		75.00	.198	.171	.127	.114	.086	.051	.047	
		80.00	.205	.131	.093	.077	.057	.022	.027	
		85.00	.178	.094	.066	.047	.019	.007	.002	
		90.00	.142	.056	.025	.007	.013	.040	.013	
		95.00	.093	.012	.016	.008	.039	.067	.030	
		1.25	.308	.357	.377	.362	.413		.376	
		2.50	.272	.280	.297	.293	.331	.353	.287	
		5.00	.217	.201	.204	.238	.240	.244	.180	
		7.50	.189	.149	.163	.194	.183	.187	.101	
		10.00	.164	.116	.127	.158	.143	.165	.054	
		15.00	.120	.069	.084	.087	.103	.092	.005	
		20.00	.071	.037	.041	.050	.061	.059	.041	
	25.00	.075	.004	.012	.019	.033	.027	.036		
	30.00	.086	.021	.012	.012	.009	.003	.071		
	35.00	.083	.020	.013	.017	.006	.026	.075		
	40.00	.087	.024	.015	.021	.019	.041	.080		
	45.00	.087	.027	.018	.025	.035	.055	.083		
	50.00	.052	.083	.070	.051	.040	.064	.082		
	55.00	.061	.081	.069	.052	.045	.065	.082		
	60.00	.066	.077	.068	.052	.039	.056	.073		
	65.00	.075	.075	.068	.044	.022	.043	.055		
	70.00	.068						.049		
	75.00	.067	.063	.039	.028	.007	.014	.043		
	80.00	.047	.035	.023	.007	.005	.002	.035		
	85.00	.069	.027	.000	.007	.017	.028	.003		
	90.00	.055	.016	.012	.016	.033	.027	.003		
	95.00	.054	.006	.038	.041	.057	.056	.017		

TABLE I

BASIC WING

	PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M = 0.60 α = 6.0°							
	1.00	.019	.0984	.1088	.0764	.1570	.1268	.1403
	1.25	.348	.613	.834	.511	.396	.1055	.1440
	2.50	.413	.648	.867	.438	.320	.0992	.1358
	5.00	.418	.698	.908	.367	.283	.0863	.1242
	7.50	.401	.688	.875	.371	.283	.0863	.1207
	10.00	.373	.538	.686	.363	.278	.0707	.1062
	15.00	.359	.460	.570	.332	.257	.0585	.0927
	20.00	.341	.420	.490	.286	.203	.0496	.0829
	25.00	.335	.398	.440	.280	.193	.0417	.0738
	30.00	.331	.374	.411	.288	.199	.0369	.0623
	35.00	.313	.360	.384	.270	.180	.0324	.0511
	40.00	.312	.351	.364	.243	.155	.0285	.0419
	45.00	.311	.337	.338	.233	.140	.0256	.0366
	50.00	.319	.323	.312	.227	.130	.0220	.0318
	55.00	.277	.304	.279	.208	.117	.0192	.0273
	60.00	.283	.281	.250	.208	.113	.0160	.0237
	65.00	.282	.249	.212	.199	.100	.0131	.0217
LOWER SURFACE	1.00	.222	.187	.141	.137	.116	.077	.098
	1.25	.222	.142	.105	.099	.094	.049	.081
	2.50	.190	.099	.07	.069	.062	.026	.066
	5.00	.148	.059	.028	.034	.025	.003	.055
	7.50	.088	.009	.010	.016	.001	.024	.044
	1.25	.366	.459	.454	.447	.453		.424
	2.50	.336	.391	.396	.396	.408	.410	.361
	5.00	.310	.309	.312	.337	.330	.319	.286
	7.50	.282	.254	.239	.278	.270	.262	.249
	10.00	.255	.211	.220	.233	.225	.235	.2108
	15.00	.214	.160	.169	.171	.174	.163	.1532
	20.00	.163	.123	.117	.125	.121	.109	.1008
	25.00	.154	.087	.092	.095	.091	.071	.0641
	30.00	.101	.061	.060	.070	.062	.037	.063
	35.00	.075	.034	.036	.036	.032	.008	.070
	40.00	.064	.019	.017	.017	.018	.014	.079
	45.00	.035	.004	.002	.006	.002	.031	.083
	50.00	.011	.023	.015	.005	.012	.044	.085
55.00	.004	.026	.017	.009	.022	.052	.088	
60.00	.013	.027	.013	.014	.022	.053	.080	
65.00	.016	.028	.014	.014	.013	.043	.066	
70.00	.017						.059	
75.00	.025	.024	.008	.001	.000	.026	.087	
80.00	.004	.003	.002	.011	.000	.013	.044	
85.00	.025	.001	.017	.021	.007	.016	.027	
90.00	.021	.009	.023	.022	.025	.005	.018	
95.00	.017	.021	.048	.041	.036	.023	.005	
UPPER SURFACE	M = 0.60 α = 8.0°							
	1.00	.003	.1274	.1541	.0768	.1252	.1157	.1483
	1.25	.600	.213	.282	.042	.053	.036	.172
	2.50	.653	.134	.227	.006	.004	.007	.195
	5.00	.624	.030	.137	.006	.005	.002	.082
	7.50	.594	.005	.030	.001	.001	.003	.065
	10.00	.548	.829	.035	.007	.007	.009	.039
	15.00	.503	.713	.026	.009	.008	.007	.026
	20.00	.466	.625	.017	.009	.007	.006	.022
	25.00	.480	.560	.013	.007	.006	.001	.035
	30.00	.442	.511	.002	.003	.001	.001	.027
	35.00	.411	.459	.002	.002	.001	.001	.027
	40.00	.405	.441	.001	.001	.001	.001	.025
	45.00	.396	.412	.001	.001	.001	.001	.024
	50.00	.398	.388	.001	.001	.001	.001	.022
	55.00	.345	.360	.001	.001	.001	.001	.021
	60.00	.348	.333	.001	.001	.001	.001	.020
	65.00	.341	.297	.001	.001	.001	.001	.019
70.00	.272					.190	.182	
75.00	.283	.225	.168	.176	.118	.125	.178	
80.00	.266	.180	.131	.142	.094	.098	.166	
85.00	.230	.132	.081	.095	.062	.076	.143	
90.00	.181	.087	.052	.065	.034	.044	.131	
95.00	.113	.035	.025	.040	.012	.021	.117	
LOWER SURFACE	1.25	.368	.484	.467	.476	.467		.425
	2.50	.353	.448	.449	.439	.449	.461	.393
	5.00	.369	.382	.381	.389	.388	.380	.308
	7.50	.365	.327	.337	.336	.334	.322	.222
	10.00	.337	.285	.289	.304	.301	.302	.166
	15.00	.288	.223	.234	.232	.238	.216	.083
	20.00	.233	.182	.183	.185	.190	.165	.017
	25.00	.220	.146	.141	.155	.151	.124	.000
	30.00	.160	.112	.117	.121	.118	.089	.030
	35.00	.129	.082	.081	.093	.090	.053	.035
	40.00	.111	.063	.060	.071	.067	.026	.053
	45.00	.080	.035	.039	.049	.043	.004	.056
	50.00	.051	.017	.026	.030	.029	.015	.058
	55.00	.030	.009	.019	.020	.017	.026	.063
	60.00	.017	.004	.017	.014	.008	.033	.061
	65.00	.013	.002	.006	.014	.013	.026	.052
	70.00	.008						.044
	75.00	.011	.008	.009	.014	.013	.022	.047
80.00	.011	.007	.012	.017	.010	.010	.039	
85.00	.010	.001	.022	.027	.013	.014	.027	
90.00	.013	.011	.022	.027	.023	.008	.036	
95.00	.009	.015	.032	.035	.023	.006	.045	

TABLE I

BASIC WING

PERCENT CHORD		PRESSURE COEFFICIENT, P, AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
M = 0.60 $\alpha = 11.2^\circ$								
UPPER SURFACE	.00	.001	.007	.1789	.1385	.820	.550	.552
	1.25	.057	.1304	.1271	.1227	.794	.536	.459
	2.50	.123	.1336	.1267	.1239	.794	.525	.461
	3.75	.183	.1336	.1267	.1246	.791	.510	.463
	5.00	.249	.1353	.1267	.1266	.780	.514	.455
	7.50	.348	.1368	.1231	.1281	.770	.504	.439
	10.00	.448	.1428	.1226	.1202	.758	.487	.385
	12.50	.548	.1488	.1192	.1111	.737	.465	.336
	15.00	.648	.1548	.1159	.1024	.719	.437	.298
	17.50	.748	.1608	.1111	.936	.705	.422	.271
	20.00	.848	.1668	.973	.864	.689	.401	.257
	22.50	.948	.1728	.888	.798	.672	.382	.230
	25.00	.1048	.1788	.787	.739	.661	.369	.215
	27.50	.1108	.1848	.698	.675	.641	.354	.210
	30.00	.1168	.1908	.591	.623	.618	.341	.200
	32.50	.1228	.1968	.520	.534	.588	.325	.195
	35.00	.1288	.2028	.432	.499	.548	.325	.193
	LOWER SURFACE	1.25	.380	.491	.438	.500	.469	.477
2.50		.351	.527	.493	.492	.499	.477	.398
3.75		.442	.492	.459	.472	.451	.414	.319
5.00		.488	.445	.418	.432	.399	.368	.245
7.50		.471	.412	.382	.397	.356	.345	.195
10.00		.428	.340	.325	.317	.302	.266	.129
12.50		.362	.298	.272	.275	.244	.207	.057
15.00		.334	.250	.230	.228	.205	.152	.018
17.50		.281	.217	.197	.194	.170	.117	.005
20.00		.239	.183	.168	.160	.131	.074	.015
22.50		.214	.154	.137	.132	.105	.043	.035
25.00		.180	.126	.113	.108	.071	.010	.037
27.50		.154	.101	.087	.080	.051	.020	.069
30.00		.130	.087	.071	.059	.021	.040	.076
32.50		.099	.074	.071	.046	.008	.055	.076
35.00		.093	.068	.053	.040	.005	.055	.075
37.50		.077	.055	.035	.015	.025	.075	.092
40.00		.043	.055	.036	.005	.051	.077	.087
M = 0.60 $\alpha = 15.5^\circ$								
UPPER SURFACE	.00	.007	.324	.1264	.1045	.666	.468	.371
	1.25	.051	.1981	.1234	.857	.647	.454	.335
	2.50	.1209	.1993	.1233	.854	.649	.453	.334
	3.75	.1888	.1993	.1233	.850	.642	.440	.331
	5.00	.2613	.1993	.1233	.847	.638	.443	.330
	7.50	.3294	.1993	.1233	.848	.632	.432	.328
	10.00	.3964	.1993	.1233	.833	.621	.426	.318
	12.50	.4644	.1993	.1233	.822	.608	.415	.309
	15.00	.5324	.1993	.1233	.804	.595	.401	.304
	17.50	.6004	.1993	.1233	.798	.584	.404	.308
	20.00	.6684	.1993	.1233	.787	.575	.394	.305
	22.50	.7364	.1993	.1233	.785	.564	.385	.300
	25.00	.8044	.1993	.1233	.774	.557	.379	.299
	27.50	.8724	.1993	.1233	.772	.551	.373	.298
	30.00	.9404	.1993	.1233	.762	.547	.373	.291
	32.50	.1008	.1993	.1233	.754	.546	.368	.287
	35.00	.1072	.1993	.1233	.734	.540	.360	.284
	LOWER SURFACE	1.25	.257	.424	.378	.476	.433	.487
2.50		.275	.557	.518	.530	.489	.445	.358
3.75		.473	.575	.528	.514	.489	.445	.358
5.00		.597	.543	.504	.491	.458	.408	.329
7.50		.595	.517	.471	.461	.415	.386	.315
10.00		.558	.447	.428	.392	.354	.309	.265
12.50		.491	.403	.359	.344	.303	.246	.209
15.00		.450	.352	.319	.295	.254	.193	.154
17.50		.398	.315	.275	.257	.207	.146	.103
20.00		.358	.277	.240	.215	.167	.108	.067
22.50		.321	.245	.207	.181	.138	.081	.040
25.00		.280	.209	.178	.151	.107	.057	.016
27.50		.239	.178	.141	.113	.088	.049	.009
30.00		.210	.158	.113	.088	.053	.071	.016
32.50		.184	.131	.088	.053	.029	.087	.014
35.00		.164	.112	.073	.028	.007	.087	.013
37.50		.145	.078	.031	.009	.071	.111	.013
40.00		.108	.070	.014	.009	.106	.130	.013

TABLE I

BASIC WING

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M = 0.60 α = 19.5°								
	.00	.081	1.259	1.070	1.060	.604	.507	.453	
	1.25	.030	1.248	1.037	.801	.599	.498	.433	
	2.50	1.236	1.236	1.038	.794	.596	.498	.428	
	3.75	1.383	1.231	1.088	.793	.591	.489	.424	
	5.00	1.344	1.226	.996	.790	.591	.493	.415	
	7.50	1.315	1.219	1.003	.783	.587	.492	.424	
	10.00	1.277	1.205	.986	.766	.580	.486	.421	
	15.00	1.234	1.198	.975	.757	.574	.480	.419	
	20.00	1.192	1.182	.955	.740	.569	.471	.424	
	25.00	1.140	1.167	.947	.730	.563	.476	.424	
	30.00	1.081	1.146	.938	.720	.554	.472	.423	
	35.00	1.022	1.124	.930	.714	.552	.470	.423	
	40.00	.968	1.095	.915	.707	.551	.466	.421	
	45.00	.923	1.063	.904	.705	.547	.459	.414	
	50.00	.850	1.030	.888	.706	.546	.454	.409	
	55.00	.816	.997	.874	.703	.542	.450	.403	
	60.00	.785	.955	.861	.691	.537	.438	.393	
	65.00	.707	.889	.825	.670	.526	.423	.378	
	70.00	.660	.836	.782	.650	.516	.410	.366	
	75.00	.619	.778	.736	.636	.500	.400	.356	
	80.00	.555	.724	.704	.608	.484	.385	.346	
95.00	.481	.668	.687	.591	.450	.368	.331		
LOWER SURFACE	1.25	.155	.408	.331	.423	.387	.486	.358	
	2.50	.224	.571	.506	.518	.495	.470	.408	
	3.75	.512	.606	.548	.527	.507	.441	.374	
	5.00	.673	.589	.535	.510	.483	.441	.322	
	7.50	.669	.566	.512	.488	.455	.426	.273	
	10.00	.633	.508	.467	.430	.397	.353	.200	
	15.00	.565	.466	.421	.381	.345	.292	.116	
	20.00	.523	.414	.372	.332	.297	.241	.068	
	25.00	.471	.371	.329	.287	.249	.186	.021	
	30.00	.419	.329	.287	.243	.202	.139	.017	
	35.00	.387	.293	.250	.204	.164	.094	.023	
	40.00	.344	.254	.216	.156	.116	.051	.070	
	45.00	.305	.216	.158	.116	.083	.015	.093	
	50.00	.264	.184	.125	.080	.044	.022	.116	
	55.00	.232	.152	.093	.048	.010	.048	.119	
	60.00	.209	.127	.052	.012	.012	.075	.128	
	65.00	.178	.057	.030	.059	.069	.105	.156	
	70.00	.118	.033	.047	.091	.107	.139	.164	
	75.00	.113	.021	.108	.127	.145	.160	.181	
	80.00	.076	.080	.180	.195	.153	.185	.191	
	95.00	.007	.184	.269	.264	.249	.222	.217	
	UPPER SURFACE	M = 0.60 α = 25.8°							
.00		.055	1.024	1.008	.923	.740	.604	.524	
1.25		.924	1.018	.983	.879	.735	.593	.519	
2.50		1.073	1.015	.982	.879	.733	.599	.516	
3.75		1.034	1.021	.987	.880	.729	.590	.511	
5.00		1.026	1.023	.985	.875	.725	.591	.512	
7.50		1.022	1.023	.985	.874	.723	.591	.509	
10.00		1.023	1.023	.985	.868	.710	.587	.506	
15.00		1.023	1.023	.985	.863	.708	.582	.509	
20.00		1.024	1.034	.980	.847	.705	.571	.512	
25.00		1.022	1.035	.985	.843	.695	.574	.513	
30.00		1.017	1.032	.981	.833	.690	.571	.515	
35.00		1.008	1.032	.976	.825	.684	.566	.514	
40.00		.999	1.035	.969	.819	.680	.561	.508	
45.00		.984	1.033	.963	.812	.671	.552	.500	
50.00		.978	1.031	.954	.803	.665	.543	.492	
55.00		.963	1.026	.942	.793	.658	.536	.476	
60.00		.962	1.021	.927	.776	.646	.524	.465	
65.00		.961	1.007	.927	.776	.646	.524	.465	
70.00		.935	.978	.897	.750	.625	.500	.444	
75.00		.918	.957	.884	.732	.616	.486	.433	
80.00		.893	.927	.864	.724	.604	.472	.419	
85.00	.842	.897	.853	.712	.589	.452	.405		
95.00	.792	.839	.858	.703	.580	.438	.398		
LOWER SURFACE	1.25	.069	.314	.177	.285	.217	.451	.239	
	2.50	.088	.558	.460	.468	.445	.503	.377	
	3.75	.501	.658	.567	.538	.523	.500	.403	
	5.00	.709	.668	.584	.554	.530	.495	.375	
	7.50	.725	.660	.583	.547	.517	.495	.339	
	10.00	.731	.619	.552	.510	.480	.441	.274	
	15.00	.643	.584	.509	.469	.436	.385	.196	
	20.00	.641	.534	.464	.424	.394	.335	.137	
	25.00	.595	.495	.433	.383	.343	.282	.081	
	30.00	.552	.454	.382	.339	.299	.230	.040	
	35.00	.514	.416	.337	.296	.254	.187	.016	
	40.00	.471	.372	.297	.245	.208	.138	.031	
	45.00	.425	.326	.250	.206	.159	.089	.060	
	50.00	.386	.291	.211	.160	.119	.048	.084	
	55.00	.346	.254	.178	.116	.078	.014	.097	
	60.00	.314	.213	.117	.073	.045	.020	.112	
	65.00	.278	.186	.016	.016	.033	.076	.128	
	70.00	.209	.087	.031	.061	.081	.093	.147	
	75.00	.186	.006	.094	.117	.132	.150	.164	
	80.00	.138	.057	.189	.201	.161	.190	.189	
	95.00	.009	.201	.307	.292	.281	.237	.243	

TABLE I

BASIC WING

		PRESSURE COEFFICIENT, P, AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
PERCENT CHORD								
M = 0.85 α = 0.0°								
UPPER SURFACE	.00	.026	.516	.458	.712	.467	.487	.454
	1.25	.193	.097	.127	.144	.048	.092	.098
	2.50	.182	.080	.110	.138	.113	.156	.195
	5.00	.050	.078	.115	.146	.082	.085	.112
	7.50	.011	.093	.120	.141	.120	.107	.145
	10.00	.004	.107	.138	.140	.135	.123	.178
	15.00	.041	.121	.155	.164	.135	.152	.162
	20.00	.066	.138	.172	.171	.154	.156	.176
	25.00	.093	.157	.188	.179	.167	.163	.163
	30.00	.115	.170	.201	.190	.170	.174	.154
	35.00	.116	.187	.210	.191	.177	.178	.148
	40.00	.136	.202	.215	.194	.176	.181	.139
	45.00	.158	.212	.221	.193	.175	.176	.130
	50.00	.187	.224	.216	.190	.171	.163	.116
	55.00	.161	.231	.198	.181	.162	.149	.100
	60.00	.188	.217	.184	.126	.143	.122	.070
	65.00	.206	.196	.155	.127	.118	.099	.056
	70.00	.157						.036
	75.00	.149	.163	.106	.091	.072	.043	.013
80.00	.185	.125	.075	.059	.043	.012	.015	
85.00	.174	.088	.021	.032	.003	.020	.041	
90.00	.145	.055	.009	.003	.027	.049	.061	
95.00	.103	.008	.027	.019	.053	.078	.084	
LOWER SURFACE	1.25	.176	.102	.160	.167	.183		.219
	2.50	.114	.073	.117	.138	.143	.117	.180
	5.00	.045	.073	.119	.087	.147	.140	.170
	7.50	.026	.093	.132	.120	.161	.150	.204
	10.00	.005	.104	.146	.141	.163	.147	.215
	15.00	.038	.130	.165	.181	.156	.156	.216
	20.00	.083	.141	.181	.183	.173	.172	.191
	25.00	.055	.159	.195	.191	.179	.177	.169
	30.00	.126	.178	.204	.193	.183	.182	.160
	35.00	.134	.196	.212	.195	.185	.187	.151
	40.00	.140	.206	.223	.199	.188	.184	.143
	45.00	.166	.223	.223	.200	.190	.177	.132
	50.00	.204	.233	.221	.193	.179	.166	.118
	55.00	.200	.223	.205	.182	.168	.148	.102
	60.00	.208	.213	.185	.161	.146	.121	.078
	65.00	.219	.201	.168	.139	.116	.095	.052
	70.00	.197						.038
	75.00	.185	.161	.114	.087	.060	.039	.014
	80.00	.153	.121	.085	.053	.041	.004	.005
85.00	.175	.094	.049	.023	.011	.033	.028	
90.00	.158	.055	.018	.001	.017	.035	.049	
95.00	.150	.010	.031	.046	.048	.066	.077	
M = 0.85 α = 4.0°								
UPPER SURFACE	.00	.036	.098	.057	.175	.338	.410	.281
	1.25	.023	.098	.176	.243	.284	.205	.170
	2.50	.124	.186	.080	.186	.216	.1206	.136
	5.00	.193	.424	.547	.743	.949	.843	.882
	7.50	.213	.380	.484	.512	.681	.655	.688
	10.00	.212	.360	.469	.465	.621	.593	.549
	15.00	.229	.338	.435	.455	.430	.472	.356
	20.00	.236	.328	.414	.433	.405	.408	.275
	25.00	.254	.332	.406	.404	.381	.362	.229
	30.00	.269	.338	.406	.377	.350	.327	.212
	35.00	.261	.345	.396	.349	.332	.285	.199
	40.00	.276	.357	.380	.330	.309	.255	.186
	45.00	.295	.361	.371	.306	.285	.224	.173
	50.00	.328	.389	.350	.281	.260	.200	.152
	55.00	.349	.388	.339	.257	.233	.174	.135
	60.00	.321	.330	.262	.186	.196	.141	.103
	65.00	.338	.299	.220	.180	.158	.111	.086
	70.00	.265						.069
	75.00	.273	.223	.140	.114	.086	.052	.046
80.00	.274	.164	.101	.073	.052	.019	.022	
85.00	.251	.110	.046	.036	.012	.018	.000	
90.00	.195	.062	.018	.008	.022	.042	.011	
95.00	.123	.010	.022	.019	.049	.066	.020	
LOWER SURFACE	1.25	.335	.356	.357	.342	.387		.360
	2.50	.309	.283	.278	.276	.307	.328	.279
	5.00	.253	.211	.194	.245	.219	.230	.179
	7.50	.222	.163	.149	.187	.164	.173	.094
	10.00	.196	.137	.114	.130	.123	.160	.043
	15.00	.149	.078	.069	.070	.088	.084	.043
	20.00	.098	.046	.027	.040	.042	.039	.028
	25.00	.101	.013	.002	.009	.014	.003	.103
	30.00	.044	.013	.030	.012	.009	.030	.108
	35.00	.016	.034	.048	.031	.031	.060	.108
	40.00	.006	.051	.065	.045	.046	.077	.107
	45.00	.023	.076	.077	.061	.064	.096	.105
	50.00	.052	.092	.087	.068	.068	.106	.100
	55.00	.084	.092	.085	.069	.074	.104	.088
	60.00	.073	.092	.077	.062	.067	.093	.070
	65.00	.086	.089	.073	.053	.055	.076	.050
	70.00	.080						.038
	75.00	.080	.076	.050	.034	.028	.036	.024
	80.00	.064	.051	.026	.013	.019	.007	.010
85.00	.082	.040	.003	.008	.001	.007	.001	
90.00	.071	.020	.011	.028	.020	.023	.025	
95.00	.088	.008	.044	.049	.036	.051	.044	

TABLE I

BASIC WING

PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:						
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
$M = 0.85 \quad \alpha = 5.0^\circ$							
UPPER SURFACE							
1.00	.0285	.207	.347	.035	.710	.822	.718
1.25	.158	1.194	1.298	1.331	1.465	1.326	1.007
2.50	.273	1.098	1.115	1.239	1.435	1.273	.994
5.00	.344	.788	.935	1.073	1.348	1.168	.930
7.50	.368	.628	.817	.944	1.254	1.093	.883
10.00	.348	.563	.756	.868	1.175	1.002	.828
15.00	.349	.483	.664	.752	.944	.777	.728
20.00	.363	.446	.600	.668	.614	.633	.632
25.00	.360	.445	.565	.598	.475	.492	.543
30.00	.351	.454	.523	.559	.346	.338	.450
35.00	.363	.466	.500	.405	.317	.294	.372
40.00	.378	.470	.469	.357	.289	.253	.312
45.00	.419	.479	.425	.317	.261	.216	.271
50.00	.375	.451	.352	.285	.230	.183	.216
55.00	.415	.410	.299	.211	.190	.147	.189
60.00	.443	.369	.250	.202	.155	.117	.172
65.00	.352						.168
70.00	.350	.262	.156	.135	.089	.062	.134
75.00	.338	.193	.116	.099	.055	.033	.109
80.00	.302	.131	.059	.061	.021	.006	.090
85.00	.227	.079	.030	.023	.015	.021	.070
90.00	.136	.019	.005	.006	.039	.043	
95.00							
LOWER SURFACE							
1.25	.381	.454	.437	.415	.453		.420
2.50	.375	.381	.368	.345	.390	.410	.354
5.00	.338	.303	.284	.308	.307	.317	.256
7.50	.311	.247	.235	.255	.250	.260	.174
10.00	.282	.217	.199	.198	.209	.219	.114
15.00	.232	.153	.145	.143	.163	.163	.015
20.00	.174	.113	.098	.105	.114	.108	.038
25.00	.169	.079	.067	.074	.080	.068	.082
30.00	.112	.052	.032	.051	.051	.031	.101
35.00	.079	.023	.013	.022	.027	.004	.107
40.00	.065	.003	.009	.002	.007	.030	.104
45.00	.034	.023	.028	.017	.015	.056	.106
50.00	.002	.042	.041	.028	.025	.074	.102
55.00	.016	.047	.042	.033	.037	.081	.094
60.00	.027	.050	.040	.033	.036	.074	.077
65.00	.040	.083	.039	.029	.028	.065	.062
70.00	.035						.048
75.00	.044	.047	.033	.017	.013	.031	.034
80.00	.029	.026	.013	.000	.011	.008	.021
85.00	.050	.021	.003	.012	.001	.033	.110
90.00	.045	.009	.014	.021	.011	.016	.005
95.00	.053	.010	.044	.041	.029	.036	.012
UPPER SURFACE							
1.00	.022	.494	.599	.228	.960	.834	.957
1.25	.272	1.215	1.478	1.466	1.277	.749	1.105
2.50	.404	1.198	1.417	1.427	1.262	.742	1.088
5.00	.493	1.190	1.371	1.319	1.228	.713	1.067
7.50	.503	1.139	1.290	1.223	1.197	.717	1.095
10.00	.502	1.097	1.069	1.139	1.172	.699	1.111
15.00	.481	.935	.939	1.034	1.091	.661	.964
20.00	.442	.658	.846	.901	1.023	.628	.463
25.00	.454	.483	.781	.818	.967	.584	.231
30.00	.457	.490	.726	.718	.897	.559	.236
35.00	.434	.526	.685	.610	.830	.523	.250
40.00	.429	.546	.640	.518	.753	.487	.243
45.00	.423	.548	.586	.445	.673	.449	.243
50.00	.488	.571	.466	.384	.588	.414	.238
55.00	.467	.567	.363	.333	.498	.384	.223
60.00	.527	.511	.313	.247	.402	.356	.207
65.00	.560	.428	.289	.235	.301	.319	.204
70.00	.460						.200
75.00	.452	.262	.167	.161	.158	.277	.191
80.00	.410	.189	.126	.127	.113	.230	.186
85.00	.337	.126	.074	.091	.065	.213	.170
90.00	.238	.074	.047	.050	.024	.193	.161
95.00	.142	.015	.018	.023	.002	.181	.152
LOWER SURFACE							
1.25	.403	.520	.484	.460	.478		.427
2.50	.427	.462	.431	.411	.440	.448	.378
5.00	.419	.386	.387	.368	.369	.365	.289
7.50	.403	.327	.306	.318	.314	.306	.210
10.00	.369	.292	.270	.268	.268	.287	.153
15.00	.317	.226	.216	.216	.221	.205	.062
20.00	.286	.188	.164	.171	.169	.146	.011
25.00	.248	.146	.128	.141	.132	.100	.028
30.00	.185	.114	.097	.109	.099	.064	.027
35.00	.143	.084	.068	.077	.073	.023	.103
40.00	.128	.061	.046	.057	.047	.006	.105
45.00	.092	.034	.030	.027	.021	.038	.113
50.00	.062	.013	.006	.021	.008	.062	.114
55.00	.039	.003	.001	.011	.012	.075	.109
60.00	.021	.003	.000	.001	.012	.079	.097
65.00	.014	.008	.008	.001	.012	.073	.084
70.00	.009						.079
75.00	.009	.012	.006	.008	.011	.060	.076
80.00	.011	.004	.004	.013	.014	.049	.064
85.00	.010	.008	.018	.024	.028	.038	.062
90.00	.014	.009	.018	.024	.021	.049	.062
95.00	.016	.020	.027	.028	.012	.027	.078

TABLE I

BASIC WING

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
M = 0.85 α = 11.3°									
UPPER SURFACE		.00	.007	-.887	-1.015	-.660	-.873	-.514	-.515
		1.25	.451	-1.479	-1.384	-1.146	-.787	-.483	-.421
		2.50	.607	-1.464	-1.267	-1.158	-.783	-.481	-.422
		5.00	.749	-1.499	-1.246	-1.136	-.764	-.465	-.430
		7.50	.767	-1.492	-1.203	-1.134	-.755	-.468	-.430
		10.00	.749	-1.461	-1.193	-1.131	-.747	-.463	-.427
		15.00	.735	-1.378	-1.163	-1.093	-.731	-.454	-.397
		20.00	.670	-1.330	-1.111	-1.050	-.711	-.441	-.356
		25.00	.648	-1.268	-1.063	-1.007	-.687	-.420	-.323
		30.00	.627	-.970	-1.017	-.962	-.675	-.415	-.302
		35.00	.601	-.546	-.967	-.921	-.660	-.405	-.287
		40.00	.588	-.566	-.912	-.876	-.647	-.391	-.276
		45.00	.588	-.596	-.850	-.829	-.630	-.384	-.274
		50.00	.624	-.620	-.783	-.785	-.614	-.377	-.270
		55.00	.568	-.619	-.702	-.737	-.598	-.372	-.269
		60.00	.616	-.577	-.637	-.701	-.582	-.367	-.266
		65.00	.656	-.490	-.581	-.642	-.566	-.367	-.274
	LOWER SURFACE		70.00	.579					
		75.00	.579	-.307	-.446	-.535	-.520	-.360	-.269
		80.00	.498	-.240	-.388	-.495	-.505	-.352	-.265
		85.00	.379	-.173	-.310	-.450	-.478	-.345	-.268
		90.00	.273	-.111	-.243	-.403	-.444	-.333	-.261
		95.00	.186	-.054	-.190	-.362	-.418	-.323	-.256
		1.25	.383	.576	.511	.515	.489		.437
		2.50	.443	.556	.507	.490	.491	.482	.412
		5.00	.519	.501	.456	.464	.441	.412	.331
		7.50	.541	.448	.413	.418	.389	.362	.255
		10.00	.509	.411	.376	.373	.351	.299	.202
		15.00	.454	.343	.323	.308	.297	.260	.113
		20.00	.383	.298	.269	.263	.243	.198	.009
		25.00	.356	.256	.222	.225	.196	.147	.015
		30.00	.294	.220	.192	.190	.161	.099	.059
		35.00	.250	.182	.159	.153	.122	.053	.013
		40.00	.228	.156	.130	.125	.090	.012	.124
		45.00	.188	.124	.104	.094	.054	.030	.136
	50.00	.155	.096	.067	.067	.037	.068	.144	
	55.00	.124	.079	.048	.048	.001	.098	.146	
	60.00	.099	.065	.033	.033	.021	.121	-.146	
	65.00	.077	.052	.039	.016	-.037	-.141	-.156	
	70.00	.077						-.159	
	75.00	.037	.034	.082	-.011	-.069	-.162	-.160	
	80.00	.049	.039	.023	-.021	-.100	-.188	-.177	
	85.00	.030	.011	.023	-.036	-.125	-.197	-.183	
	90.00	.018	.018	.006	-.073	-.172	-.212	-.184	
	95.00	.017	.008	-.001	-.107	-.217	-.229	-.199	
M = 0.85 α = 15.8°									
UPPER SURFACE		.00	.007	1.294	1.090	.950	.666	.466	.419
		1.25	.644	1.561	1.086	.791	.654	.456	.398
		2.50	.863	1.529	1.006	.784	.652	.456	.399
		5.00	.052	1.505	1.035	.784	.644	.448	.409
		7.50	1.071	1.482	1.035	.788	.642	.452	.407
		10.00	1.053	1.454	1.040	.791	.639	.448	.394
		15.00	1.024	1.318	1.023	.767	.632	.441	.396
		20.00	.931	1.269	.991	.753	.620	.435	.395
		25.00	.832	1.224	.959	.743	.605	.417	.389
		30.00	.716	1.171	.944	.738	.598	.423	.392
		35.00	.638	1.106	.928	.728	.587	.419	.390
		40.00	.637	1.031	.909	.715	.578	.414	.387
		45.00	.609	.967	.894	.705	.569	.412	.385
		50.00	.643	.897	.879	.697	.560	.411	.385
		55.00	.676	.836	.855	.687	.550	.406	.381
		60.00	.699	.790	.838	.684	.544	.403	.377
		65.00	.623		.841	.680	.546	.412	.388
	LOWER SURFACE		70.00	.663					
		75.00	.663	.663	.783	.645	.512	.397	.367
		80.00	.601	.607	.763	.629	.503	.390	.359
		85.00	.545	.537	.731	.611	.488	.386	.358
		90.00	.449	.466	.707	.582	.475	.382	.347
		95.00	.360	.403	.696	.569	.462	.377	.336
		1.25	.340	.579	.489	.527	.459		.413
		2.50	.438	.619	.550	.520	.502	.495	.415
		5.00	.607	.594	.588	.509	.476	.449	.385
		7.50	.684	.554	.496	.474	.436	.404	.328
		10.00	.654	.519	.466	.437	.398	.358	.253
		15.00	.589	.453	.413	.374	.347	.302	.207
		20.00	.510	.405	.358	.326	.289	.244	.100
		25.00	.475	.361	.314	.278	.242	.190	.003
		30.00	.410	.318	.270	.240	.206	.155	.066
		35.00	.361	.275	.238	.216	.182	.135	.081
		40.00	.333	.245	.216	.196	.168	.128	.083
		45.00	.288	.204	.188	.171	.148	.114	.078
	50.00	.245	.174	.168	.158	.138	.108	.078	
	55.00	.214	.146	.146	.140	.122	.098	.078	
	60.00	.186	.128	.128	.122	.107	.088	.078	
	65.00	.163	.094	.096	.103	.088	.073	.078	
	70.00	.144						.078	
	75.00	.144	.055	.019	.082	.114	.213	.230	
	80.00	.090	.043	.045	.109	.154	.240	.263	
	85.00	.059	.010	.079	.145	.185	.266	.289	
	90.00	.028	.040	.146	.201	.232	.281	.299	
	95.00	.003	.100	.217	.258	.287	.288	.277	

TABLE I

BASIC WING

PERCENT CHORD		PRESSURE COEFFICIENT, P, AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M = 0.85 $\alpha = 19.6^\circ$							
	.00	.030	1.080	.895	.841	.666	.567	.492
	1.25	.648	1.010	.870	.737	.662	.552	.477
	2.50	.818	1.008	.851	.734	.663	.558	.476
	5.00	.957	.997	.863	.748	.652	.548	.481
	7.50	1.039	.989	.844	.752	.649	.548	.474
	10.00	1.027	.984	.846	.748	.648	.548	.467
	15.00	1.023	.963	.844	.726	.643	.549	.461
	20.00	.987	.954	.835	.720	.639	.541	.462
	25.00	.947	.942	.827	.717	.628	.523	.459
	30.00	.904	.936	.823	.715	.624	.530	.466
	35.00	.858	.925	.818	.709	.618	.528	.469
	40.00	.817	.910	.809	.701	.612	.520	.468
	45.00	.785	.895	.805	.696	.608	.519	.467
	50.00	.781	.886	.797	.690	.603	.519	.466
	55.00	.729	.873	.786	.686	.595	.513	.460
	60.00	.738	.857	.778	.683	.596	.507	.453
	65.00	.734	.858	.783	.679	.599	.508	.460
	70.00	.686						.440
75.00	.772	.818	.751	.661	.579	.488	.435	
80.00	.686	.804	.735	.655	.572	.477	.421	
85.00	.672	.784	.722	.646	.563	.465	.417	
90.00	.641	.763	.703	.631	.552	.453	.402	
95.00	.584	.740	.689	.621	.538	.436	.386	
LOWER SURFACE	1.25	.217	.573	.459	.509	.421		.374
	2.50	.402	.661	.573	.541	.506	.492	.421
	5.00	.661	.668	.583	.542	.512	.475	.394
	7.50	.774	.640	.563	.522	.486	.446	.337
	10.00	.754	.612	.539	.500	.455	.396	.297
	15.00	.696	.554	.492	.444	.409	.361	.209
	20.00	.621	.505	.437	.398	.353	.305	.122
	25.00	.580	.460	.393	.349	.306	.248	.050
	30.00	.538	.413	.352	.307	.260	.195	.018
	35.00	.489	.376	.308	.266	.218	.138	.074
	40.00	.437	.338	.271	.219	.173	.090	.105
	45.00	.392	.299	.231	.179	.125	.039	.142
	50.00	.350	.260	.191	.136	.087	.014	.169
	55.00	.311	.227	.152	.097	.045	.062	.194
	60.00	.276	.196	.126	.061	.011	.100	.204
	65.00	.253	.165	.078	.024	.025	.141	.231
	70.00	.222						.242
	75.00	.154	.103	.001	.048	.089	.198	.278
	80.00	.151	.076	.038	.082	.134	.234	.285
85.00	.111	.010	.085	.128	.172	.267	.285	
90.00	.053	.053	.155	.184	.239	.296	.285	
95.00	.020	.163	.230	.254	.297	.319	.306	
M = 0.85 $\alpha = 26.0^\circ$								
UPPER SURFACE	.00	.115	.881	.880	.811	.780	.697	.600
	1.25	.554	.885	.879	.796	.781	.694	.587
	2.50	.821	.880	.865	.798	.781	.709	.587
	5.00	.897	.884	.860	.803	.788	.706	.591
	7.50	.889	.886	.868	.807	.791	.732	.586
	10.00	.885	.887	.872	.819	.790	.750	.577
	15.00	.887	.881	.875	.812	.813	.711	.573
	20.00	.882	.887	.873	.811	.824	.682	.572
	25.00	.878	.886	.877	.815	.799	.662	.569
	30.00	.880	.888	.887	.820	.786	.671	.575
	35.00	.873	.889	.886	.817	.776	.669	.578
	40.00	.865	.889	.879	.816	.769	.662	.578
	45.00	.859	.889	.873	.813	.769	.657	.577
	50.00	.853	.889	.870	.813	.765	.655	.573
	55.00	.839	.887	.865	.809	.759	.649	.565
	60.00	.837	.885	.861	.808	.753	.639	.552
	65.00	.836	.890	.868	.804	.753	.644	.556
	70.00	.805						.510
	75.00	.898	.871	.848	.783	.726	.611	.515
80.00	.811	.867	.840	.776	.718	.594	.501	
85.00	.804	.857	.830	.767	.703	.577	.494	
90.00	.783	.843	.826	.758	.693	.562	.476	
95.00	.736	.825	.823	.753	.678	.541	.465	
LOWER SURFACE	1.25	.046	.511	.358	.420	.293		.256
	2.50	.292	.686	.567	.530	.475	.460	.395
	5.00	.661	.749	.640	.577	.541	.498	.420
	7.50	.842	.743	.639	.576	.543	.498	.395
	10.00	.840	.732	.632	.571	.538	.471	.366
	15.00	.816	.687	.600	.543	.499	.444	.301
	20.00	.758	.649	.557	.508	.454	.396	.209
	25.00	.717	.605	.518	.468	.413	.347	.144
	30.00	.668	.565	.478	.428	.368	.294	.070
	35.00	.619	.526	.441	.388	.326	.242	.009
	40.00	.583	.488	.398	.334	.288	.192	.042
	45.00	.541	.443	.356	.294	.238	.138	.079
	50.00	.500	.404	.316	.251	.193	.083	.118
	55.00	.460	.369	.277	.208	.149	.029	.151
	60.00	.423	.335	.240	.167	.107	.015	.173
	65.00	.399	.298	.192	.128	.070	.067	.208
	70.00	.352						.237
	75.00	.292	.222	.096	.041	.003	.143	.246
	80.00	.259	.185	.050	.001	.064	.160	.279
85.00	.210	.103	.001	.001	.118	.227	.300	
90.00	.138	.037	.001	.129	.148	.269	.317	
95.00	.076	.092	.173	.214	.274	.341	.349	

TABLE I

BASIC WING

PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:						
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
M = 0.90 $\alpha = 0.0^\circ$							
UPPER SURFACE	0.00	.030	.530	.459	.720	.464	.487
	1.25	.030	.530	.459	.720	.464	.487
	2.50	.030	.530	.459	.720	.464	.487
	3.75	.030	.530	.459	.720	.464	.487
	5.00	.030	.530	.459	.720	.464	.487
	7.50	.030	.530	.459	.720	.464	.487
	10.00	.030	.530	.459	.720	.464	.487
	15.00	.030	.530	.459	.720	.464	.487
	20.00	.030	.530	.459	.720	.464	.487
	25.00	.030	.530	.459	.720	.464	.487
	30.00	.030	.530	.459	.720	.464	.487
	35.00	.030	.530	.459	.720	.464	.487
	40.00	.030	.530	.459	.720	.464	.487
	45.00	.030	.530	.459	.720	.464	.487
	50.00	.030	.530	.459	.720	.464	.487
	55.00	.030	.530	.459	.720	.464	.487
	60.00	.030	.530	.459	.720	.464	.487
	65.00	.030	.530	.459	.720	.464	.487
LOWER SURFACE	0.00	.030	.530	.459	.720	.464	.487
	1.25	.030	.530	.459	.720	.464	.487
	2.50	.030	.530	.459	.720	.464	.487
	3.75	.030	.530	.459	.720	.464	.487
	5.00	.030	.530	.459	.720	.464	.487
	7.50	.030	.530	.459	.720	.464	.487
	10.00	.030	.530	.459	.720	.464	.487
	15.00	.030	.530	.459	.720	.464	.487
	20.00	.030	.530	.459	.720	.464	.487
	25.00	.030	.530	.459	.720	.464	.487
	30.00	.030	.530	.459	.720	.464	.487
	35.00	.030	.530	.459	.720	.464	.487
	40.00	.030	.530	.459	.720	.464	.487
	45.00	.030	.530	.459	.720	.464	.487
	50.00	.030	.530	.459	.720	.464	.487
	55.00	.030	.530	.459	.720	.464	.487
	60.00	.030	.530	.459	.720	.464	.487
	65.00	.030	.530	.459	.720	.464	.487
M = 0.90 $\alpha = 4.0^\circ$							
UPPER SURFACE	0.00	.036	.165	.038	.280	.215	.309
	1.25	.036	.165	.038	.280	.215	.309
	2.50	.036	.165	.038	.280	.215	.309
	3.75	.036	.165	.038	.280	.215	.309
	5.00	.036	.165	.038	.280	.215	.309
	7.50	.036	.165	.038	.280	.215	.309
	10.00	.036	.165	.038	.280	.215	.309
	15.00	.036	.165	.038	.280	.215	.309
	20.00	.036	.165	.038	.280	.215	.309
	25.00	.036	.165	.038	.280	.215	.309
	30.00	.036	.165	.038	.280	.215	.309
	35.00	.036	.165	.038	.280	.215	.309
	40.00	.036	.165	.038	.280	.215	.309
	45.00	.036	.165	.038	.280	.215	.309
	50.00	.036	.165	.038	.280	.215	.309
	55.00	.036	.165	.038	.280	.215	.309
	60.00	.036	.165	.038	.280	.215	.309
	65.00	.036	.165	.038	.280	.215	.309
LOWER SURFACE	0.00	.036	.165	.038	.280	.215	.309
	1.25	.036	.165	.038	.280	.215	.309
	2.50	.036	.165	.038	.280	.215	.309
	3.75	.036	.165	.038	.280	.215	.309
	5.00	.036	.165	.038	.280	.215	.309
	7.50	.036	.165	.038	.280	.215	.309
	10.00	.036	.165	.038	.280	.215	.309
	15.00	.036	.165	.038	.280	.215	.309
	20.00	.036	.165	.038	.280	.215	.309
	25.00	.036	.165	.038	.280	.215	.309
	30.00	.036	.165	.038	.280	.215	.309
	35.00	.036	.165	.038	.280	.215	.309
	40.00	.036	.165	.038	.280	.215	.309
	45.00	.036	.165	.038	.280	.215	.309
	50.00	.036	.165	.038	.280	.215	.309
	55.00	.036	.165	.038	.280	.215	.309
	60.00	.036	.165	.038	.280	.215	.309
	65.00	.036	.165	.038	.280	.215	.309

TABLE I

BASIC WING

		PRESSURE COEFFICIENT, P, AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
M = 0.90 $\alpha = 6.0^\circ$								
UPPER SURFACE	PERCENT CHORD							
	0.00	.0285	-.073	-.197	-.086	-.511	-.627	-.581
	1.25	.087	-.019	-.1257	-.1.277	-.1.311	-.1.297	-.1.138
	2.50	.199	-.008	-.1.204	-.1.313	-.1.317	-.1.363	-.1.128
	3.75	.284	-.0735	-.1.096	-.1.235	-.1.255	-.1.260	-.1.070
	5.00	.308	-.168	-.961	-.1.121	-.1.199	-.1.238	-.1.018
	7.50	.308	-.266	-.771	-.1.058	-.1.157	-.1.202	-.832
	10.00	.313	-.473	-.535	-.1.885	-.1.052	-.1.127	-.728
	12.50	.313	-.435	-.491	-.1.545	-.1.027	-.1.058	-.646
	15.00	.313	-.421	-.497	-.1.541	-.814	-.588	-.540
	17.50	.313	-.416	-.516	-.1.558	-.622	-.475	-.439
	20.00	.313	-.426	-.522	-.1.597	-.563	-.359	-.336
	22.50	.313	-.442	-.536	-.1.616	-.386	-.252	-.288
	25.00	.313	-.450	-.555	-.1.623	-.217	-.169	-.251
	27.50	.313	-.474	-.565	-.1.537	-.153	-.145	-.214
	30.00	.313	-.492	-.565	-.1.378	-.119	-.111	-.186
	32.50	.313	-.489	-.565	-.1.288	-.095	-.075	-.187
	35.00	.313	-.452	-.516	-.1.172	-.047	-.026	-.167
LOWER SURFACE	37.50	.313	-.502	-.219	-.079	-.047	-.001	-.158
	40.00	.313	-.455	-.128	-.045	-.024	-.026	-.129
	42.50	.313	-.297	-.057	-.011	-.008	-.050	-.124
	45.00	.313	-.150	-.018	-.021	-.037	-.050	-.091
	47.50	.313	-.058	-.017	-.031	-.056	-.071	-.091
	50.00	.313						
	52.50	.367	.448	.426	.399	.435	.389	.407
	55.00	.367	.378	.350	.317	.367	.293	.339
	57.50	.344	.298	.270	.292	.284	.239	.243
	60.00	.315	.241	.219	.236	.228	.201	.165
	62.50	.284	.213	.188	.172	.186	.146	.107
	65.00	.234	.149	.132	.122	.147	.091	.069
	67.50	.179	.110	.085	.082	.098	.053	.132
	70.00	.113	.073	.051	.052	.065	.017	.152
	72.50	.076	.042	.022	.030	.038	.021	.142
	75.00	.065	.014	.006	.001	.015	.050	.134
	77.50	.030	.005	.027	.015	.004	.080	.128
	80.00	.005	.037	.046	.035	.026	.100	.109
	82.50	.005	.057	.061	.046	.035	.106	.097
	85.00	.005	.063	.061	.049	.046	.109	.077
	87.50	.005	.063	.061	.049	.043	.109	.056
	90.00	.005	.073	.062	.040	.033	.083	.049
	92.50	.051						.026
	95.00	.048						.011
	97.50	.057	.071	.047	.022	.013	.034	.000
	100.00	.046	.053	.027	.003	.008	.028	.014
	102.50	.070	.049	.006	.011	.007	.032	.016
	105.00	.071	.040	.005	.025	.021	.032	
	107.50	.097	.021	.037	.048	.046	.062	
M = 0.90 $\alpha = 7.0^\circ$								
UPPER SURFACE	PERCENT CHORD							
	0.00	.018	-.330	-.413	-.095	-.738	-.876	-.750
	1.25	.188	-.1.139	-.1.419	-.1.411	-.1.403	-.1.079	-.798
	2.50	.317	-.1.135	-.1.351	-.1.433	-.1.401	-.1.098	-.797
	3.75	.415	-.1.119	-.1.336	-.1.433	-.1.355	-.098	-.802
	5.00	.442	-.1.031	-.1.250	-.1.380	-.1.313	-.097	-.782
	7.50	.433	-.957	-.1.145	-.1.322	-.1.287	-.095	-.713
	10.00	.436	-.801	-.881	-.1.273	-.1.232	-.092	-.644
	12.50	.410	-.600	-.689	-.1.223	-.1.190	-.087	-.527
	15.00	.423	-.478	-.623	-.1.186	-.1.133	-.084	-.474
	17.50	.423	-.468	-.609	-.1.153	-.1.081	-.084	-.450
	20.00	.409	-.463	-.615	-.1.129	-.1.021	-.083	-.421
	22.50	.410	-.404	-.627	-.1.107	-.096	-.083	-.395
	25.00	.418	-.338	-.636	-.1.074	-.071	-.083	-.375
	27.50	.428	-.257	-.634	-.1.037	-.049	-.083	-.358
	30.00	.420	-.184	-.642	-.1.003	-.033	-.083	-.338
	32.50	.474	-.119	-.598	-.095	-.024	-.083	-.327
	35.00	.523						.321
	37.50	.464						.307
	40.00	.459	.578	.286	.075	.162	.356	.296
	42.50	.502	.546	.179	.044	.124	.317	.284
	45.00	.533	.412	.099	.024	.075	.279	.272
	47.50	.533	.199	.082	.002	.032	.231	.260
	50.00	.523	.092	.013	.013	.004	.196	
	52.50	.420						.432
LOWER SURFACE	55.00	.413	.517	.489	.449	.472	.435	.380
	57.50	.444	.452	.427	.386	.422	.353	.294
	60.00	.428	.378	.351	.356	.348	.293	.213
	62.50	.407	.322	.302	.296	.292	.247	.154
	65.00	.376	.288	.265	.242	.248	.200	.061
	67.50	.323	.221	.210	.193	.202	.147	.016
	70.00	.260	.176	.158	.153	.154	.100	.080
	72.50	.247	.137	.125	.118	.116	.068	.120
	75.00	.185	.103	.088	.087	.087	.028	.134
	77.50	.144	.075	.059	.060	.062	.011	.138
	80.00	.127	.053	.035	.035	.037	.044	.136
	82.50	.092	.032	.013	.016	.014	.073	.127
	85.00	.058	.003	.008	.003	.002	.089	.118
	87.50	.034	.014	.018	.018	.018	.092	.105
	90.00	.014	.003	.023	.014	.021	.088	.092
	92.50	.004	.033	.083	.018	.017		.089
	95.00	.004						.083
	97.50	.004	.039	.022	.006	.010	.060	.076
	100.00	.007	.034	.007	.004	.011	.040	.084
	102.50	.007	.036	.004	.017	.008	.031	.091
	105.00	.033	.038	.010	.020	.005	.066	
	107.50	.043	.030	.029	.038	.019		.110
	110.00	.065						

TABLE I

BASIC WING

PERCENT CHORD		PRESSURE COEFFICIENT, P_i AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
M = 0.90 $\alpha = 11.4^\circ$								
UPPER SURFACE	.00	.005	-.713	-.800	-.436	-.568	-.361	-.478
	1.25	.346	1.358	1.426	1.073	.523	.529	.400
	2.50	.493	1.344	1.349	1.073	.503	.446	.402
	5.00	.636	1.380	1.388	1.085	.463	.425	.417
	7.50	.684	1.348	1.282	1.115	.403	.373	.407
	10.00	.657	1.303	1.249	1.129	.367	.329	.385
	15.00	.658	1.208	1.189	1.102	.299	.255	.370
	20.00	.604	1.143	1.131	1.064	.246	.208	.353
	25.00	.592	1.076	1.087	1.048	.203	.157	.327
	30.00	.576	.934	1.047	1.019	.165	.105	.318
	35.00	.556	.683	1.009	.981	.126	.060	.311
	40.00	.550	.574	.971	.934	.089	.012	.307
	45.00	.547	.595	.923	.887	.059	.034	.309
	50.00	.578	.631	.879	.837	.025	.076	.312
	55.00	.525	.651	.826	.788	.009	.116	.312
	60.00	.571	.661	.781	.756	.036	.141	.314
	65.00	.618	.631	.712	.708	.062	.167	.312
	LOWER SURFACE	70.00	.573					
75.00		.525	.651	.492	.619	.114	.193	.322
80.00		.643	.593	.416	.581	.138	.201	.329
85.00		.643	.383	.338	.546	.154	.209	.328
90.00		.587	.244	.280	.497	.187	.231	.315
95.00		.433	.159	.237	.467	.243	.251	.311
1.25		.388	.587	.522	.508	.624		.439
2.50		.474	.558	.508	.473	.811	.489	.414
5.00		.540	.498	.448	.445	.775	.506	.334
7.50		.557	.444	.404	.396	.765	.506	.255
10.00		.524	.410	.368	.353	.758	.501	.203
15.00		.464	.339	.313	.288	.746	.487	.109
20.00		.388	.293	.257	.244	.736	.467	.009
25.00		.363	.250	.216	.200	.702	.447	.032
30.00		.297	.210	.179	.168	.702	.445	.087
35.00		.251	.176	.146	.132	.680	.444	.129
40.00		.229	.148	.113	.099	.663	.437	.152
45.00		.188	.113	.088	.069	.647	.426	.155
50.00	.150	.082	.059	.043	.628	.426	.172	
55.00	.120	.063	.040	.022	.614	.423	.185	
60.00	.096	.047	.029	.005	.598	.423	.187	
65.00	.080	.029	.006	.018	.584	.442	.209	
70.00	.068						.213	
75.00	.015	.009	.017	.041	.522	.436	.211	
80.00	.037	.008	.018	.051	.513	.426	.244	
85.00	.016	.018	.025	.071	.497	.426	.243	
90.00	.001	.038	.044	.110	.488	.407	.242	
95.00	.008	.053	.056	.155	.458	.375	.255	
M = 0.90 $\alpha = 15.8^\circ$								
UPPER SURFACE	.00	.000	1.065	1.027	.865	.684	.517	.460
	1.25	.491	1.468	.951	.773	.670	.505	.336
	2.50	.700	1.443	.925	.771	.667	.509	.335
	5.00	.880	1.452	.954	.770	.680	.496	.345
	7.50	.907	1.457	.946	.782	.656	.499	.345
	10.00	.896	1.439	.954	.792	.655	.497	.3430
	15.00	.882	1.352	.962	.780	.650	.492	.328
	20.00	.812	1.291	.948	.769	.640	.482	.328
	25.00	.764	1.206	.932	.758	.629	.468	.320
	30.00	.718	1.154	.923	.751	.622	.468	.321
	35.00	.670	1.108	.908	.742	.612	.466	.320
	40.00	.622	1.008	.893	.730	.604	.458	.318
	45.00	.566	.903	.881	.718	.594	.454	.315
	50.00	.533	.833	.867	.709	.587	.453	.315
	55.00	.500	.758	.846	.699	.578	.446	.310
	60.00	.465	.738	.827	.695	.572	.442	.313
	65.00	.431	.727	.834	.690	.578	.449	.310
	LOWER SURFACE	70.00	.393					
75.00		.628	.673	.777	.660	.549	.433	.391
80.00		.611	.611	.757	.646	.537	.424	.381
85.00		.619	.551	.731	.633	.525	.417	.381
90.00		.569	.460	.712	.607	.511	.412	.367
95.00		.448	.419	.705	.599	.497	.404	.356
1.25		.357	.621	.521	.548	.472		.419
2.50		.482	.566	.531	.531	.510	.498	.428
5.00		.643	.541	.507	.516	.481	.448	.378
7.50		.678	.507	.474	.478	.448	.406	.303
10.00		.612	.471	.441	.441	.406	.391	.284
15.00		.534	.423	.383	.383	.351	.310	.161
20.00		.498	.368	.335	.335	.298	.250	.073
25.00		.435	.324	.289	.289	.248	.197	.008
30.00		.384	.288	.250	.250	.205	.143	.071
35.00		.356	.269	.237	.237	.185	.128	.187
40.00		.314	.230	.207	.207	.153	.088	.168
45.00		.273	.198	.175	.175	.128	.043	.191
50.00	.239	.171	.157	.157	.100	.014	.191	
55.00	.208	.148	.136	.136	.086	.063	.175	
60.00	.190	.121	.106	.106	.066	.109	.144	
65.00	.166		.081	.081	.055	.187	.128	
70.00	.102						.128	
75.00	.115	.080	.008	.069	.109	.237	.128	
80.00	.083	.063	.036	.094	.153	.283	.128	
85.00	.083	.083	.067	.131	.186	.315	.128	
90.00	.044	.083	.097	.190	.239	.325	.128	
95.00	.023	.086	.196	.251	.298	.334	.128	

TABLE 1

BASIC WING

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:							
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
		M = 0.90 α = 19.8°								
UPPER SURFACE		.00	-.043	-1.106	-.932	-.859	-.784	-.896	-.533	
		1.25	-.601	-1.086	-.911	-.763	-.728	-.889	-.520	
		2.50	-.834	-1.087	-.885	-.766	-.730	-.892	-.518	
		5.00	-1.127	-1.086	-.905	-.778	-.697	-.880	-.526	
		7.50	-1.117	-1.072	-.887	-.792	-.687	-.894	-.528	
		10.00	-1.083	-1.060	-.887	-.788	-.682	-.896	-.518	
		15.00	-1.063	-1.027	-.887	-.757	-.680	-.892	-.509	
		20.00	-.981	-1.017	-.873	-.752	-.674	-.883	-.507	
		25.00	-.920	-.997	-.861	-.752	-.665	-.865	-.503	
		30.00	-.854	-.989	-.856	-.749	-.664	-.872	-.509	
		35.00	-.798	-.973	-.848	-.745	-.656	-.871	-.517	
		40.00	-.765	-.954	-.837	-.737	-.651	-.866	-.513	
		45.00	-.738	-.932	-.829	-.730	-.645	-.863	-.514	
		50.00	-.746	-.916	-.821	-.726	-.639	-.862	-.510	
		55.00	-.695	-.896	-.813	-.721	-.631	-.856	-.504	
		60.00	-.719	-.875	-.801	-.717	-.625	-.852	-.496	
		65.00	-.726	-.866	-.808	-.715	-.632	-.858	-.503	
		70.00	-.665						-.490	
		75.00	-.725	-.822	-.773	-.699	-.609	-.839	-.471	
LOWER SURFACE		80.00	-.688	-.807	-.764	-.689	-.604	-.828	-.459	
		85.00	-.661	-.779	-.748	-.682	-.597	-.815	-.453	
		90.00	-.624	-.755	-.732	-.670	-.590	-.804	-.439	
		95.00	-.550	-.727	-.723	-.664	-.586	-.807	-.428	
		1.25	.233	.604	.487	.533	.423		.378	
		2.50	.443	.686	.591	.552	.514	.499	.431	
		5.00	.699	.690	.601	.553	.518	.484	.406	
		7.50	.807	.662	.583	.532	.495	.455	.354	
		10.00	.779	.636	.560	.507	.466	.427	.313	
		15.00	.719	.574	.514	.457	.421	.373	.288	
		20.00	.644	.529	.461	.412	.368	.317	.140	
		25.00	.603	.488	.417	.363	.320	.262	.068	
		30.00	.546	.440	.376	.326	.277	.209	.011	
		35.00	.496	.402	.337	.284	.233	.159	.070	
		40.00	.464	.366	.296	.241	.190	.109	.115	
		45.00	.420	.328	.261	.197	.145	.055	.155	
		50.00	.377	.288	.218	.157	.103	.008	.190	
		55.00	.339	.257	.186	.120	.062	.051	.228	
		60.00	.306	.228	.158	.083	.027	.098	.238	
	65.00	.289	.200	.115	.046	.007	.138	.268		
	70.00	.252						.291		
	75.00	.188	.143	.040	.027	.072	.201	.299		
	80.00	.181	.118	.002	.059	.117	.252	.327		
	85.00	.140	.044	.043	.099	.160	.291	.343		
	90.00	.089	.002	.110	.162	.230	.332	.348		
	95.00	.052	.104	.185	.232	.288	.372	.360		
		M = 0.90 α = 26.2°								
UPPER SURFACE		.00	-.161	-.920	-.901	-.804	-.811	-.781	-.640	
		1.25	-.475	-.903	-.881	-.777	-.790	-.763	-.611	
		2.50	-.925	-.898	-.862	-.780	-.783	-.789	-.605	
		5.00	-.912	-.899	-.881	-.788	-.778	-.803	-.584	
		7.50	-.898	-.900	-.864	-.792	-.783	-.859	-.595	
		10.00	-.900	-.905	-.872	-.787	-.783	-.893	-.603	
		15.00	-.899	-.896	-.867	-.785	-.774	-.767	-.596	
		20.00	-.904	-.909	-.876	-.785	-.776	-.701	-.599	
		25.00	-.905	-.912	-.886	-.780	-.779	-.677	-.588	
		30.00	-.896	-.906	-.886	-.817	-.775	-.680	-.589	
		35.00	-.888	-.905	-.884	-.817	-.774	-.678	-.610	
		40.00	-.871	-.906	-.874	-.818	-.776	-.679	-.613	
		45.00	-.867	-.904	-.868	-.817	-.778	-.675	-.614	
		50.00	-.867	-.896	-.861	-.813	-.776	-.674	-.612	
		55.00	-.842	-.895	-.854	-.812	-.778	-.674	-.613	
		60.00	-.846	-.895	-.856	-.804	-.769	-.672	-.600	
		65.00	-.842	-.856	-.827	-.779	-.739	-.642	-.573	
		70.00	-.798						-.590	
		75.00	-.885	-.875	-.845	-.789	-.748	-.650	-.570	
	80.00	-.815	-.869	-.845	-.782	-.744	-.640	-.558		
LOWER SURFACE		85.00	-.806	-.852	-.840	-.772	-.731	-.624	-.530	
		90.00	-.781	-.841	-.829	-.763	-.718	-.599	-.516	
		95.00	-.723	-.824	-.822	-.760	-.701	-.578	-.500	
		1.25	.069	.539	.394	.440	.320		.273	
		2.50	.327	.706	.629	.560	.492	.474	.409	
		5.00	.691	.766	.660	.601	.557	.518	.437	
		7.50	.865	.782	.665	.603	.560	.518	.413	
		10.00	.863	.780	.655	.598	.548	.492	.386	
		15.00	.838	.705	.626	.569	.518	.470	.334	
		20.00	.777	.668	.583	.533	.474	.422	.286	
		25.00	.737	.626	.545	.491	.435	.373	.165	
		30.00	.689	.583	.505	.453	.393	.327	.099	
		35.00	.641	.543	.470	.414	.352	.278	.053	
		40.00	.606	.508	.430	.373	.314	.227	.011	
		45.00	.563	.462	.393	.331	.266	.174	.061	
		50.00	.528	.425	.350	.284	.226	.121	.103	
		55.00	.486	.394	.313	.247	.182	.067	.143	
		60.00	.449	.361	.278	.206	.142	.023	.174	
		65.00	.431	.331	.236	.173	.115	.008	.186	
	70.00	.377						.228		
	75.00	.319	.253	.142	.083	.031	.083	.261		
	80.00	.286	.219	.097	.042	.027	.152	.253		
	85.00	.239	.150	.046	.001	.071	.186	.299		
	90.00	.168	.082	.026	.076	.097	.230	.319		
	95.00	.112	.038	.110	.158	.227	.313	.362		

TABLE I

BASIC WING

	PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.96b/2
$M = 0.94 \quad \alpha = 0.0^\circ$								
UPPER SURFACE	1.00	.027	.041	.068	.073	.045	.485	.465
	1.25	.034	.053	.088	.093	.077	.072	.081
	2.50	.166	.042	.084	.109	.150	.171	.080
	5.00	.095	.045	.098	.120	.122	.095	.097
	7.50	.054	.062	.105	.135	.167	.118	.137
	10.00	.038	.076	.148	.143	.191	.131	.178
	15.00	.005	.096	.174	.197	.200	.151	.220
	20.00	.035	.119	.195	.218	.235	.157	.279
	25.00	.067	.141	.235	.246	.259	.171	.270
	30.00	.090	.161	.251	.254	.249	.190	.234
	35.00	.095	.187	.250	.270	.219	.202	.233
	40.00	.119	.208	.272	.295	.172	.220	.218
	45.00	.146	.228	.285	.325	.157	.244	.208
	50.00	.185	.257	.296	.325	.171	.253	.208
	55.00	.164	.273	.323	.255	.175	.205	.208
	60.00	.204	.267	.338	.103	.161	.115	.255
	65.00	.243	.264	.309	.092	.138	.070	.208
	70.00	.196						.217
LOWER SURFACE	1.00	.027	.041	.068	.073	.045	.485	.465
	1.25	.034	.053	.088	.093	.077	.072	.081
	2.50	.166	.042	.084	.109	.150	.171	.080
	5.00	.095	.045	.098	.120	.122	.095	.097
	7.50	.054	.062	.105	.135	.167	.118	.137
	10.00	.038	.076	.148	.143	.191	.131	.178
	15.00	.005	.096	.174	.197	.200	.151	.220
	20.00	.035	.119	.195	.218	.235	.157	.279
	25.00	.067	.141	.235	.246	.259	.171	.270
	30.00	.090	.161	.251	.254	.249	.190	.234
	35.00	.095	.187	.250	.270	.219	.202	.233
	40.00	.119	.208	.272	.295	.172	.220	.218
	45.00	.146	.228	.285	.325	.157	.244	.208
	50.00	.185	.257	.296	.325	.171	.253	.208
	55.00	.164	.273	.323	.255	.175	.205	.208
	60.00	.204	.267	.338	.103	.161	.115	.255
	65.00	.243	.264	.309	.092	.138	.070	.208
	70.00	.196						.217
UPPER SURFACE	1.00	.027	.041	.068	.073	.045	.485	.465
	1.25	.034	.053	.088	.093	.077	.072	.081
	2.50	.166	.042	.084	.109	.150	.171	.080
	5.00	.095	.045	.098	.120	.122	.095	.097
	7.50	.054	.062	.105	.135	.167	.118	.137
	10.00	.038	.076	.148	.143	.191	.131	.178
	15.00	.005	.096	.174	.197	.200	.151	.220
	20.00	.035	.119	.195	.218	.235	.157	.279
	25.00	.067	.141	.235	.246	.259	.171	.270
	30.00	.090	.161	.251	.254	.249	.190	.234
	35.00	.095	.187	.250	.270	.219	.202	.233
	40.00	.119	.208	.272	.295	.172	.220	.218
	45.00	.146	.228	.285	.325	.157	.244	.208
	50.00	.185	.257	.296	.325	.171	.253	.208
	55.00	.164	.273	.323	.255	.175	.205	.208
	60.00	.204	.267	.338	.103	.161	.115	.255
	65.00	.243	.264	.309	.092	.138	.070	.208
	70.00	.196						.217
LOWER SURFACE	1.00	.027	.041	.068	.073	.045	.485	.465
	1.25	.034	.053	.088	.093	.077	.072	.081
	2.50	.166	.042	.084	.109	.150	.171	.080
	5.00	.095	.045	.098	.120	.122	.095	.097
	7.50	.054	.062	.105	.135	.167	.118	.137
	10.00	.038	.076	.148	.143	.191	.131	.178
	15.00	.005	.096	.174	.197	.200	.151	.220
	20.00	.035	.119	.195	.218	.235	.157	.279
	25.00	.067	.141	.235	.246	.259	.171	.270
	30.00	.090	.161	.251	.254	.249	.190	.234
	35.00	.095	.187	.250	.270	.219	.202	.233
	40.00	.119	.208	.272	.295	.172	.220	.218
	45.00	.146	.228	.285	.325	.157	.244	.208
	50.00	.185	.257	.296	.325	.171	.253	.208
	55.00	.164	.273	.323	.255	.175	.205	.208
	60.00	.204	.267	.338	.103	.161	.115	.255
	65.00	.243	.264	.309	.092	.138	.070	.208
	70.00	.196						.217

TABLE I

BASIC WING

		PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:							
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
		$M = 0.94 \quad \alpha = 5.9^\circ$								
UPPER SURFACE		0.00	.024	.017	.093	.204	.332	.410	.368	
		1.25	.035	.076	.186	.165	.193	.161	.269	
		2.50	.147	1.002	.146	.197	.211	.243	.261	
		5.00	.234	.678	.099	.130	.144	.141	.175	
		7.50	.266	.548	.011	.062	.090	.117	.152	
		10.00	.269	.492	.886	.016	.065	.083	.132	
		15.00	.281	.428	.464	.917	1.013	1.055	1.106	
		20.00	.274	.405	.431	.531	.976	1.021	1.077	
		25.00	.298	.394	.445	.482	.849	.990	.989	
		30.00	.307	.390	.464	.504	.608	.980	.821	
		35.00	.307	.398	.477	.527	.567	.948	.522	
		40.00	.317	.413	.495	.554	.584	.820	.426	
		45.00	.330	.424	.515	.575	.610	.649	.399	
		50.00	.372	.445	.530	.599	.645	.447	.349	
		55.00	.329	.465	.541	.624	.678	.263	.257	
		60.00	.374	.470	.543	.559	.669	.113	.169	
		65.00	.420	.439	.530	.602	.347	.000	.116	
		70.00	.368						.098	
		75.00	.361	.484	.519	.290	.102	.061	.076	
LOWER SURFACE		80.00	.408	.488	.374	.173	.042	.074	.068	
		85.00	.448	.437	.224	.100	.010	.085	.061	
		90.00	.437	.334	.142	.039	.050	.096	.047	
		95.00	.413	.182	.080	.004	.077	.108	.034	
		1.25	.397	.452	.425	.387	.409		.396	
		2.50	.398	.380	.345	.307	.334		.321	
		5.00	.356	.309	.265	.287	.250	.360	.226	
		7.50	.330	.253	.214	.222	.192	.262	.148	
		10.00	.296	.221	.180	.160	.153	.206	.092	
		15.00	.246	.154	.126	.104	.112	.162	.007	
		20.00	.186	.111	.078	.064	.061	.068	.089	
		25.00	.181	.074	.039	.031	.028	.025	.193	
		30.00	.121	.041	.005	.004	.000	.007	.246	
		35.00	.082	.011	.021	.029	.024	.043	.288	
		40.00	.068	.013	.077	.053	.048	.076	.286	
		45.00	.031	.045	.072	.076	.065	.113	.251	
		50.00	.000	.071	.091	.089	.070	.149	.258	
		55.00	.026	.082	.101	.096	.080	.172	.169	
		60.00	.043	.090	.106	.096	.079	.179	.088	
	65.00	.060	.095	.113	.084	.067	.159	.049		
	70.00	.065						.026		
	75.00	.078	.105	.111	.069	.046	.070	.005		
	80.00	.072	.099	.089	.044	.040	.015	.011		
	85.00	.100	.105	.071	.024	.020	.013	.022		
	90.00	.111	.112	.059	.031	.003	.046	.040		
	95.00	.150	.099	.034	.020	.037	.080	.056		
		$M = 0.94 \quad \alpha = 7.9^\circ$								
UPPER SURFACE		0.00	.022	.202	.282	.033	.561	.675	.627	
		1.25	.117	1.073	1.275	1.253	1.286	1.259	.731	
		2.50	.242	1.067	1.219	1.282	1.296	1.317	.734	
		5.00	.343	1.023	1.207	1.228	1.241	1.228	.759	
		7.50	.375	.908	1.139	1.173	1.196	1.216	.742	
		10.00	.372	.843	1.099	1.131	1.173	1.186	.726	
		15.00	.379	.684	.921	1.092	1.130	1.152	.729	
		20.00	.360	.528	.646	1.020	1.101	1.104	.729	
		25.00	.376	.441	.554	.979	1.079	.985	.707	
		30.00	.379	.425	.533	.936	1.028	.993	.683	
		35.00	.369	.435	.537	.856	1.005	.915	.635	
		40.00	.373	.455	.548	.826	.978	.886	.590	
		45.00	.381	.469	.563	.823	.849	.852	.552	
		50.00	.423	.490	.584	.837	.854	.805	.523	
		55.00	.378	.507	.583	.856	.866	.882	.493	
		60.00	.427	.522	.590	.800	.863	.854	.460	
		65.00	.475	.486	.581	.649	.842	.493	.418	
		70.00	.426						.398	
		75.00	.414	.533	.542	.345	.130	.426	.378	
LOWER SURFACE		80.00	.465	.514	.414	.213	.077	.380	.360	
		85.00	.492	.492	.274	.125	.054	.328	.330	
		90.00	.493	.359	.190	.060	.041	.262	.312	
		95.00	.458	.200	.127	.019	.028	.211	.288	
		1.25	.430	.528	.490	.449	.453		.431	
		2.50	.460	.466	.424	.379	.394	.408	.375	
		5.00	.443	.395	.345	.345	.314	.319	.284	
		7.50	.424	.335	.297	.288	.256	.264	.206	
		10.00	.388	.301	.257	.229	.217	.219	.150	
		15.00	.335	.229	.205	.176	.168	.170	.043	
		20.00	.269	.185	.154	.133	.118	.114	.029	
		25.00	.259	.145	.114	.096	.078	.071	.126	
		30.00	.195	.109	.076	.060	.049	.034	.186	
		35.00	.151	.077	.046	.033	.020	.005	.229	
		40.00	.133	.053	.020	.005	.005	.041	.240	
		45.00	.096	.021	.009	.021	.030	.079	.246	
		50.00	.060	.010	.037	.044	.045	.120	.240	
		55.00	.031	.021	.045	.060	.058	.145	.203	
		60.00	.011	.034	.086	.064	.058	.161	.166	
	65.00	.000	.045	.066	.064	.054	.160	.133		
	70.00	.007						.117		
	75.00	.030	.058	.080	.062	.038	.120	.104		
	80.00	.019	.060	.072	.038	.038	.082	.080		
	85.00	.080	.085	.063	.020	.028	.074	.075		
	90.00	.063	.095	.063	.011	.019	.067	.083		
	95.00	.107	.098	.088	.018	.005	.076	.103		

TABLE I

BASIC WING

PERCENT CHORD		PRESSURE COEFFICIENT, P, AT:							
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
M = 0.94 α = 11.4°									
UPPER SURFACE	1.00	.011	-	.585	-.648	-.253	-.671	-.738	-.564
	1.25	.264	1.242	1.277	1.156	.964	.699	.506	
	2.50	.405	1.228	1.208	1.155	.970	.705	.505	
	5.00	.549	1.261	1.182	1.142	.954	.680	.538	
	7.50	.579	1.222	1.089	1.142	.957	.686	.502	
	10.00	.577	1.174	1.038	1.145	.951	.677	.472	
	15.00	.584	1.079	.991	1.101	.959	.670	.464	
	20.00	.536	1.019	.941	1.059	.930	.638	.453	
	25.00	.527	.962	.904	1.027	.887	.604	.419	
	30.00	.516	.851	.871	.986	.866	.607	.420	
	35.00	.496	.760	.840	.933	.834	.593	.405	
	40.00	.494	.676	.813	.890	.791	.572	.383	
	45.00	.499	.584	.793	.847	.740	.553	.379	
	50.00	.499	.535	.781	.799	.702	.543	.370	
	55.00	.530	.564	.763	.757	.670	.522	.368	
	60.00	.483	.591	.765	.731	.644	.503	.389	
	65.00	.530	.580	.752	.699	.645	.503	.390	
	70.00	.574	-	-	-	-	.460	.360	
	75.00	.525	.601	.610	.622	.564	.440	.354	
	80.00	.530	.591	.536	.592	.507	.426	.360	
85.00	.582	.567	.460	.551	.475	.413	.346		
90.00	.571	.404	.405	.505	.447	.397	.341		
95.00	.530	.234	.362	.471	-	-	.445		
LOWER SURFACE	1.25	.408	.607	.540	.521	.492	.469	.345	
	2.50	.508	.574	.516	.479	.476	.396	.259	
	5.00	.566	.513	.458	.449	.417	.344	.205	
	7.50	.581	.461	.410	.397	.365	.295	.113	
	10.00	.547	.426	.374	.348	.326	.247	.012	
	15.00	.468	.356	.318	.289	.271	.187	.059	
	20.00	.411	.308	.265	.242	.219	.135	.120	
	25.00	.366	.266	.220	.197	.174	.088	.166	
	30.00	.318	.228	.184	.163	.137	.039	.191	
	35.00	.273	.191	.149	.127	.093	.004	.209	
	40.00	.250	.162	.116	.093	.065	.054	.231	
	45.00	.209	.127	.088	.062	.032	.105	.241	
	50.00	.173	.094	.057	.032	.000	.146	.239	
	55.00	.140	.075	.038	.009	.031	.181	.257	
	60.00	.111	.035	.020	.031	.072	.212	.255	
	65.00	.096	.038	.003	-	-	.246	.278	
	70.00	.079	-	.034	-.063	-.106	.270	.277	
	75.00	.026	.015	.037	-.077	-.140	.292	.277	
	80.00	.048	.008	.041	-.096	-.164	.311	.277	
	85.00	.024	.019	.066	-.137	-.208	.315	.286	
90.00	.008	.058	.098	-.179	.254	-	.286		
95.00	.001	.090	.098	-	-	-	.286		
M = 0.94 α = 15.7°									
UPPER SURFACE	1.00	.018	-.953	1.047	-.776	-.759	-.600	-.528	
	1.25	.418	1.383	1.113	.861	.755	.594	.512	
	2.50	.624	1.361	1.105	.860	.758	.596	.509	
	5.00	.806	1.375	1.105	.856	.745	.583	.513	
	7.50	.833	1.385	1.080	.862	.741	.591	.510	
	10.00	.829	1.369	1.081	.878	.735	.583	.503	
	15.00	.820	1.311	1.077	.864	.734	.579	.504	
	20.00	.756	1.302	1.041	.851	.720	.571	.499	
	25.00	.714	1.282	1.010	.826	.703	.551	.493	
	30.00	.676	1.262	.978	.828	.703	.556	.497	
	35.00	.643	1.230	.948	.808	.696	.545	.498	
	40.00	.631	1.190	.928	.792	.681	.539	.491	
	45.00	.628	.721	.943	.780	.681	.541	.488	
	50.00	.653	.583	.930	.771	.674	.541	.485	
	55.00	.601	.599	.909	.760	.660	.533	.480	
	60.00	.614	.640	.888	.757	.654	.526	.478	
	65.00	.607	.640	.920	.756	.667	.537	.488	
	70.00	.540	-	-	-	-	.512	.463	
	75.00	.577	.679	.832	.715	.630	.503	.454	
	80.00	.574	.644	.808	.706	.618	.496	.458	
85.00	.610	.555	.773	.692	.605	.494	.448		
90.00	.603	.397	.750	.669	.593	.485	.438		
95.00	.505	.339	.735	.655	.580	-	.422		
LOWER SURFACE	1.25	.358	.642	.542	.567	.477	.496	.381	
	2.50	.502	.663	.581	.548	.453	.454	.316	
	5.00	.663	.632	.555	.527	.447	.411	.267	
	7.50	.731	.586	.521	.487	.410	.361	.181	
	10.00	.695	.555	.489	.446	.359	.319	.081	
	15.00	.629	.486	.440	.392	.302	.260	.005	
	20.00	.546	.438	.392	.346	.259	.208	.071	
	25.00	.511	.394	.353	.314	.214	.156	.128	
	30.00	.448	.353	.320	.280	.174	.104	.176	
	35.00	.397	.315	.284	.244	.134	.057	.204	
	40.00	.370	.280	.251	.211	.106	.005	.240	
	45.00	.326	.243	.215	.175	.074	.016	.286	
	50.00	.287	.207	.178	.138	.043	.015	.307	
	55.00	.252	.184	.156	.116	.040	.015	.326	
	60.00	.219	.160	.132	.092	-	.015	.333	
	65.00	.206	.134	.107	.067	-	.015	.363	
	70.00	.176	-	.094	.046	.094	.243	.390	
	75.00	.110	.094	.006	.073	.141	.282	.399	
	80.00	.122	.080	.034	.104	.173	.351	.407	
	85.00	.090	.037	.091	.158	.241	.393	.407	
90.00	.048	.010	.153	.222	.286	-	.407		
95.00	.031	.070	-	-	-	-	.407		

TABLE 1

BASIC WING

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:							
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
		M = 0.94 α = 20.0°								
UPPER SURFACE		.00	-.062	-1.170	-.984	-.899	-.821	-.638	-.573	
		1.25	-.507	-1.251	-.965	-.791	-.818	-.632	-.566	
		2.50	-.773	-1.237	-.931	-.790	-.808	-.641	-.563	
		5.00	-1.031	-1.232	-.965	-.795	-.742	-.625	-.570	
		7.50	-1.059	-1.218	-.953	-.800	-.725	-.636	-.564	
		10.00	-1.054	-1.200	-.960	-.820	-.718	-.638	-.558	
		15.00	-1.027	-1.154	-.957	-.809	-.718	-.631	-.555	
		20.00	-.904	-1.140	-.941	-.811	-.715	-.620	-.556	
		25.00	-.800	-1.116	-.926	-.805	-.707	-.602	-.555	
		30.00	-.734	-1.100	-.916	-.794	-.705	-.612	-.561	
		35.00	-.693	-1.078	-.905	-.779	-.700	-.610	-.566	
		40.00	-.688	-1.055	-.897	-.770	-.693	-.605	-.567	
		45.00	-.693	-1.027	-.889	-.762	-.689	-.601	-.566	
		50.00	-.718	-.997	-.884	-.758	-.682	-.603	-.564	
		55.00	-.677	-.958	-.870	-.753	-.673	-.598	-.561	
		60.00	-.704	-.932	-.858	-.749	-.670	-.596	-.552	
		65.00	-.723	-.892	-.864	-.749	-.675	-.600	-.556	
		70.00	-.673						-.535	
		75.00	-.715	-.836	-.822	-.730	-.648	-.584	-.533	
LOWER SURFACE		80.00	-.689	-.811	-.812	-.720	-.645	-.578	-.519	
		85.00	-.686	-.760	-.793	-.715	-.638	-.572	-.512	
		90.00	-.666	-.720	-.785	-.702	-.629	-.567	-.499	
		95.00	-.559	-.683	-.779	-.693	-.620	-.558	-.483	
		1.25	-.253	-.633	-.512	-.564	-.448		-.390	
		2.50	-.484	-.712	-.615	-.579	-.535	-.516	-.452	
		5.00	-.735	-.716	-.684	-.581	-.544	-.504	-.434	
		7.50	-.836	-.688	-.607	-.556	-.523	-.479	-.384	
		10.00	-.807	-.658	-.583	-.529	-.497	-.430	-.344	
		15.00	-.750	-.604	-.539	-.486	-.452	-.403	-.328	
		20.00	-.674	-.560	-.489	-.444	-.399	-.348	-.289	
		25.00	-.634	-.512	-.446	-.396	-.357	-.299	-.253	
		30.00	-.576	-.473	-.405	-.363	-.313	-.247	-.202	
		35.00	-.526	-.430	-.370	-.321	-.271	-.195	-.158	
		40.00	-.494	-.398	-.331	-.275	-.229	-.146	-.122	
		45.00	-.452	-.360	-.294	-.237	-.186	-.095	-.132	
		50.00	-.411	-.321	-.258	-.199	-.148	-.041	-.171	
		55.00	-.374	-.293	-.226	-.163	-.107	-.010	-.203	
		60.00	-.342	-.263	-.197	-.129	-.073	-.054	-.223	
	65.00	-.325	-.240	-.160	-.094	-.045	-.098	-.254		
	70.00	-.284						-.282		
	75.00	-.224	-.184	-.081	-.023	-.022	-.165	-.301		
	80.00	-.214	-.163	-.080	-.007	-.070	-.217	-.328		
	85.00	-.176	-.102	-.011	-.046	-.106	-.257	-.364		
	90.00	-.121	-.052	-.057	-.108	-.167	-.297	-.392		
	95.00	-.084	-.039	-.125	-.172	-.237	-.352	-.421		
		M = 0.94 α = 24.2°								
UPPER SURFACE		.00	-.200	-.998	-.955	-.899	-.789	-.717	-.647	
		1.25	-.456	-.947	-.893	-.863	-.756	-.695	-.606	
		2.50	-.682	-.941	-.867	-.871	-.751	-.704	-.600	
		5.00	-.962	-.947	-.868	-.896	-.752	-.694	-.567	
		7.50	-.940	-.943	-.874	-.898	-.748	-.719	-.579	
		10.00	-.942	-.950	-.887	-.825	-.749	-.738	-.597	
		15.00	-.940	-.937	-.889	-.788	-.736	-.702	-.588	
		20.00	-.949	-.950	-.895	-.783	-.741	-.674	-.588	
		25.00	-.946	-.958	-.892	-.808	-.748	-.660	-.604	
		30.00	-.910	-.940	-.883	-.793	-.741	-.660	-.597	
		35.00	-.877	-.940	-.879	-.793	-.735	-.659	-.596	
		40.00	-.834	-.936	-.868	-.791	-.732	-.660	-.602	
		45.00	-.815	-.931	-.857	-.788	-.727	-.661	-.601	
		50.00	-.824	-.917	-.845	-.776	-.721	-.649	-.601	
		55.00	-.776	-.907	-.837	-.776	-.723	-.654	-.603	
		60.00	-.752	-.904	-.841	-.764	-.719	-.656	-.602	
		65.00	-.728	-.849	-.796	-.739	-.683	-.614	-.587	
		70.00	-.741						-.601	
		75.00	-.803	-.864	-.822	-.755	-.700	-.634	-.574	
LOWER SURFACE		80.00	-.762	-.853	-.821	-.752	-.705	-.629	-.567	
		85.00	-.748	-.829	-.810	-.745	-.696	-.618	-.555	
		90.00	-.720	-.816	-.792	-.736	-.687	-.609	-.545	
		95.00	-.657	-.802	-.782	-.729	-.677	-.576	-.509	
		1.25	-.154	-.601	-.460	-.500	-.384		-.330	
		2.50	-.408	-.734	-.617	-.585	-.526	-.507	-.440	
		5.00	-.735	-.773	-.667	-.610	-.570	-.532	-.454	
		7.50	-.864	-.761	-.664	-.607	-.567	-.524	-.424	
		10.00	-.872	-.742	-.648	-.592	-.550	-.493	-.394	
		15.00	-.832	-.691	-.616	-.559	-.513	-.464	-.340	
		20.00	-.769	-.650	-.570	-.521	-.470	-.416	-.281	
		25.00	-.725	-.609	-.529	-.481	-.425	-.368	-.263	
		30.00	-.675	-.569	-.493	-.447	-.387	-.321	-.206	
		35.00	-.625	-.528	-.455	-.404	-.348	-.271	-.154	
		40.00	-.591	-.496	-.418	-.363	-.307	-.224	-.100	
		45.00	-.548	-.454	-.379	-.323	-.262	-.174	-.058	
		50.00	-.512	-.418	-.340	-.281	-.224	-.122	-.008	
		55.00	-.472	-.384	-.304	-.241	-.182	-.071	-.143	
		60.00	-.437	-.353	-.274	-.206	-.145	-.027	-.166	
	65.00	-.484	-.328	-.236	-.178	-.122	-.003	-.176		
	70.00	-.365						-.200		
	75.00	-.314	-.258	-.146	-.090	-.044	-.070	-.252		
	80.00	-.283	-.228	-.107	-.055	-.010	-.161	-.239		
	85.00	-.256	-.189	-.089	-.016	-.051	-.198	-.286		
	90.00	-.172	-.100	-.008	-.053	-.120	-.206	-.332		
	95.00	-.124	-.009	-.082	-.126	-.189	-.285	-.362		

TABLE I

BASIC WING

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:							
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
		M = 0.98 α = 0.0°								
UPPER SURFACE		.00	.036	.526	.442	.707	.375	.383	.377	
		1.25	.229	.146	.201	.204	.202	.220	.266	
		2.50	.159	.103	.181	.200	.238	.320	.266	
		5.00	.082	.074	.129	.175	.188	.223	.234	
		7.50	.045	.082	.132	.158	.219	.244	.266	
		10.00	.027	.094	.181	.159	.239	.267	.301	
		15.00	.010	.104	.166	.212	.247	.303	.316	
		20.00	.037	.126	.186	.231	.278	.318	.406	
		25.00	.071	.145	.214	.263	.311	.341	.424	
		30.00	.089	.159	.236	.283	.326	.367	.436	
		35.00	.092	.184	.254	.303	.347	.390	.446	
		40.00	.111	.211	.276	.328	.369	.419	.462	
		45.00	.140	.226	.302	.342	.391	.440	.482	
		50.00	.180	.254	.314	.361	.428	.467	.494	
		55.00	.158	.276	.320	.390	.453	.516	.465	
		60.00	.197	.275	.335	.344	.450	.493	.172	
		65.00	.238	.254	.334	.376	.389	.543	.048	
		70.00	.195						.042	
		75.00	.221	.299	.346	.300	.231	.146	.083	
	80.00	.255	.287	.292	.208	.171	.026	.108		
	85.00	.280	.276	.189	.150	.123	.045	.130		
	90.00	.282	.240	.137	.103	.079	.094	.143		
	95.00	.264	.155	.083	.056	.048	.131	.159		
LOWER SURFACE		1.25	.262	.038	.022	.072	.114		.180	
		2.50	.212	.042	.003	.052	.102	.105	.177	
		5.00	.145	.021	.029	.020	.135	.163	.187	
		7.50	.115	.004	.054	.071	.170	.191	.234	
		10.00	.091	.021	.073	.103	.191	.214	.253	
		15.00	.049	.058	.106	.155	.204	.232	.321	
		20.00	.002	.071	.133	.184	.231	.268	.361	
		25.00	.023	.094	.153	.205	.258	.287	.404	
		30.00	.052	.124	.186	.230	.287	.322	.425	
		35.00	.067	.150	.208	.259	.306	.353	.437	
		40.00	.079	.169	.237	.281	.331	.378	.448	
		45.00	.109	.196	.263	.309	.368	.408	.455	
		50.00	.155	.228	.287	.336	.389	.424	.455	
		55.00	.166	.237	.290	.347	.406	.449	.441	
		60.00	.182	.246	.296	.349	.394	.431	.230	
		65.00	.212	.253	.310	.353	.351	.342	.047	
		70.00	.193						.040	
		75.00	.215	.260	.306	.254	.187	.182	.072	
		80.00	.201	.245	.260	.180	.153	.046	.089	
	85.00	.248	.237	.186	.133	.122	.036	.105		
	90.00	.255	.227	.136	.097	.101	.069	.131		
	95.00	.301	.154	.071	.051	.059	.113	.147		
		M = 0.98 α = 3.9°								
UPPER SURFACE		.00	.041	.305	.191	.423	.042	.027	.105	
		1.25	.107	.750	.848	.809	.823	.867	.917	
		2.50	.011	.638	.782	.856	.863	.901	.902	
		5.00	.068	.277	.528	.751	.789	.809	.783	
		7.50	.105	.266	.380	.457	.784	.778	.779	
		10.00	.108	.260	.389	.349	.819	.746	.773	
		15.00	.136	.253	.381	.374	.400	.646	.745	
		20.00	.148	.252	.319	.379	.439	.474	.683	
		25.00	.178	.254	.310	.398	.439	.472	.487	
		30.00	.194	.281	.335	.409	.473	.511	.348	
		35.00	.180	.281	.362	.421	.485	.529	.483	
		40.00	.205	.300	.379	.442	.507	.553	.510	
		45.00	.226	.312	.402	.465	.524	.576	.544	
		50.00	.244	.338	.424	.486	.551	.599	.567	
		55.00	.238	.350	.432	.510	.560	.645	.574	
		60.00	.274	.365	.429	.439	.593	.665	.506	
		65.00	.318	.335	.420	.483	.562	.559	.308	
		70.00	.271						.122	
		75.00	.268	.383	.442	.431	.317	.172	.052	
	80.00	.318	.368	.372	.263	.228	.105	.004		
	85.00	.354	.357	.237	.183	.168	.055	.052		
	90.00	.353	.282	.158	.118	.121	.004	.075		
	95.00	.321	.166	.099	.062	.083	.046	.089		
LOWER SURFACE		1.25	.368	.356	.332	.284	.290	.222	.278	
		2.50	.348	.288	.259	.231	.215	.121	.198	
		5.00	.292	.220	.184	.204	.131	.068	.118	
		7.50	.261	.176	.142	.143	.076	.030	.041	
		10.00	.234	.152	.107	.098	.039	.015	.006	
		15.00	.184	.087	.058	.021	.007	.015	.096	
		20.00	.130	.058	.015	.015	.046	.060	.148	
		25.00	.135	.024	.023	.048	.082	.092	.260	
		30.00	.069	.006	.037	.071	.111	.110	.306	
		35.00	.039	.035	.079	.108	.140	.135	.342	
		40.00	.026	.057	.111	.131	.167	.158	.347	
		45.00	.010	.096	.135	.161	.196	.192	.355	
		50.00	.044	.124	.160	.181	.199	.229	.373	
		55.00	.065	.132	.166	.195	.193	.257	.377	
		60.00	.081	.144	.170	.208	.178	.283	.364	
		65.00	.109	.152	.182	.195	.185	.292	.304	
		70.00	.109						.805	
		75.00	.123	.158	.190	.145	.118	.273	.123	
		80.00	.119	.148	.156	.114	.120	.207	.025	
	85.00	.144	.157	.122	.093	.113	.131	.043		
	90.00	.149	.161	.100	.073	.110	.053	.079		
	95.00	.210	.126	.062	.046	.096	.013	.099		

TABLE 1

BASIC WING

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:								
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2		
		M = 0.98 α = 5.9°									
UPPER SURFACE		0.00	.038	-	-	.006	-	-	-	.162	
		1.25	.017	-	.924	1.044	-	1.032	-	1.086	
		2.50	.087	-	.926	1.014	-	1.058	-	1.079	
		5.00	.175	-	.588	.957	-	1.000	-	.974	
		7.50	.210	-	.466	.880	-	.956	-	.973	
		10.00	.219	-	.419	.801	-	.927	-	.964	
		15.00	.235	-	.366	.422	-	.875	-	.945	
		20.00	.228	-	.351	.380	-	.854	-	.926	
		25.00	.254	-	.345	.394	-	.783	-	.878	
		30.00	.264	-	.343	.409	-	.577	-	.819	
		35.00	.262	-	.352	.426	-	.515	-	.811	
		40.00	.272	-	.368	.444	-	.533	-	.811	
		45.00	.290	-	.385	.464	-	.560	-	.808	
		50.00	.331	-	.402	.484	-	.591	-	.768	
		55.00	.290	-	.420	.494	-	.626	-	.697	
		60.00	.335	-	.430	.501	-	.652	-	.554	
		65.00	.378	-	.399	.482	-	.646	-	.406	
	LOWER SURFACE		0.00	.038	-	-	.006	-	-	-	.359
		1.25	.017	.443	-	.503	-	.414	.247	.301	
		2.50	.087	.432	-	.427	.325	.281	.191	.242	
		5.00	.175	.416	-	.278	.224	.216	.145	.149	
		7.50	.210	.331	-	.185	.157	.161	.089	.108	
		10.00	.219	.190	-	.125	.104	.123	.045	.104	
		15.00	.235	-	.460	.429	.379	.387	-	.378	
		20.00	.228	-	.388	.352	.309	.312	.327	.290	
		25.00	.254	-	.320	.272	.282	.226	.227	.212	
		30.00	.264	-	.267	.223	.225	.170	.174	.131	
		35.00	.262	-	.234	.189	.162	.129	.133	.081	
		40.00	.272	-	.168	.136	.097	.087	.087	.016	
		45.00	.290	-	.126	.085	.056	.032	.037	.094	
		50.00	.331	-	.088	.046	.022	.006	.003	.203	
		55.00	.290	-	.056	.012	.007	.037	.036	.260	
		60.00	.335	-	.024	.017	.040	.061	.076	.304	
		65.00	.378	-	.003	.045	.070	.087	.103	.328	
		70.00	.331	.103	-	.069	.096	.110	.141	.349	
	75.00	.071	.100	.140	.093	.122	.119	.184	.369		
	80.00	.066	.112	.100	.105	.138	.136	.219	.381		
	85.00	.095	.125	.088	.112	.140	.140	.252	.375		
	90.00	.101	.115	.070	.121	.140	.126	.267	.357		
	95.00	.151	-	-	-	-	-	-	.324		
		M = 0.98 α = 7.8°									
UPPER SURFACE		0.00	.030	-	.181	-	.133	-	.429	.488	.440
		1.25	.061	1.030	1.148	1.111	1.139	1.116	1.176	1.171	
		2.50	.184	1.023	1.105	1.149	1.160	1.176	1.176	1.171	
		5.00	.285	.910	1.093	1.094	1.108	1.088	1.088	1.084	
		7.50	.318	.749	1.028	1.042	1.066	1.082	1.079	1.079	
		10.00	.320	.674	1.001	1.006	1.044	1.052	1.069	1.069	
		15.00	.330	.557	.905	.974	1.005	1.033	1.042	1.042	
		20.00	.315	.475	.634	.917	.984	1.007	1.023	1.023	
		25.00	.334	.428	.495	.902	.973	.975	.975	.975	
		30.00	.335	.409	.468	.717	.927	.984	.930	.930	
		35.00	.329	.410	.477	.584	.906	.972	.915	.915	
		40.00	.336	.423	.491	.555	.894	.961	.905	.905	
		45.00	.346	.437	.511	.557	.823	.949	.893	.893	
		50.00	.387	.457	.531	.574	.714	.936	.839	.839	
		55.00	.343	.470	.544	.605	.680	.951	.630	.630	
		60.00	.368	.484	.549	.547	.674	.764	.659	.659	
		65.00	.434	.453	.541	.606	.666	.609	.663	.663	
	LOWER SURFACE		0.00	.030	-	.181	-	.133	-	.429	.488
		1.25	.061	1.030	1.148	1.111	1.139	1.116	1.176	1.171	
		2.50	.184	1.023	1.105	1.149	1.160	1.176	1.176	1.171	
		5.00	.285	.910	1.093	1.094	1.108	1.088	1.088	1.084	
		7.50	.318	.749	1.028	1.042	1.066	1.082	1.079	1.079	
		10.00	.320	.674	1.001	1.006	1.044	1.052	1.069	1.069	
		15.00	.330	.557	.905	.974	1.005	1.033	1.042	1.042	
		20.00	.315	.475	.634	.917	.984	1.007	1.023	1.023	
		25.00	.334	.428	.495	.902	.973	.975	.975	.975	
		30.00	.335	.409	.468	.717	.927	.984	.930	.930	
		35.00	.329	.410	.477	.584	.906	.972	.915	.915	
		40.00	.336	.423	.491	.555	.894	.961	.905	.905	
		45.00	.346	.437	.511	.557	.823	.949	.893	.893	
		50.00	.387	.457	.531	.574	.714	.936	.839	.839	
		55.00	.343	.470	.544	.605	.680	.951	.630	.630	
		60.00	.368	.484	.549	.547	.674	.764	.659	.659	
		65.00	.434	.453	.541	.606	.666	.609	.663	.663	

TABLE I

BASIC WING

		PRESSURE COEFFICIENT, P , AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
$M = 0.98 \quad \alpha = 11.5^\circ$								
UPPER SURFACE	PERCENT CHORD							
	.00	.006	.465	.511	.121	.758	.861	.794
	1.25	.180	.125	.176	.284	.310	.1080	.1017
	2.50	.313	.112	.139	.291	.313	.1086	.1027
	5.00	.456	.142	.152	.254	.281	.1045	.992
	7.50	.489	.105	.1083	.218	.252	.1067	.910
	10.00	.493	.052	.087	.198	.230	.1048	.837
	15.00	.499	.062	.081	.137	.190	.1030	.836
	20.00	.460	.07	.061	.076	.147	.990	.837
	25.00	.454	.049	.017	.052	.118	.938	.819
	30.00	.446	.075	.070	.098	.1093	.933	.804
	35.00	.426	.001	.069	.022	.1067	.894	.772
	40.00	.424	.460	.755	.081	1.035	.844	.736
	45.00	.430	.441	.744	.856	.954	.798	.702
	50.00	.462	.474	.737	.814	.864	.751	.673
	55.00	.421	.501	.719	.807	.839	.707	.640
	60.00	.464	.525	.719	.755	.828	.665	.610
	65.00	.504	.520	.716	.686	.812	.653	.602
LOWER SURFACE	70.00	.464						.577
	75.00	.459	.543	.660	.531	.781	.620	.545
	80.00	.498	.540	.579	.459	.773	.610	.527
	85.00	.517	.533	.482	.385	.737	.594	.522
	90.00	.516	.450	.404	.320	.683	.581	.510
	95.00	.496	.255	.381	.276	.611	.556	.501
	1.25	.439	.640	.563	.558	.503		.460
	2.50	.545	.602	.536	.495	.478	.481	.433
	5.00	.598	.541	.471	.465	.417	.408	.356
	7.50	.611	.467	.427	.408	.364	.356	.281
	10.00	.578	.454	.391	.352	.325	.305	.231
	15.00	.515	.383	.337	.299	.273	.259	.145
	20.00	.440	.336	.282	.253	.219	.203	.002
	25.00	.415	.293	.241	.209	.175	.157	.046
	30.00	.351	.255	.203	.179	.140	.110	.118
	35.00	.302	.220	.170	.140	.104	.065	.176
	40.00	.281	.188	.137	.104	.072	.022	.217
	45.00	.239	.153	.109	.072	.033	.023	.248
	50.00	.206	.119	.079	.043	.002	.073	.277
	55.00	.169	.100	.062	.019	.002	.117	.298
	60.00	.141	.081	.045	.008	.003	.154	.300
	65.00	.131	.067	.021	.014	.039	.181	.310
	70.00	.108						.312
	75.00	.045	.045	.022	.048	.082	.229	.316
	80.00	.072	.040	.027	.060	.108	.259	.325
	85.00	.045	.001	.030	.081	.123	.288	.334
	90.00	.025	.033	.051	.108	.158	.307	.364
	95.00	.021	.080	.078	.134	.190	.344	.392
$M = 0.98 \quad \alpha = 15.9^\circ$								
UPPER SURFACE	PERCENT CHORD							
	.00	.157	.839	.933	.736	1.040	.665	.503
	1.25	.325	.1263	.1193	.1152	.937	.660	.481
	2.50	.532	.1241	.1237	.1128	.938	.662	.482
	5.00	.704	.1287	.1171	.1111	.933	.641	.462
	7.50	.741	.1263	.1150	.1114	.933	.647	.497
	10.00	.737	.1247	.1168	.1118	.930	.642	.509
	15.00	.729	.1189	.1176	.1098	.925	.632	.528
	20.00	.673	.1181	.1177	.1083	.911	.625	.547
	25.00	.631	.1167	.1173	.1089	.892	.608	.563
	30.00	.597	.1158	.1170	.1078	.878	.622	.578
	35.00	.565	.1145	.1164	.1059	.858	.619	.587
	40.00	.556	.1129	.1154	.1035	.834	.615	.589
	45.00	.556	.1079	.1144	.1009	.811	.610	.592
	50.00	.581	.596	.1135	.980	.791	.608	.593
	55.00	.546	.577	.1117	.954	.770	.605	.588
	60.00	.580	.593	.1124	.930	.751	.604	.580
	65.00	.613	.596	.1118	.896	.739	.611	.584
LOWER SURFACE	70.00	.582						.580
	75.00	.606	.640	.1058	.825	.696	.611	.570
	80.00	.623	.638	.962	.810	.684	.610	.562
	85.00	.634	.630	.862	.787	.670	.607	.558
	90.00	.629	.491	.780	.756	.655	.604	.548
	95.00	.562	.314	.564	.727	.641	.597	.540
	1.25	.378	.678	.574	.589	.502		.453
	2.50	.539	.694	.604	.565	.537	.529	.473
	5.00	.702	.660	.578	.550	.513	.488	.422
	7.50	.764	.617	.546	.509	.475	.448	.363
	10.00	.729	.583	.515	.470	.441	.436	.321
	15.00	.658	.518	.462	.421	.395	.362	.235
	20.00	.580	.471	.410	.374	.342	.307	.141
	25.00	.546	.427	.366	.333	.300	.255	.018
	30.00	.483	.385	.328	.296	.261	.208	
	35.00	.433	.346	.292	.255	.224	.162	.071
	40.00	.401	.316	.257	.222	.190	.116	.113
	45.00	.362	.276	.224	.186	.147	.066	.154
	50.00	.325	.244	.192	.155	.117	.014	.198
	55.00	.288	.219	.167	.127	.084	.034	.226
	60.00	.256	.196	.144	.106	.056	.077	.246
	65.00	.247	.175	.112	.082	.030	.113	.270
	70.00	.208						.293
	75.00	.148	.135	.062	.039	.023	.179	.313
	80.00	.119	.119	.049	.018	.066	.193	.323
	85.00	.113	.087	.039	.016	.099	.243	.323
	90.00	.063	.021	.020	.071	.107	.292	.323
	95.00	.039	.052	.007	.133	.212	.341	.432

TABLE I

BASIC WING

		PRESSURE COEFFICIENT, P, AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
M = 0.98 $\alpha = 20.2^\circ$								
UPPER SURFACE	.00	.258	1.067	1.185	.931	.808	.727	.662
	1.25	.408	1.217	1.070	.850	.787	.710	.638
	2.50	.685	1.206	1.088	.851	.780	.714	.635
	5.00	.933	1.211	1.064	.856	.780	.694	.612
	7.50	.951	1.210	1.047	.853	.780	.708	.625
	10.00	.939	1.214	1.063	.843	.773	.706	.632
	15.00	.918	1.183	1.050	.863	.773	.692	.627
	20.00	.841	1.191	1.057	.856	.786	.684	.625
	25.00	.769	1.200	1.052	.856	.778	.687	.631
	30.00	.714	1.152	1.030	.854	.772	.680	.620
	35.00	.674	1.160	1.016	.847	.770	.675	.615
	40.00	.659	1.158	.992	.840	.764	.668	.613
	45.00	.654	1.123	.970	.829	.759	.668	.610
	50.00	.674	1.014	.952	.825	.757	.667	.602
	55.00	.645	.860	.931	.818	.750	.667	.582
	60.00	.664	.747	.930	.791	.725	.641	.553
	65.00	.684	.689	.885				.589
	70.00	.649		.892	.797	.728	.653	.583
	75.00	.673	.660	.886	.793	.727	.647	.568
	80.00	.664	.583	.865	.781	.715	.639	.560
LOWER SURFACE	85.00	.633	.544	.882	.769	.696	.624	.553
	90.00	.597	.460	.850	.761	.686	.610	.553
	95.00	.546	.385					.410
	1.25	.279	.665	.542	.569	.464	.534	.474
	2.50	.517	.740	.641	.602	.553	.528	.459
	5.00	.768	.742	.650	.603	.543	.501	.417
	7.50	.868	.712	.631	.580	.517	.463	.376
	10.00	.836	.683	.608	.556	.476	.429	.303
	15.00	.778	.626	.564	.513	.427	.379	.208
	20.00	.705	.585	.516	.473	.383	.326	.134
	25.00	.685	.542	.473	.429	.343	.282	.060
	30.00	.608	.500	.435	.393	.305	.236	.006
	35.00	.528	.460	.398	.351	.305	.184	.056
	40.00	.425	.428	.361	.313	.264	.135	.095
	45.00	.389	.389	.329	.275	.222	.084	.134
	50.00	.449	.351	.291	.237	.184	.033	.169
	55.00	.408	.325	.262	.200	.147	.008	.198
	60.00	.376	.298	.235	.171	.111	.044	.210
	65.00	.344	.277	.200	.141	.089		.259
	70.00	.315		.131	.071	.024	.111	.270
	75.00	.283	.223	.100	.042	.021	.181	.273
	80.00	.237	.201	.064	.008	.062	.218	.320
	85.00	.199	.143	.008	.053	.120	.234	.353
	90.00	.140	.094	.008	.116	.180	.296	.389
	95.00	.094	.010	.061				
M = 1.00 $\alpha = 0.0^\circ$								
UPPER SURFACE	.00	.074	.582	.503	.747	.472	.473	.434
	1.25	.288	.006	.045	.074	.047	.063	.123
	2.50	.227	.015	.034	.082	.121	.180	.129
	5.00	.157	.011	.044	.097	.138	.187	.148
	7.50	.117	.005	.050	.096	.138	.161	.168
	10.00	.097	.021	.075	.103	.163	.186	.207
	15.00	.057	.036	.096	.144	.183	.226	.228
	20.00	.028	.057	.121	.164	.216	.239	.328
	25.00	.005	.079	.147	.195	.249	.263	.374
	30.00	.030	.100	.175	.222	.283	.298	.392
	35.00	.034	.127	.196	.245	.307	.326	.414
	40.00	.055	.152	.220	.267	.338	.360	.433
	45.00	.083	.168	.248	.288	.373	.384	.445
	50.00	.124	.197	.289	.305	.418	.418	.468
	55.00	.103	.224	.265	.339	.456	.456	.418
	60.00	.144	.222	.286	.326	.464	.464	.418
	65.00	.187	.205	.286	.326	.464	.464	.418
	70.00	.147		.298	.286	.411	.411	.417
	75.00	.137	.247	.259	.184	.355	.355	.391
	80.00	.200	.242	.161	.131	.311	.311	.328
	85.00	.232	.237	.114	.086	.270	.270	.300
	90.00	.234	.210	.062	.053	.245	.245	.266
	95.00	.220	.132	.062				.066
LOWER SURFACE	1.25	.263	.013	.079	.118	.180	.160	.245
	2.50	.211	.012	.045	.103	.144	.197	.213
	5.00	.140	.011	.050	.081	.129	.185	.202
	7.50	.116	.007	.069	.093	.155	.201	.225
	10.00	.094	.084	.083	.112	.164	.215	.240
	15.00	.053	.082	.114	.164	.234	.278	.304
	20.00	.001	.084	.138	.187	.263	.294	.371
	25.00	.088	.083	.155	.208	.287	.324	.408
	30.00	.044	.113	.179	.222	.306	.352	.428
	35.00	.088	.138	.224	.279	.366	.403	.444
	40.00	.065	.138	.253	.308	.387	.433	.489
	45.00	.097	.164	.281	.338	.408	.473	.488
	50.00	.136	.213	.288	.343	.404	.497	.470
	55.00	.151	.224	.290	.348	.404	.418	.477
	60.00	.169	.233	.298	.348	.389		.404
	65.00	.197	.235					.366
	70.00	.190		.299	.292	.299	.227	.136
	75.00	.204	.246	.264	.267	.261	.149	.089
	80.00	.184	.233	.221	.228	.228	.122	.021
	85.00	.237	.214	.189	.184	.184	.086	.021
	90.00	.238	.214	.060	.038	.088	.039	.059
	95.00	.287	.138					

TABLE I

BASIC WING

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
M = 1.00 α = 3.0°									
UPPER SURFACE	1.00	.067	.333	.821	.456	.073	.053	.176	
	1.25	.131	.704	.804	.768	.073	.821	.857	
	2.50	.041	.681	.739	.813	.017	.854	.846	
	5.00	.038	.246	.510	.714	.746	.766	.735	
	7.50	.074	.239	.298	.441	.686	.735	.727	
	10.00	.078	.233	.301	.333	.596	.709	.783	
	15.00	.108	.226	.299	.358	.372	.621	.702	
	20.00	.118	.225	.296	.355	.406	.449	.644	
	25.00	.151	.230	.306	.378	.437	.450	.463	
	30.00	.167	.235	.320	.385	.452	.468	.430	
	35.00	.163	.281	.338	.398	.463	.506	.455	
	40.00	.179	.273	.326	.420	.463	.528	.479	
	45.00	.201	.289	.376	.441	.503	.553	.511	
	50.00	.239	.313	.396	.464	.538	.573	.536	
	55.00	.206	.333	.409	.490	.558	.616	.568	
	60.00	.249	.342	.409	.426	.576	.639	.571	
	65.00	.293	.312	.396	.460	.550	.609	.565	
	70.00	.250						.326	
	75.00	.242	.360	.426	.458	.364	.250	.182	
80.00	.290	.348	.368	.281	.242	.183	.133		
85.00	.330	.339	.238	.186	.179	.136	.087		
90.00	.329	.275	.154	.124	.132	.092	.049		
95.00	.299	.158	.092	.063	.100	.062	.016		
LOWER SURFACE	1.25	.385	.372	.340	.292	.295		.281	
	2.50	.368	.307	.271	.240	.219	.225	.203	
	5.00	.313	.242	.199	.136	.127	.127	.124	
	7.50	.281	.195	.153	.149	.083	.074	.051	
	10.00	.254	.169	.121	.108	.042	.040	.006	
	15.00	.207	.109	.077	.033	.011	.007	.078	
	20.00	.151	.079	.029	.003	.039	.056	.142	
	25.00	.156	.047	.009	.039	.078	.097	.236	
	30.00	.093	.016	.038	.066	.105	.127	.283	
	35.00	.059	.015	.063	.098	.134	.153	.324	
	40.00	.049	.038	.093	.124	.162	.164	.330	
	45.00	.013	.074	.119	.154	.195	.189	.346	
	50.00	.024	.105	.147	.173	.212	.221	.353	
	55.00	.043	.113	.157	.185	.212	.251	.359	
	60.00	.062	.126	.160	.200	.191	.275	.350	
	65.00	.089	.138	.169	.200	.174	.283	.333	
	70.00	.089						.305	
	75.00	.104	.141	.180	.151	.127	.275	.275	
	80.00	.100	.132	.182	.119	.119	.209	.197	
85.00	.130	.141	.119	.097	.107	.191	.143		
90.00	.138	.146	.099	.083	.105	.165	.108		
95.00	.193	.112	.059	.054	.098	.101	.036		
M = 1.00 α = 5.0°									
UPPER SURFACE	1.00	.044	.142	.030	.333	.154	.187	.2091	
	1.25	.083	.273	.076	.939	.950	.941	.12018	
	2.50	.051	.262	.046	.984	.984	.1008	.18011	
	5.00	.138	.245	.028	.927	.927	.919	.9920	
	7.50	.173	.2414	.009	.856	.883	.899	.9909	
	10.00	.183	.2372	.755	.806	.860	.875	.9900	
	15.00	.199	.225	.323	.765	.804	.856	.9881	
	20.00	.221	.2114	.149	.493	.785	.826	.9861	
	25.00	.230	.210	.066	.401	.743	.804	.9821	
	30.00	.230	.2109	.179	.423	.547	.803	.9768	
	35.00	.240	.2134	.112	.443	.480	.783	.9753	
	40.00	.256	.2149	.033	.471	.496	.726	.9753	
	45.00	.260	.2189	.443	.491	.523	.625	.9753	
	50.00	.260	.2187	.443	.512	.554	.690	.9744	
	55.00	.260	.2187	.443	.512	.554	.690	.9744	
	60.00	.260	.2187	.443	.512	.554	.690	.9744	
	65.00	.260	.2187	.443	.512	.554	.690	.9744	
	70.00	.260	.2187	.443	.512	.554	.690	.9744	
	75.00	.260	.2187	.443	.512	.554	.690	.9744	
80.00	.260	.2187	.443	.512	.554	.690	.9744		
85.00	.260	.2187	.443	.512	.554	.690	.9744		
90.00	.260	.2187	.443	.512	.554	.690	.9744		
95.00	.260	.2187	.443	.512	.554	.690	.9744		
LOWER SURFACE	1.25	.431	.480	.441	.395	.396		.387	
	2.50	.436	.408	.367	.385	.385	.330	.332	
	5.00	.397	.339	.288	.299	.299	.254	.227	
	7.50	.368	.286	.240	.240	.244	.180	.152	
	10.00	.338	.255	.203	.179	.144	.144	.107	
	15.00	.290	.188	.151	.114	.103	.095	.003	
	20.00	.238	.149	.102	.076	.049	.044	.076	
	25.00	.224	.113	.083	.041	.010	.008	.170	
	30.00	.163	.080	.029	.011	.021	.025	.232	
	35.00	.123	.048	.003	.020	.022	.057	.282	
	40.00	.108	.020	.028	.053	.077	.088	.297	
	45.00	.078	.014	.021	.077	.103	.125	.317	
	50.00	.037	.024	.021	.101	.118	.165	.339	
	55.00	.010	.028	.028	.116	.124	.200	.352	
	60.00	.009	.029	.028	.131	.121	.229	.352	
	65.00	.031	.083	.107	.131	.116	.245	.349	
	70.00	.031						.342	
	75.00	.023	.086	.125	.105	.099	.257	.326	
	80.00	.047	.081	.109	.094	.103	.202	.286	
85.00	.076	.097	.098	.078	.099	.197	.234		
90.00	.084	.108	.084	.072	.100	.188	.2171		
95.00	.129	.097	.063	.080	.108	.126	.2111		

TABLE I

BASIC WING

PERCENT CHORD		PRESSURE COEFFICIENT, P , AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
$M = 1.00 \quad \alpha = 7.8^\circ$								
UPPER SURFACE	.00	.001	.061	.136	.168	.370	.431	.374
	1.25	.086	1.002	1.076	1.042	1.066	1.061	1.104
	2.50	.143	.984	1.035	1.079	1.082	1.105	1.099
	5.00	.243	.858	1.085	1.031	1.035	1.021	1.015
	7.50	.276	.671	.962	.984	.996	1.010	1.002
	10.00	.280	.580	.937	.943	.978	.985	.995
	15.00	.290	.477	.836	.915	.939	.965	.979
	20.00	.278	.426	.607	.858	.918	.943	.963
	25.00	.296	.398	.463	.842	.911	.914	.917
	30.00	.298	.383	.434	.702	.865	.921	.874
	35.00	.293	.382	.437	.552	.837	.908	.867
	40.00	.300	.395	.457	.522	.827	.900	.865
	45.00	.319	.405	.475	.524	.780	.890	.868
	50.00	.354	.423	.499	.541	.685	.873	.872
	55.00	.315	.439	.506	.570	.643	.893	.848
	60.00	.356	.452	.517	.574	.635	.885	.870
	65.00	.398	.423	.505	.576	.627	.868	.879
	70.00	.358						.821
	75.00	.335	.461	.520	.565	.463	.544	.660
	80.00	.403	.449	.460	.391	.341	.546	.669
	85.00	.422	.439	.395	.260	.285	.509	.632
LOWER SURFACE	90.00	.419	.360	.205	.186	.238	.393	.609
	95.00	.404	.201	.140	.134	.203	.220	.563
	1.25	.463	.552	.505	.463	.450		.435
	2.50	.494	.483	.441	.394	.388	.395	.371
	5.00	.471	.416	.361	.348	.306	.302	.288
	7.50	.447	.360	.311	.295	.251	.250	.212
	10.00	.418	.385	.277	.245	.212	.201	.160
	15.00	.367	.252	.220	.183	.162	.158	.064
	20.00	.301	.214	.167	.142	.121	.104	.005
	25.00	.291	.173	.132	.104	.072	.062	.121
	30.00	.266	.135	.089	.073	.038	.026	.190
	35.00	.182	.102	.062	.040	.008	.015	.243
	40.00	.163	.076	.031	.005	.019	.049	.271
	45.00	.125	.039	.008	.023	.046	.086	.297
	50.00	.090	.012	.018	.054	.067	.128	.323
	55.00	.061	.002	.030	.070	.086	.167	.341
	60.00	.035	.017	.043	.078	.094	.202	.344
	65.00	.020	.029	.089	.086	.092	.222	.344
	70.00	.013						.341
	75.00	.022	.037	.087	.084	.086	.249	.341
	80.00	.002	.031	.082	.075	.097	.204	.281
	85.00	.024	.003	.074	.071	.101	.207	.233
UPPER SURFACE	90.00	.036	.079	.073	.071	.114	.212	.190
	95.00	.053	.089	.064	.065	.126	.109	.200
	.00	.118	.440	.472	.083	.713	.822	.756
	1.25	.164	1.103	1.141	1.227	1.268	1.210	1.085
	2.50	.296	1.090	1.092	1.240	1.267	1.231	1.078
	5.00	.440	1.115	1.126	1.203	1.238	1.166	1.053
	7.50	.475	1.083	1.039	1.170	1.207	1.185	1.019
	10.00	.481	1.024	.977	1.141	1.184	1.157	.984
	15.00	.488	.934	.903	1.090	1.156	1.125	.952
	20.00	.450	.881	.839	1.031	1.106	1.080	.935
	25.00	.443	.826	.802	.938	1.031	1.017	.904
	30.00	.433	.735	.774	.853	1.093	1.016	.870
	35.00	.416	.587	.755	.882	1.081	.965	.825
	40.00	.415	.455	.751	.839	1.070	.895	.782
	45.00	.419	.432	.751	.817	.993	.840	.751
	50.00	.446	.463	.755	.800	.842	.807	.730
	55.00	.413	.491	.786	.774	.789	.777	.709
	60.00	.452	.514	.708	.729	.777	.754	.684
	65.00	.494	.510	.703	.666	.775	.740	.677
	70.00	.463						.645
	75.00	.438	.533	.661	.534	.777	.715	.619
	80.00	.502	.525	.585	.476	.780	.696	.600
LOWER SURFACE	85.00	.508	.528	.584	.412	.754	.669	.592
	90.00	.507	.454	.402	.358	.709	.643	.572
	95.00	.493	.255	.348	.293	.629	.606	.561
	1.25	.438	.644	.575	.570	.516		.474
	2.50	.547	.608	.544	.508	.490	.488	.444
	5.00	.604	.547	.482	.477	.426	.409	.365
	7.50	.615	.492	.437	.422	.375	.360	.293
	10.00	.579	.458	.401	.364	.338	.313	.245
	15.00	.520	.389	.347	.311	.287	.267	.158
	20.00	.443	.340	.293	.265	.235	.210	.068
	25.00	.416	.297	.251	.226	.189	.168	.034
	30.00	.382	.260	.212	.192	.154	.117	.108
	35.00	.304	.224	.180	.154	.119	.078	.164
	40.00	.279	.192	.148	.121	.087	.031	.207
	45.00	.248	.156	.121	.087	.053	.014	.243
	50.00	.208	.123	.090	.058	.023	.064	.274
	55.00	.171	.102	.073	.036	.004	.107	.301
	60.00	.141	.084	.058	.018	.024	.146	.305
	65.00	.130	.069	.031	.004	.043	.176	.319
	70.00	.103						.317
	75.00	.035	.048	.011	.029	.065	.222	.320
	80.00	.057	.045	.017	.042	.096	.206	.327
	85.00	.030	.005	.017	.063	.114	.255	.348
	90.00	.004	.029	.036	.096	.149	.297	.358
	95.00	.012	.074	.062	.131	.174	.334	.392

TABLE I

BASIC WING

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:												
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2						
M = 1.00 α = 15.9°															
UPPER SURFACE	.00	-	.285	-	.788	-	.879	-	.696	-	1.003	-	.835	-	.403
	1.25	-	.286	-	1.207	-	1.138	-	1.127	-	.975	-	.784	-	.380
	2.50	-	.490	-	1.186	-	1.074	-	1.105	-	.973	-	.788	-	.382
	5.00	-	.664	-	1.199	-	1.114	-	1.088	-	.969	-	.754	-	.391
	7.50	-	.699	-	1.206	-	1.092	-	1.088	-	.960	-	.762	-	.414
	10.00	-	.697	-	1.192	-	1.115	-	1.088	-	.954	-	.753	-	.440
	15.00	-	.691	-	1.137	-	1.120	-	1.081	-	.943	-	.731	-	.475
	20.00	-	.634	-	1.129	-	1.122	-	1.072	-	.933	-	.721	-	.512
	25.00	-	.595	-	1.116	-	1.120	-	1.096	-	.920	-	.711	-	.545
	30.00	-	.564	-	1.105	-	1.114	-	1.094	-	.904	-	.713	-	.567
	35.00	-	.534	-	1.092	-	1.109	-	1.091	-	.888	-	.698	-	.579
	40.00	-	.522	-	1.030	-	1.099	-	1.088	-	.870	-	.679	-	.587
	45.00	-	.522	-	.732	-	1.091	-	1.077	-	.855	-	.659	-	.591
	50.00	-	.542	-	.567	-	1.081	-	1.038	-	.830	-	.645	-	.590
	55.00	-	.518	-	.545	-	1.067	-	.979	-	.810	-	.633	-	.593
60.00	-	.545	-	.562	-	1.079	-	.947	-	.794	-	.624	-	.586	
65.00	-	.581	-	.565	-	1.066	-	.911	-	.773	-	.618	-	.589	
70.00	-	.556	-	-	-	-	-	-	-	-	-	-	-	.575	
75.00	-	.556	-	.609	-	1.064	-	.852	-	.728	-	.617	-	.583	
80.00	-	.598	-	.607	-	1.017	-	.831	-	.712	-	.613	-	.569	
85.00	-	.600	-	.603	-	.941	-	.808	-	.688	-	.609	-	.567	
90.00	-	.600	-	.499	-	.835	-	.794	-	.671	-	.605	-	.559	
95.00	-	.547	-	.315	-	.626	-	.774	-	.653	-	.599	-	.550	
LOWER SURFACE	1.25	-	.391	-	.689	-	.588	-	.595	-	.519	-	.540	-	.482
	2.50	-	.558	-	.703	-	.621	-	.575	-	.548	-	.500	-	.494
	5.00	-	.718	-	.670	-	.590	-	.556	-	.522	-	.461	-	.444
	7.50	-	.784	-	.629	-	.557	-	.521	-	.484	-	.418	-	.385
	10.00	-	.745	-	.593	-	.525	-	.480	-	.449	-	.418	-	.341
	15.00	-	.677	-	.529	-	.476	-	.427	-	.403	-	.377	-	.258
	20.00	-	.597	-	.483	-	.423	-	.384	-	.351	-	.320	-	.153
	25.00	-	.563	-	.437	-	.381	-	.341	-	.307	-	.274	-	.083
	30.00	-	.500	-	.397	-	.342	-	.307	-	.269	-	.229	-	.013
	35.00	-	.449	-	.359	-	.305	-	.267	-	.232	-	.180	-	.043
	40.00	-	.421	-	.328	-	.268	-	.229	-	.197	-	.137	-	.092
	45.00	-	.379	-	.289	-	.238	-	.193	-	.157	-	.086	-	.132
	50.00	-	.344	-	.254	-	.206	-	.158	-	.128	-	.038	-	.173
	55.00	-	.305	-	.230	-	.180	-	.131	-	.095	-	.007	-	.200
	60.00	-	.274	-	.210	-	.158	-	.106	-	.069	-	.052	-	.222
65.00	-	.268	-	.190	-	.124	-	.086	-	.048	-	.087	-	.243	
70.00	-	.221	-	-	-	-	-	-	-	-	-	-	-	.285	
75.00	-	.165	-	.151	-	.068	-	.041	-	.002	-	.145	-	.290	
80.00	-	.157	-	.135	-	.048	-	.022	-	.041	-	.204	-	.309	
85.00	-	.124	-	.086	-	.033	-	.005	-	.076	-	.241	-	.339	
90.00	-	.067	-	.039	-	.000	-	.058	-	.123	-	.258	-	.367	
95.00	-	.041	-	.033	-	.025	-	.118	-	.182	-	.302	-	.404	
M = 1.03 α = 0.0°															
UPPER SURFACE	.00	-	.043	-	.578	-	.501	-	.746	-	.486	-	.505	-	.471
	1.25	-	.240	-	.007	-	.038	-	.048	-	.017	-	.015	-	.079
	2.50	-	.183	-	.002	-	.029	-	.066	-	.086	-	.133	-	.087
	5.00	-	.118	-	.004	-	.036	-	.080	-	.059	-	.079	-	.101
	7.50	-	.081	-	.018	-	.041	-	.078	-	.106	-	.115	-	.129
	10.00	-	.065	-	.032	-	.070	-	.090	-	.130	-	.139	-	.173
	15.00	-	.035	-	.043	-	.089	-	.130	-	.147	-	.185	-	.189
	20.00	-	.011	-	.065	-	.116	-	.147	-	.179	-	.202	-	.289
	25.00	-	.022	-	.083	-	.139	-	.178	-	.210	-	.224	-	.321
	30.00	-	.042	-	.099	-	.166	-	.199	-	.231	-	.252	-	.334
	35.00	-	.044	-	.124	-	.181	-	.221	-	.250	-	.274	-	.360
	40.00	-	.060	-	.149	-	.209	-	.253	-	.274	-	.299	-	.370
	45.00	-	.089	-	.167	-	.233	-	.274	-	.286	-	.328	-	.398
	50.00	-	.118	-	.194	-	.252	-	.290	-	.314	-	.361	-	.411
	55.00	-	.114	-	.219	-	.259	-	.307	-	.345	-	.411	-	.438
60.00	-	.140	-	.224	-	.267	-	.308	-	.350	-	.423	-	.408	
65.00	-	.179	-	.203	-	.263	-	.300	-	.310	-	.351	-	.271	
70.00	-	.154	-	-	-	-	-	-	-	-	-	-	-	.189	
75.00	-	.115	-	.240	-	.281	-	.287	-	.164	-	.196	-	.111	
80.00	-	.207	-	.230	-	.247	-	.167	-	.102	-	.140	-	.059	
85.00	-	.214	-	.227	-	.165	-	.100	-	.057	-	.084	-	.000	
90.00	-	.218	-	.206	-	.098	-	.046	-	.017	-	.031	-	.056	
95.00	-	.210	-	.128	-	.042	-	.024	-	.004	-	.015	-	.087	
LOWER SURFACE	1.25	-	.207	-	.048	-	.104	-	.131	-	.181	-	.132	-	.237
	2.50	-	.164	-	.017	-	.059	-	.106	-	.137	-	.163	-	.203
	5.00	-	.095	-	.011	-	.060	-	.052	-	.140	-	.163	-	.189
	7.50	-	.069	-	.028	-	.070	-	.091	-	.163	-	.177	-	.218
	10.00	-	.056	-	.037	-	.087	-	.106	-	.178	-	.195	-	.236
	15.00	-	.024	-	.069	-	.108	-	.157	-	.191	-	.205	-	.273
	20.00	-	.020	-	.077	-	.135	-	.178	-	.206	-	.238	-	.313
	25.00	-	.007	-	.094	-	.153	-	.195	-	.233	-	.254	-	.345
	30.00	-	.050	-	.114	-	.178	-	.213	-	.258	-	.282	-	.361
	35.00	-	.073	-	.141	-	.194	-	.243	-	.272	-	.309	-	.368
	40.00	-	.073	-	.157	-	.221	-	.268	-	.293	-	.331	-	.384
	45.00	-	.102	-	.185	-	.245	-	.298	-	.328	-	.364	-	.414
	50.00	-	.132	-	.213	-	.276	-	.328	-	.348	-	.383	-	.433
	55.00	-	.156	-	.224	-	.278	-	.329	-	.370	-	.432	-	.481
	60.00	-	.166	-	.233	-	.278	-	.331	-	.370	-	.460	-	.459
65.00	-	.188	-	.238	-	.285	-	.325	-	.344	-	.419	-	.306	
70.00	-	.183	-	-	-	-	-	-	-	-	-	-	-	.204	
75.00	-	.199	-	.242	-	.290	-	.302	-	.159	-	.181	-	.102	
80.00	-	.179	-	.230	-	.257	-	.173	-	.109	-	.128	-	.053	
85.00	-	.224	-	.225	-	.197	-	.096	-	.069	-	.083	-	.004	
90.00	-	.212	-	.211	-	.117	-	.048	-	.034	-	.040	-	.046	
95.00	-	.268	-	.189	-	.042	-	.004	-	.004	-	.003	-	.088	

TABLE I

BASIC WING

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:							
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
		M = 1.03 α = 3.9°								
UPPER SURFACE	.00	.054	.358	.250	.497	.116	.105	.200		
	1.25	.113	.681	.741	.698	.703	.752	.789		
	2.50	.028	.602	.696	.750	.746	.778	.776		
	5.00	.054	.226	.570	.681	.684	.695	.670		
	7.50	.092	.232	.274	.529	.636	.668	.670		
	10.00	.102	.226	.281	.337	.593	.646	.667		
	15.00	.120	.218	.285	.318	.352	.606	.646		
	20.00	.122	.216	.278	.322	.345	.443	.628		
	30.00	.150	.222	.286	.342	.380	.383	.488		
	35.00	.164	.226	.287	.352	.396	.422	.387		
	40.00	.161	.238	.314	.368	.413	.447	.403		
	45.00	.172	.261	.332	.383	.430	.469	.428		
	50.00	.193	.273	.349	.403	.446	.493	.465		
	55.00	.224	.296	.370	.424	.471	.516	.491		
	60.00	.207	.315	.383	.447	.505	.555	.521		
	65.00	.237	.328	.390	.396	.528	.578	.528		
	70.00	.248	.303	.369	.432	.506	.551	.527		
	75.00	.209	.334	.396	.423	.333	.221	.345		
	80.00	.294	.325	.344	.250	.199	.152	.177		
	85.00	.304	.318	.215	.144	.135	.110	.128		
	90.00	.307	.258	.115	.063	.088	.070	.086		
	95.00	.283	.132	.055	.062	.056	.037	.044		
LOWER SURFACE	1.25	.320	.361	.344	.300	.329		.307		
	2.50	.330	.299	.276	.253	.257	.272	.235		
	5.00	.285	.233	.204	.231	.176	.174	.152		
	7.50	.262	.194	.161	.172	.123	.123	.077		
	10.00	.238	.167	.127	.128	.086	.087	.035		
	15.00	.198	.111	.088	.053	.057	.045	.052		
	20.00	.145	.083	.043	.022	.010	.004	.124		
	25.00	.155	.080	.010	.006	.027	.041	.207		
	30.00	.095	.019	.022	.035	.056	.076	.252		
	35.00	.089	.007	.048	.065	.083	.106	.287		
	40.00	.049	.029	.075	.086	.106	.127	.302		
	45.00	.018	.061	.097	.111	.137	.144	.311		
	50.00	.015	.087	.125	.135	.155	.173	.329		
	55.00	.041	.100	.130	.143	.163	.204	.329		
	60.00	.061	.109	.133	.146	.144	.226	.316		
	65.00	.072	.123	.140	.150	.127	.236	.301		
	70.00	.072						.290		
	75.00	.100	.123	.135	.101	.083	.226	.256		
	80.00	.079	.107	.111	.073	.075	.185	.187		
	85.00	.107	.113	.083	.050	.060	.164	.136		
	90.00	.115	.111	.056	.036	.056	.130	.093		
	95.00	.133	.084	.021	.012	.050	.074	.032		
		M = 1.03 α = 5.9°								
UPPER SURFACE	.00	.066	.183	.074	.378	.105	.134	.040		
	1.25	.060	.855	.896	.861	.874	.886	.925		
	2.50	.035	.812	.875	.912	.906	.932	.921		
	5.00	.126	.676	.829	.850	.855	.847	.827		
	7.50	.167	.375	.741	.789	.815	.830	.822		
	10.00	.177	.326	.709	.733	.797	.806	.817		
	15.00	.191	.294	.428	.709	.731	.793	.802		
	20.00	.163	.287	.314	.577	.712	.759	.791		
	25.00	.107	.286	.334	.377	.718	.735	.745		
	30.00	.216	.284	.346	.374	.600	.732	.686		
	35.00	.220	.292	.358	.395	.460	.721	.682		
	40.00	.230	.309	.374	.420	.442	.714	.684		
	45.00	.239	.321	.395	.442	.467	.654	.694		
	50.00	.269	.340	.416	.464	.498	.577	.699		
	55.00	.247	.359	.427	.493	.534	.564	.694		
	60.00	.278	.371	.438	.439	.556	.578	.635		
	65.00	.319	.346	.422	.490	.555	.573	.508		
	70.00	.289						.432		
	75.00	.242	.378	.436	.473	.442	.244	.331		
	80.00	.335	.366	.384	.301	.248	.183	.310		
	85.00	.339	.355	.233	.175	.176	.147	.253		
	90.00	.339	.276	.133	.110	.125	.108	.203		
95.00	.326	.136	.073	.073	.089	.073	.135			
LOWER SURFACE	1.25	.376	.481	.454	.417	.426		.406		
	2.50	.411	.414	.382	.350	.354	.358	.333		
	5.00	.389	.346	.303	.323	.270	.265	.250		
	7.50	.373	.294	.257	.264	.216	.210	.178		
	10.00	.342	.262	.226	.206	.175	.174	.130		
	15.00	.297	.199	.176	.144	.139	.124	.034		
	20.00	.238	.162	.127	.106	.087	.075	.065		
	25.00	.238	.129	.092	.070	.050	.035	.141		
	30.00	.177	.095	.056	.042	.019	.002	.201		
	35.00	.136	.070	.032	.009	.013	.032	.243		
	40.00	.128	.047	.003	.017	.040	.056	.263		
	45.00	.089	.012	.026	.038	.071	.091	.286		
	50.00	.054	.017	.049	.055	.088	.130	.306		
	55.00	.029	.026	.060	.068	.097	.162	.317		
	60.00	.008	.044	.064	.085	.095	.191	.314		
	65.00	.006	.058	.072	.095	.086	.208	.311		
	70.00	.009						.304		
	75.00	.063	.063	.074	.077	.066	.219	.290		
	80.00	.033	.047	.068	.058	.065	.184	.241		
	85.00	.054	.054	.083	.042	.060	.173	.196		
	90.00	.069	.058	.044	.038	.059	.160	.143		
	95.00	.061	.055	.023	.017	.068	.098	.096		

TABLE I

BASIC WING

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:												
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2						
UPPER SURFACE	M = 1.03 α = 7.9°														
	.00	-	.106	-	.019	-	.094	-	.211	-	.318	-	.368	-	.319
	1.25	-	.002	1.001	.038	-	1.005	-	1.032	-	1.008	-	1.008	-	1.062
	2.50	-	.104	.972	.991	-	1.043	-	1.046	-	1.056	-	1.056	-	1.056
	5.00	-	.212	.922	.987	-	.990	-	.998	-	.971	-	.981	-	.981
	7.50	-	.252	.801	.921	-	.948	-	.957	-	.966	-	.968	-	.968
	10.00	-	.258	.584	.911	-	.910	-	.939	-	.939	-	.953	-	.953
	15.00	-	.273	.415	.860	-	.878	-	.903	-	.921	-	.935	-	.935
	20.00	-	.257	.368	.683	-	.847	-	.879	-	.894	-	.915	-	.915
	25.00	-	.275	.352	.456	-	.839	-	.869	-	.862	-	.869	-	.869
	30.00	-	.277	.344	.399	-	.767	-	.854	-	.876	-	.831	-	.831
	35.00	-	.270	.349	.396	-	.575	-	.838	-	.869	-	.825	-	.825
	40.00	-	.274	.359	.410	-	.495	-	.829	-	.866	-	.824	-	.824
	45.00	-	.289	.370	.427	-	.480	-	.810	-	.869	-	.835	-	.835
	50.00	-	.319	.389	.450	-	.487	-	.728	-	.867	-	.841	-	.841
	55.00	-	.292	.403	.461	-	.513	-	.635	-	.866	-	.811	-	.811
	60.00	-	.327	.412	.474	-	.462	-	.599	-	.863	-	.518	-	.518
	65.00	-	.364	.391	.465	-	.523	-	.577	-	.550	-	.529	-	.529
	70.00	-	.335	-	-	-	-	-	-	-	-	-	.587	-	.587
75.00	-	.284	.418	.473	-	.521	-	.406	-	.521	-	.643	-	.643	
80.00	-	.383	.407	.483	-	.353	-	.305	-	.548	-	.649	-	.649	
85.00	-	.384	.397	.269	-	.218	-	.252	-	.543	-	.644	-	.644	
90.00	-	.378	.313	.164	-	.146	-	.205	-	.473	-	.627	-	.627	
95.00	-	.369	.164	.096	-	.112	-	.167	-	.336	-	.601	-	.601	
LOWER SURFACE	M = 1.03 α = 11.5°														
	.00	-	.174	.383	.408	-	.026	-	.638	-	.741	-	.678	-	.678
	1.25	-	.114	1.026	1.061	-	1.148	-	1.162	-	1.099	-	1.028	-	1.028
	2.50	-	.238	1.014	1.012	-	1.157	-	1.167	-	1.122	-	1.015	-	1.015
	5.00	-	.389	1.038	1.040	-	1.123	-	1.142	-	1.069	-	.978	-	.978
	7.50	-	.425	1.007	.952	-	1.098	-	1.115	-	1.082	-	.986	-	.986
	10.00	-	.433	.952	.893	-	1.063	-	1.096	-	1.068	-	.992	-	.992
	15.00	-	.440	.866	.823	-	1.016	-	1.065	-	1.047	-	.941	-	.941
	20.00	-	.400	.816	.765	-	.958	-	1.035	-	1.024	-	.910	-	.910
	25.00	-	.393	.767	.726	-	.936	-	1.020	-	.976	-	.893	-	.893
	30.00	-	.381	.684	.701	-	.887	-	1.002	-	.981	-	.836	-	.836
	35.00	-	.365	.549	.681	-	.819	-	.994	-	.953	-	.790	-	.790
	40.00	-	.358	.407	.669	-	.774	-	.986	-	.973	-	.736	-	.736
	45.00	-	.363	.370	.659	-	.753	-	.940	-	.968	-	.734	-	.734
	50.00	-	.387	.399	.653	-	.738	-	.803	-	.925	-	.725	-	.725
	55.00	-	.363	.426	.636	-	.713	-	.733	-	.808	-	.713	-	.713
	60.00	-	.395	.450	.638	-	.678	-	.722	-	.697	-	.690	-	.690
	65.00	-	.432	.445	.635	-	.614	-	.715	-	.678	-	.658	-	.658
	70.00	-	.408	-	-	-	-	-	-	-	-	-	.658	-	.658
75.00	-	.367	.468	.601	-	.490	-	.723	-	.661	-	.632	-	.632	
80.00	-	.450	.465	.530	-	.433	-	.728	-	.645	-	.604	-	.604	
85.00	-	.447	.460	.452	-	.366	-	.707	-	.620	-	.588	-	.588	
90.00	-	.445	.402	.389	-	.306	-	.669	-	.591	-	.562	-	.562	
95.00	-	.433	.208	.304	-	.261	-	.601	-	.559	-	.562	-	.562	
UPPER SURFACE	M = 1.03 α = 11.5°														
	.00	-	.174	.383	.408	-	.026	-	.638	-	.741	-	.678	-	.678
	1.25	-	.114	1.026	1.061	-	1.148	-	1.162	-	1.099	-	1.028	-	1.028
	2.50	-	.238	1.014	1.012	-	1.157	-	1.167	-	1.122	-	1.015	-	1.015
	5.00	-	.389	1.038	1.040	-	1.123	-	1.142	-	1.069	-	.978	-	.978
	7.50	-	.425	1.007	.952	-	1.098	-	1.115	-	1.082	-	.986	-	.986
	10.00	-	.433	.952	.893	-	1.063	-	1.096	-	1.068	-	.992	-	.992
	15.00	-	.440	.866	.823	-	1.016	-	1.065	-	1.047	-	.941	-	.941
	20.00	-	.400	.816	.765	-	.958	-	1.035	-	1.024	-	.910	-	.910
	25.00	-	.393	.767	.726	-	.936	-	1.020	-	.976	-	.893	-	.893
	30.00	-	.381	.684	.701	-	.887	-	1.002	-	.981	-	.836	-	.836
	35.00	-	.365	.549	.681	-	.819	-	.994	-	.953	-	.790	-	.790
	40.00	-	.358	.407	.669	-	.774	-	.986	-	.973	-	.736	-	.736
	45.00	-	.363	.370	.659	-	.753	-	.940	-	.968	-	.734	-	.734
	50.00	-	.387	.399	.653	-	.738	-	.803	-	.925	-	.725	-	.725
	55.00	-	.363	.426	.636	-	.713	-	.733	-	.808	-	.713	-	.713
	60.00	-	.395	.450	.638	-	.678	-	.722	-	.697	-	.690	-	.690
	65.00	-	.432	.445	.635	-	.614	-	.715	-	.678	-	.658	-	.658
	70.00	-	.408	-	-	-	-	-	-	-	-	-	.658	-	.658
75.00	-	.367	.468	.601	-	.490	-	.723	-	.661	-	.632	-	.632	
80.00	-	.450	.465	.530	-	.433	-	.728	-	.645	-	.604	-	.604	
85.00	-	.447	.460	.452	-	.366	-	.707	-	.620	-	.588	-	.588	
90.00	-	.445	.402	.389	-	.306	-	.669	-	.591	-	.562	-	.562	
95.00	-	.433	.208	.304	-	.261	-	.601	-	.559	-	.562	-	.562	
LOWER SURFACE	M = 1.03 α = 11.5°														
	.00	-	.174	.383	.408	-	.026	-	.638	-	.741	-	.678	-	.678
	1.25	-	.114	1.026	1.061	-	1.148	-	1.162	-	1.099	-	1.028	-	1.028
	2.50	-	.238	1.014	1.012	-	1.157	-	1.167	-	1.122	-	1.015	-	1.015
	5.00	-	.389	1.038	1.040	-	1.123	-	1.142	-	1.069	-	.978	-	.978
	7.50	-	.425	1.007	.952	-	1.098	-	1.115	-	1.082	-	.986	-	.986
	10.00	-	.433	.952	.893	-	1.063	-	1.096	-	1.068	-	.992	-	.992
	15.00	-	.440	.866	.823	-	1.016	-	1.065	-	1.047	-	.941	-	.941
	20.00	-	.400	.816	.765	-	.958	-	1.035	-	1.024	-	.910	-	.910
	25.00	-	.393	.767	.726	-	.936	-	1.020	-	.976	-	.893	-	.893
	30.00	-	.381	.684	.701	-	.887	-	1.002	-	.981	-	.836	-	.836
	35.00	-	.365	.549	.681	-	.819	-	.994	-	.953	-	.790	-	.790
	40.00	-	.358	.407	.669	-	.774	-	.986	-	.973	-	.736	-	.736
	45.00	-	.363	.370	.659	-	.753	-	.940	-	.968	-	.734	-	.734
	50.00	-	.387	.399	.653	-	.738	-	.803	-	.925	-	.725	-	.725
	55.00	-	.363	.426	.636	-	.713	-	.733	-	.808	-	.713	-	.713
	60.00	-	.395	.450	.638	-	.678	-	.722	-	.697	-	.690	-	.690
	65.00	-	.432	.445	.635	-	.614	-	.715	-	.678	-	.658	-	.658
	70.00	-	.408	-	-	-	-	-	-	-	-	-	.658	-	.658
75.00	-	.367	.468	.601	-	.490	-	.723	-	.661	-	.632	-	.632	
80.00	-	.450	.465	.530	-	.433	-	.728	-	.645	-	.604	-	.604	
85.00	-	.447	.460	.452	-	.366	-	.707	-	.620	-	.588	-	.588	
90.00	-	.445	.402	.389	-	.306	-	.669	-	.591	-	.562	-	.562	
95.00	-	.433	.208	.304	-	.261	-	.601	-	.559	-	.562	-	.562	
UPPER SURFACE	M = 1.03 α = 11.5°														
	.00	-	.174	.383	.408	-	.026	-	.638	-	.741	-	.678	-	.678
	1.25	-	.114	1.026	1.061	-	1.148	-	1.162	-	1.099	-	1.028	-	1.028
	2.50	-	.238	1.014	1.012	-	1.157	-	1.167	-	1.122	-	1.015	-	1.015
	5.00	-	.389	1.038	1.040	-	1.123	-	1.142	-	1.069	-	.978	-	.978
	7.50	-	.425	1.007	.952	-	1.098	-	1.115	-	1.082	-	.986	-	.986
	10.00	-	.433	.952	.893	-	1.063	-	1.096	-	1.068	-	.992	-	.992
	15.00	-	.440	.866	.823	-	1.016	-	1.065	-	1.047	-	.941	-	.941
	20.00	-	.400	.816	.765	-	.958	-	1.035	-	1.024	-	.910	-	.910
	25.00	-	.393	.767	.726	-	.936	-	1.020	-	.976	-	.893	-	.893
	30.00	-	.381	.684	.701	-	.887	-	1.002	-	.981	-	.836	-	.836
	35.00	-	.365	.549	.681	-	.819	-	.994	-	.953	-	.790	-	.790
	40.00	-	.358	.407	.669	-	.774	-	.986	-	.973	-	.736	-	.736
	45.00	-	.363	.370	.659	-	.753	-	.940	-	.968	-	.734	-	.734
	50.00	-	.387	.399	.653	-	.738	-	.803	-	.925	-	.725	-	.725
	55.00	-	.363	.426	.636	-	.713	-	.733	-	.808	-	.713	-	.713
	60.00	-	.395	.450	.638	-	.678	-	.722	-	.697	-	.690	-	.690
	65.00	-	.432	.445	.635	-	.614	-	.715	-	.678	-	.658	-	.658
	70.00	-	.408	-	-	-	-	-	-	-	-	-	.658	-	

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, p , AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	$M = 0.60 \quad \alpha = 0.0^\circ$								
	.00	-	.233	.476	.451	.687	.437	.435	.330
	1.25	-	.354	.080	.029	.039	.171	.189	.154
	2.50	-	.083	.066	.027	.016	.077	.085	.103
	5.00	-	.021	.069	.034	.004	.061	.089	.048
	7.50	-	.009	.075	.031	.010	.033	.055	.016
	10.00	-	.021	.084	.052	.005	.015	.040	.016
	15.00	-	.046	.082	.084	.011	.013	.014	.030
	20.00	-	.059	.087	.051	.009	.009	.016	.045
	25.00	-	.075	.087	.044	.002	.008	.019	.033
	30.00	-	.086	.081	.034	.010	.019	.019	.042
	35.00	-	.079	.062	.009	.033	.042	.033	.040
	40.00	-	.083	.039	.028	.054	.072	.054	.041
	45.00	-	.080	.006	.068	.095	.108	.084	.033
	50.00	-	.072	.076	.133	.151	.162	.129	.034
	55.00	-	.018	.184	.225	.230	.240	.194	.042
	60.00	-	.070	.272	.312	.317	.320	.254	.038
	65.00	-	.277	.266	.319	.317	.315	.246	.035
LOWER SURFACE	70.00	-	.523	.230	.328	.325	.364	.228	.042
	75.00	1	.514	.834	.389	.336	.269	.174	.040
	80.00	-	.606	.015	.413	.346	.268	.164	.047
	85.00	-	.275	.918	.443	.352	.267	.153	.068
	90.00	-	.036	.659	.451	.361	.271	.138	.081
	95.00	-	.032	.297	.414	.339	.263	.138	.079
	1.25	-	.126	.127	.188	.236	.364	.448	.467
	2.50	-	.068	.094	.135	.172	.255	.292	.324
	5.00	-	.028	.091	.120	.106	.218	.255	.252
	7.50	-	.010	.105	.133	.133	.210	.235	.246
	10.00	-	.033	.111	.139	.149	.204	.235	.238
	15.00	-	.060	.130	.158	.174	.191	.208	.216
	20.00	-	.098	.146	.164	.176	.202	.215	.205
	25.00	-	.074	.152	.169	.183	.212	.216	.178
	30.00	-	.125	.164	.174	.184	.213	.217	.171
	35.00	-	.135	.176	.179	.197	.216	.222	.171
	40.00	-	.141	.179	.188	.200	.222	.222	.175
	45.00	-	.156	.184	.190	.206	.225	.218	.163
50.00	-	.175	.190	.191	.208	.225	.218	.156	
55.00	-	.177	.180	.183	.203	.222	.214	.134	
60.00	-	.179	.172	.155	.196	.214	.197	.117	
65.00	-	.178	.158	.159	.186	.199	.178	.099	
70.00	-	.165						.087	
75.00	-	.159	.108	.124	.177	.181	.130	.083	
80.00	-	.141	.091	.144	.158	.169	.127	.061	
85.00	-	.130	.072	.142	.151	.164	.122		
90.00	-	.094	.031	.157	.161	.134	.127		
95.00	-	.065	.003	.167	.178	.176	.121		
$M = 0.60 \quad \alpha = 4.0^\circ$									
UPPER SURFACE	.00	-	.160	.151	.251	.071	.358	.238	.060
	1.25	-	.142	.964	1.042	.920	.857	.866	.684
	2.50	-	.214	.553	.570	.603	.567	.544	.456
	5.00	-	.244	.419	.426	.416	.338	.313	.310
	7.50	-	.249	.363	.350	.333	.296	.266	.267
	10.00	-	.238	.336	.321	.291	.276	.232	.229
	15.00	-	.238	.291	.271	.242	.202	.196	.160
	20.00	-	.231	.265	.230	.200	.169	.149	.141
	25.00	-	.231	.244	.194	.165	.133	.110	.110
	30.00	-	.230	.217	.163	.131	.103	.087	.088
	35.00	-	.209	.183	.121	.088	.067	.047	.077
	40.00	-	.203	.143	.071	.045	.012	.008	.063
	45.00	-	.188	.076	.014	.015	.046	.049	.057
	50.00	-	.171	.013	.066	.083	.112	.111	.044
	55.00	-	.094	.120	.134	.143	.175	.155	.038
	60.00	-	.008	.155	.155	.187	.186	.164	.026
	65.00	-	.201	.157	.192	.175	.180	.147	.027
	70.00	-	.431	.097	.160	.188	.194	.144	.028
75.00	1	.390	.764	.387	.329	.247	.155	.030	
80.00	-	.779	.889	.398	.330	.248	.144	.036	
85.00	-	.362	.837	.419	.342	.243	.136	.038	
90.00	-	.166	.693	.428	.353	.247	.128	.042	
95.00	-	.045	.473	.405	.340	.258	.121	.035	
LOWER SURFACE	1.25	-	.303	.346	.363	.331	.341	.330	.256
	2.50	-	.270	.264	.270	.257	.257	.246	.175
	5.00	-	.222	.186	.179	.191	.168	.147	.097
	7.50	-	.179	.140	.138	.140	.116	.101	.033
	10.00	-	.146	.107	.099	.099	.075	.090	.015
	15.00	-	.115	.062	.082	.043	.041	.028	.067
	20.00	-	.065	.030	.019	.007	.003	.021	.087
	25.00	-	.062	.000	.016	.027	.038	.049	.090
	30.00	-	.016	.023	.037	.048	.056	.074	.107
	35.00	-	.007	.050	.059	.071	.076	.092	.110
	40.00	-	.022	.058	.074	.088	.094	.110	.121
	45.00	-	.045	.079	.086	.107	.110	.127	.119
	50.00	-	.072	.091	.102	.121	.126	.134	.123
	55.00	-	.079	.093	.106	.128	.136	.146	.117
	60.00	-	.087	.090	.109	.129	.136	.142	.101
	65.00	-	.096	.094	.106	.134	.134	.133	.099
	70.00	-	.085						.094
	75.00	-	.093	.062	.134	.139	.132	.080	.070
80.00	-	.083	.056	.115	.128	.131	.088	.065	
85.00	-	.083	.045	.117	.130	.132	.088	.064	
90.00	-	.065	.039	.144	.146	.104	.097	.041	
95.00	-	.065	.042	.154	.170	.158	.096	.028	

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:													
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2							
		$M = 0.60 \quad \alpha = 6.0^\circ$														
UPPER SURFACE		.00	-	.138	-	.652	-	.743	-	.331	-	1.080	-	.834	-	.739
		1.25	-	.337	-	1.463	-	.877	-	.758	-	1.023	-	.817	-	.927
		2.50	-	.400	-	.912	-	.807	-	.716	-	.903	-	.650	-	.686
		5.00	-	.404	-	.676	-	.723	-	.657	-	.705	-	.560	-	.538
		7.50	-	.393	-	.570	-	.622	-	.588	-	.595	-	.492	-	.454
		10.00	-	.364	-	.508	-	.570	-	.527	-	.497	-	.444	-	.353
		15.00	-	.345	-	.422	-	.463	-	.433	-	.357	-	.350	-	.235
		20.00	-	.325	-	.370	-	.369	-	.352	-	.278	-	.272	-	.174
		25.00	-	.314	-	.330	-	.305	-	.285	-	.214	-	.201	-	.132
		30.00	-	.307	-	.289	-	.248	-	.227	-	.165	-	.153	-	.108
		35.00	-	.276	-	.248	-	.196	-	.177	-	.119	-	.097	-	.087
		40.00	-	.263	-	.198	-	.137	-	.126	-	.066	-	.052	-	.064
		45.00	-	.245	-	.126	-	.080	-	.066	-	.014	-	.004	-	.047
		50.00	-	.224	-	.036	-	.015	-	.011	-	.034	-	.043	-	.040
		55.00	-	.135	-	.065	-	.058	-	.046	-	.084	-	.086	-	.040
		60.00	-	.039	-	.009	-	.105	-	.115	-	.120	-	.112	-	.044
		65.00	-	.173	-	.086	-	.115	-	.110	-	.135	-	.116	-	.048
		70.00	-	.418	-	.034	-	.130	-	.130	-	.170	-	.116	-	.058
		75.00	-	1.273	-	.721	-	.380	-	.314	-	.255	-	.155	-	.065
LOWER SURFACE		80.00	-	.919	-	.629	-	.396	-	.318	-	.259	-	.146	-	.070
		85.00	-	.435	-	.775	-	.410	-	.332	-	.267	-	.134	-	.070
		90.00	-	.413	-	.669	-	.419	-	.331	-	.270	-	.126	-	.077
		95.00	-	.094	-	.512	-	.392	-	.312	-	.261	-	.121	-	.068
		1.25	-	.360	-	.450	-	.447	-	.426	-	.440	-	.437	-	.387
		2.50	-	.340	-	.381	-	.376	-	.363	-	.374	-	.370	-	.303
		5.00	-	.302	-	.298	-	.288	-	.296	-	.288	-	.259	-	.205
		7.50	-	.276	-	.244	-	.230	-	.248	-	.221	-	.208	-	.134
		10.00	-	.243	-	.203	-	.195	-	.197	-	.184	-	.197	-	.071
		15.00	-	.200	-	.148	-	.149	-	.130	-	.129	-	.124	-	.014
		20.00	-	.147	-	.107	-	.096	-	.089	-	.081	-	.067	-	.007
		25.00	-	.140	-	.070	-	.060	-	.049	-	.043	-	.023	-	.055
		30.00	-	.091	-	.044	-	.032	-	.026	-	.012	-	.008	-	.077
		35.00	-	.061	-	.017	-	.008	-	.006	-	.018	-	.037	-	.081
		40.00	-	.043	-	.001	-	.017	-	.027	-	.035	-	.059	-	.101
		45.00	-	.014	-	.025	-	.035	-	.051	-	.063	-	.083	-	.097
		50.00	-	.013	-	.041	-	.054	-	.069	-	.079	-	.101	-	.100
		55.00	-	.026	-	.047	-	.062	-	.087	-	.092	-	.111	-	.100
		60.00	-	.035	-	.047	-	.052	-	.093	-	.095	-	.113	-	.089
	65.00	-	.051	-	.050	-	.075	-	.094	-	.099	-	.108	-	.082	
	70.00	-	.043	-		-		-		-		-		-	.084	
	75.00	-	.056	-	.037	-	.090	-	.108	-	.103	-	.066	-	.062	
	80.00	-	.049	-	.039	-	.092	-	.102	-	.104	-	.074	-	.062	
	85.00	-	.055	-	.025	-	.102	-	.106	-	.107	-	.082	-	.064	
	90.00	-	.045	-	.037	-	.124	-	.121	-	.084	-	.092	-	.042	
	95.00	-	.035	-	.069	-	.137	-	.146	-	.134	-	.092	-	.042	
		$M = 0.60 \quad \alpha = 8.0^\circ$														
UPPER SURFACE		.00	-	.127	-	1.176	-	1.101	-	.689	-	1.337	-	.914	-	.987
		1.25	-	.575	-	1.129	-	.931	-	1.090	-	.983	-	.667	-	.814
		2.50	-	.618	-	1.069	-	.894	-	.999	-	.985	-	.650	-	.786
		5.00	-	.596	-	.987	-	.847	-	.900	-	.920	-	.611	-	.740
		7.50	-	.566	-	.875	-	.782	-	.760	-	.872	-	.596	-	.714
		10.00	-	.520	-	.776	-	.739	-	.655	-	.824	-	.564	-	.655
		15.00	-	.473	-	.645	-	.639	-	.550	-	.640	-	.517	-	.551
		20.00	-	.435	-	.552	-	.544	-	.493	-	.460	-	.447	-	.436
		25.00	-	.411	-	.470	-	.469	-	.447	-	.353	-	.387	-	.323
		30.00	-	.397	-	.407	-	.409	-	.387	-	.286	-	.344	-	.236
		35.00	-	.357	-	.345	-	.345	-	.335	-	.237	-	.297	-	.186
		40.00	-	.337	-	.286	-	.277	-	.280	-	.194	-	.236	-	.137
		45.00	-	.313	-	.221	-	.219	-	.230	-	.159	-	.179	-	.113
		50.00	-	.289	-	.187	-	.186	-	.176	-	.111	-	.122	-	.094
		55.00	-	.187	-	.057	-	.087	-	.132	-	.080	-	.072	-	.094
		60.00	-	.078	-	.007	-	.063	-	.087	-	.056	-	.036	-	.094
		65.00	-	.116	-	.043	-	.066	-	.105	-	.037	-	.049	-	.094
		70.00	-	.385	-	.069	-	.007	-	.036	-	.029	-	.018	-	.106
		75.00	-	1.193	-	.675	-	.398	-	.337	-	.292	-	.194	-	.106
	80.00	-	.933	-	.768	-	.430	-	.351	-	.302	-	.179	-	.105	
	85.00	-	.535	-	.696	-	.410	-	.324	-	.291	-	.160	-	.095	
	90.00	-	.274	-	.566	-	.372	-	.280	-	.263	-	.137	-	.099	
	95.00	-	.160	-	.431	-	.324	-	.253	-	.239	-	.127	-	.089	
LOWER SURFACE		1.25	-	.374	-	.484	-	.466	-	.473	-	.461	-	.459	-	.431
		2.50	-	.367	-	.444	-	.425	-	.434	-	.436	-	.428	-	.374
		5.00	-	.371	-	.370	-	.359	-	.383	-	.365	-	.340	-	.285
		7.50	-	.366	-	.321	-	.310	-	.326	-	.304	-	.278	-	.204
		10.00	-	.327	-	.278	-	.272	-	.279	-	.265	-	.268	-	.138
		15.00	-	.287	-	.218	-	.215	-	.203	-	.211	-	.182	-	.065
		20.00	-	.227	-	.174	-	.162	-	.158	-	.158	-	.129	-	.014
		25.00	-	.210	-	.130	-	.120	-	.113	-	.110	-	.079	-	.009
		30.00	-	.155	-	.101	-	.087	-	.074	-	.080	-	.047	-	.038
		35.00	-	.121	-	.070	-	.060	-	.054	-	.050	-	.009	-	.052
		40.00	-	.099	-	.049	-	.027	-	.023	-	.023	-	.019	-	.071
		45.00	-	.070	-	.023	-	.011	-	.008	-	.004	-	.046	-	.068
		50.00	-	.042	-	.003	-	.008	-	.018	-	.023	-	.060	-	.069
		55.00	-	.023	-	.005	-	.019	-	.033	-	.040	-	.082	-	.068
		60.00	-	.005	-	.011	-	.014	-	.045	-	.049	-	.086	-	.063
		65.00	-	.006	-	.023	-	.045	-	.058	-	.052	-	.083	-	.061
		70.00	-	.001	-		-		-		-		-		-	.062
		75.00	-	.023	-	.008	-	.069	-	.069	-	.062	-	.066	-	.044
		80.00	-	.012	-	.020	-	.066	-	.065	-	.070	-	.072	-	.044
	85.00	-	.020	-	.028	-	.071	-	.069	-	.072	-	.072	-	.048	
	90.00	-	.024	-	.028	-	.097	-	.087	-	.056	-	.089	-	.043	
	95.00	-	.012	-	.063	-	.103	-	.106	-	.104	-	.087	-	.044	

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

PRESSURE COEFFICIENT, P, AT:								
PERCENT CHORD	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
M = 0.60 $\alpha = 11.3^\circ$								
UPPER SURFACE	0.00	0.034	1.770	1.529	1.124	0.786	0.470	0.416
	1.25	1.063	1.393	1.100	0.957	0.675	0.439	0.327
	2.50	1.134	1.422	1.079	0.964	0.675	0.434	0.330
	5.00	1.079	1.455	1.098	0.969	0.670	0.431	0.314
	7.50	0.980	1.533	1.073	0.971	0.666	0.425	0.306
	10.00	0.865	1.640	1.070	0.965	0.661	0.420	0.303
	15.00	0.737	1.582	1.040	0.954	0.652	0.408	0.293
	20.00	0.650	1.390	1.007	0.944	0.642	0.395	0.281
	25.00	0.596	1.065	0.976	0.930	0.635	0.377	0.278
	30.00	0.571	0.658	0.934	0.899	0.627	0.360	0.254
	35.00	0.519	0.358	0.892	0.863	0.616	0.345	0.268
	40.00	0.478	0.234	0.839	0.820	0.603	0.333	0.262
	45.00	0.429	0.175	0.772	0.771	0.585	0.321	0.249
	50.00	0.376	0.117	0.697	0.714	0.562	0.309	0.232
	55.00	0.262	0.016	0.604	0.654	0.535	0.297	0.215
	60.00	0.141	0.074	0.521	0.560	0.502	0.285	0.200
	65.00	0.002	0.001	0.485	0.535	0.466	0.274	0.004
	70.00	0.305	0.678	0.471	0.482	0.411	0.277	0.183
	75.00	1.130	0.738	0.442	0.463	0.412	0.271	0.182
	80.00	0.905	0.682	0.339	0.440	0.435	0.268	0.173
LOWER SURFACE	85.00	0.606	0.554	0.261	0.383	0.403	0.250	0.176
	90.00	0.372	0.390	0.197	0.344	0.375	0.245	0.169
	95.00	0.234	0.390	0.197	0.344	0.375	0.245	0.169
	1.25	0.366	0.487	0.461	0.501	0.476	0.477	0.411
	2.50	0.382	0.524	0.504	0.508	0.491	0.477	0.378
	5.00	0.534	0.490	0.467	0.470	0.438	0.403	0.310
	7.50	0.494	0.446	0.485	0.429	0.390	0.351	0.237
	10.00	0.465	0.405	0.429	0.388	0.348	0.335	0.176
	15.00	0.433	0.328	0.388	0.312	0.288	0.250	0.120
	20.00	0.364	0.287	0.276	0.260	0.240	0.194	0.046
	25.00	0.331	0.243	0.232	0.221	0.194	0.146	0.018
	30.00	0.285	0.205	0.197	0.185	0.155	0.100	0.031
	35.00	0.236	0.172	0.171	0.149	0.115	0.060	0.021
	40.00	0.214	0.145	0.135	0.128	0.092	0.028	0.056
	45.00	0.174	0.110	0.112	0.091	0.032	0.030	0.081
	50.00	0.128	0.089	0.081	0.067	0.035	0.052	0.087
	55.00	0.127	0.069	0.070	0.045	0.035	0.067	0.084
	60.00	0.095	0.052	0.059	0.038	0.023	0.073	0.077
	65.00	0.084	0.052	0.043	0.021	0.062	0.061	0.081
	70.00	0.070	0.042	0.013	0.019	0.074	0.082	0.075
75.00	0.033	0.021	0.020	0.016	0.094	0.108	0.087	
80.00	0.039	0.005	0.013	0.023	0.099	0.124	0.090	
85.00	0.041	0.002	0.005	0.055	0.097	0.146	0.090	
90.00	0.017	0.002	0.002	0.091	0.170	0.146	0.104	
95.00	0.013	0.035	0.002	0.091	0.170	0.146	0.104	
M = 0.80 $\alpha = 15.5^\circ$								
UPPER SURFACE	0.00	0.088	2.006	1.803	0.642	0.526	0.462	0.457
	1.25	1.570	1.948	1.167	0.763	0.516	0.446	0.395
	2.50	1.809	1.948	1.167	0.759	0.516	0.448	0.394
	5.00	1.866	1.946	1.194	0.754	0.524	0.443	0.398
	7.50	1.594	1.994	1.184	0.745	0.522	0.450	0.387
	10.00	1.274	2.070	1.182	0.745	0.524	0.451	0.381
	15.00	1.053	2.025	1.163	0.734	0.525	0.451	0.382
	20.00	0.941	1.939	1.157	0.721	0.519	0.442	0.380
	25.00	0.770	1.751	1.164	0.716	0.515	0.446	0.373
	30.00	0.681	1.433	1.166	0.714	0.507	0.437	0.367
	35.00	0.618	1.058	1.156	0.714	0.498	0.429	0.357
	40.00	0.580	0.739	1.129	0.703	0.492	0.421	0.351
	45.00	0.534	0.518	1.073	0.686	0.484	0.409	0.340
	50.00	0.478	0.362	1.015	0.673	0.482	0.403	0.336
	55.00	0.362	0.190	0.947	0.657	0.479	0.407	0.326
	60.00	0.266	0.198	0.872	0.605	0.467	0.405	0.320
	65.00	0.205	0.183	0.841	0.623	0.455	0.393	0.309
	70.00	0.085	0.198	0.849	0.634	0.434	0.344	0.302
	75.00	1.179	0.618	0.657	0.551	0.434	0.338	0.290
	80.00	0.731	0.617	0.610	0.564	0.429	0.325	0.280
LOWER SURFACE	85.00	0.496	0.448	0.565	0.543	0.418	0.325	0.265
	90.00	0.368	0.247	0.589	0.529	0.404	0.319	0.265
	95.00	0.247	0.247	0.589	0.529	0.404	0.319	0.265
	1.25	0.257	0.435	0.396	0.476	0.445	0.430	0.397
	2.50	0.317	0.570	0.589	0.528	0.502	0.480	0.400
	5.00	0.530	0.883	0.840	0.583	0.477	0.436	0.352
	7.50	0.613	0.555	0.810	0.498	0.436	0.392	0.287
	10.00	0.593	0.526	0.489	0.448	0.401	0.377	0.287
	15.00	0.569	0.456	0.371	0.339	0.343	0.297	0.158
	20.00	0.497	0.417	0.384	0.388	0.289	0.238	0.080
	25.00	0.458	0.366	0.371	0.339	0.242	0.184	0.048
	30.00	0.411	0.324	0.348	0.294	0.208	0.140	0.000
	35.00	0.389	0.288	0.323	0.250	0.188	0.092	0.030
	40.00	0.330	0.257	0.213	0.173	0.121	0.051	0.000
	45.00	0.291	0.223	0.180	0.138	0.079	0.008	0.000
	50.00	0.236	0.189	0.151	0.108	0.041	0.022	0.000
	55.00	0.280	0.169	0.127	0.070	0.006	0.056	0.000
	60.00	0.195	0.146	0.108	0.039	0.017	0.074	0.000
	65.00	0.170	0.128	0.076	0.014	0.038	0.093	0.000
	70.00	0.146	0.111	0.034	0.068	0.090	0.093	0.000
75.00	0.106	0.077	0.014	0.076	0.115	0.112	0.000	
80.00	0.107	0.077	0.011	0.104	0.146	0.147	0.000	
85.00	0.108	0.089	0.089	0.164	0.146	0.175	0.000	
90.00	0.061	0.089	0.189	0.237	0.238	0.201	0.000	
95.00	0.049	0.040	0.189	0.237	0.238	0.201	0.000	

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

		PRESSURE COEFFICIENT, P, AT:						
PERCENT CHORD		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
M = 0.85 $\alpha = 0.0^\circ$								
UPPER SURFACE	.00	.177	.512	.478	.749	.476	.465	.366
	1.25	.209	.068	.041	.008	.161	.200	.173
	2.50	.133	.058	.033	.008	.066	.087	.123
	5.00	.061	.060	.039	.013	.054	.095	.062
	7.50	.022	.071	.040	.001	.025	.052	.032
	10.00	.007	.082	.051	.002	.009	.048	.014
	15.00	.028	.087	.051	.015	.011	.018	.045
	20.00	.050	.090	.042	.009	.017	.020	.058
	25.00	.071	.090	.033	.000	.031	.023	.044
	30.00	.087	.080	.019	.039	.054	.026	.038
	35.00	.080	.056	.003	.067	.083	.026	.034
	40.00	.066	.013	.037	.112	.126	.095	.021
	45.00	.028	.053	.126	.174	.185	.147	.039
	50.00	.007	.251	.347	.260	.270	.224	.048
	55.00	.006	.277	.385	.333	.323	.266	.049
	60.00	.003	.277	.385	.349	.352	.240	.053
	65.00	.001	.277	.385	.321	.272	.189	.057
	70.00	.001	.277	.385	.321	.272	.182	.071
	75.00	.001	.277	.385	.322	.268	.170	.091
80.00	.001	.277	.385	.326	.270	.165	.100	
85.00	.001	.277	.385	.331	.275	.159	.097	
90.00	.001	.277	.385	.332	.281			
95.00	.001	.277	.385					
LOWER SURFACE	.00	.177	.512	.478	.749	.476	.465	.366
	1.25	.209	.068	.041	.008	.161	.200	.173
	2.50	.133	.058	.033	.008	.066	.087	.123
	5.00	.061	.060	.039	.013	.054	.095	.062
	7.50	.022	.071	.040	.001	.025	.052	.032
	10.00	.007	.082	.051	.002	.009	.048	.014
	15.00	.028	.087	.051	.015	.011	.018	.045
	20.00	.050	.090	.042	.009	.017	.020	.058
	25.00	.071	.090	.033	.000	.031	.023	.044
	30.00	.087	.080	.019	.039	.054	.026	.038
	35.00	.080	.056	.003	.067	.083	.026	.034
	40.00	.066	.013	.037	.112	.126	.095	.021
	45.00	.028	.053	.126	.174	.185	.147	.039
	50.00	.007	.251	.347	.260	.270	.224	.048
	55.00	.006	.277	.385	.333	.323	.266	.049
	60.00	.003	.277	.385	.349	.352	.240	.053
	65.00	.001	.277	.385	.321	.272	.189	.057
	70.00	.001	.277	.385	.321	.272	.182	.071
	75.00	.001	.277	.385	.322	.268	.170	.091
80.00	.001	.277	.385	.326	.270	.165	.100	
85.00	.001	.277	.385	.331	.275	.159	.097	
90.00	.001	.277	.385	.332	.281			
95.00	.001	.277	.385					
M = 0.85 $\alpha = 4.0^\circ$								
UPPER SURFACE	.00	.196	.110	.000	.251	.123	.096	.043
	1.25	.207	.020	.019	.006	.217	.088	.026
	2.50	.123	.020	.019	.006	.217	.088	.026
	5.00	.061	.020	.019	.006	.217	.088	.026
	7.50	.022	.020	.019	.006	.217	.088	.026
	10.00	.007	.020	.019	.006	.217	.088	.026
	15.00	.028	.020	.019	.006	.217	.088	.026
	20.00	.050	.020	.019	.006	.217	.088	.026
	25.00	.071	.020	.019	.006	.217	.088	.026
	30.00	.087	.020	.019	.006	.217	.088	.026
	35.00	.080	.020	.019	.006	.217	.088	.026
	40.00	.066	.020	.019	.006	.217	.088	.026
	45.00	.028	.020	.019	.006	.217	.088	.026
	50.00	.007	.020	.019	.006	.217	.088	.026
	55.00	.006	.020	.019	.006	.217	.088	.026
	60.00	.003	.020	.019	.006	.217	.088	.026
	65.00	.001	.020	.019	.006	.217	.088	.026
	70.00	.001	.020	.019	.006	.217	.088	.026
	75.00	.001	.020	.019	.006	.217	.088	.026
80.00	.001	.020	.019	.006	.217	.088	.026	
85.00	.001	.020	.019	.006	.217	.088	.026	
90.00	.001	.020	.019	.006	.217	.088	.026	
95.00	.001	.020	.019	.006	.217	.088	.026	
LOWER SURFACE	.00	.196	.110	.000	.251	.123	.096	.043
	1.25	.207	.020	.019	.006	.217	.088	.026
	2.50	.123	.020	.019	.006	.217	.088	.026
	5.00	.061	.020	.019	.006	.217	.088	.026
	7.50	.022	.020	.019	.006	.217	.088	.026
	10.00	.007	.020	.019	.006	.217	.088	.026
	15.00	.028	.020	.019	.006	.217	.088	.026
	20.00	.050	.020	.019	.006	.217	.088	.026
	25.00	.071	.020	.019	.006	.217	.088	.026
	30.00	.087	.020	.019	.006	.217	.088	.026
	35.00	.080	.020	.019	.006	.217	.088	.026
	40.00	.066	.020	.019	.006	.217	.088	.026
	45.00	.028	.020	.019	.006	.217	.088	.026
	50.00	.007	.020	.019	.006	.217	.088	.026
	55.00	.006	.020	.019	.006	.217	.088	.026
	60.00	.003	.020	.019	.006	.217	.088	.026
	65.00	.001	.020	.019	.006	.217	.088	.026
	70.00	.001	.020	.019	.006	.217	.088	.026
	75.00	.001	.020	.019	.006	.217	.088	.026
80.00	.001	.020	.019	.006	.217	.088	.026	
85.00	.001	.020	.019	.006	.217	.088	.026	
90.00	.001	.020	.019	.006	.217	.088	.026	
95.00	.001	.020	.019	.006	.217	.088	.026	

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M = 0.85 $\alpha = 5.9^\circ$								
	0.00	.206	-.178	-.275	-.109	-.530	-.444	-.368	
	1.25	.139	1.194	1.134	.779	1.040	.742	.821	
	2.50	.251	1.169	.895	.724	.869	.628	.798	
	5.00	.381	.768	.765	.665	.732	.574	.726	
	7.50	.338	.574	.665	.607	.636	.544	.683	
	10.00	.326	.491	.600	.551	.566	.499	.639	
	15.00	.331	.441	.484	.473	.452	.435	.447	
	20.00	.329	.420	.372	.386	.338	.358	.275	
	25.00	.336	.381	.300	.315	.249	.288	.151	
	30.00	.331	.327	.239	.251	.190	.225	.095	
	35.00	.317	.253	.183	.194	.134	.167	.054	
	40.00	.312	.169	.130	.143	.080	.108	.049	
	45.00	.286	.072	.075	.091	.031	.054	.042	
	50.00	.247	.039	.011	.038	.019	.002	.041	
	55.00	.112	.119	.048	.012	.057	.038	.053	
	60.00	.015	.113	.080	.074	.095	.067	.057	
	65.00	.227	.048	.083	.047	.115	.051	.064	
	70.00	.434	.005	.111	.085	.158	.085	.075	
	75.00	1.012	.716	.404	.338	.287	.191	.078	
80.00	.771	.741	.419	.347	.293	.183	.078		
85.00	.645	.701	.434	.357	.296	.176	.078		
90.00	.443	.638	.439	.350	.289	.163	.075		
95.00	.295	.545	.413	.338	.265	.149	.068		
LOWER SURFACE	1.25	.378	.441	.438	.420	.429	.428	.386	
	2.50	.373	.366	.360	.344	.359	.353	.303	
	5.00	.337	.288	.276	.292	.264	.247	.207	
	7.50	.299	.234	.223	.241	.206	.189	.128	
	10.00	.263	.202	.186	.193	.162	.180	.069	
	15.00	.220	.143	.136	.116	.117	.095	.031	
	20.00	.182	.102	.085	.074	.062	.039	.072	
	25.00	.154	.067	.048	.034	.022	.004	.106	
	30.00	.098	.032	.013	.008	.008	.038	.123	
	35.00	.068	.011	.001	.023	.040	.078	.128	
	40.00	.048	.015	.001	.051	.066	.110	.134	
	45.00	.018	.040	.060	.077	.092	.136	.134	
	50.00	.014	.063	.081	.096	.112	.158	.130	
	55.00	.031	.071	.089	.115	.130	.167	.126	
	60.00	.049	.077	.093	.122	.135	.163	.108	
	65.00	.063	.084	.100	.130	.137	.154	.098	
	70.00	.059	.077	.130	.147	.139	.128	.093	
	75.00	.076	.086	.134	.140	.141	.102	.068	
	80.00	.092	.095	.139	.142	.145	.091	.063	
	90.00	.083	.109	.162	.157	.104	.110	.046	
95.00	.111	.145	.181	.176	.159	.109	.041		
UPPER SURFACE	M = 0.85 $\alpha = 8.0^\circ$								
	0.00	.204	-.493	-.543	-.070	-.843	-.797	-.785	
	1.25	.277	1.370	.956	.946	1.008	.693	.829	
	2.50	.407	1.388	.897	.877	.972	.679	.815	
	3.75	.500	1.243	.859	.828	.943	.645	.805	
	5.00	.522	.842	.774	.781	.922	.631	.801	
	7.50	.499	.735	.733	.750	.896	.610	.788	
	10.00	.484	.647	.668	.696	.772	.572	.717	
	15.00	.452	.600	.589	.661	.567	.533	.624	
	20.00	.447	.531	.518	.637	.438	.492	.500	
	25.00	.433	.424	.456	.585	.365	.462	.372	
	30.00	.441	.323	.389	.507	.323	.420	.260	
	35.00	.448	.243	.320	.407	.280	.378	.183	
	40.00	.391	.157	.259	.300	.243	.324	.155	
	45.00	.319	.068	.201	.215	.201	.271	.140	
	50.00	.170	.016	.142	.149	.166	.225	.140	
	55.00	.036	.083	.118	.078	.129	.166	.140	
	60.00	.173	.073	.124	.103	.089	.178	.144	
	65.00	.381	.075	.062	.038	.012	.107	.147	
	70.00	.977	.716	.420	.350	.313	.241	.143	
75.00	.796	.727	.451	.366	.317	.228	.137		
80.00	.683	.682	.443	.342	.288	.193	.128		
85.00	.531	.617	.412	.305	.257	.162	.120		
90.00	.399	.529	.370	.263	.227	.143	.109		
LOWER SURFACE	1.25	.403	.511	.492	.503	.473	.473	.433	
	2.50	.434	.452	.433	.409	.428	.428	.375	
	3.75	.414	.376	.356	.367	.348	.334	.285	
	5.00	.394	.320	.304	.314	.287	.277	.210	
	7.50	.356	.286	.288	.268	.248	.267	.143	
	10.00	.311	.218	.207	.191	.193	.179	.051	
	15.00	.243	.174	.158	.142	.138	.119	.020	
	20.00	.228	.138	.114	.108	.100	.075	.063	
	25.00	.170	.104	.079	.071	.060	.033	.093	
	30.00	.138	.069	.053	.036	.032	.010	.104	
	35.00	.108	.044	.024	.006	.003	.045	.111	
	40.00	.077	.017	.001	.025	.085	.079	.114	
	45.00	.046	.008	.028	.046	.047	.105	.110	
	50.00	.023	.018	.041	.063	.069	.122	.104	
	55.00	.001	.034	.048	.074	.081	.128	.091	
	60.00	.011	.043	.089	.089	.081	.121	.083	
	65.00	.018	.018	.018	.018	.018	.018	.078	
	70.00	.038	.046	.121	.108	.090	.083	.060	
	75.00	.033	.082	.099	.101	.083	.083	.059	
	80.00	.065	.088	.107	.106	.088	.088	.057	
90.00	.065	.100	.128	.123	.088	.089	.049		
95.00	.107	.149	.140	.133	.120	.088	.055		

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:											
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2					
UPPER SURFACE	M = 0.85 α = 11.3°													
	0.00	.170	-	.886	-	.990	-	.535	-	.730	-	.459	-	.457
	1.25	.450	-	.468	-	.135	-	.975	-	.658	-	.424	-	.359
	2.50	.614	-	.446	-	.090	-	.970	-	.654	-	.422	-	.360
	3.75	.735	-	.483	-	.108	-	.956	-	.650	-	.410	-	.359
	5.00	.769	-	.476	-	.108	-	.963	-	.642	-	.418	-	.353
	7.50	.749	-	.440	-	.054	-	.957	-	.636	-	.412	-	.351
	10.00	.723	-	.350	-	.959	-	.944	-	.617	-	.404	-	.348
	15.00	.670	-	.298	-	.862	-	.934	-	.599	-	.395	-	.345
	20.00	.641	-	.216	-	.830	-	.927	-	.581	-	.377	-	.340
	25.00	.621	-	.1975	-	.850	-	.913	-	.569	-	.377	-	.332
	30.00	.598	-	.152	-	.855	-	.899	-	.556	-	.368	-	.328
	35.00	.588	-	.103	-	.808	-	.875	-	.540	-	.361	-	.317
	40.00	.589	-	.042	-	.808	-	.841	-	.524	-	.351	-	.306
	45.00	.582	-	.067	-	.768	-	.795	-	.504	-	.346	-	.289
	50.00	.561	-	.001	-	.701	-	.738	-	.483	-	.340	-	.276
	55.00	.525	-	.009	-	.627	-	.622	-	.464	-	.334	-	.262
	60.00	.496	-	.132	-	.575	-	.619	-	.442	-	.328	-	.253
	65.00	.452	-	.159	-	.477	-	.581	-	.400	-	.321	-	.235
	70.00	.395	-	.141	-	.477	-	.491	-	.409	-	.307	-	.238
75.00	.331	-	.138	-	.533	-	.518	-	.426	-	.302	-	.232	
80.00	.271	-	.105	-	.454	-	.506	-	.428	-	.296	-	.230	
85.00	.254	-	.038	-	.383	-	.464	-	.407	-	.283	-	.230	
90.00	.254	-	.512	-	.311	-	.415	-	.389	-	.275	-	.224	
95.00	.463	-	-	-	-	-	-	-	-	-	-	-	-	
LOWER SURFACE	1.25	.411	-	.568	-	.515	-	.522	-	.498	-	.489	-	.441
	2.50	.484	-	.550	-	.509	-	.495	-	.489	-	.479	-	.402
	3.75	.528	-	.491	-	.455	-	.466	-	.434	-	.409	-	.329
	5.00	.551	-	.439	-	.410	-	.416	-	.386	-	.346	-	.259
	7.50	.506	-	.407	-	.378	-	.369	-	.338	-	.337	-	.197
	10.00	.461	-	.339	-	.318	-	.305	-	.288	-	.252	-	.099
	15.00	.382	-	.292	-	.261	-	.251	-	.232	-	.193	-	.004
	20.00	.355	-	.245	-	.223	-	.211	-	.188	-	.141	-	.032
	25.00	.302	-	.209	-	.185	-	.178	-	.166	-	.092	-	.079
	30.00	.248	-	.175	-	.156	-	.141	-	.113	-	.043	-	.111
	35.00	.227	-	.149	-	.120	-	.109	-	.078	-	.005	-	.130
	40.00	.191	-	.114	-	.092	-	.077	-	.041	-	.038	-	.142
	45.00	.135	-	.085	-	.062	-	.048	-	.016	-	.077	-	.151
	50.00	.133	-	.066	-	.047	-	.024	-	.014	-	.110	-	.160
	55.00	.100	-	.053	-	.027	-	.009	-	.034	-	.128	-	.154
	60.00	.088	-	.035	-	.007	-	.008	-	.054	-	.146	-	.152
	65.00	.066	-	-	-	-	-	-	-	-	-	-	-	.155
	70.00	.022	-	.022	-	.045	-	.051	-	.093	-	.108	-	.149
	75.00	.031	-	.007	-	.029	-	.042	-	.114	-	.134	-	.158
	80.00	.029	-	.033	-	.035	-	.056	-	.140	-	.160	-	.163
85.00	.005	-	.062	-	.051	-	.090	-	.140	-	.191	-	.162	
90.00	.009	-	.125	-	.064	-	.137	-	.223	-	.201	-	.176	
95.00	-	-	-	-	-	-	-	-	-	-	-	-	-	
M = 0.35 α = 15.5°														
UPPER SURFACE	0.00	.055	-	1.282	-	1.085	-	.910	-	.559	-	.495	-	.491
	1.25	.640	-	1.646	-	1.035	-	.766	-	.555	-	.484	-	.454
	2.50	.862	-	1.612	-	1.009	-	.759	-	.560	-	.491	-	.449
	3.75	1.040	-	1.617	-	1.043	-	.747	-	.565	-	.483	-	.450
	5.00	1.065	-	1.621	-	1.042	-	.739	-	.569	-	.493	-	.446
	7.50	1.040	-	1.598	-	1.048	-	.728	-	.571	-	.495	-	.435
	10.00	1.010	-	1.483	-	1.038	-	.700	-	.573	-	.495	-	.429
	15.00	.929	-	1.397	-	1.016	-	.688	-	.573	-	.492	-	.429
	20.00	.866	-	1.314	-	.991	-	.675	-	.567	-	.479	-	.425
	25.00	.805	-	1.257	-	.979	-	.664	-	.562	-	.485	-	.434
	30.00	.737	-	1.182	-	.987	-	.646	-	.559	-	.477	-	.434
	35.00	.628	-	1.082	-	.935	-	.630	-	.555	-	.473	-	.431
	40.00	.579	-	.953	-	.898	-	.615	-	.548	-	.466	-	.427
	45.00	.603	-	.800	-	.863	-	.607	-	.538	-	.462	-	.420
	50.00	.544	-	.680	-	.827	-	.601	-	.527	-	.463	-	.411
	55.00	.420	-	.644	-	.800	-	.544	-	.521	-	.476	-	.401
	60.00	.318	-	.702	-	.820	-	.596	-	.526	-	.495	-	.403
	65.00	.096	-	.525	-	.775	-	.557	-	.493	-	.479	-	.008
	70.00	1.147	-	.844	-	.735	-	.552	-	.475	-	.410	-	.372
	75.00	.894	-	.814	-	.708	-	.565	-	.477	-	.421	-	.357
80.00	.734	-	.591	-	.693	-	.573	-	.473	-	.418	-	.351	
85.00	.565	-	.457	-	.669	-	.560	-	.470	-	.406	-	.334	
90.00	.430	-	.371	-	.653	-	.553	-	.462	-	.387	-	.312	
95.00	-	-	-	-	-	-	-	-	-	-	-	-	-	
LOWER SURFACE	1.25	.331	-	.575	-	.500	-	.563	-	.469	-	.449	-	.420
	2.50	.476	-	.614	-	.556	-	.541	-	.504	-	.488	-	.420
	3.75	.599	-	.592	-	.539	-	.532	-	.475	-	.443	-	.374
	5.00	.698	-	.546	-	.504	-	.491	-	.433	-	.392	-	.314
	7.50	.650	-	.520	-	.472	-	.447	-	.395	-	.383	-	.255
	10.00	.597	-	.449	-	.423	-	.393	-	.346	-	.302	-	.160
	15.00	.512	-	.404	-	.365	-	.348	-	.293	-	.242	-	.074
	20.00	.478	-	.358	-	.325	-	.295	-	.244	-	.191	-	.005
	25.00	.484	-	.318	-	.284	-	.254	-	.199	-	.136	-	.055
	30.00	.382	-	.278	-	.248	-	.218	-	.154	-	.082	-	.106
	35.00	.340	-	.243	-	.209	-	.174	-	.113	-	.037	-	.136
	40.00	.298	-	.207	-	.175	-	.134	-	.072	-	.014	-	.155
	45.00	.233	-	.174	-	.141	-	.095	-	.034	-	.063	-	.173
	50.00	.233	-	.148	-	.116	-	.061	-	.005	-	.105	-	.187
	55.00	.193	-	.127	-	.091	-	.038	-	.036	-	.138	-	.190
	60.00	.178	-	.099	-	.060	-	.004	-	.066	-	.174	-	.213
	65.00	.145	-	-	-	-	-	-	-	-	-	-	-	.208
	70.00	.087	-	.075	-	.011	-	.086	-	.123	-	.131	-	.203
	75.00	.088	-	.045	-	.021	-	.090	-	.153	-	.171	-	.233
	80.00	.088	-	.018	-	.088	-	.127	-	.189	-	.218	-	.228
85.00	.033	-	.015	-	.116	-	.181	-	.187	-	.279	-	.227	
90.00	.007	-	.062	-	.194	-	.244	-	.287	-	.286	-	.242	
95.00	-	-	-	-	-	-	-	-	-	-	-	-	-	

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, C_p , AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	$M = 0.90 \quad \alpha = 0.0^\circ$								
	.00	.451	.522	.484	.754	.480	.477	.372	
	1.25	.441	.071	.056	.001	.146	.203	.152	
	2.50	.147	.061	.045	.009	.055	.073	.101	
	5.00	.074	.061	.044	.009	.046	.092	.041	
	7.50	.034	.074	.046	.010	.020	.057	.022	
	10.00	.014	.084	.051	.005	.003	.042	.036	
	15.00	.022	.093	.049	.027	.007	.007	.077	
	20.00	.046	.096	.037	.018	.007	.012	.087	
	25.00	.070	.096	.026	.009	.015	.022	.071	
	30.00	.087	.081	.012	.004	.030	.022	.054	
	35.00	.087	.052	.011	.030	.051	.035	.044	
	40.00	.088	.003	.040	.058	.082	.058	.036	
	45.00	.086	.073	.085	.107	.124	.092	.055	
	50.00	.069	.170	.158	.170	.186	.148	.076	
	55.00	.012	.271	.251	.256	.270	.226	.089	
	60.00	.160	.280	.310	.348	.318	.257	.074	
	65.00	.359	.274	.314	.337	.320	.228	.068	
	70.00	.587	.233	.334	.353	.329	.230	.073	
LOWER SURFACE	.00	.159	.793	.385	.336	.284	.200	.077	
	1.25	.592	.842	.420	.336	.281	.193	.087	
	2.50	.495	.802	.432	.339	.281	.186	.109	
	5.00	.322	.729	.439	.340	.285	.177	.117	
	7.50	.218	.592	.426	.330	.286	.172	.115	
	10.00	.400	.074	.125	.194	.325	.404	.503	
	15.00	.137	.056	.100	.168	.243	.301	.387	
	20.00	.095	.061	.104	.105	.220	.265	.298	
	25.00	.039	.074	.120	.137	.228	.268	.323	
	30.00	.014	.093	.131	.164	.233	.212	.335	
	35.00	.022	.116	.156	.217	.219	.250	.335	
	40.00	.072	.137	.181	.222	.237	.276	.307	
	45.00	.052	.158	.206	.241	.258	.297	.253	
	50.00	.112	.182	.227	.244	.258	.305	.235	
	55.00	.129	.199	.240	.256	.278	.316	.208	
	60.00	.143	.213	.259	.264	.290	.341	.217	
	65.00	.189	.245	.276	.272	.313	.327	.221	
	70.00	.178	.265	.286	.274	.316	.311	.212	
	75.00	.213	.262	.268	.282	.317	.271	.191	
80.00	.229	.259	.243	.277	.295	.244	.169		
85.00	.231	.230	.206	.265	.274	.201	.134		
90.00	.238						.132		
95.00	.245	.207	.212	.267	.247	.125	.119		
100.00	.239	.184	.216	.247	.228	.132	.097		
105.00	.235	.147	.223	.226	.206	.128	.090		
110.00	.204	.117	.236	.234	.152	.128	.081		
115.00	.171	.117	.240	.237	.207	.131	.081		
$M = 0.90 \quad \alpha = 3.9^\circ$									
UPPER SURFACE	.00	.249	.175	.060	.309	.067	.063	.064	
	1.25	.017	.903	.039	.981	.998	.940	.899	
	2.50	.087	.794	.926	.860	.896	.882	.805	
	5.00	.160	.390	.509	.458	.415	.344	.371	
	7.50	.188	.355	.419	.346	.352	.308	.353	
	10.00	.214	.339	.368	.323	.327	.292	.321	
	15.00	.235	.323	.283	.283	.256	.240	.191	
	20.00	.241	.307	.217	.426	.207	.184	.142	
	25.00	.241	.298	.177	.185	.161	.138	.102	
	30.00	.247	.264	.147	.146	.116	.100	.082	
	35.00	.247	.183	.106	.100	.066	.054	.067	
	40.00	.246	.091	.060	.045	.016	.001	.052	
	45.00	.237	.015	.002	.019	.042	.052	.052	
	50.00	.194	.127	.073	.081	.107	.102	.052	
	55.00	.064	.192	.128	.138	.148	.137	.064	
	60.00	.070	.167	.145	.191	.172	.150	.060	
	65.00	.148	.014	.144	.181	.177	.140	.064	
	70.00	.148	.081	.163	.193		.134	.064	
	75.00	.082	.755	.430	.339	.269	.191	.063	
80.00	.588	.767	.433	.340	.269	.182	.062		
85.00	.586	.733	.442	.344	.269	.176	.069		
90.00	.454	.672	.449	.351	.271	.171	.070		
95.00	.306	.583	.436	.346	.264	.165	.065		
LOWER SURFACE	.00	.337	.341	.349	.328	.332	.338	.283	
	1.25	.314	.270	.267	.249	.254	.255	.200	
	2.50	.260	.200	.186	.197	.166	.150	.111	
	5.00	.284	.148	.136	.142	.109	.097	.031	
	7.50	.188	.121	.104	.098	.071	.097	.027	
	10.00	.148	.073	.058	.037	.031	.011	.119	
	15.00	.093	.036	.013	.008	.016	.038	.154	
	20.00	.089	.004	.023	.038	.035	.081	.172	
	25.00	.035	.084	.051	.059	.079	.113	.180	
	30.00	.008	.056	.075	.088	.113	.153	.177	
	35.00	.011	.069	.099	.115	.127	.186	.179	
	40.00	.036	.100	.120	.139	.162	.217	.180	
	45.00	.047	.119	.137	.158	.175	.239	.170	
	50.00	.087	.123	.143	.178	.189	.234	.159	
	55.00	.104	.131	.142	.179	.200	.223	.128	
	60.00	.121	.134	.144	.184	.202	.204	.125	
	65.00	.131	.119	.184	.198	.202	.132	.119	
	70.00	.131	.119	.173	.188	.198	.128	.095	
	75.00	.136	.123	.180	.185	.192	.125	.084	
80.00	.119	.124	.202	.200	.151	.140	.060		
85.00	.141	.169	.218	.219	.200	.133	.049		

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P_i AT:										
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2				
UPPER SURFACE	M = 0.90 $\alpha = 5.9^\circ$												
	0.00	.259	-	-	.189	-	.165	-	.440	-	.404	-	.345
	1.25	.085	-	1.067	.206	-	1.086	-	1.130	-	.748	-	.877
	2.50	.200	-	1.093	.127	-	1.026	-	1.022	-	.622	-	.861
	3.75	.283	-	.782	.910	-	.919	-	.836	-	.576	-	.813
	5.00	.310	-	.636	.778	-	.788	-	.712	-	.560	-	.788
	7.50	.303	-	.549	.693	-	.657	-	.609	-	.526	-	.747
	10.00	.315	-	.462	.560	-	.496	-	.507	-	.466	-	.603
	15.00	.315	-	.427	.383	-	.370	-	.396	-	.396	-	.411
	20.00	.309	-	.411	.266	-	.297	-	.281	-	.324	-	.211
	25.00	.329	-	.404	.202	-	.242	-	.202	-	.262	-	.090
	30.00	.344	-	.363	.158	-	.189	-	.147	-	.207	-	.056
	35.00	.333	-	.160	.119	-	.137	-	.094	-	.152	-	.039
	40.00	.334	-	.031	.072	-	.075	-	.053	-	.098	-	.040
	45.00	.342	-	.072	.017	-	.014	-	.014	-	.046	-	.044
	50.00	.335	-	.146	.036	-	.047	-	.025	-	.004	-	.064
	55.00	.135	-	.134	.060	-	.121	-	.061	-	.023	-	.071
	60.00	.025	-	.021	.055	-	.113	-	.082	-	.008	-	.088
65.00	.232	-	.000	.083	-	.134	-	.082	-	.045	-	.098	
70.00	.434	-	.751	.437	-	.354	-	.280	-	.209	-	.098	
75.00	.066	-	.768	.444	-	.356	-	.287	-	.207	-	.096	
80.00	.594	-	.740	.454	-	.363	-	.294	-	.193	-	.092	
85.00	.470	-	.677	.459	-	.363	-	.290	-	.172	-	.086	
90.00	.332	-	.592	.440	-	.351	-	.273	-	.153	-	.075	
95.00		-			-		-		-		-		
LOWER SURFACE	1.25	.394	.452	.439	.417	.430	.442	.388					
	2.50	.396	.357	.362	.354	.362	.368	.307					
	3.75	.364	.300	.278	.293	.271	.260	.213					
	5.00	.323	.248	.224	.241	.214	.201	.131					
	7.50	.285	.216	.187	.191	.171	.197	.064					
	10.00	.242	.155	.137	.113	.123	.107	.036					
	15.00	.181	.115	.085	.076	.067	.048	.100					
	20.00	.173	.078	.047	.030	.027	.004	.153					
	25.00	.115	.046	.016	.006	.005	.036	.164					
	30.00	.083	.016	.011	.007	.038	.079	.166					
	35.00	.062	.007	.039	.059	.064	.117	.162					
	40.00	.031	.037	.062	.086	.100	.155	.161					
	45.00	.001	.058	.084	.111	.117	.182	.156					
	50.00	.022	.067	.096	.128	.138	.194	.145					
	55.00	.042	.074	.101	.140	.147	.191	.130					
	60.00	.053	.085	.107	.144	.150	.171	.107					
	65.00	.055						.102					
	70.00	.076	.078	.143	.171	.162	.116	.085					
75.00	.073	.089	.153	.162	.162	.104	.077						
80.00	.093	.102	.165	.163	.161	.107	.070						
85.00	.091	.115	.194	.185	.124	.111	.055						
90.00	.125	.171	.211	.205	.178	.105	.047						
95.00													
M = 0.90 $\alpha = 8.0^\circ$													
UPPER SURFACE	0.00	.257	.340	.401	.031	.689	.765	.683					
	1.25	.193	1.165	1.146	1.172	.904	.837	.774					
	2.50	.319	1.176	1.001	1.111	.868	.837	.764					
	3.75	.420	1.109	.989	1.000	.836	.720	.751					
	5.00	.458	1.003	.830	.909	.808	.702	.748					
	7.50	.441	.926	.793	.843	.780	.692	.738					
	10.00	.441	.778	.688	.755	.723	.662	.701					
	15.00	.416	.637	.592	.682	.602	.628	.643					
	20.00	.416	.538	.516	.652	.508	.587	.537					
	25.00	.426	.494	.431	.600	.445	.534	.453					
	30.00	.423	.481	.358	.523	.403	.514	.354					
	35.00	.419	.335	.294	.418	.361	.445	.324					
	40.00	.420	.106	.245	.293	.326	.361	.291					
	45.00	.433	.012	.207	.184	.287	.281	.221					
	50.00	.468	.111	.175	.108	.255	.208	.210					
	55.00	.068	.114	.171	.026	.221	.152	.201					
	60.00	.152	.034	.101	.063	.177	.132	.193					
	65.00	.370	.006	.118	.004	.156	.061	.190					
70.00	1.061	.774	.471	.362	.340	.248	.185						
75.00	.686	.776	.808	.374	.341	.243	.172						
80.00	.609	.753	.482	.363	.310	.201	.158						
85.00	.000	.688	.448	.333	.278	.171	.148						
90.00	.421	.602	.409	.302	.244	.146	.135						
95.00													
LOWER SURFACE	1.25	.422	.521	.490	.476	.470	.468	.433					
	2.50	.453	.454	.428	.400	.418	.420	.371					
	3.75	.438	.385	.381	.360	.337	.323	.287					
	5.00	.413	.329	.298	.308	.279	.265	.204					
	7.50	.371	.297	.260	.257	.235	.259	.143					
	10.00	.326	.234	.203	.185	.183	.170	.046					
	15.00	.258	.184	.150	.139	.127	.108	.017					
	20.00	.242	.143	.109	.098	.085	.059	.091					
	25.00	.180	.110	.076	.068	.052	.022	.123					
	30.00	.142	.077	.043	.030	.017	.024	.137					
	35.00	.119	.050	.016	.000	.013	.063	.154					
	40.00	.089	.007	.011	.012	.047	.098	.143					
	45.00	.053	.007	.038	.052	.067	.131	.133					
	50.00	.025	.019	.051	.078	.089	.150	.128					
	55.00	.008	.033	.082	.089	.100	.154	.107					
	60.00	.009	.043	.074	.101	.108	.150	.092					
	65.00	.012						.089					
	70.00	.036	.043	.135	.123	.118	.106	.067					
75.00	.036	.068	.126	.119	.122	.102	.067						
80.00	.060	.088	.137	.120	.122	.099	.065						
85.00	.065	.112	.159	.137	.087	.101	.057						
90.00	.113	.175	.173	.159	.140	.099	.065						
95.00													

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:						
			0.135h/2	0.25h/2	0.40h/2	0.55h/2	0.70h/2	0.85h/2	0.95h/2
UPPER SURFACE	M = 0.90 $\alpha = 11.3^\circ$								
	.00	.214	-.694	-.777	-.352	-.917	-.548	-.468	
	1.25	.344	1.388	1.288	1.209	.840	.504	.380	
	2.50	.497	1.311	1.113	1.193	.834	.505	.377	
	5.00	.619	1.346	1.088	1.168	.825	.491	.374	
	7.50	.689	1.328	1.039	1.189	.820	.494	.378	
	10.00	.648	1.275	1.010	1.208	.818	.487	.371	
	12.50	.640	1.180	.948	1.149	.812	.477	.370	
	15.00	.601	1.128	.885	1.108	.804	.470	.368	
	17.50	.578	1.041	.825	1.079	.794	.462	.365	
	20.00	.543	.951	.765	1.039	.784	.449	.364	
	22.50	.508	.851	.684	1.000	.777	.438	.361	
	25.00	.468	.751	.647	.957	.759	.430	.356	
	27.50	.423	.651	.607	.917	.739	.418	.350	
	30.00	.378	.548	.574	.878	.706	.408	.343	
	32.50	.333	.443	.549	.861	.668	.394	.329	
	35.00	.288	.338	.518	.895	.618	.383	.329	
	LOWER SURFACE	.00	.201	-.184	-.802	-.318	-.919	-.544	-.473
1.25		.103	.797	.864	.367	.498	.162	.265	
2.50		.010	.794	.810	.379	.519	.162	.259	
5.00		.034	.766	.480	.356	.487	.150	.256	
7.50		.059	.710	.484	.321	.448	.140	.254	
10.00		.085	.617	.365	.297	.410	.128	.244	
12.50		.112	.530	.258	.248	.390	.120	.240	
15.00		.140	.452	.151	.204	.477	.108	.214	
17.50		.169	.380	.043	.153	.419	.100	.216	
20.00		.198	.307	-.070	.100	.368	.093	.219	
22.50		.227	.234	-.178	.048	.325	.086	.208	
25.00		.256	.161	-.285	.000	.281	.079	.205	
27.50		.285	.088	-.392	-.052	.238	.072	.202	
30.00		.314	.015	-.499	-.104	.195	.065	.200	
32.50		.343	-.058	-.606	-.156	.152	.058	.199	
35.00		.372	-.135	-.713	-.208	.109	.051	.196	
37.50		.401	-.212	-.820	-.260	.066	.044	.193	
40.00		.430	-.289	-.927	-.312	.023	.037	.190	
UPPER SURFACE	M = 0.90 $\alpha = 15.5^\circ$								
	.00	.119	1.063	1.117	.818	.545	.510	.510	
	1.25	.494	1.478	1.082	.776	.545	.508	.467	
	2.50	.708	1.447	1.017	.777	.531	.511	.465	
	5.00	.880	1.488	1.043	.770	.526	.498	.467	
	7.50	.914	1.477	1.026	.768	.528	.498	.470	
	10.00	.891	1.426	1.011	.761	.525	.499	.468	
	12.50	.876	1.381	.993	.744	.517	.491	.454	
	15.00	.817	1.300	.970	.728	.510	.481	.453	
	17.50	.713	1.199	.930	.715	.503	.478	.448	
	20.00	.589	1.079	.878	.700	.498	.470	.438	
	22.50	.472	.948	.818	.678	.492	.464	.437	
	25.00	.364	.808	.757	.657	.485	.458	.424	
	27.50	.264	.664	.699	.636	.478	.451	.418	
	30.00	.178	.521	.644	.622	.471	.444	.415	
	32.50	.103	.381	.598	.614	.464	.437	.408	
	35.00	.043	.248	.557	.606	.457	.430	.403	
	LOWER SURFACE	.00	.356	.617	.530	.573	.484	.456	.428
1.25		.519	.642	.474	.543	.514	.486	.458	
2.50		.680	.611	.408	.533	.484	.456	.428	
5.00		.718	.571	.318	.491	.448	.420	.392	
7.50		.669	.536	.228	.451	.408	.380	.352	
10.00		.617	.478	.138	.412	.369	.341	.313	
12.50		.524	.428	.048	.372	.329	.301	.273	
15.00		.438	.379	-.042	.346	.303	.275	.247	
17.50		.344	.339	-.132	.320	.277	.249	.221	
20.00		.244	.289	-.222	.299	.256	.228	.200	
22.50		.144	.207	-.312	.282	.239	.211	.183	
25.00		.044	.124	-.402	.264	.221	.193	.165	
27.50		-.056	.044	-.492	.246	.203	.175	.147	
30.00		-.156	-.064	-.582	.228	.185	.157	.129	
32.50		-.256	-.164	-.672	.210	.167	.139	.111	
35.00		-.356	-.264	-.762	.192	.149	.121	.093	
37.50		-.456	-.364	-.852	.174	.131	.103	.075	
40.00		-.556	-.464	-.942	.156	.113	.085	.057	

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M = 0.94 α = 0.0°								
	.00	.291	.536	.489	.779	.479	.471	.378	
	1.25	.241	.060	.068	.014	.116	.180	.145	
	2.50	.170	.047	.049	.019	.026	.050	.100	
	5.00	.096	.051	.048	.018	.018	.072	.042	
	7.50	.053	.064	.047	.003	.009	.037	.023	
	10.00	.036	.076	.054	.010	.026	.025	.040	
	15.00	.007	.086	.045	.036	.023	.009	.100	
	20.00	.037	.095	.029	.029	.020	.007	.107	
	25.00	.080	.101	.015	.023	.011	.009	.083	
	30.00	.079	.086	.002	.009	.008	.020	.053	
	35.00	.080	.048	.019	.013	.028	.037	.056	
	40.00	.089	.018	.088	.087	.112	.074	.078	
	45.00	.083	.058	.156	.152	.177	.135	.106	
	50.00	.066	.196	.248	.248	.262	.211	.127	
	55.00	.025	.295	.302	.337	.308	.241	.103	
	60.00	.187	.297	.302	.330	.311	.216	.088	
	65.00	.375	.248	.323	.346	.323	.216	.093	
70.00	.604	.852	.424	.368	.309	.225	.096		
75.00	1.192	.895	.462	.369	.305	.221	.109		
80.00	.648	.895	.465	.367	.306	.214	.131		
85.00	.519	.849	.468	.367	.310	.205	.136		
90.00	.397	.781	.468	.367	.310	.205	.136		
95.00	.318	.650	.486	.360	.317	.198	.135		
LOWER SURFACE	M = 0.94 α = 0.0°								
	1.25	.218	.054	.100	.171	.320	.412	.535	
	2.50	.161	.034	.079	.153	.240	.319	.409	
	5.00	.110	.038	.084	.096	.231	.289	.315	
	7.50	.059	.059	.105	.134	.247	.291	.342	
	10.00	.035	.072	.116	.160	.261	.240	.334	
	15.00	.000	.104	.147	.214	.253	.297	.355	
	20.00	.052	.125	.172	.237	.277	.325	.374	
	25.00	.035	.144	.197	.262	.309	.350	.381	
	30.00	.095	.166	.224	.269	.322	.362	.349	
	35.00	.108	.184	.247	.295	.343	.378	.298	
	40.00	.126	.200	.271	.320	.365	.384	.293	
	45.00	.151	.231	.297	.352	.399	.382	.283	
	50.00	.170	.258	.319	.373	.409	.395	.266	
	55.00	.202	.267	.329	.383	.389	.392	.162	
	60.00	.220	.269	.329	.370	.309	.286	.129	
	65.00	.229	.264	.314	.324			.127	
	70.00	.237						.121	
75.00	.253	.276	.262	.308	.300	.127	.107		
80.00	.260	.279	.251	.292	.281	.117	.103		
85.00	.273	.252	.254	.270	.278	.119	.091		
90.00	.275	.197	.278	.262	.229	.135	.095		
95.00	.269	.162	.290	.268	.276	.148			
UPPER SURFACE	M = 0.94 α = 3.9°								
	.00	.277	.235	.115	.352	.011	.028	.078	
	1.25	.054	.850	.973	.847	.915	.907	.970	
	2.50	.043	.793	.893	.705	.854	.882	.889	
	5.00	.125	.352	.544	.497	.503	.560	.543	
	7.50	.160	.334	.430	.393	.383	.367	.434	
	10.00	.161	.325	.413	.379	.354	.329	.339	
	15.00	.188	.305	.391	.322	.276	.277	.296	
	20.00	.200	.302	.301	.235	.234	.216	.174	
	25.00	.230	.306	.250	.201	.202	.174	.109	
	30.00	.239	.308	.193	.158	.147	.130	.092	
	35.00	.234	.304	.081	.119	.098	.077	.080	
	40.00	.233	.304	.044	.072	.041	.025	.073	
	45.00	.233	.269	.010	.012	.021	.040	.075	
	50.00	.248	.060	.081	.055	.092	.095	.087	
	55.00	.080	.212	.127	.117	.132	.127	.102	
	60.00	.086	.168	.142	.177	.156	.139	.092	
	65.00	.266	.135	.136	.163	.160	.129	.094	
70.00	.473	.024	.154	.173	.175	.119	.090		
75.00	.218	.863	.497	.386	.283	.210	.087		
80.00	.717	.908	.495	.382	.283	.206	.085		
85.00	.688	.854	.486	.383	.285	.203	.085		
90.00	.527	.768	.481	.383	.289	.197	.087		
95.00	.395	.660	.469	.376	.286	.191	.076		
LOWER SURFACE	M = 0.94 α = 3.9°								
	1.25	.347	.342	.338	.323	.323	.331	.280	
	2.50	.326	.273	.258	.246	.248	.251	.199	
	5.00	.279	.204	.179	.188	.160	.140	.112	
	7.50	.235	.156	.130	.135	.104	.089	.036	
	10.00	.200	.128	.098	.090	.066	.090	.020	
	15.00	.153	.080	.052	.028	.030	.004	.124	
	20.00	.102	.043	.007	.011	.025	.050	.085	
	25.00	.100	.007	.032	.048	.059	.093	.108	
	30.00	.042	.022	.060	.070	.091	.127	.108	
	35.00	.016	.050	.085	.100	.118	.166	.127	
	40.00	.002	.072	.113	.124	.151	.204	.168	
	45.00	.034	.104	.134	.147	.185	.253	.245	
	50.00	.063	.128	.154	.170	.210	.299	.285	
	55.00	.086	.135	.161	.187	.235	.330	.309	
	60.00	.107	.144	.185	.199	.247	.347	.316	
	65.00	.127	.147	.148	.216	.247	.321	.273	
	70.00	.119						.182	
75.00	.144	.136	.155	.253	.239	.145	.107		
80.00	.137	.137	.179	.251	.229	.130	.099		
85.00	.141	.125	.204	.253	.238	.133	.090		
90.00	.141	.100	.243	.256	.204	.143	.066		
95.00	.174	.122	.282	.258	.265	.142	.053		

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

PERCENT CHORD		PRESSURE COEFFICIENT, P, AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M = 0.94 α = 5.9°							
	.00	.289	.014	-.096	-.253	-.361	-.381	-.313
	1.25	.028	.997	-.180	-.155	-.176	-.013	-.967
	2.50	.139	.013	-.137	-.183	-.176	.947	-.954
	5.00	.229	.697	-.109	-.117	-.103	.834	-.872
	7.50	.263	.542	-.002	-.065	-.031	.759	-.817
	10.00	.260	.476	-.864	-.013	.906	.665	-.758
	15.00	.276	.417	-.484	.668	.468	.496	-.625
	20.00	.273	.398	-.441	.266	.317	.472	-.505
	25.00	.291	.391	-.366	.197	.257	.318	-.371
	30.00	.300	.388	-.179	.148	.203	.257	-.238
	35.00	.305	.392	-.111	.118	.153	.197	-.137
	40.00	.313	.327	-.062	.081	.103	.137	-.084
	45.00	.326	.054	-.011	.027	.041	.080	-.072
	50.00	.361	.083	-.043	.037	.017	.023	-.096
	55.00	.211	.144	-.086	.107	.071	.025	-.071
	60.00	.036	.117	-.066	.183	.116	.055	-.096
	65.00	.187	.079	-.106	.178	.142	.054	-.112
	70.00	.410	.004	-.129	.188	.152	.068	-.115
75.00	.185	.839	-.510	.395	.283	.218	-.115	
80.00	.689	.876	-.510	.393	.284	.221	-.109	
85.00	.529	.837	-.503	.390	.286	.218	-.104	
90.00	.397	.758	-.498	.387	.289	.206	-.076	
95.00	.289	.653	-.488	.378	.286	.189	-.076	
LOWER SURFACE	M = 0.94 α = 5.9°							
	1.25	.406	.458	.437	.412	.426	.431	.385
	2.50	.409	.384	.359	.316	.356	.358	.306
	5.00	.371	.311	.279	.284	.265	.249	.217
	7.50	.337	.259	.224	.236	.210	.194	.133
	10.00	.300	.227	.188	.185	.168	.186	.075
	15.00	.257	.166	.144	.114	.122	.096	.034
	20.00	.193	.156	.086	.073	.066	.038	.126
	25.00	.182	.083	.049	.031	.025	.009	.215
	30.00	.185	.053	.015	.010	.012	.050	.266
	35.00	.091	.024	.013	.024	.046	.093	.284
	40.00	.068	.000	.068	.053	.083	.135	.295
	45.00	.036	.030	.064	.081	.115	.181	.251
	50.00	.006	.057	.086	.108	.146	.227	.204
	55.00	.014	.070	.096	.128	.172	.267	.150
	60.00	.037	.079	.098	.145	.184	.265	.114
	65.00	.059	.086	.100	.164	.189	.311	.108
	70.00	.054						
	75.00	.078	.078	.130	.203	.199	.173	.094
80.00	.077	.085	.149	.205	.208	.171	.090	
85.00	.098	.082	.171	.214	.218	.148	.081	
90.00	.091	.085	.219	.224	.192	.115	.061	
95.00	.129	.119	.257	.241	.253	.125	.048	
UPPER SURFACE	M = 0.94 α = 7.9°							
	.00	.286	.211	-.280	-.121	-.585	-.612	-.599
	1.25	.130	1.096	-.244	-.288	-.297	-.725	-.843
	2.50	.242	1.101	-.191	-.316	-.290	-.690	-.835
	5.00	.346	1.032	-.157	-.261	-.246	-.657	-.824
	7.50	.382	.887	-.031	-.215	-.181	-.640	-.819
	10.00	.377	.802	.934	-.178	1.077	.609	-.811
	15.00	.385	.665	.773	-.064	.838	.572	-.779
	20.00	.367	.558	.614	.720	.838	.538	-.729
	25.00	.379	.494	.464	.552	.628	.506	-.645
	30.00	.375	.454	.382	.464	.561	.497	-.548
	35.00	.375	.463	.300	.349	.489	.479	-.426
	40.00	.381	.467	.243	.233	.414	.459	-.322
	45.00	.389	.176	.200	.148	.345	.439	-.251
	50.00	.432	.013	.158	.079	.285	.418	-.235
	55.00	.310	.105	.133	.028	.232	.397	-.220
	60.00	.130	.102	.133	.059	.194	.367	-.205
	65.00	.111	.103	.146	.065	.178	.328	-.202
	70.00	.355	.007	.098	.104	.132	.311	-.199
75.00	1.166	.853	.535	.430	.332	.267	-.186	
80.00	.714	.880	.548	.427	.346	.301	-.174	
85.00	.567	.864	.534	.427	.334	.261	-.162	
90.00	.456	.788	.511	.415	.312	.226	-.149	
95.00	.431	.677	.487	.397	.281	.194	-.137	
LOWER SURFACE	M = 0.94 α = 7.9°							
	1.25	.436	.528	.488	.451	.454	.455	.419
	2.50	.472	.463	.485	.376	.398	.400	.350
	5.00	.449	.389	.445	.345	.316	.303	.267
	7.50	.424	.335	.389	.289	.237	.242	.193
	10.00	.383	.299	.286	.244	.216	.242	.127
	15.00	.338	.237	.202	.171	.162	.148	.021
	20.00	.270	.191	.146	.127	.109	.090	.051
	25.00	.254	.130	.186	.085	.064	.038	.159
	30.00	.193	.115	.070	.056	.027	.003	.211
	35.00	.153	.082	.010	.020	.006	.050	.255
	40.00	.129	.054	.017	.014	.043	.088	.278
	45.00	.095	.050	.041	.079	.106	.166	.288
	50.00	.061	.037	.036	.098	.140	.224	.238
	55.00	.037	.030	.064	.120	.148	.255	.152
	60.00	.010	.034	.074	.135	.150	.275	.108
	65.00	.006						
	70.00	.006						
	75.00	.037	.043	.130	.179	.174	.144	.081
80.00	.034	.089	.145	.182	.193	.154	.081	
85.00	.058	.074	.167	.186	.203	.133	.083	
90.00	.060	.090	.208	.194	.178	.095	.070	
95.00	.103	.145	.242	.218	.228	.099	.072	

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:							
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
		$M = 0.94 \quad \alpha = 11.3^\circ$								
UPPER SURFACE		.00	.243	-.563	-.625	-.181	-.878	-.738	-.521	
		1.25	.258	-.1.210	-.1.224	-.1.350	-.1.094	-.672	-.444	
		2.50	.406	-.1.197	-.1.101	-.1.357	-.1.094	-.675	-.437	
		5.00	.528	-.1.228	-.1.046	-.1.319	-.1.075	-.653	-.445	
		7.50	.571	-.1.201	-.980	-.1.279	-.1.060	-.656	-.438	
		10.00	.562	-.1.140	-.951	-.1.229	-.1.046	-.643	-.426	
		15.00	.569	-.1.049	-.893	-.1.069	-.1.021	-.622	-.418	
		20.00	.530	-.936	-.814	-.944	-.992	-.593	-.410	
		25.00	.513	-.918	-.719	-.912	-.967	-.563	-.401	
		30.00	.503	-.798	-.632	-.873	-.937	-.558	-.400	
		35.00	.491	-.624	-.561	-.840	-.908	-.544	-.397	
		40.00	.482	-.504	-.505	-.806	-.875	-.525	-.391	
		45.00	.490	-.432	-.529	-.757	-.844	-.504	-.387	
		50.00	.522	-.057	-.597	-.685	-.808	-.487	-.380	
		55.00	.421	-.033	-.628	-.575	-.766	-.465	-.374	
		60.00	.271	-.055	-.635	-.407	-.706	-.451	-.363	
		65.00	.113	-.025	-.669	-.403	-.656	-.449	-.367	
		70.00	.208	-.081	-.733	-.351	-.547	-.421	-.357	
LOWER SURFACE		75.00	1.121	-.422	-.733	-.462	-.547	-.413	-.335	
		80.00	.834	-.453	-.633	-.469	-.569	-.416	-.324	
		85.00	.537	-.410	-.558	-.435	-.530	-.411	-.323	
		90.00	.510	-.516	-.516	-.397	-.475	-.398	-.317	
		95.00	.472	-.731	-.477	-.361	-.428	-.382	-.308	
		1.25	.426	-.607	-.546	-.552	-.488	-.478	-.456	
		2.50	.537	-.507	-.518	-.483	-.470	-.465	-.415	
		5.00	.646	-.457	-.459	-.451	-.408	-.391	-.343	
		7.50	.537	-.422	-.375	-.395	-.356	-.330	-.276	
		10.00	.484	-.351	-.318	-.344	-.310	-.322	-.211	
		15.00	.408	-.308	-.264	-.283	-.259	-.238	-.114	
		20.00	.361	-.263	-.222	-.238	-.203	-.177	-.021	
		25.00	.324	-.225	-.184	-.192	-.157	-.125	-.050	
		30.00	.268	-.185	-.135	-.155	-.119	-.079	-.117	
		35.00	.249	-.159	-.117	-.117	-.078	-.030	-.162	
		40.00	.209	-.121	-.087	-.077	-.044	-.012	-.191	
		45.00	.154	-.080	-.056	-.043	-.004	-.062	-.212	
		50.00	.147	-.071	-.037	-.011	-.024	-.108	-.238	
	55.00	.110	-.052	-.017	-.020	-.053	-.147	-.242		
	60.00	.096	-.036	-.009	-.072	-.092	-.207	-.252		
	65.00	.075						-.249		
	70.00	.024	-.026	-.093	-.131	-.126	-.137	-.237		
	75.00	.035	-.001	-.082	-.115	-.146	-.180	-.264		
	80.00	.032	-.024	-.113	-.125	-.157	-.226	-.257		
	85.00	.008	-.057	-.160	-.157	-.135	-.287	-.249		
	90.00	.000	-.130	-.202	-.198	-.203	-.299	-.249		
		$M = 0.94 \quad \alpha = 15.6^\circ$								
UPPER SURFACE		.00	.160	-.925	-.1.042	-.741	-.615	-.546	-.556	
		1.25	.160	-.925	-.1.042	-.741	-.615	-.546	-.556	
		2.50	.160	-.925	-.1.042	-.741	-.615	-.546	-.556	
		5.00	.160	-.925	-.1.042	-.741	-.615	-.546	-.556	
		7.50	.160	-.925	-.1.042	-.741	-.615	-.546	-.556	
		10.00	.160	-.925	-.1.042	-.741	-.615	-.546	-.556	
		15.00	.160	-.925	-.1.042	-.741	-.615	-.546	-.556	
		20.00	.160	-.925	-.1.042	-.741	-.615	-.546	-.556	
		25.00	.160	-.925	-.1.042	-.741	-.615	-.546	-.556	
		30.00	.160	-.925	-.1.042	-.741	-.615	-.546	-.556	
		35.00	.160	-.925	-.1.042	-.741	-.615	-.546	-.556	
		40.00	.160	-.925	-.1.042	-.741	-.615	-.546	-.556	
		45.00	.160	-.925	-.1.042	-.741	-.615	-.546	-.556	
		50.00	.160	-.925	-.1.042	-.741	-.615	-.546	-.556	
		55.00	.160	-.925	-.1.042	-.741	-.615	-.546	-.556	
		60.00	.160	-.925	-.1.042	-.741	-.615	-.546	-.556	
		65.00	.160	-.925	-.1.042	-.741	-.615	-.546	-.556	
		70.00	.160	-.925	-.1.042	-.741	-.615	-.546	-.556	
LOWER SURFACE		75.00	1.102	-.922	-.791	-.637	-.561	-.507	-.479	
		80.00	.859	-.950	-.748	-.642	-.564	-.522	-.468	
		85.00	.637	-.898	-.723	-.633	-.556	-.522	-.447	
		90.00	.368	-.788	-.704	-.633	-.553	-.514	-.434	
		95.00	.465	-.617	-.664	-.626	-.550	-.495	-.413	
		1.25	.372	.645	.550	.587	.493	.464	.435	
		2.50	.546	.667	.567	.558	.522	.501	.441	
		5.00	.688	.635	.558	.545	.490	.456	.403	
		7.50	.743	.592	.584	.501	.450	.411	.343	
		10.00	.690	.559	.495	.461	.412	.402	.286	
		15.00	.638	.494	.442	.403	.365	.324	.190	
		20.00	.555	.446	.385	.351	.314	.266	.091	
		25.00	.517	.399	.343	.309	.264	.212	.015	
		30.00	.463	.359	.305	.271	.223	.164	-.053	
		35.00	.402	.322	.269	.233	.184	.113	-.115	
		40.00	.377	.286	.233	.193	.144	.066	-.159	
		45.00	.337	.246	.193	.153	.101	.014	-.192	
		50.00	.278	.219	.167	.118	.067	.043	-.228	
	55.00	.271	.195	.141	.084	.083	.090	-.254		
	60.00	.234	.171	.114	.056	-.001	.131	-.265		
	65.00	.216	.152	.085	.028	-.027	.171	-.287		
	70.00	.184						-.301		
	75.00	.122	.127	.011	-.053	-.093	.166	-.316		
	80.00	.128	.087	.003	-.057	-.126	.210	-.340		
	85.00	.126	.050	-.003	-.090	-.163	.265	-.360		
	90.00	.068	.014	-.071	-.139	-.167	.338	-.377		
	95.00	.044	-.051	-.129	-.211	-.276	.379	-.373		

CONFIDENTIAL

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

PERCENT CHORD		PRESSURE COEFFICIENT, P , AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
$M = 0.98 \quad \alpha = -0.1^\circ$								
UPPER SURFACE	.00	.344	.561	.507	.803	.494	.486	.430
	1.25	.270	.034	.064	.010	.121	.145	.108
	2.50	.201	.016	.039	.003	.034	.032	.078
	5.00	.128	.019	.033	.010	.028	.053	.028
	7.50	.072	.030	.029	.032	.000	.018	.008
	10.00	.024	.054	.031	.016	.020	.006	.037
	12.50	.006	.081	.038	.012	.020	.032	.091
	15.00	.000	.097	.033	.007	.020	.013	.164
	17.50	.000	.077	.034	.004	.011	.007	.123
	20.00	.000	.058	.033	.005	.002	.006	.085
	22.50	.000	.037	.033	.025	.051	.019	.093
	25.00	.000	.016	.033	.048	.097	.046	.139
	27.50	.000	.000	.115	.090	.159	.101	.207
	30.00	.000	.076	.138	.152	.249	.185	.215
	32.50	.000	.083	.138	.152	.249	.185	.215
	35.00	.000	.051	.138	.152	.249	.185	.215
	37.50	.000	.054	.138	.152	.249	.185	.215
	40.00	.000	.225	.327	.310	.323	.327	.173
LOWER SURFACE	.00	.344	.561	.507	.803	.494	.486	.430
	1.25	.270	.034	.064	.010	.121	.145	.108
	2.50	.201	.016	.039	.003	.034	.032	.078
	5.00	.128	.019	.033	.010	.028	.053	.028
	7.50	.072	.030	.029	.032	.000	.018	.008
	10.00	.024	.054	.031	.016	.020	.006	.037
	12.50	.006	.081	.038	.012	.020	.032	.091
	15.00	.000	.097	.033	.007	.020	.013	.164
	17.50	.000	.077	.034	.004	.011	.007	.123
	20.00	.000	.058	.033	.005	.002	.006	.085
	22.50	.000	.037	.033	.025	.051	.019	.093
	25.00	.000	.016	.033	.048	.097	.046	.139
	27.50	.000	.000	.115	.090	.159	.101	.207
	30.00	.000	.076	.138	.152	.249	.185	.215
	32.50	.000	.083	.138	.152	.249	.185	.215
	35.00	.000	.051	.138	.152	.249	.185	.215
	37.50	.000	.054	.138	.152	.249	.185	.215
	40.00	.000	.225	.327	.310	.323	.327	.173
	42.50	.000	.398	.377	.388	.407	.319	.187
UPPER SURFACE	.00	.344	.561	.507	.803	.494	.486	.430
	1.25	.270	.034	.064	.010	.121	.145	.108
	2.50	.201	.016	.039	.003	.034	.032	.078
	5.00	.128	.019	.033	.010	.028	.053	.028
	7.50	.072	.030	.029	.032	.000	.018	.008
	10.00	.024	.054	.031	.016	.020	.006	.037
	12.50	.006	.081	.038	.012	.020	.032	.091
	15.00	.000	.097	.033	.007	.020	.013	.164
	17.50	.000	.077	.034	.004	.011	.007	.123
	20.00	.000	.058	.033	.005	.002	.006	.085
	22.50	.000	.037	.033	.025	.051	.019	.093
	25.00	.000	.016	.033	.048	.097	.046	.139
	27.50	.000	.000	.115	.090	.159	.101	.207
	30.00	.000	.076	.138	.152	.249	.185	.215
	32.50	.000	.083	.138	.152	.249	.185	.215
	35.00	.000	.051	.138	.152	.249	.185	.215
	37.50	.000	.054	.138	.152	.249	.185	.215
	40.00	.000	.225	.327	.310	.323	.327	.173
	42.50	.000	.398	.377	.388	.407	.319	.187
LOWER SURFACE	.00	.344	.561	.507	.803	.494	.486	.430
	1.25	.270	.034	.064	.010	.121	.145	.108
	2.50	.201	.016	.039	.003	.034	.032	.078
	5.00	.128	.019	.033	.010	.028	.053	.028
	7.50	.072	.030	.029	.032	.000	.018	.008
	10.00	.024	.054	.031	.016	.020	.006	.037
	12.50	.006	.081	.038	.012	.020	.032	.091
	15.00	.000	.097	.033	.007	.020	.013	.164
	17.50	.000	.077	.034	.004	.011	.007	.123
	20.00	.000	.058	.033	.005	.002	.006	.085
	22.50	.000	.037	.033	.025	.051	.019	.093
	25.00	.000	.016	.033	.048	.097	.046	.139
	27.50	.000	.000	.115	.090	.159	.101	.207
	30.00	.000	.076	.138	.152	.249	.185	.215
	32.50	.000	.083	.138	.152	.249	.185	.215
	35.00	.000	.051	.138	.152	.249	.185	.215
	37.50	.000	.054	.138	.152	.249	.185	.215
	40.00	.000	.225	.327	.310	.323	.327	.173
	42.50	.000	.398	.377	.388	.407	.319	.187

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

	PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
$M = 0.98 \quad \alpha = 5.9^\circ$								
UPPER SURFACE	.00	.319	.094	.010	.326	.251	.301	.217
	1.25	.083	.060	1.071	1.050	1.070	1.042	1.112
	2.50	.082	.045	1.041	1.102	1.100	1.127	1.106
	5.00	.174	.055	1.004	1.037	1.038	1.033	1.090
	7.50	.210	.045	.923	.985	.994	.996	.974
	10.00	.210	.409	.854	.943	.960	.952	.946
	15.00	.234	.367	.439	.867	.795	.582	.496
	20.00	.229	.355	.391	.341	.353	.288	.267
	25.00	.250	.353	.404	.180	.167	.235	.186
	30.00	.258	.351	.202	.139	.139	.203	.163
	35.00	.263	.359	.095	.104	.100	.158	.150
	40.00	.270	.354	.044	.062	.062	.111	.138
	45.00	.285	.062	.009	.007	.017	.051	.147
	50.00	.327	.096	.070	.060	.035	.002	.172
	55.00	.211	.134	.110	.129	.091	.046	.214
	60.00	.045	.118	.126	.127	.137	.084	.235
	65.00	.188	.129	.121	.127	.121	.099	.258
	70.00	.113	.011	.145	.127	.187	.100	.252
	75.00	1.252	1.010	.417	.496	.440	.337	.245
80.00	.732	1.032	.688	.501	.435	.332	.235	
85.00	.476	.940	.633	.507	.430	.324	.237	
90.00	.438	.850	.634	.509	.427	.319	.231	
95.00	.400	.679	.623	.504	.419	.310	.228	
LOWER SURFACE	1.25	.420	.465	.443	.403	.416	.430	.395
	2.50	.427	.396	.355	.308	.346	.359	.324
	5.00	.397	.319	.284	.278	.258	.253	.237
	7.50	.356	.271	.233	.235	.199	.199	.159
	10.00	.316	.242	.193	.193	.161	.152	.103
	15.00	.273	.179	.093	.114	.128	.103	.005
	20.00	.211	.139	.083	.070	.085	.048	.094
	25.00	.203	.103	.057	.038	.027	.002	.193
	30.00	.142	.069	.081	.010	.004	.035	.257
	35.00	.111	.037	.005	.023	.034	.079	.306
	40.00	.088	.012	.037	.048	.071	.116	.335
	45.00	.057	.021	.061	.068	.105	.168	.347
	50.00	.023	.051	.083	.097	.130	.217	.358
	55.00	.002	.063	.092	.113	.156	.260	.346
	60.00	.024	.075	.096	.123	.174	.303	.332
	65.00	.046	.083	.093	.138	.185	.328	.320
	70.00	.044						.317
	75.00	.071	.074	.182	.182	.214	.298	.310
	80.00	.070	.089	.141	.185	.232	.318	.309
85.00	.100	.076	.133	.202	.248	.356	.302	
90.00	.085	.062	.176	.234	.226	.386	.280	
95.00	.124	.054	.214	.277	.316	.369	.218	
$M = 0.98 \quad \alpha = 7.9^\circ$								
UPPER SURFACE	.00	.316	.114	.175	.203	.452	.529	.502
	1.25	.060	1.053	1.180	1.164	1.195	1.051	1.054
	2.50	.179	1.052	1.130	1.209	1.212	1.080	1.052
	5.00	.282	.969	1.126	1.145	1.157	.998	1.006
	7.50	.320	.765	1.061	1.099	1.115	.965	.964
	10.00	.318	.671	1.039	1.073	1.086	.903	.914
	15.00	.330	.551	.964	1.027	.979	.709	.702
	20.00	.317	.484	.893	.903	.903	.495	.629
	25.00	.332	.446	.872	.922	.953	.455	.584
	30.00	.334	.427	.840	.944	.909	.475	.537
	35.00	.330	.426	.770	.944	.839	.461	.460
	40.00	.338	.434	.702	.917	.887	.439	.487
	45.00	.350	.201	.089	.080	.842	.409	.408
	50.00	.290	.015	.080	.012	.803	.392	.408
	55.00	.293	.101	.026	.083	.178	.375	.377
	60.00	.128	.089	.036	.176	.153	.354	.328
	65.00	.108	.100	.042	.182	.140	.345	.328
	70.00	.350	.004	.037	.199	.099	.388	.377
	75.00	.250	1.014	.633	.516	.500	.448	.356
80.00	.779	1.027	.658	.520	.519	.440	.334	
85.00	.773	.943	.660	.526	.514	.412	.315	
90.00	.789	.849	.654	.524	.501	.382	.300	
95.00	.430	.708	.637	.520	.482	.353	.278	
LOWER SURFACE	1.25	.452	.543	.505	.465	.465	.463	.425
	2.50	.490	.476	.436	.383	.407	.407	.362
	5.00	.465	.402	.359	.348	.325	.310	.278
	7.50	.444	.349	.301	.304	.269	.250	.203
	10.00	.401	.317	.269	.255	.228	.251	.144
	15.00	.353	.251	.215	.187	.177	.159	.046
	20.00	.287	.208	.180	.144	.128	.100	.037
	25.00	.272	.165	.121	.098	.081	.051	.155
	30.00	.208	.126	.087	.072	.042	.008	.283
	35.00	.172	.095	.085	.037	.010	.039	.278
	40.00	.145	.070	.084	.008	.025	.060	.305
	45.00	.110	.033	.090	.022	.062	.127	.334
	50.00	.079	.009	.085	.048	.090	.175	.353
	55.00	.053	.004	.040	.071	.120	.222	.371
	60.00	.025	.022	.045	.087	.142	.263	.366
	65.00	.007	.031	.049	.104	.154	.293	.361
	70.00	.004						.364
	75.00	.023	.026	.088	.152	.187	.258	.349
	80.00	.019	.044	.094	.163	.204	.285	.340
85.00	.044	.033	.117	.184	.224	.322	.318	
90.00	.044	.081	.163	.217	.268	.344	.318	
95.00	.082	.064	.207	.262	.300	.396	.258	

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	$M = 0.98 \quad \alpha = 11.4^\circ$								
	.00	.291	.453	.496	.050	.759	.850	.742	
	1.25	.185	1.112	1.120	1.263	1.223	.901	.714	
	2.50	.327	1.096	1.051	1.277	1.214	.910	.708	
	5.00	.448	1.121	1.004	1.228	1.197	.881	.699	
	7.50	.494	1.093	.925	1.196	1.167	.894	.682	
	10.00	.488	1.031	.876	1.160	1.130	.887	.663	
	15.00	.496	.939	.820	1.036	1.006	.877	.640	
	20.00	.463	.889	.747	.797	.877	.865	.612	
	25.00	.449	.880	.656	.744	.811	.836	.589	
	30.00	.441	.852	.573	.723	.788	.816	.565	
	35.00	.424	.822	.496	.690	.763	.792	.543	
	40.00	.424	.773	.438	.637	.745	.761	.524	
	45.00	.433	.716	.385	.571	.724	.728	.508	
	50.00	.465	.647	.335	.498	.709	.693	.490	
	55.00	.382	.566	.283	.389	.689	.654	.471	
	60.00	.233	.485	.231	.308	.659	.619	.465	
	65.00	.083	.403	.182	.257	.607	.579	.450	
	70.00	.223	.330	.130	.257	.611	.567	.440	
	75.00	1.041	.980	.750	.557	.632	.555	.433	
80.00	.770	.980	.696	.554	.632	.555	.433		
85.00	.473	.902	.614	.536	.585	.529	.421		
90.00	.442	.814	.583	.509	.553	.501	.409		
95.00	.443	.694	.563	.475					
LOWER SURFACE	1.25	.451	.634	.573	.593	.510	.494	.466	
	2.50	.569	.598	.542	.508	.486	.475	.427	
	5.00	.716	.531	.481	.471	.420	.402	.361	
	7.50	.605	.480	.433	.417	.370	.346	.295	
	10.00	.567	.448	.399	.367	.324	.336	.233	
	15.00	.518	.379	.342	.305	.277	.252	.138	
	20.00	.438	.332	.289	.259	.222	.193	.009	
	25.00	.407	.287	.244	.212	.177	.146	.048	
	30.00	.352	.248	.211	.182	.138	.101	.122	
	35.00	.296	.216	.178	.147	.103	.055	.175	
	40.00	.272	.185	.147	.109	.065	.009	.208	
	45.00	.235	.148	.117	.074	.025	.041	.239	
	50.00	.182	.117	.088	.043	.007	.089	.268	
	55.00	.175	.100	.066	.016	.042	.134	.268	
	60.00	.138	.080	.050	.009	.063	.176	.291	
	65.00	.123	.066	.029	.037	.080	.210	.307	
	70.00	.100						.311	
	75.00	.047	.063	.020	.102	.118	.198	.311	
	80.00	.056	.035	.033	.100	.143	.223	.322	
	85.00	.053	.006	.066	.124	.164	.275	.340	
90.00	.034	.016	.111	.161	.160	.324	.351		
95.00	.020	.067	.159	.209	.251	.354	.360		
UPPER SURFACE	$M = 0.98 \quad \alpha = 15.8^\circ$								
	.00	.196	.814	.912	.698	.695	.642	.607	
	1.25	.316	1.235	1.172	.903	.678	.659	.586	
	2.50	.521	1.206	1.092	.901	.670	.645	.581	
	5.00	.691	1.232	1.131	.888	.653	.619	.574	
	7.50	.732	1.232	1.111	.893	.641	.629	.575	
	10.00	.719	1.214	1.100	.899	.631	.626	.574	
	15.00	.712	1.149	1.132	.928	.629	.627	.574	
	20.00	.618	1.142	1.092	.979	.645	.631	.569	
	25.00	.584	1.121	1.054	1.035	.687	.619	.570	
	30.00	.584	1.103	1.015	1.053	.713	.630	.577	
	35.00	.553	1.092	.983	1.052	.719	.621	.577	
	40.00	.543	1.034	.962	1.034	.723	.610	.576	
	45.00	.545	.741	.970	.974	.713	.602	.575	
	50.00	.565	.284	1.012	.903	.713	.595	.573	
	55.00	.488	.106	1.091	.847	.704	.589	.571	
	60.00	.391	.099	1.149	.773	.700	.582	.575	
	65.00	.280	.357	1.086	.733	.687	.581	.571	
	70.00	.078	.115	.963	.723	.629	.581	.566	
	75.00	1.062	1.048	.963	.765	.650	.582	.564	
80.00	.831	1.030	.988	.793	.647	.580	.559		
85.00	.620	.944	.854	.755	.637	.578	.538		
90.00	.521	.846	.780	.691	.625	.573	.537		
95.00	.523	.673	.688						
LOWER SURFACE	1.25	.395	.676	.583	.600	.519	.490	.461	
	2.50	.585	.694	.614	.579	.547	.530	.476	
	5.00	.734	.661	.583	.563	.516	.489	.439	
	7.50	.777	.616	.547	.520	.475	.443	.378	
	10.00	.726	.584	.518	.481	.439	.436	.327	
	15.00	.667	.519	.466	.424	.391	.361	.243	
	20.00	.583	.472	.423	.380	.340	.307	.154	
	25.00	.548	.428	.368	.336	.298	.255	.060	
	30.00	.493	.385	.334	.300	.256	.213	.013	
	35.00	.434	.315	.298	.260	.216	.164	.072	
	40.00	.405	.315	.262	.223	.185	.119	.120	
	45.00	.366	.279	.232	.188	.140	.068	.153	
	50.00	.313	.244	.194	.148	.109	.018	.192	
	55.00	.304	.171	.171	.119	.077	.029	.215	
	60.00	.383	.300	.145	.090	.049	.069	.234	
	65.00	.251	.180	.119	.063	.030	.109	.259	
	70.00	.212						.276	
	75.00	.145	.135	.041	.010	.026	.143	.296	
	80.00	.130	.119	.036	.008	.037	.183	.315	
	85.00	.150	.081	.004	.026	.092	.236	.348	
90.00	.080	.034	.048	.073	.096	.279	.374		
95.00	.047	.030	.095	.133	.200	.322	.412		

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	$M = 1.00 \quad \alpha = 0.0^\circ$								
	.00	.410	.583	.518	.821	.511	.494	.438	
	1.25	.291	.007	.051	.028	.138	.162	.123	
	2.50	.226	.006	.029	.028	.056	.038	.082	
	5.00	.159	.006	.024	.033	.048	.067	.045	
	7.50	.115	.013	.025	.058	.022	.031	.011	
	10.00	.096	.025	.024	.045	.000	.019	.016	
	15.00	.052	.045	.034	.016	.002	.017	.070	
	20.00	.025	.075	.029	.020	.000	.006	.154	
	25.00	.010	.086	.046	.023	.009	.001	.160	
	30.00	.030	.077	.088	.031	.022	.008	.126	
	35.00	.035	.005	.077	.050	.044	.021	.091	
	40.00	.056	.078	.102	.072	.072	.033	.065	
	45.00	.070	.164	.141	.113	.111	.058	.084	
	50.00	.042	.262	.206	.173	.174	.107	.139	
	55.00	.079	.353	.285	.259	.260	.189	.234	
	60.00	.251	.346	.328	.355	.312	.237	.279	
	65.00	.421	.356	.324	.348	.318	.219	.305	
70.00	.631	.300	.341	.361	.325	.278	.272		
75.00	1.148	.336	.580	.464	.466	.372	.261		
80.00	.648	.001	.581	.471	.473	.369	.268		
85.00	.568	.423	.599	.482	.483	.362	.280		
90.00	.405	.859	.609	.487	.499	.362	.280		
95.00	.317	.580	.587	.483	.510	.353	.277		
LOWER SURFACE	$M = 1.00 \quad \alpha = 0.0^\circ$								
	1.25	.269	.002	.032	.092	.236	.303	.402	
	2.50	.220	.025	.016	.068	.163	.227	.294	
	5.00	.173	.020	.026	.031	.166	.228	.241	
	7.50	.120	.002	.048	.069	.182	.230	.244	
	10.00	.096	.018	.057	.094	.195	.241	.281	
	15.00	.059	.045	.084	.145	.204	.241	.285	
	20.00	.005	.065	.111	.169	.224	.293	.327	
	25.00	.023	.083	.136	.198	.255	.311	.357	
	30.00	.039	.108	.159	.211	.272	.337	.341	
	35.00	.051	.131	.181	.239	.286	.361	.368	
	40.00	.064	.144	.206	.263	.307	.394	.391	
	45.00	.089	.171	.233	.289	.349	.418	.397	
	50.00	.111	.199	.258	.318	.365	.462	.423	
	55.00	.144	.209	.285	.338	.389	.483	.427	
	60.00	.166	.217	.262	.331	.360	.437	.383	
	65.00	.179	.211	.262	.319	.360	.437	.358	
	70.00	.197	.238	.284	.305	.322	.312	.295	
75.00	.197	.238	.284	.305	.322	.312	.295		
80.00	.199	.239	.286	.306	.323	.310	.290		
85.00	.210	.240	.287	.307	.324	.311	.291		
90.00	.227	.257	.304	.324	.341	.328	.308		
95.00	.225	.122	.247	.306	.336	.327	.276		
UPPER SURFACE	$M = 1.00 \quad \alpha = 3.0^\circ$								
	.00	.376	.323	.220	.446	.077	.082	.165	
	1.25	.135	.733	.855	.835	.752	.797	.857	
	2.50	.042	.677	.784	.898	.692	.817	.802	
	5.00	.038	.256	.472	.799	.469	.659	.662	
	7.50	.074	.284	.312	.605	.348	.531	.469	
	10.00	.074	.240	.312	.350	.278	.476	.274	
	15.00	.108	.230	.312	.137	.230	.247	.237	
	20.00	.116	.234	.312	.110	.203	.192	.231	
	25.00	.149	.234	.312	.093	.169	.148	.150	
	30.00	.170	.240	.312	.066	.186	.116	.103	
	35.00	.164	.254	.312	.044	.076	.071	.089	
	40.00	.178	.254	.312	.007	.023	.029	.081	
	45.00	.178	.254	.312	.045	.036	.020	.013	
	50.00	.239	.211	.131	.110	.101	.077	.133	
	55.00	.239	.211	.131	.164	.161	.114	.131	
	60.00	.239	.211	.131	.216	.198	.126	.131	
	65.00	.239	.211	.131	.216	.198	.126	.131	
70.00	.239	.211	.131	.216	.198	.126	.131		
75.00	.239	.211	.131	.216	.198	.126	.131		
80.00	.239	.211	.131	.216	.198	.126	.131		
85.00	.239	.211	.131	.216	.198	.126	.131		
90.00	.239	.211	.131	.216	.198	.126	.131		
95.00	.239	.211	.131	.216	.198	.126	.131		
LOWER SURFACE	$M = 1.00 \quad \alpha = 3.0^\circ$								
	.00	.376	.323	.220	.446	.077	.082	.165	
	1.25	.390	.379	.364	.331	.340	.331	.307	
	2.50	.377	.311	.286	.284	.269	.256	.234	
	5.00	.341	.247	.215	.206	.180	.158	.158	
	7.50	.288	.203	.160	.156	.124	.103	.091	
	10.00	.257	.172	.138	.114	.088	.058	.042	
	15.00	.217	.125	.089	.047	.030	.038	.055	
	20.00	.153	.090	.041	.010	.004	.018	.143	
	25.00	.124	.055	.003	.020	.046	.080	.225	
	30.00	.097	.084	.028	.051	.071	.119	.289	
	35.00	.068	.005	.084	.021	.015	.149	.300	
	40.00	.023	.087	.083	.107	.138	.192	.297	
	45.00	.020	.064	.107	.137	.143	.246	.308	
	50.00	.018	.096	.141	.149	.169	.288	.321	
	55.00	.036	.103	.149	.145	.166	.321	.308	
	60.00	.061	.113	.161	.162	.186	.321	.301	
	65.00	.085	.124	.149	.147	.197	.343	.302	
70.00	.081	.129	.115	.172	.226	.270	.298		
75.00	.106	.132	.111	.177	.231	.290	.305		
80.00	.111	.132	.111	.177	.231	.290	.305		
85.00	.111	.132	.111	.177	.231	.290	.305		
90.00	.111	.132	.111	.177	.231	.290	.305		
95.00	.111	.132	.111	.177	.231	.290	.305		

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

PERCENT CHORD		PRESSURE COEFFICIENT, P, AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M = 1.00 α = 5.9°							
	.00	.363	.135	.032	.344	.191	.249	.166
	1.25	.051	.927	1.004	.979	.999	.989	1.039
	2.50	.051	.891	.979	1.029	1.029	1.063	1.036
	5.00	.145	.626	.939	.963	.965	.976	.939
	7.50	.179	.410	.858	.915	.925	.942	.920
	10.00	.179	.362	.806	.871	.901	.902	.893
	15.00	.203	.330	.486	.816	.803	.663	.611
	20.00	.251	.316	.354	.369	.415	.277	.234
	25.00	.251	.317	.368	.155	.142	.209	.149
	30.00	.234	.319	.288	.118	.119	.184	.128
	35.00	.236	.325	.073	.080	.083	.143	.124
	40.00	.243	.331	.018	.037	.041	.094	.125
	45.00	.259	.044	.034	.028	.003	.043	.139
	50.00	.304	.123	.091	.085	.085	.021	.172
	55.00	.207	.177	.129	.138	.108	.057	.223
	60.00	.035	.126	.145	.218	.157	.093	.258
	65.00	.208	.119	.138	.205	.194	.114	.277
	70.00	.426	.034	.163	.215	.214	.105	.279
	75.00	1.202	.995	.615	.505	.479	.391	.272
80.00	.699	.912	.632	.510	.475	.377	.272	
85.00	.440	.913	.633	.524	.476	.365	.272	
90.00	.407	.814	.687	.529	.476	.356	.264	
95.00	.373	.653	.612	.525	.479	.356	.264	
LOWER SURFACE	1.25	.433	.486	.460	.423	.420	.430	.411
	2.50	.441	.410	.385	.329	.351	.359	.333
	5.00	.398	.340	.304	.302	.265	.255	.254
	7.50	.371	.288	.282	.268	.207	.200	.182
	10.00	.336	.260	.218	.205	.169	.200	.124
	15.00	.288	.159	.166	.138	.124	.114	.032
	20.00	.229	.159	.116	.092	.072	.058	.070
	25.00	.219	.118	.077	.053	.033	.021	.167
	30.00	.168	.087	.044	.026	.010	.062	.225
	35.00	.128	.053	.016	.005	.022	.099	.304
	40.00	.107	.029	.018	.034	.091	.143	.328
	45.00	.074	.005	.041	.059	.113	.195	.328
	50.00	.040	.033	.072	.085	.139	.235	.328
	55.00	.009	.043	.077	.087	.158	.279	.308
	60.00	.011	.060	.078	.108	.170	.306	.294
	65.00	.034	.068	.078	.108	.170	.306	.294
	70.00	.033	.070	.070	.151	.196	.279	.291
	75.00	.065	.076	.080	.155	.208	.300	.295
	80.00	.057	.076	.080	.171	.223	.314	.307
	85.00	.086	.081	.140	.202	.283	.378	.309
90.00	.079	.041	.182	.245	.291	.377	.289	
95.00	.111	.041	.182	.245	.291	.377	.289	
UPPER SURFACE	M = 1.00 α = 7.8°							
	.00	.364	.053	.125	.255	.395	.470	.433
	1.25	.019	1.018	1.126	1.092	1.131	1.062	1.088
	2.50	.134	1.002	1.069	1.131	1.152	1.139	1.086
	5.00	.235	.931	1.068	1.074	1.101	1.048	1.019
	7.50	.277	.696	1.000	1.033	1.059	1.036	.984
	10.00	.273	.467	.921	1.004	1.031	.980	.935
	15.00	.286	.467	.639	.967	.972	.788	.752
	20.00	.274	.422	.467	.900	.772	.580	.583
	25.00	.287	.397	.385	.581	.558	.463	.488
	30.00	.290	.383	.147	.396	.351	.436	.456
	35.00	.290	.386	.069	.296	.280	.414	.443
	40.00	.296	.395	.021	.178	.231	.320	.443
	45.00	.307	.177	.021	.048	.190	.165	.224
	50.00	.349	.041	.031	.046	.151	.165	.416
	55.00	.270	.123	.070	.117	.120	.322	.409
	60.00	.105	.098	.058	.207	.097	.299	.404
	65.00	.139	.107	.108	.213	.060	.246	.400
	70.00	.389	.028	.432	.504	.517	.483	.391
	75.00	1.181	.449	.647	.509	.537	.474	.375
80.00	.717	.895	.647	.515	.541	.456	.361	
85.00	.429	.800	.638	.519	.527	.433	.349	
90.00	.408	.666	.618	.511	.512	.407	.334	
95.00	.408	.666	.618	.511	.512	.407	.334	
LOWER SURFACE	1.25	.475	.561	.519	.484	.474	.469	.444
	2.50	.509	.499	.453	.396	.415	.412	.376
	5.00	.486	.422	.373	.321	.333	.315	.300
	7.50	.467	.369	.321	.281	.277	.258	.289
	10.00	.425	.338	.288	.257	.233	.228	.170
	15.00	.380	.272	.251	.201	.189	.169	.078
	20.00	.315	.226	.191	.159	.135	.115	.017
	25.00	.295	.191	.157	.115	.093	.063	.140
	30.00	.230	.125	.104	.089	.058	.025	.190
	35.00	.168	.077	.077	.055	.025	.021	.242
	40.00	.141	.060	.044	.028	.007	.053	.280
	45.00	.107	.027	.029	.029	.042	.104	.305
	50.00	.077	.018	.029	.047	.102	.153	.328
	55.00	.053	.004	.026	.060	.121	.201	.371
	60.00	.038	.013	.034	.083	.133	.269	.334
	65.00	.038	.013	.034	.083	.133	.269	.334
	70.00	.015	.003	.052	.124	.164	.253	.330
	75.00	.001	.020	.064	.136	.181	.274	.328
	80.00	.017	.021	.087	.154	.201	.316	.330
	85.00	.017	.021	.087	.154	.201	.316	.330
90.00	.017	.021	.087	.154	.201	.316	.330	
95.00	.017	.021	.087	.154	.201	.316	.330	

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

PERCENT CHORD		PRESSURE COEFFICIENT, P_i AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
$M = 1.00 \quad \alpha = 11.4^\circ$								
UPPER SURFACE	.00	.303	.421	.453	.008	.713	.811	.732
	1.25	.158	1.085	1.088	1.217	1.203	.969	.790
	2.50	.299	1.065	1.031	1.432	1.197	.972	.785
	5.00	.420	1.089	.998	1.190	1.175	.938	.773
	7.50	.469	1.059	.916	1.155	1.145	.956	.769
	10.00	.462	.995	.857	1.120	1.104	.943	.759
	15.00	.470	.903	.799	1.014	.992	.918	.737
	20.00	.438	.852	.725	.763	.836	.883	.717
	25.00	.424	.789	.636	.723	.792	.840	.689
	30.00	.419	.688	.551	.704	.768	.832	.671
	35.00	.406	.567	.481	.688	.746	.810	.651
	40.00	.402	.464	.424	.613	.725	.789	.629
	45.00	.410	.419	.439	.613	.708	.763	.615
	50.00	.439	.036	.517	.553	.693	.743	.598
	55.00	.500	.077	.566	.484	.679	.721	.580
	60.00	.518	.091	.579	.373	.668	.694	.554
	65.00	.507	.078	.621	.391	.644	.663	.540
	70.00	.445	.020	.563	.342	.630	.614	.525
LOWER SURFACE	75.00	1.033	.957	.735	.548	.618	.610	.515
	80.00	.747	.957	.691	.548	.642	.606	.504
	85.00	.451	.877	.600	.534	.629	.596	.496
	90.00	.425	.795	.568	.507	.604	.576	.486
	95.00	.423	.676	.542	.489	.574	.550	.476
	1.25	.458	.647	.581	.597	.518	.503	.477
	2.50	.582	.606	.548	.516	.491	.484	.442
	5.00	.781	.543	.488	.479	.426	.412	.377
	7.50	.617	.492	.439	.428	.376	.352	.306
	10.00	.568	.457	.406	.376	.333	.345	.350
	15.00	.523	.389	.349	.315	.281	.260	.253
	20.00	.443	.342	.298	.270	.233	.203	.049
	25.00	.413	.296	.255	.227	.186	.153	.036
	30.00	.357	.261	.219	.191	.148	.109	.106
	35.00	.303	.223	.186	.155	.108	.064	.164
	40.00	.280	.193	.153	.122	.075	.020	.198
	45.00	.241	.156	.127	.086	.038	.029	.236
	50.00	.191	.125	.096	.056	.006	.078	.266
	55.00	.180	.104	.079	.028	.028	.125	.290
	60.00	.141	.085	.064	.005	.050	.163	.293
UPPER SURFACE	65.00	.130	.074	.042	.024	.068	.195	.302
	70.00	.101	.071	.004	.085	.107	.185	.314
	75.00	.038	.045	.018	.085	.133	.230	.325
	80.00	.050	.019	.049	.107	.152	.274	.340
	85.00	.064	.000	.092	.143	.145	.320	.353
	90.00	.008	.031	.139	.192	.236	.351	.378
	95.00	.006	.051	.139	.192	.236	.351	.378
$M = 1.03 \quad \alpha = 0.0^\circ$								
UPPER SURFACE	.00	.679	.576	.513	.832	.551	.529	.468
	1.25	.245	.080	.065	.037	.185	.242	.169
	2.50	.191	.009	.047	.042	.101	.084	.133
	5.00	.125	.009	.050	.055	.095	.109	.091
	7.50	.083	.028	.055	.084	.067	.074	.049
	10.00	.069	.042	.083	.075	.046	.064	.027
	15.00	.035	.059	.090	.056	.043	.023	.017
	20.00	.012	.074	.042	.064	.043	.037	.018
	25.00	.020	.094	.089	.071	.051	.048	.086
	30.00	.042	.110	.089	.080	.063	.060	.049
	35.00	.044	.108	.112	.096	.083	.073	.025
	40.00	.061	.059	.145	.116	.112	.097	.041
	45.00	.086	.086	.125	.155	.155	.144	.094
	50.00	.109	.080	.125	.215	.214	.225	.190
	55.00	.015	.388	.330	.294	.298	.267	.245
	60.00	.015	.432	.368	.369	.346	.248	.269
	65.00	.408	.421	.383	.364	.349	.248	.260
	70.00	.600	.369	.265	.376	.270	.243	.231
LOWER SURFACE	75.00	1.114	.883	.588	.422	.480	.342	.231
	80.00	.574	.939	.519	.429	.425	.339	.219
	85.00	.383	.866	.534	.441	.435	.334	.232
	90.00	.200	.800	.547	.448	.448	.325	.243
	95.00	.124	.602	.534	.442	.457	.316	.243
	1.25	.220	.013	.033	.043	.203	.273	.368
	2.50	.184	.005	.021	.045	.129	.193	.265
	5.00	.126	.005	.045	.004	.129	.193	.203
	7.50	.088	.014	.049	.038	.145	.194	.231
	10.00	.067	.030	.059	.062	.157	.146	.247
	15.00	.008	.056	.085	.114	.166	.204	.259
	20.00	.009	.076	.112	.137	.187	.237	.290
	25.00	.039	.092	.131	.166	.216	.257	.307
	30.00	.063	.106	.153	.180	.234	.278	.315
	35.00	.067	.133	.166	.206	.253	.296	.342
	40.00	.091	.145	.192	.235	.275	.321	.352
	45.00	.119	.174	.218	.260	.308	.357	.360
	50.00	.141	.205	.242	.282	.333	.386	.388
	55.00	.165	.211	.249	.298	.350	.407	.389
	60.00	.184	.220	.249	.298	.350	.447	.384
	65.00	.183	.217	.249	.298	.350	.433	.351
	70.00	.201	.223	.261	.279	.286	.294	.275
	75.00	.201	.223	.261	.279	.286	.294	.275
	80.00	.214	.223	.261	.279	.286	.294	.275
	85.00	.214	.223	.261	.279	.286	.294	.275
	90.00	.214	.223	.261	.279	.286	.294	.275
	95.00	.208	.213	.209	.269	.293	.293	.245

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:					
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2
UPPER SURFACE	M = 1.03 α = 3.9°							
	.00	.547	.353	.246	.482	.123	.106	.202
	1.25	.114	.713	.785	.767	.761	.722	.782
	2.50	.029	.644	.743	.823	.792	.732	.734
	5.00	.056	.259	.634	.742	.715	.537	.603
	7.50	.092	.249	.307	.645	.616	.296	.440
	10.00	.098	.235	.297	.419	.310	.246	.241
	15.00	.121	.228	.297	.281	.084	.217	.194
	20.00	.125	.224	.292	.093	.091	.165	.186
	25.00	.150	.227	.292	.073	.082	.118	.100
	30.00	.162	.231	.062	.040	.059	.082	.056
	35.00	.162	.244	.012	.009	.025	.036	.043
	40.00	.175	.251	.049	.035	.022	.008	.036
	45.00	.194	.083	.090	.092	.078	.056	.049
	50.00	.227	.217	.138	.156	.144	.107	.086
	55.00	.185	.235	.168	.204	.208	.149	.155
	60.00	.068	.135	.178	.204	.239	.166	.196
	65.00	.264	.114	.174	.237	.246	.160	.227
70.00	.469	.112	.204	.248	.416	.150	.229	
75.00	1	.914	.551	.453	.416	.350	.217	
80.00	.644	.944	.570	.457	.415	.338	.207	
85.00	.414	.845	.566	.472	.414	.323	.211	
90.00	.334	.738	.556	.478	.417	.310	.216	
95.00	.309	.580	.538	.471	.420	.301	.215	
LOWER SURFACE	1.25	.324	.366	.360	.346	.353	.374	.341
	2.50	.336	.304	.290	.269	.285	.299	.269
	5.00	.325	.241	.217	.223	.201	.195	.200
	7.50	.273	.198	.170	.174	.148	.147	.131
	10.00	.237	.167	.142	.133	.111	.151	.086
	15.00	.206	.125	.098	.068	.080	.070	.011
	20.00	.149	.089	.054	.035	.029	.022	.100
	25.00	.149	.057	.018	.000	.008	.008	.180
	30.00	.097	.029	.013	.024	.035	.036	.227
	35.00	.071	.000	.041	.056	.063	.075	.255
	40.00	.053	.018	.066	.077	.083	.107	.257
	45.00	.024	.052	.089	.105	.108	.148	.257
	50.00	.005	.078	.116	.125	.106	.198	.267
	55.00	.028	.091	.124	.130	.122	.243	.281
	60.00	.050	.098	.121	.128	.140	.278	.267
	65.00	.073	.113	.123	.113	.154	.299	.261
	70.00	.073						.259
	75.00	.099	.106	.077	.121	.178	.259	.257
80.00	.099	.112	.079	.125	.186	.270	.265	
85.00	.105	.098	.077	.140	.197	.290	.269	
90.00	.105	.077	.111	.165	.175	.306	.272	
95.00	.106	.018	.148	.209	.257	.311	.256	
M = 1.03 α = 5.8°								
UPPER SURFACE	.00	.508	.182	.071	.382	.139	.190	.123
	1.25	.061	.879	.930	.908	.926	.923	.970
	2.50	.032	.833	.908	.968	.964	.984	.972
	5.00	.127	.766	.876	.898	.910	.895	.874
	7.50	.171	.381	.802	.852	.867	.868	.851
	10.00	.175	.314	.770	.812	.835	.837	.833
	15.00	.195	.296	.779	.747	.747	.715	.627
	20.00	.190	.293	.322	.583	.546	.321	.229
	25.00	.209	.291	.340	.153	.128	.172	.136
	30.00	.219	.291	.280	.079	.064	.137	.124
	35.00	.218	.297	.055	.040	.031	.097	.120
	40.00	.223	.314	.006	.001	.007	.048	.106
	45.00	.239	.033	.060	.008	.049	.004	.106
	50.00	.278	.152	.116	.122	.095	.043	.128
	55.00	.222	.191	.150	.187	.146	.085	.177
	60.00	.021	.185	.166	.252	.192	.121	.211
	65.00	.224	.111	.160	.240	.229	.153	.243
	70.00	.438	.070	.190	.252	.412	.188	.247
75.00	1	.912	.572	.467	.436	.364	.248	
80.00	.674	.931	.591	.477	.437	.352	.245	
85.00	.396	.840	.588	.485	.440	.328	.244	
90.00	.386	.744	.574	.492	.445	.329	.244	
95.00	.310	.600	.557	.487	.445	.321	.239	
LOWER SURFACE	1.25	.379	.486	.468	.449	.451	.461	.432
	2.50	.420	.395	.395	.344	.383	.390	.360
	5.00	.393	.348	.317	.320	.298	.288	.287
	7.50	.378	.298	.267	.278	.243	.235	.211
	10.00	.341	.275	.235	.228	.202	.232	.159
	15.00	.304	.212	.184	.159	.163	.148	.064
	20.00	.237	.170	.138	.116	.111	.093	.070
	25.00	.235	.139	.101	.080	.073	.044	.121
	30.00	.178	.107	.070	.056	.047	.014	.187
	35.00	.147	.077	.042	.028	.020	.025	.234
	40.00	.123	.058	.012	.006	.009	.061	.264
	45.00	.096	.025	.012	.023	.028	.104	.284
	50.00	.066	.002	.036	.040	.073	.151	.298
	55.00	.041	.013	.048	.051	.097	.192	.291
	60.00	.012	.024	.047	.055	.115	.212	.270
	65.00	.002	.039	.047	.069	.127	.259	.260
	70.00	.002						.256
	75.00	.047	.034	.036	.110	.151	.239	.252
80.00	.043	.038	.039	.117	.161	.269	.257	
85.00	.036	.035	.085	.136	.175	.305	.276	
90.00	.065	.017	.098	.160	.159	.331	.277	
95.00	.056	.000	.138	.200	.240	.334	.263	

TABLE II

WING WITH UPPER SURFACE SPOILER (NO GAP)

PERCENT CHORD		PRESSURE COEFFICIENT, P , AT:												
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2						
UPPER SURFACE	$M = 1.03 \quad \alpha = 7.8^\circ$													
	.00	.470	-	.009	-	.082	-	.297	-	.339	-	.411	-	.377
	1.25	.009	-	1.002	-	1.043	-	1.025	-	1.047	-	1.000	-	1.038
	2.50	.093	-	.974	-	.999	-	1.062	-	1.066	-	1.072	-	1.038
	5.00	.199	-	.935	-	.999	-	1.007	-	1.019	-	.977	-	.968
	7.50	.246	-	.827	-	.936	-	.968	-	.980	-	.965	-	.948
	10.00	.251	-	.603	-	.822	-	.940	-	.958	-	.920	-	.908
	15.00	.268	-	.409	-	.871	-	.903	-	.897	-	.763	-	.764
	20.00	.253	-	.359	-	.722	-	.856	-	.730	-	.551	-	.562
	25.00	.266	-	.347	-	.473	-	.586	-	.512	-	.427	-	.420
	30.00	.269	-	.339	-	.379	-	.403	-	.314	-	.391	-	.380
	35.00	.265	-	.346	-	.152	-	.303	-	.245	-	.370	-	.372
	40.00	.267	-	.355	-	.045	-	.197	-	.208	-	.352	-	.371
	45.00	.280	-	.160	-	.019	-	.051	-	.176	-	.333	-	.376
	50.00	.316	-	.081	-	.082	-	.077	-	.146	-	.316	-	.376
	55.00	.264	-	.148	-	.121	-	.160	-	.125	-	.300	-	.373
60.00	.084	-	.103	-	.142	-	.249	-	.110	-	.283	-	.373	
65.00	.151	-	.105	-	.142	-	.254	-	.101	-	.279	-	.372	
70.00	.430	-	.037	-	.171	-	.270	-	.110	-	.225	-	.368	
75.00	1.092	-	.898	-	.580	-	.450	-	.485	-	.446	-	.356	
80.00	.685	-	.910	-	.601	-	.457	-	.511	-	.437	-	.346	
85.00	.388	-	.826	-	.598	-	.462	-	.505	-	.419	-	.326	
90.00	.302	-	.735	-	.582	-	.463	-	.487	-	.393	-	.314	
95.00	.337	-	.609	-	.563	-	.461	-	.469	-	.373	-	.305	
LOWER SURFACE	1.25	.418	-	.575	-	.537	-	.515	-	.506	-	.501	-	.473
	2.50	.499	-	.509	-	.467	-	.428	-	.448	-	.443	-	.412
	5.00	.491	-	.441	-	.394	-	.392	-	.368	-	.349	-	.334
	7.50	.486	-	.388	-	.348	-	.350	-	.313	-	.293	-	.283
	10.00	.447	-	.360	-	.314	-	.299	-	.274	-	.293	-	.208
	15.00	.402	-	.246	-	.261	-	.233	-	.226	-	.207	-	.117
	20.00	.334	-	.251	-	.210	-	.193	-	.174	-	.150	-	.003
	25.00	.319	-	.215	-	.170	-	.158	-	.132	-	.103	-	.077
	30.00	.261	-	.179	-	.135	-	.123	-	.101	-	.065	-	.150
	35.00	.225	-	.147	-	.108	-	.090	-	.067	-	.019	-	.200
	40.00	.199	-	.123	-	.078	-	.063	-	.031	-	.017	-	.231
	45.00	.166	-	.087	-	.053	-	.035	-	.003	-	.060	-	.266
	50.00	.136	-	.062	-	.029	-	.007	-	.030	-	.108	-	.294
	55.00	.107	-	.049	-	.018	-	.008	-	.057	-	.151	-	.306
	60.00	.078	-	.030	-	.016	-	.028	-	.078	-	.189	-	.305
	65.00	.065	-	.019	-	.009	-	.047	-	.089	-	.221	-	.302
70.00	.052	-		-		-		-		-		-	.297	
75.00	.005	-	.031	-	.016	-	.083	-	.119	-	.231	-	.295	
80.00	.010	-	.019	-	.026	-	.094	-	.134	-	.247	-	.296	
85.00	.020	-	.016	-	.051	-	.115	-	.152	-	.280	-	.296	
90.00	.017	-	.018	-	.091	-	.143	-	.138	-	.324	-	.301	
95.00	.018	-	.000	-	.130	-	.186	-	.223	-	.341	-	.285	

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M 0.60 $\alpha = 0.0^\circ$								
	1.00	.220	.482	.439	.692	.441	.424	.315	
	1.25	.059	.064	.004	.074	.189	.276	.172	
	2.50	.042	.058	.006	.051	.103	.131	.117	
	5.00	.028	.054	.009	.027	.075	.102	.058	
	7.50	.005	.061	.006	.027	.048	.070	.020	
	10.00	.013	.072	.033	.023	.034	.058	.006	
	15.00	.036	.074	.033	.004	.030	.031	.026	
	20.00	.058	.075	.027	.010	.026	.036	.042	
	25.00	.072	.072	.023	.021	.031	.036	.035	
	30.00	.074	.062	.011	.030	.041	.036	.034	
	35.00	.066	.044	.011	.048	.056	.053	.034	
	40.00	.070	.013	.053	.073	.090	.069	.034	
	45.00	.064	.037	.090	.108	.123	.100	.026	
	50.00	.050	.116	.154	.166	.173	.146	.030	
	55.00	.000	.228	.246	.240	.252	.209	.039	
	60.00	.094	.296	.316	.334	.320	.271	.031	
	65.00	.180	.278	.322	.343	.320	.244	.050	
	70.00	.312	.240	.327	.287	.230	.154	.040	
75.00	.556	.661	.338	.284	.230	.143	.052		
80.00	.509	.727	.353	.288	.232	.134	.072		
85.00	.395	.693	.353	.288	.234	.124	.087		
90.00	.247	.622	.374	.293	.238	.121	.080		
95.00	.158	.502	.364	.299					
LOWER SURFACE	1.25	.125	.144	.223	.295	.411	.471	.491	
	2.50	.069	.109	.158	.213	.286	.307	.347	
	5.00	.027	.101	.145	.132	.231	.263	.256	
	7.50	.007	.112	.149	.150	.223	.230	.245	
	10.00	.031	.122	.154	.163	.211	.207	.233	
	15.00	.062	.141	.168	.192	.207	.219	.218	
	20.00	.095	.153	.182	.194	.212	.218	.218	
	25.00	.081	.160	.192	.205	.219	.216	.218	
	30.00	.124	.172	.210	.209	.226	.216	.218	
	35.00	.138	.190	.217	.217	.227	.216	.218	
	40.00	.149	.200	.224	.223	.231	.216	.218	
	45.00	.157	.219	.226	.228	.238	.216	.218	
	50.00	.204	.236	.233	.234	.240	.213	.218	
	55.00	.190	.234	.234	.234	.241	.208	.215	
	60.00	.207	.247	.227	.233	.232	.201	.210	
	65.00	.239	.278	.244	.234	.210	.178	.177	
	70.00	.273							
	75.00	.191	.024	.079	.152	.152	.160	.121	
	80.00	.069	.067	.134	.154	.157	.142	.093	
85.00	.098	.079	.149	.154	.157	.110	.093		
90.00	.065	.085	.166	.168	.144	.115	.065		
95.00	.062	.113	.175	.181	.159	.114	.059		
GAP			.784	.453	.273	.262	.177		
UPPER SURFACE	M = 0.60 $\alpha = 4.0^\circ$								
	1.00	.195	.134					.001	
	1.25	.453	.913					.624	
	2.50	.410	.487					.416	
	5.00	.323	.393					.262	
	7.50	.257	.336					.242	
	10.00	.220	.309					.200	
	15.00	.221	.265					.132	
	20.00	.220	.238					.123	
	25.00	.220	.212					.091	
	30.00	.212	.188					.069	
	35.00	.187	.151					.049	
	40.00	.183	.102					.038	
	45.00	.165	.032					.028	
	50.00	.144	.062					.029	
	55.00	.067	.161					.017	
	60.00	.013	.181					.017	
	65.00	.096	.184					.023	
	70.00	.211	.112					.023	
75.00	.788	.678					.028		
80.00	.651	.690					.031		
85.00	.467	.654					.038		
90.00	.294	.592					.033		
95.00	.185	.497					.033		
LOWER SURFACE	1.25	.310	.338					.245	
	2.50	.278	.256					.165	
	5.00	.238	.184					.092	
	7.50	.191	.134					.026	
	10.00	.153	.107					.009	
	15.00	.121	.054					.069	
	20.00	.072	.027					.085	
	25.00	.056	.005					.088	
	30.00	.024	.033					.099	
	35.00	.007	.058					.108	
	40.00	.026	.077					.127	
	45.00	.041	.104					.119	
	50.00	.095	.126					.115	
	55.00	.087	.139					.111	
	60.00	.108	.160					.097	
	65.00	.144	.199					.085	
	70.00	.190						.093	
	75.00	.129	.081					.069	
	80.00	.085	.021					.060	
85.00	.045	.042					.063		
90.00	.046	.068					.032		
95.00	.062	.111					.023		
GAP			.775						

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:						
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
M = 0.80 $\alpha = 8.0^\circ$							
UPPER SURFACE	0.00	.158	.015	.874	.290	.896	.738
	1.25	.588	.280	.165	.759	.919	.106
	2.50	.533	.922	.723	.699	.754	.686
	5.00	.474	.691	.783	.629	.613	.504
	7.50	.398	.570	.893	.543	.523	.428
	10.00	.345	.496	.909	.479	.449	.326
	15.00	.318	.403	.900	.374	.334	.211
	20.00	.306	.350	.896	.296	.257	.165
	25.00	.307	.307	.899	.230	.200	.121
	30.00	.289	.266	.893	.175	.147	.092
	35.00	.255	.221	.881	.123	.096	.073
	40.00	.243	.163	.866	.073	.046	.053
	45.00	.223	.088	.820	.016	.005	.037
	50.00	.198	.005	.744	.037	.056	.057
	55.00	.111	.103	.614	.089	.104	.037
	60.00	.028	.129	.451	.153	.128	.048
	65.00	.126	.107	.264	.150	.142	.051
	70.00	.214	.037	.127	.161	.157	.068
	75.00	.824	.707	.347	.251	.224	.069
	80.00	.710	.711	.363	.281	.228	.073
LOWER SURFACE	85.00	.505	.602	.362	.283	.235	.076
	90.00	.335	.609	.362	.283	.235	.079
	95.00	.213	.509	.355	.284	.233	.069
	1.25	.363	.442	.455	.423	.442	.391
	2.50	.343	.364	.385	.357	.366	.327
	5.00	.306	.293	.296	.307	.275	.257
	7.50	.282	.234	.238	.240	.219	.198
	10.00	.239	.209	.197	.194	.174	.158
	15.00	.207	.137	.146	.128	.118	.101
	20.00	.150	.100	.093	.083	.074	.052
	25.00	.135	.060	.052	.047	.037	.016
	30.00	.095	.032	.017	.017	.004	.016
	35.00	.056	.000	.009	.014	.001	.051
	40.00	.032	.022	.037	.037	.046	.070
	45.00	.015	.056	.083	.068	.072	.093
	50.00	.037	.083	.077	.084	.089	.108
	55.00	.043	.096	.086	.099	.101	.120
	60.00	.102	.125	.101	.116	.114	.126
	65.00	.157	.165	.143	.134	.126	.119
	70.00	.101	.105	.030	.037	.045	.084
	75.00	.002	.002	.055	.084	.078	.073
UPPER SURFACE	80.00	.015	.029	.083	.083	.088	.063
	85.00	.083	.066	.169	.117	.093	.067
	90.00	.046	.110	.130	.131	.087	.045
	95.00	.046	.793	.522	.256	.279	.039
	GAP						.161
M = 0.80 $\alpha = 8.0^\circ$							
UPPER SURFACE	0.00	.105	1.171	1.398	.441	1.086	.714
	1.25	.654	1.080	1.181	.784	.930	.557
	2.50	.616	1.014	1.106	.753	.913	.520
	5.00	.581	.926	1.091	.712	.893	.497
	7.50	.532	.853	.910	.666	.872	.483
	10.00	.504	.812	.818	.626	.861	.459
	15.00	.459	.694	.594	.541	.808	.416
	20.00	.425	.571	.428	.467	.729	.374
	25.00	.401	.477	.339	.419	.685	.325
	30.00	.380	.395	.279	.366	.624	.298
	35.00	.339	.327	.229	.309	.507	.263
	40.00	.220	.256	.181	.249	.406	.230
	45.00	.222	.176	.134	.180	.336	.198
	50.00	.265	.095	.080	.124	.282	.171
	55.00	.165	.005	.059	.075	.206	.158
	60.00	.059	.014	.084	.013	.108	.145
	65.00	.029	.015	.073	.026	.124	.132
	70.00	.125	.697	.335	.027	.196	.093
	75.00	.744	.700	.343	.306	.254	.229
	80.00	.674	.772	.355	.313	.263	.244
	85.00	.536	.605	.369	.314	.263	.226
LOWER SURFACE	90.00	.378	.605	.369	.311	.254	.187
	95.00	.258	.485	.385	.287	.231	.164
	1.25	.389	.480	.471	.495	.476	.413
	2.50	.381	.437	.440	.433	.412	.385
	5.00	.377	.369	.362	.381	.356	.364
	7.50	.370	.314	.325	.321	.322	.266
	10.00	.325	.282	.214	.277	.246	.238
	15.00	.298	.214	.161	.205	.184	.164
	20.00	.227	.131	.114	.158	.134	.109
	25.00	.210	.094	.077	.107	.096	.067
	30.00	.117	.059	.046	.079	.049	.028
	35.00	.093	.031	.017	.046	.024	.005
	40.00	.070	.000	.009	.018	.004	.031
	45.00	.013	.031	.033	.010	.033	.025
	50.00	.014	.052	.085	.033	.086	.077
	55.00	.018	.081	.073	.064	.078	.091
	60.00	.056	.132	.118	.087	.096	.101
	65.00	.119	.133	.082	.105	.117	.097
	70.00	.064	.020	.035	.008	.006	.033
	75.00	.036	.010	.009	.004	.000	.051
	80.00	.015	.048	.088	.073	.051	.037
	85.00	.003	.048	.183	.083	.071	.088
	90.00	.013	.782	.628	.106	.111	.068
	95.00	.013			.256	.312	.218
	GAP						

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

	PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:							
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
UPPER SURFACE	M = 0.60 α = 11.3°								
	.00	.009	1.708	1.506	.988	.645	.454	.396	
	1.25	1.041	1.344	1.072	.904	.576	.402	.294	
	2.50	1.110	1.356	1.060	.893	.572	.389	.291	
	3.75	1.058	1.368	1.064	.885	.572	.375	.279	
	5.00	.946	1.404	1.030	.869	.567	.375	.274	
	7.50	.838	1.467	1.019	.857	.562	.371	.268	
	10.00	.709	1.403	.973	.834	.550	.362	.261	
	15.00	.632	1.217	.930	.811	.539	.352	.247	
	20.00	.584	.925	.886	.791	.531	.341	.236	
	25.00	.556	.624	.838	.764	.524	.335	.233	
	30.00	.504	.436	.776	.733	.512	.322	.229	
	35.00	.461	.314	.707	.698	.502	.311	.223	
	40.00	.411	.216	.629	.648	.493	.301	.215	
	45.00	.358	.120	.555	.601	.470	.285	.209	
	50.00	.243	.012	.476	.559	.446	.272	.203	
	55.00	.119	.058	.406	.466	.412	.263	.189	
	60.00	.013	.020	.353	.451	.381	.247	.185	
	65.00	.336	.681	.401	.412	.471	.232	.182	
	70.00	1.088	.683	.487	.469	.488	.226	.175	
	75.00	.726	.685	.439	.497	.461	.236	.171	
	80.00	.577	.609	.439	.446	.381	.239	.169	
	85.00	.425	.490	.372	.399	.351	.234	.165	
	90.00	.295							
	95.00								
LOWER SURFACE	1.25	.390	.497	.468	.504	.490	.482	.413	
	2.50	.377	.524	.506	.497	.485	.465	.378	
	3.75	.560	.490	.462	.467	.433	.392	.309	
	5.00	.498	.442	.417	.417	.380	.338	.240	
	7.50	.463	.407	.381	.370	.337	.312	.188	
	10.00	.432	.333	.319	.297	.273	.237	.106	
	15.00	.361	.288	.260	.244	.220	.175	.039	
	20.00	.324	.236	.214	.197	.171	.125	.015	
	25.00	.283	.194	.177	.157	.134	.086	.016	
	30.00	.228	.159	.139	.119	.099	.045	.037	
	35.00	.192	.125	.103	.083	.062	.010	.035	
	40.00	.156	.084	.076	.058	.038	.024	.067	
	45.00	.103	.053	.043	.029	.014	.002	.076	
	50.00	.097	.023	.015	.008	.003	.002	.083	
	55.00	.053	.007	.009	.039	.056	.090	.085	
	60.00	.020	.059	.061	.073	.090	.093	.081	
	65.00	.063						.084	
	70.00	.029	.185	.138	.093	.039	.049	.092	
	75.00	.103	.064	.038	.001	.040	.064	.091	
	80.00	.066	.004	.003	.023	.079	.084	.091	
	85.00	.030	.008	.033	.068	.085	.125	.103	
	90.00	.011	.061	.052	.113	.163	.146	.103	
	95.00		.779	.624	.412	.423	.211	.119	
	UPPER SURFACE	M = 0.60 α = 15.6°							
		.00	.146	1.951	1.140	.879	.483	.423	.548
1.25		1.541	1.094	1.101	.698	.478	.420	.483	
2.50		1.787	1.087	1.097	.688	.479	.411	.473	
3.75		1.838	1.006	1.116	.682	.482	.406	.467	
5.00		1.532	1.951	1.096	.672	.486	.413	.476	
7.50		1.244	1.997	1.097	.671	.487	.412	.469	
10.00		1.038	1.944	1.073	.657	.490	.413	.471	
15.00		.750	1.853	1.069	.647	.491	.413	.473	
20.00		.684	1.675	1.075	.645	.488	.407	.473	
25.00		.618	1.077	1.082	.644	.482	.407	.467	
30.00		.569	.774	1.045	.636	.471	.403	.459	
35.00		.523	.539	1.001	.623	.464	.397	.454	
40.00		.461	.371	.950	.602	.458	.385	.447	
45.00		.344	.233	.898	.581	.452	.372	.442	
50.00		.243	.186	.832	.562	.446	.377	.438	
55.00		.183	.133	.853	.507	.437	.381	.430	
60.00		.010	.010	.855	.530	.420	.376	.422	
65.00		1.068	.680	.854	.476	.418	.376	.426	
70.00		.679	.680	.873	.508	.391	.385	.414	
75.00		.571	.659	.822	.526	.411	.377	.403	
80.00		.446	.570	.862	.521	.405	.377	.389	
85.00		.348	.425	.810	.499	.390	.317	.379	
90.00					.481	.375	.307	.356	
95.00									
LOWER SURFACE	1.25	.274	.436	.415	.486	.459	.446	.310	
	2.50	.311	.553	.532	.536	.506	.483	.313	
	3.75	.548	.580	.533	.532	.480	.436	.283	
	5.00	.610	.547	.506	.494	.442	.398	.200	
	7.50	.595	.521	.475	.461	.403	.374	.148	
	10.00	.570	.453	.420	.388	.344	.329	.035	
	15.00	.501	.404	.361	.334	.289	.254	.038	
	20.00	.438	.353	.321	.288	.237	.181	.059	
	25.00	.407	.304	.272	.247	.194	.139	.059	
	30.00	.351	.260	.228	.201	.154	.098	.135	
	35.00	.306	.231	.192	.166	.119	.047	.178	
	40.00	.277	.185	.154	.124	.073	.007	.175	
	45.00	.212	.144	.119	.086	.037	.027	.193	
	50.00	.211	.116	.088	.047	.001	.061	.209	
	55.00	.188	.074	.050	.007	.042	.090	.207	
	60.00	.119	.008	.020	.022	.086	.102	.216	
	65.00	.016						.238	
	70.00	.048	.247	.202	.105	.047	.033	.229	
	75.00	.173	.120	.077	.006	.048	.077	.237	
	80.00	.127	.083	.027	.053	.098	.105	.243	
	85.00	.076	.038	.027	.123	.110	.123	.250	
	90.00	.051	.008	.029	.198	.196	.183	.256	
	95.00	.081	.761	.766	.480	.498	.536		

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

		PRESSURE COEFFICIENT, P, AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
PERCENT CHORD								
M = 0.60 α = 19.5°								
UPPER SURFACE	0.00	.194	1.145	1.014	.853	.668	.533	.463
	1.25	.831	1.122	.943	.768	.659	.527	.436
	2.50	1.122	1.117	.925	.756	.654	.527	.423
	5.00	1.256	1.116	.935	.754	.651	.518	.421
	7.50	1.220	1.114	.924	.747	.647	.512	.417
	10.00	1.195	1.101	.921	.740	.646	.507	.416
	15.00	1.167	1.092	.913	.734	.639	.495	.417
	20.00	1.134	1.064	.907	.723	.630	.490	.418
	25.00	1.092	1.076	.898	.715	.609	.480	.412
	30.00	1.062	1.064	.883	.703	.596	.471	.406
	35.00	.967	1.052	.853	.694	.586	.475	.403
	40.00	.923	1.010	.809	.679	.567	.494	.396
	45.00	.877	.984	.794	.655	.543	.488	.389
	50.00	.809	.957	.787	.643	.531	.489	.379
	55.00	.743	.936	.780	.628	.516	.489	.370
	60.00	.648	.949	.755	.648	.537	.489	.364
	65.00	.465	.949	.716	.650	.549	.489	.351
	70.00	1.077	.743	.644	.557	.457	.387	.330
75.00	.573	.781	.657	.557	.449	.380	.319	
80.00	.573	.753	.656	.548	.438	.364	.319	
85.00	.530	.722	.648	.531	.419	.346	.305	
90.00	.493	.680	.634	.515				
95.00	.493	.680	.634	.515				
GAP								
LOWER SURFACE	1.25	.182	.427	.355	.432	.383	.387	.369
	2.50	.283	.576	.509	.517	.494	.488	.414
	5.00	.563	.614	.547	.545	.508	.476	.397
	7.50	.700	.593	.508	.514	.486	.446	.344
	10.00	.673	.569	.508	.487	.455	.428	.297
	15.00	.646	.507	.464	.422	.402	.357	.208
	20.00	.576	.467	.399	.375	.349	.298	.128
	25.00	.536	.425	.355	.326	.295	.243	.078
	30.00	.481	.370	.311	.284	.250	.193	.035
	35.00	.426	.327	.267	.237	.203	.143	.004
	40.00	.382	.281	.224	.198	.161	.097	.013
	45.00	.348	.241	.184	.157	.115	.053	.075
	50.00	.279	.193	.138	.107	.072	.024	.087
	55.00	.268	.151	.096	.053	.029	.024	.095
	60.00	.213	.100	.046	.006	.014	.061	.106
	65.00	.165	.022	.038	.058	.077	.080	.118
	70.00	.080	.247	.165	.099	.079	.006	.128
	75.00	.066	.103	.080	.015	.028	.065	.141
80.00	.186	.004	.082	.070	.092	.099	.147	
85.00	.128	.039	.115	.148	.107	.163	.163	
90.00	.088	.163	.216	.223	.207	.193	.193	
95.00	.024	.072	.063	.561	.614	.374	.193	
GAP								
M = 0.60 α = 25.0°								
UPPER SURFACE	0.00	.340	.878	.900	.829	.704	.607	.505
	1.25	.790	.874	.882	.798	.702	.591	.492
	2.50	.714	.873	.882	.799	.699	.591	.494
	5.00	.871	.878	.881	.800	.698	.582	.486
	7.50	.865	.880	.879	.799	.693	.585	.480
	10.00	.868	.882	.874	.798	.686	.577	.477
	15.00	.867	.889	.872	.780	.677	.568	.477
	20.00	.867	.894	.872	.771	.672	.561	.480
	25.00	.871	.893	.865	.765	.665	.551	.483
	30.00	.873	.896	.859	.761	.663	.543	.484
	35.00	.873	.896	.859	.761	.663	.546	.484
	40.00	.873	.896	.859	.761	.663	.546	.484
	45.00	.873	.896	.859	.761	.663	.546	.484
	50.00	.871	.888	.858	.759	.655	.546	.484
	55.00	.865	.885	.857	.755	.653	.546	.484
	60.00	.856	.883	.836	.747	.646	.546	.484
	65.00	.838	.835	.804	.735	.635	.546	.484
	70.00	.797	.831	.785	.635	.546	.484	.484
75.00	.914	.833	.788	.638	.548	.487	.403	
80.00	.834	.833	.788	.638	.548	.487	.394	
85.00	.812	.833	.788	.625	.548	.487	.375	
90.00	.798	.809	.716	.625	.548	.487	.375	
95.00	.783	.790	.710	.625	.548	.487	.375	
GAP								
LOWER SURFACE	1.25	.040	.341	.227	.319	.240	.247	.260
	2.50	.160	.570	.475	.477	.450	.453	.383
	5.00	.582	.858	.774	.866	.830	.823	.415
	7.50	.723	.864	.785	.854	.824	.823	.388
	10.00	.722	.864	.785	.854	.824	.823	.354
	15.00	.741	.814	.804	.808	.808	.808	.272
	20.00	.687	.830	.804	.808	.808	.808	.209
	25.00	.641	.830	.804	.808	.808	.808	.140
	30.00	.598	.830	.804	.808	.808	.808	.087
	35.00	.544	.830	.804	.808	.808	.808	.013
	40.00	.497	.830	.804	.808	.808	.808	.023
	45.00	.461	.830	.804	.808	.808	.808	.046
	50.00	.396	.830	.804	.808	.808	.808	.069
	55.00	.376	.830	.804	.808	.808	.808	.088
	60.00	.319	.830	.804	.808	.808	.808	.101
	65.00	.265	.830	.804	.808	.808	.808	.118
	70.00	.151	.830	.804	.808	.808	.808	.130
	75.00	.133	.830	.804	.808	.808	.808	.146
80.00	.173	.830	.804	.808	.808	.808	.166	
85.00	.088	.830	.804	.808	.808	.808	.190	
90.00	.008	.830	.804	.808	.808	.808	.224	
95.00	.008	.830	.804	.808	.808	.808	.224	
GAP								

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M = 0.85 α = 0.0°								
	.00	.251	.518	.477	.764	.478	.468	.332	
	1.25	.118	.057	.036	.019	.168	.278	.180	
	2.50	.086	.045	.024	.011	.085	.125	.129	
	5.00	.064	.046	.019	.004	.065	.100	.129	
	7.50	.033	.057	.026	.017	.039	.069	.083	
	10.00	.012	.070	.045	.015	.022	.057	.053	
	15.00	.020	.078	.038	.001	.021	.053	.059	
	20.00	.047	.083	.026	.007	.020	.031	.069	
	25.00	.069	.076	.014	.015	.031	.035	.057	
	30.00	.079	.062	.001	.029	.045	.047	.047	
	35.00	.071	.039	.022	.051	.065	.068	.042	
	40.00	.077	.010	.020	.078	.091	.068	.036	
	45.00	.074	.050	.104	.120	.132	.102	.051	
	50.00	.056	.174	.178	.184	.189	.151	.059	
	55.00	.014	.283	.266	.264	.271	.221	.070	
	60.00	.014	.300	.303	.356	.321	.252	.058	
	65.00	.006	.280	.273	.337	.318	.237	.062	
	70.00	.337	.231	.304	.355	.326	.233	.067	
	75.00	.681	.697	.388	.296	.281	.183	.070	
80.00	.562	.701	.369	.298	.251	.174	.083		
85.00	.415	.668	.370	.298	.251	.164	.100		
90.00	.344	.611	.387	.305	.251	.156	.115		
95.00	.220	.525	.385	.307	.253	.152	.112		
LOWER SURFACE	M = 0.85 α = 0.0°								
	1.25	.183	.102	.169	.268	.407	.509	.608	
	2.50	.121	.076	.126	.197	.302	.356	.464	
	5.00	.068	.048	.123	.123	.248	.290	.324	
	7.50	.031	.087	.133	.158	.245	.269	.333	
	10.00	.002	.096	.144	.179	.241	.225	.330	
	15.00	.034	.117	.169	.225	.234	.253	.298	
	20.00	.077	.136	.191	.239	.250	.277	.273	
	25.00	.063	.155	.214	.240	.260	.279	.251	
	30.00	.117	.173	.236	.247	.273	.283	.228	
	35.00	.135	.196	.254	.257	.291	.285	.229	
	40.00	.148	.214	.273	.257	.300	.274	.217	
	45.00	.167	.243	.281	.285	.300	.270	.207	
	50.00	.227	.270	.273	.285	.299	.229	.193	
	55.00	.209	.285	.273	.285	.282	.244	.166	
	60.00	.233	.327	.266	.274	.264	.213	.149	
	65.00	.268	.336	.265	.279	.264	.213	.147	
	70.00	.104	.104	.095	.210	.208	.203	.125	
	75.00	.163	.094	.163	.206	.208	.147	.114	
	80.00	.157	.100	.178	.196	.195	.138	.105	
85.00	.142	.111	.202	.208	.189	.132	.087		
90.00	.129	.147	.211	.211	.194	.132	.088		
GAP		.756	.466	.298	.279	.203			
UPPER SURFACE	M = 0.85 α = 4.0°								
	.00	.247	.101	.081	.277	.106	.081	.066	
	1.25	.251	.847	1.089	.930	.841	.884	.794	
	2.50	.214	.717	.825	.717	.724	.668	.627	
	5.00	.188	.403	.437	.409	.388	.325	.319	
	7.50	.177	.364	.374	.333	.326	.290	.295	
	10.00	.200	.338	.341	.277	.291	.252	.262	
	15.00	.215	.301	.273	.239	.272	.212	.168	
	20.00	.232	.282	.219	.172	.172	.156	.139	
	25.00	.242	.255	.179	.146	.131	.119	.101	
	30.00	.239	.220	.132	.100	.095	.081	.075	
	35.00	.223	.220	.087	.057	.048	.040	.061	
	40.00	.221	.095	.032	.009	.006	.009	.046	
	45.00	.204	.001	.025	.051	.064	.063	.045	
	50.00	.173	.113	.094	.109	.120	.114	.042	
	55.00	.068	.125	.156	.154	.158	.145	.044	
	60.00	.040	.177	.178	.218	.173	.151	.039	
	65.00	.118	.108	.180	.208	.171	.141	.052	
	70.00	.270	.055	.196	.216	.168	.135	.053	
	75.00	.775	.698	.380	.299	.241	.168	.056	
80.00	.648	.683	.380	.299	.241	.158	.063		
85.00	.489	.647	.380	.299	.241	.151	.069		
90.00	.427	.593	.384	.305	.241	.145	.063		
95.00	.314	.519	.378	.306	.241	.145	.063		
LOWER SURFACE	M = 0.85 α = 4.0°								
	1.25	.336	.332	.345	.330	.323	.320	.255	
	2.50	.307	.259	.269	.255	.244	.243	.167	
	5.00	.274	.195	.183	.204	.185	.137	.090	
	7.50	.217	.140	.133	.142	.100	.086	.015	
	10.00	.178	.125	.101	.105	.064	.058	.015	
	15.00	.143	.066	.051	.027	.021	.044	.018	
	20.00	.087	.031	.008	.006	.023	.081	.018	
	25.00	.085	.001	.026	.035	.055	.111	.016	
	30.00	.033	.029	.066	.053	.113	.149	.016	
	35.00	.001	.029	.066	.053	.113	.149	.016	
	40.00	.017	.022	.116	.143	.163	.171	.016	
	45.00	.041	.113	.155	.163	.177	.204	.016	
	50.00	.107	.146	.183	.177	.191	.209	.016	
	55.00	.093	.127	.173	.188	.196	.202	.016	
	60.00	.117	.159	.198	.207	.199	.202	.016	
	65.00	.128	.198	.207	.207	.199	.202	.016	
	70.00	.223	.068	.099	.099	.130	.160	.016	
	75.00	.064	.042	.105	.138	.158	.184	.016	
	80.00	.081	.076	.131	.143	.168	.184	.016	
85.00	.076	.076	.131	.143	.168	.184	.016		
90.00	.101	.143	.181	.181	.174	.184	.016		
GAP		.753	.519	.296	.280	.196			

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M = 0.85 α = 6.0°								
	.00	.242	.217	.209	.128	.474	.378	.304	
	1.25	.518	1.041	1.203	.731	.807	.628	.776	
	2.50	.460	1.028	.975	.680	.762	.500	.646	
	5.00	.392	.900	.791	.626	.780	.453	.552	
	7.50	.356	.794	.671	.573	.689	.434	.514	
	10.00	.330	.674	.599	.516	.656	.405	.453	
	15.00	.333	.487	.469	.443	.573	.361	.340	
	20.00	.340	.414	.382	.361	.478	.315	.253	
	25.00	.348	.367	.376	.289	.367	.264	.180	
	30.00	.333	.310	.311	.227	.251	.232	.133	
	35.00	.313	.243	.184	.173	.153	.195	.104	
	40.00	.307	.188	.096	.120	.071	.165	.083	
	45.00	.281	.060	.042	.067	.004	.132	.070	
	50.00	.240	.049	.010	.016	.057	.106	.068	
	55.00	.114	.143	.064	.030	.113	.088	.075	
	60.00	.012	.142	.097	.097	.147	.075	.073	
	65.00	.092	.092	.098	.078	.186	.069	.088	
	70.00	.223	.032	.128	.108	.191	.030	.092	
	75.00	.768	.715	.374	.307	.253	.201	.097	
	80.00	.649	.693	.381	.313	.259	.210	.092	
85.00	.516	.655	.381	.317	.260	.197	.090		
90.00	.448	.608	.383	.317	.260	.178	.085		
95.00	.351	.536	.376	.308	.248	.154	.075		
LOWER SURFACE	1.25	.390	.443	.448	.447	.434	.424	.377	
	2.50	.384	.369	.374	.357	.362	.345	.291	
	5.00	.352	.293	.286	.305	.269	.244	.204	
	7.50	.314	.240	.233	.249	.209	.187	.121	
	10.00	.270	.214	.197	.205	.168	.149	.082	
	15.00	.235	.148	.145	.128	.100	.097	.042	
	20.00	.178	.104	.092	.086	.060	.038	.078	
	25.00	.160	.068	.081	.048	.081	.005	.113	
	30.00	.113	.037	.017	.019	.014	.043	.131	
	35.00	.071	.005	.013	.013	.043	.081	.136	
	40.00	.050	.081	.041	.045	.073	.111	.142	
	45.00	.027	.057	.065	.071	.104	.141	.137	
	50.00	.038	.092	.093	.097	.123	.165	.135	
	55.00	.031	.114	.107	.119	.145	.171	.126	
	60.00	.056	.148	.136	.136	.152	.171	.109	
	65.00	.099	.187	.186	.167	.167	.160	.098	
	70.00	.180	.098	.027	.044	.077	.101	.095	
	75.00	.167	.081	.074	.096	.119	.095	.070	
	80.00	.018	.055	.103	.098	.132	.095	.066	
	85.00	.040	.094	.136	.132	.135	.100	.044	
	90.00	.049	.144	.157	.145	.152	.095	.042	
GAP		.761	.540		.275	.300			
UPPER SURFACE	M = 0.85 α = 8.0°								
	.00	.226	.518	.522	.039	.812	.634	.544	
	1.25	.667	1.207	.902	.838	.900	.574	.656	
	2.50	.624	1.177	.856	.782	.888	.523	.618	
	5.00	.566	1.147	.837	.741	.880	.498	.597	
	7.50	.529	1.105	.775	.697	.875	.489	.597	
	10.00	.500	1.053	.783	.653	.870	.476	.590	
	15.00	.486	.893	.632	.601	.845	.457	.552	
	20.00	.463	.710	.585	.588	.803	.440	.487	
	25.00	.468	.539	.486	.587	.747	.421	.373	
	30.00	.457	.391	.488	.532	.660	.418	.268	
	35.00	.429	.277	.364	.451	.535	.410	.209	
	40.00	.425	.187	.307	.355	.341	.415	.186	
	45.00	.401	.088	.207	.268	.262	.419	.188	
	50.00	.332	.009	.214	.193	.062	.421	.189	
	55.00	.171	.118	.177	.137	.014	.421	.201	
	60.00	.069	.183	.180	.063	.067	.400	.192	
	65.00	.029	.084	.145	.073	.092	.341	.198	
	70.00	.168	.016	.095	.008	.157	.298	.189	
	75.00	.740	.704	.397	.336	.570	.355	.174	
	80.00	.668	.686	.419	.343	.276	.366	.160	
85.00	.559	.659	.409	.342	.267	.282	.149		
90.00	.485	.623	.399	.327	.250	.218	.143		
95.00	.433	.559	.381	.298	.223	.171	.137		
LOWER SURFACE	1.25	.410	.506	.493	.497	.475	.473	.421	
	2.50	.438	.445	.431	.418	.421	.414	.353	
	5.00	.420	.376	.347	.368	.342	.323	.269	
	7.50	.399	.318	.298	.312	.282	.266	.188	
	10.00	.350	.292	.261	.268	.240	.243	.133	
	15.00	.315	.214	.203	.189	.178	.168	.089	
	20.00	.245	.170	.149	.142	.125	.107	.005	
	25.00	.225	.131	.105	.098	.082	.059	.069	
	30.00	.175	.094	.067	.068	.047	.016	.099	
	35.00	.130	.060	.039	.038	.016	.026	.111	
	40.00	.099	.030	.004	.000	.019	.057	.123	
	45.00	.076	.009	.018	.030	.049	.091	.180	
	50.00	.013	.045	.053	.060	.073	.121	.117	
	55.00	.014	.070	.070	.088	.100	.137	.110	
	60.00	.016	.108	.090	.104	.113	.147	.094	
	65.00	.059	.153	.130	.135	.135	.147	.087	
	70.00	.114						.086	
	75.00	.132	.121	.053	.004	.080	.018	.071	
	80.00	.003	.006	.047	.063	.078	.059	.067	
	85.00	.011	.060	.082	.073	.094	.062	.070	
	90.00	.029	.101	.118	.109	.099	.069	.060	
95.00	.081	.160	.133	.129	.120	.065	.068		
GAP		.756	.579	.293	.313	.296			

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:						
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
$M = 0.85 \quad \alpha = 11.3^\circ$							
UPPER SURFACE							
0.00	1.77	1.912	1.983	1.433	1.622	1.466	1.500
1.25	1.451	1.476	1.135	1.812	1.538	1.396	1.394
2.50	1.622	1.458	1.084	1.802	1.838	1.379	1.395
5.00	1.743	1.489	1.094	1.796	1.832	1.378	1.395
7.50	1.744	1.480	1.063	1.794	1.822	1.377	1.406
10.00	1.756	1.452	1.033	1.785	1.814	1.378	1.401
15.00	1.731	1.377	1.033	1.777	1.805	1.376	1.399
20.00	1.675	1.341	1.033	1.760	1.801	1.370	1.374
25.00	1.645	1.260	1.033	1.780	1.494	1.368	1.361
30.00	1.621	1.205	1.033	1.779	1.489	1.363	1.347
35.00	1.601	1.438	1.033	1.769	1.479	1.357	1.335
40.00	1.592	1.258	1.033	1.741	1.468	1.345	1.326
45.00	1.598	1.114	1.033	1.708	1.451	1.337	1.311
50.00	1.573	0.040	1.033	1.658	1.431	1.324	1.298
55.00	1.345	0.042	1.033	1.551	1.414	1.316	1.286
60.00	1.169	0.038	1.033	1.551	1.399	1.306	1.283
65.00	0.132	0.047	1.033	1.533	1.399	1.291	1.284
70.00	0.230	0.076	1.033	1.556	1.356	1.272	1.271
75.00	0.998	0.704	1.033	1.613	1.384	1.283	1.266
80.00	0.676	0.685	1.033	1.587	1.396	1.287	1.262
85.00	0.568	0.654	1.033	1.528	1.381	1.281	1.259
90.00	0.502	0.612	1.033	1.469	1.366	1.274	1.250
95.00	0.427	0.545	1.033				
LOWER SURFACE							
1.25	1.422	1.567	1.516	1.532	1.508	1.489	1.445
2.50	1.469	1.544	1.506	1.504	1.487	1.468	1.403
5.00	1.530	1.489	1.447	1.463	1.427	1.389	1.326
7.50	1.538	1.437	1.404	1.410	1.375	1.340	1.257
10.00	1.499	1.406	1.365	1.362	1.334	1.318	1.195
15.00	1.452	1.334	1.308	1.293	1.275	1.234	1.090
20.00	1.374	1.284	1.250	1.242	1.216	1.173	1.012
25.00	1.340	1.240	1.206	1.192	1.170	1.121	0.950
30.00	1.288	1.200	1.166	1.156	1.131	1.076	0.905
35.00	1.239	1.161	1.131	1.117	1.094	1.026	0.856
40.00	1.201	1.128	1.095	1.081	1.056	0.913	0.749
45.00	1.172	1.088	1.063	1.044	1.018	0.857	0.698
50.00	1.115	1.047	1.028	1.010	0.912	0.748	0.589
55.00	1.111	1.018	1.001	0.984	0.847	0.680	0.521
60.00	1.067	0.923	0.899	0.886	0.722	0.554	0.395
65.00	1.031	0.774	0.778	0.799	0.610	0.442	0.283
70.00	1.045						
75.00	0.977	1.179	1.118	1.112	1.017	0.848	0.689
80.00	0.976	1.047	1.012	1.001	0.872	0.703	0.544
85.00	0.949	0.903	0.831	0.835	0.716	0.547	0.388
90.00	0.914	0.861	0.788	0.793	0.674	0.505	0.346
95.00	0.804	1.135	1.082	1.141	1.203	1.197	1.189
GAP	0.004	0.733	0.695	0.502	0.429	0.257	
$M = 0.85 \quad \alpha = 15.6^\circ$							
UPPER SURFACE							
0.00	1.889	1.282	1.042	1.829	1.843	1.482	1.532
1.25	1.628	1.468	1.993	1.662	1.538	1.471	1.469
2.50	1.856	1.441	1.978	1.658	1.840	1.471	1.465
5.00	1.035	1.480	1.011	1.644	1.840	1.463	1.463
7.50	1.057	1.395	1.000	1.633	1.842	1.473	1.468
10.00	1.036	1.371	1.998	1.625	1.842	1.473	1.460
15.00	0.998	1.298	1.972	1.614	1.845	1.473	1.457
20.00	0.907	1.269	1.939	1.619	1.845	1.473	1.456
25.00	0.798	1.229	1.913	1.614	1.845	1.473	1.453
30.00	0.661	1.184	1.870	1.608	1.845	1.473	1.456
35.00	0.593	1.186	1.847	1.593	1.845	1.473	1.456
40.00	0.631	1.080	1.809	1.582	1.845	1.473	1.456
45.00	0.681	0.974	1.773	1.578	1.845	1.473	1.456
50.00	0.694	0.883	1.747	1.573	1.845	1.473	1.456
55.00	0.559	0.709	1.707	1.511	1.845	1.473	1.456
60.00	0.482	0.615	1.692	1.560	1.845	1.473	1.456
65.00	0.307	0.529	1.683	1.526	1.845	1.473	1.456
70.00	1.101	0.883	1.610	1.538	1.845	1.473	1.456
75.00	0.757	0.959	1.683	1.538	1.845	1.473	1.456
80.00	0.635	0.723	1.641	1.537	1.845	1.473	1.456
85.00	0.555	0.470	1.628	1.526	1.845	1.473	1.456
90.00	0.408	0.338	1.619	1.511	1.845	1.473	1.456
95.00							
LOWER SURFACE							
1.25	1.384	1.584	1.503	1.571	1.479	1.463	1.429
2.50	1.477	1.619	1.535	1.557	1.507	1.492	1.427
5.00	1.626	1.598	1.535	1.535	1.477	1.445	1.380
7.50	1.693	1.558	1.501	1.490	1.437	1.401	1.310
10.00	1.682	1.528	1.470	1.450	1.399	1.377	1.263
15.00	1.601	1.461	1.417	1.384	1.347	1.300	1.152
20.00	1.519	1.414	1.362	1.338	1.288	1.240	1.043
25.00	1.476	1.363	1.311	1.288	1.238	1.188	0.988
30.00	1.425	1.321	1.271	1.238	1.193	1.138	0.968
35.00	1.368	1.277	1.229	1.196	1.150	1.080	0.916
40.00	1.331	1.248	1.190	1.158	1.111	1.032	0.871
45.00	1.296	1.199	1.154	1.108	1.061	0.982	0.821
50.00	1.233	1.151	1.128	1.066	1.021	0.942	0.781
55.00	1.288	1.119	1.078	1.028	0.982	0.903	0.742
60.00	1.178	1.078	1.041	0.919	0.864	0.785	0.624
65.00	1.136	0.918	0.885	0.779	0.719	0.640	0.479
70.00	0.939						
75.00	0.907	1.257	1.184	1.095	1.089	0.983	0.823
80.00	1.168	1.120	1.046	0.931	0.884	0.784	0.624
85.00	1.118	0.968	0.873	0.768	0.718	0.618	0.458
90.00	0.970	0.811	0.716	0.610	0.560	0.460	0.300
95.00	0.943	0.789	0.694	0.589	0.539	0.439	0.279
GAP							

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M = 0.85 α = 19.8°								
	.00	.028	1.085	.953	.821	.650	.578	.568	
	1.25	.643	1.067	.867	.711	.646	.573	.536	
	2.50	.842	1.058	.834	.705	.643	.570	.533	
	5.00	1.092	1.028	.837	.699	.640	.561	.531	
	7.50	1.090	1.020	.836	.698	.640	.571	.535	
	10.00	1.075	1.012	.836	.698	.640	.571	.522	
	15.00	1.065	.988	.827	.688	.637	.571	.515	
	20.00	1.039	.978	.819	.683	.632	.564	.515	
	25.00	.961	.966	.806	.680	.625	.552	.508	
	30.00	.907	.953	.794	.678	.623	.544	.509	
	35.00	.849	.943	.783	.670	.613	.544	.506	
	40.00	.808	.926	.766	.663	.607	.539	.507	
	45.00	.771	.907	.757	.657	.598	.539	.506	
	50.00	.761	.891	.749	.649	.598	.539	.498	
	55.00	.699	.870	.741	.639	.586	.531	.488	
	60.00	.654	.865	.737	.632	.581	.526	.483	
	LOWER SURFACE	65.00	.551	.866	.737	.632	.581	.526	.483
70.00		.571	.887	.744	.639	.586	.531	.486	
75.00		.757	.745	.684	.591	.537	.449	.451	
80.00		.647	.731	.678	.589	.535	.452	.441	
85.00		.613	.707	.670	.584	.530	.446	.428	
90.00		.527	.695	.661	.590	.516	.428	.410	
95.00		.245	.569	.470	.553	.426	.409	.387	
1.25		.439	.656	.575	.567	.507	.497	.435	
2.50		.691	.662	.585	.533	.512	.482	.416	
5.00		.784	.635	.568	.533	.485	.448	.369	
7.50		.745	.611	.539	.504	.452	.436	.318	
10.00		.700	.581	.549	.446	.405	.363	.213	
15.00		.621	.501	.435	.394	.348	.306	.117	
20.00		.573	.450	.390	.347	.302	.253	.048	
25.00		.523	.407	.347	.302	.255	.196	.023	
30.00		.467	.362	.302	.256	.211	.141	.083	
35.00		.420	.323	.260	.212	.166	.092	.116	
40.00		.386	.277	.219	.166	.118	.039	.147	
45.00	.319	.229	.173	.119	.071	.019	.170		
50.00	.309	.190	.131	.071	.026	.064	.194		
55.00	.259	.141	.089	.017	.019	.110	.210		
60.00	.213	.068	.083	.051	.095	.140	.235		
65.00	.106	.066	.194	.136	.070	.057	.254		
70.00	.214	.047	.047	.018	.058	.115	.283		
75.00	.158	.017	.083	.071	.126	.177	.294		
80.00	.090	.017	.106	.150	.188	.271	.295		
85.00	.048	.118	.183	.225	.261	.292	.319		
GAP		.732	.749	.591	.670	.414			
UPPER SURFACE	M = 0.85 α = 26.0°								
	.00	.210	.802	.825	.773	.731	.869	.624	
	1.25	.551	.801	.809	.754	.731	.658	.609	
	2.50	.845	.797	.789	.751	.729	.649	.607	
	5.00	.834	.803	.789	.751	.729	.627	.601	
	7.50	.819	.806	.798	.751	.726	.633	.602	
	10.00	.809	.806	.802	.757	.726	.633	.595	
	15.00	.796	.803	.800	.747	.726	.634	.591	
	20.00	.793	.812	.797	.743	.723	.631	.589	
	25.00	.792	.818	.793	.744	.719	.614	.591	
	30.00	.797	.823	.791	.742	.718	.619	.591	
	35.00	.796	.826	.789	.741	.718	.613	.592	
	40.00	.801	.831	.781	.741	.720	.615	.592	
	45.00	.804	.834	.779	.748	.723	.619	.600	
	50.00	.807	.831	.777	.751	.749	.653	.600	
	55.00	.805	.824	.779	.750	.755	.681	.596	
	60.00	.798	.812	.774	.734	.727	.716	.584	
	65.00	.778	.798	.766	.731	.705	.747	.584	
70.00	.719	.781	.755	.731	.724	.734	.580		
75.00	.836	.772	.731	.685	.621	.535	.562		
80.00	.774	.779	.733	.686	.629	.550	.548		
85.00	.758	.774	.729	.684	.629	.549	.537		
90.00	.747	.770	.727	.679	.623	.548	.526		
95.00	.726	.760	.722	.682	.610	.531	.511		
LOWER SURFACE	1.25	.054	.512	.379	.465	.312	.277	.279	
	2.50	.328	.677	.568	.556	.479	.466	.404	
	5.00	.735	.737	.637	.619	.541	.505	.442	
	7.50	.833	.736	.639	.590	.543	.502	.415	
	10.00	.815	.722	.629	.582	.527	.499	.382	
	15.00	.809	.675	.593	.543	.492	.447	.295	
	20.00	.750	.636	.553	.504	.448	.394	.198	
	25.00	.704	.590	.511	.459	.404	.345	.130	
	30.00	.660	.546	.470	.418	.362	.293	.059	
	35.00	.606	.503	.429	.371	.318	.240	.001	
	40.00	.563	.459	.384	.327	.275	.187	.051	
	45.00	.525	.415	.340	.288	.234	.132	.091	
	50.00	.463	.366	.294	.234	.178	.074	.184	
	55.00	.445	.322	.247	.185	.123	.018	.160	
	60.00	.393	.268	.193	.125	.072	.033	.184	
	65.00	.346	.219	.121	.051	.006	.078	.216	
	70.00	.230	.097	.075	.020	.019	.019	.265	
	75.00	.200	.375	.271	.220	.152	.116	.299	
80.00	.106	.288	.196	.164	.101	.084	.322		
85.00	.137	.138	.044	.003	.070	.260	.340		
90.00	.165	.048	.042	.090	.146	.311	.380		
95.00	.097	.075	.131	.188	.236	.311			
GAP		.660	.721	.598	.717	.459			

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

PERCENT CHORD		PRESSURE COEFFICIENT, P, AT:							
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
UPPER SURFACE	M = 0.90 α = 0.0°								
	.00	.267	.525	.481	.783	.484	.469	.348	
	1.25	.404	.063	.061	.011	.148	.257	.164	
	2.50	.159	.052	.049	.003	.065	.107	.101	
	5.00	.121	.051	.037	.007	.047	.084	.048	
	7.50	.074	.064	.039	.007	.084	.053	.021	
	10.00	.022	.076	.050	.004	.066	.048	.037	
	15.00	.013	.085	.043	.010	.009	.008	.082	
	20.00	.043	.094	.032	.033	.009	.008	.094	
	25.00	.066	.091	.017	.003	.051	.019	.075	
	30.00	.079	.078	.004	.020	.033	.023	.059	
	35.00	.074	.051	.024	.048	.056	.036	.046	
	40.00	.083	.032	.060	.071	.083	.056	.045	
	45.00	.079	.092	.108	.118	.126	.091	.063	
	50.00	.061	.181	.179	.180	.187	.144	.077	
	55.00	.019	.294	.266	.265	.272	.212	.088	
	60.00	.124	.350	.297	.354	.380	.243	.076	
	65.00	.214	.347	.263	.340	.313	.229	.075	
70.00	.379	.285	.286	.356	.348	.222	.077		
75.00	.588	.710	.388	.318	.272	.198	.087		
80.00	.536	.730	.397	.314	.273	.191	.100		
85.00	.430	.695	.403	.316	.272	.185	.120		
90.00	.378	.643	.414	.316	.272	.174	.127		
95.00	.272	.561	.408	.316	.272	.171	.124		
LOWER SURFACE	1.25	.206	.068	.118	.212	.362	.436	.533	
	2.50	.150	.047	.097	.169	.272	.338	.416	
	5.00	.100	.033	.073	.107	.247	.281	.313	
	7.50	.052	.066	.119	.144	.251	.268	.336	
	10.00	.024	.077	.134	.165	.253	.227	.337	
	15.00	.008	.101	.159	.235	.240	.262	.373	
	20.00	.056	.122	.187	.243	.254	.289	.319	
	25.00	.046	.144	.212	.265	.272	.309	.265	
	30.00	.097	.162	.234	.277	.279	.317	.245	
	35.00	.116	.167	.259	.293	.295	.333	.231	
	40.00	.131	.204	.289	.280	.316	.341	.233	
	45.00	.151	.288	.329	.270	.328	.347	.230	
	50.00	.198	.263	.375	.282	.333	.316	.221	
	55.00	.197	.266	.364	.297	.335	.264	.205	
	60.00	.223	.323	.288	.288	.320	.250	.182	
	65.00	.249	.402	.257	.290	.284	.217	.152	
	70.00	.282						.145	
	75.00	.419	.062	.080	.238	.258	.201	.129	
80.00	.228	.107	.128	.238	.238	.166	.111		
85.00	.222	.103	.123	.220	.221	.153	.102		
90.00	.161	.133	.129	.227	.210	.147	.092		
95.00	.155	.156	.139	.229	.207	.142	.097		
GAP		.770	.523	.319	.295	.216			
UPPER SURFACE	M = 0.90 α = 4.0°								
	.00	.265	.166	.084	.349	.052	.042	.091	
	1.25	.073	.842	1.084	.907	.856	.904	.824	
	2.50	.092	.766	.893	.784	.773	.819	.737	
	5.00	.115	.401	.514	.459	.484	.330	.383	
	7.50	.141	.356	.423	.356	.348	.305	.333	
	10.00	.179	.342	.380	.304	.316	.281	.312	
	15.00	.203	.318	.279	.260	.256	.235	.200	
	20.00	.223	.305	.211	.207	.197	.178	.144	
	25.00	.244	.304	.168	.165	.157	.135	.107	
	30.00	.244	.274	.131	.122	.106	.098	.084	
	35.00	.237	.181	.090	.078	.055	.052	.067	
	40.00	.242	.086	.039	.036	.002	.000	.053	
	45.00	.242	.021	.026	.036	.053	.056	.054	
	50.00	.198	.134	.094	.098	.114	.106	.057	
	55.00	.064	.195	.164	.141	.152	.139	.064	
	60.00	.060	.176	.160	.141	.160	.146	.061	
	65.00	.141	.053	.179	.190	.171	.141	.067	
70.00	.264	.032	.179	.203	.181	.132	.067		
75.00	.703	.723	.399	.322	.288	.188	.078		
80.00	.613	.709	.405	.321	.258	.181	.074		
85.00	.497	.672	.403	.321	.258	.172	.080		
90.00	.457	.623	.401	.326	.258	.166	.084		
95.00	.364	.551	.395	.323	.258	.162	.079		
LOWER SURFACE	1.25	.347	.338	.351	.314	.286	.328	.267	
	2.50	.322	.264	.268	.231	.249	.249	.178	
	5.00	.288	.197	.185	.203	.158	.142	.101	
	7.50	.233	.151	.138	.144	.101	.091	.028	
	10.00	.191	.132	.106	.107	.063	.078	.034	
	15.00	.156	.071	.056	.028	.025	.007	.148	
	20.00	.100	.037	.087	.018	.026	.090	.167	
	25.00	.094	.002	.030	.046	.063	.123	.191	
	30.00	.048	.026	.064	.070	.051	.167	.191	
	35.00	.011	.059	.092	.101	.152	.195	.188	
	40.00	.008	.081	.123	.127	.177	.227	.188	
	45.00	.031	.114	.143	.158	.197	.247	.183	
	50.00	.093	.149	.165	.175	.216	.244	.173	
	55.00	.083	.171	.170	.187	.217	.233	.153	
	60.00	.111	.235	.189	.199	.217	.233	.134	
	65.00	.149	.255	.197	.231	.218	.203	.125	
	70.00	.186						.110	
	75.00	.287	.064	.013	.105	.158	.187	.096	
80.00	.285	.050	.106	.165	.189	.155	.080		
85.00	.091	.080	.138	.161	.192	.143	.070		
90.00	.084	.106	.179	.186	.190	.136	.059		
95.00	.113	.155	.196	.198	.194				
GAP		.769	.548	.319	.296	.207			

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

		PRESSURE COEFFICIENT, P, AT:												
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2						
UPPER SURFACE	PERCENT CHORD													
	M = 0.90 α = 8.0°													
	.00	.244	-	.093	-	.195	-	.191	-	.426	-	.318	-	.266
	1.25	.345	-	.272	-	.185	-	.187	-	.429	-	.636	-	.873
	2.50	.318	-	.292	-	.182	-	.173	-	.413	-	.500	-	.724
	5.00	.292	-	.277	-	.181	-	.186	-	.460	-	.446	-	.598
	7.50	.277	-	.277	-	.180	-	.178	-	.421	-	.431	-	.561
	10.00	.277	-	.277	-	.184	-	.173	-	.422	-	.374	-	.526
	15.00	.310	-	.292	-	.184	-	.173	-	.422	-	.374	-	.434
	20.00	.313	-	.292	-	.184	-	.173	-	.422	-	.374	-	.301
	25.00	.313	-	.292	-	.184	-	.173	-	.422	-	.374	-	.203
	30.00	.313	-	.292	-	.184	-	.173	-	.422	-	.374	-	.151
	35.00	.313	-	.292	-	.184	-	.173	-	.422	-	.374	-	.126
	40.00	.313	-	.292	-	.184	-	.173	-	.422	-	.374	-	.109
	45.00	.313	-	.292	-	.184	-	.173	-	.422	-	.374	-	.103
	50.00	.313	-	.292	-	.184	-	.173	-	.422	-	.374	-	.099
	55.00	.313	-	.292	-	.184	-	.173	-	.422	-	.374	-	.106
	60.00	.313	-	.292	-	.184	-	.173	-	.422	-	.374	-	.104
	65.00	.313	-	.292	-	.184	-	.173	-	.422	-	.374	-	.113
70.00	.313	-	.292	-	.184	-	.173	-	.422	-	.374	-	.109	
75.00	.313	-	.292	-	.184	-	.173	-	.422	-	.374	-	.114	
80.00	.313	-	.292	-	.184	-	.173	-	.422	-	.374	-	.107	
85.00	.313	-	.292	-	.184	-	.173	-	.422	-	.374	-	.101	
90.00	.313	-	.292	-	.184	-	.173	-	.422	-	.374	-	.094	
95.00	.313	-	.292	-	.184	-	.173	-	.422	-	.374	-	.083	
GAP	.410	-	.555	-	.399	-	.314	-	.250	-	.163	-	.083	
LOWER SURFACE	PERCENT CHORD													
	M = 0.90 α = 8.0°													
	1.25	.401	-	.447	-	.440	-	.431	-	.432	-	.425	-	.371
	2.50	.397	-	.371	-	.352	-	.337	-	.359	-	.346	-	.284
	5.00	.368	-	.368	-	.352	-	.337	-	.359	-	.346	-	.196
	7.50	.328	-	.328	-	.352	-	.337	-	.359	-	.346	-	.116
	10.00	.279	-	.279	-	.352	-	.337	-	.359	-	.346	-	.052
	15.00	.244	-	.244	-	.352	-	.337	-	.359	-	.346	-	.064
	20.00	.188	-	.188	-	.352	-	.337	-	.359	-	.346	-	.124
	25.00	.169	-	.169	-	.352	-	.337	-	.359	-	.346	-	.170
	30.00	.118	-	.118	-	.352	-	.337	-	.359	-	.346	-	.176
	35.00	.076	-	.076	-	.352	-	.337	-	.359	-	.346	-	.175
	40.00	.054	-	.054	-	.352	-	.337	-	.359	-	.346	-	.171
	45.00	.037	-	.037	-	.352	-	.337	-	.359	-	.346	-	.171
	50.00	.034	-	.034	-	.352	-	.337	-	.359	-	.346	-	.165
	55.00	.027	-	.027	-	.352	-	.337	-	.359	-	.346	-	.151
	60.00	.057	-	.057	-	.352	-	.337	-	.359	-	.346	-	.131
	65.00	.097	-	.097	-	.352	-	.337	-	.359	-	.346	-	.117
	70.00	.139	-	.139	-	.352	-	.337	-	.359	-	.346	-	.104
75.00	.182	-	.182	-	.352	-	.337	-	.359	-	.346	-	.084	
80.00	.224	-	.224	-	.352	-	.337	-	.359	-	.346	-	.074	
85.00	.266	-	.266	-	.352	-	.337	-	.359	-	.346	-	.067	
90.00	.308	-	.308	-	.352	-	.337	-	.359	-	.346	-	.050	
95.00	.350	-	.350	-	.352	-	.337	-	.359	-	.346	-	.045	
GAP	.410	-	.555	-	.399	-	.314	-	.250	-	.163	-	.083	
UPPER SURFACE	PERCENT CHORD													
	M = 0.90 α = 7.9°													
	.00	.265	-	.348	-	.398	-	.456	-	.484	-	.577	-	.507
	1.25	.340	-	.340	-	.398	-	.456	-	.484	-	.577	-	.700
	2.50	.304	-	.304	-	.398	-	.456	-	.484	-	.577	-	.661
	5.00	.477	-	.477	-	.398	-	.456	-	.484	-	.577	-	.631
	7.50	.445	-	.445	-	.398	-	.456	-	.484	-	.577	-	.638
	10.00	.424	-	.424	-	.398	-	.456	-	.484	-	.577	-	.615
	15.00	.430	-	.430	-	.398	-	.456	-	.484	-	.577	-	.547
	20.00	.415	-	.415	-	.398	-	.456	-	.484	-	.577	-	.485
	25.00	.421	-	.421	-	.398	-	.456	-	.484	-	.577	-	.298
	30.00	.415	-	.415	-	.398	-	.456	-	.484	-	.577	-	.235
	35.00	.403	-	.403	-	.398	-	.456	-	.484	-	.577	-	.209
	40.00	.409	-	.409	-	.398	-	.456	-	.484	-	.577	-	.209
	45.00	.422	-	.422	-	.398	-	.456	-	.484	-	.577	-	.208
	50.00	.461	-	.461	-	.398	-	.456	-	.484	-	.577	-	.214
	55.00	.480	-	.480	-	.398	-	.456	-	.484	-	.577	-	.200
	60.00	.533	-	.533	-	.398	-	.456	-	.484	-	.577	-	.194
	65.00	.577	-	.577	-	.398	-	.456	-	.484	-	.577	-	.190
70.00	.657	-	.657	-	.398	-	.456	-	.484	-	.577	-	.175	
75.00	.657	-	.657	-	.398	-	.456	-	.484	-	.577	-	.167	
80.00	.607	-	.607	-	.398	-	.456	-	.484	-	.577	-	.160	
85.00	.551	-	.551	-	.398	-	.456	-	.484	-	.577	-	.158	
90.00	.471	-	.471	-	.398	-	.456	-	.484	-	.577	-	.158	
95.00	.444	-	.444	-	.398	-	.456	-	.484	-	.577	-	.149	
GAP	.444	-	.444	-	.398	-	.456	-	.484	-	.577	-	.149	
LOWER SURFACE	PERCENT CHORD													
	M = 0.90 α = 7.9°													
	1.25	.431	-	.521	-	.490	-	.475	-	.474	-	.469	-	.418
	2.50	.464	-	.456	-	.488	-	.398	-	.420	-	.410	-	.351
	5.00	.443	-	.388	-	.380	-	.357	-	.335	-	.314	-	.267
	7.50	.419	-	.352	-	.328	-	.301	-	.279	-	.260	-	.187
	10.00	.371	-	.305	-	.282	-	.260	-	.238	-	.216	-	.158
	15.00	.334	-	.231	-	.203	-	.180	-	.177	-	.159	-	.107
	20.00	.263	-	.186	-	.148	-	.135	-	.121	-	.098	-	.107
	25.00	.246	-	.145	-	.104	-	.093	-	.080	-	.044	-	.140
	30.00	.194	-	.110	-	.068	-	.060	-	.041	-	.024	-	.158
	35.00	.143	-	.073	-	.035	-	.023	-	.004	-	.001	-	.156
	40.00	.122	-	.044	-	.000	-	.011	-	.010	-	.011	-	.152
	45.00	.091	-	.007	-	.007	-	.043	-	.064	-	.109	-	.142
	50.00	.029	-	.033	-	.053	-	.074	-	.117	-	.188	-	.187
	55.00	.033	-	.060	-	.076	-	.097	-	.121	-	.191	-	.114
	60.00	.000	-	.106	-	.099	-	.121	-	.170	-	.256	-	.105
	65.00	.041	-	.147	-	.138	-	.154	-	.186	-	.274	-	.103
	70.00	.089	-	.136	-	.083	-	.015	-	.036	-	.036	-	.090
75.00	.160	-	.011	-	.085	-	.089	-	.111	-	.078	-	.083	
80.00	.009	-	.043	-	.091	-	.098	-	.127	-	.077	-	.082	
85.00	.001	-	.087	-	.132	-	.139	-	.189	-	.098	-	.076	
90.00	.016	-	.148	-	.150	-	.157	-	.235	-	.094	-	.081	
95.00	.060	-	.745	-	.617	-	.304	-	.325	-	.303	-		
GAP	.060	-	.745	-	.617	-	.304	-	.325	-	.303	-		

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:							
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
M = 0.90 α = 11.3°								
UPPER SURFACE	.00	.223	.715	.771	.333	.747	.495	.445
	1.25	.335	1.326	1.147	1.073	.478	.441	.364
	2.50	.498	1.308	1.060	1.058	.473	.429	.365
	5.00	.617	1.346	1.051	1.034	.470	.420	.363
	7.50	.658	1.323	1.014	1.040	.465	.425	.360
	10.00	.646	1.285	.989	1.039	.459	.422	.358
	15.00	.634	1.198	.904	1.019	.453	.410	.351
	20.00	.598	1.148	.815	.989	.441	.411	.344
	25.00	.573	1.063	.768	.973	.430	.395	.340
	30.00	.559	.891	.717	.941	.424	.395	.336
	35.00	.543	.560	.678	.913	.419	.389	.329
	40.00	.540	.465	.658	.869	.409	.378	.323
	45.00	.547	.217	.648	.791	.392	.365	.316
	50.00	.576	.029	.624	.681	.369	.355	.310
	55.00	.427	.050	.591	.577	.343	.343	.300
	60.00	.256	.040	.544	.451	.310	.334	.291
LOWER SURFACE	65.00	.087	.056	.488	.414	.275	.324	.289
	70.00	.184	.093	.399	.375	.238	.315	.284
	75.00	1.009	.748	.517	.444	.295	.295	.268
	80.00	.691	.733	.548	.473	.290	.303	.261
	85.00	.568	.706	.508	.453	.286	.311	.258
	90.00	.531	.665	.467	.424	.282	.302	.254
	95.00	.466	.603	.427	.383	.274	.297	.251
	GAP							
	1.25	.432	.589	.535	.541	.503	.500	.462
	2.50	.505	.557	.515	.496	.487	.480	.418
	5.00	.586	.501	.451	.453	.425	.407	.348
	7.50	.561	.447	.404	.403	.376	.354	.278
	10.00	.521	.418	.368	.359	.333	.332	.286
	15.00	.473	.346	.312	.288	.275	.251	.107
	20.00	.390	.297	.253	.241	.216	.188	.000
	25.00	.355	.253	.207	.188	.172	.135	.043
30.00	.306	.210	.168	.152	.131	.089	.095	
35.00	.255	.172	.133	.114	.093	.041	.133	
40.00	.216	.140	.096	.077	.057	.008	.149	
45.00	.187	.100	.065	.041	.016	.054	.167	
50.00	.130	.056	.038	.006	.017	.096	.174	
55.00	.126	.024	.001	.023	.048	.129	.179	
60.00	.083	.022	.032	.058	.078	.159	.179	
65.00	.044	.071	.083	.105	.129	.157	.189	
70.00	.023						.186	
75.00	.093	.192	.121	.074	.046	.100	.181	
80.00	.080	.057	.089	.043	.051	.127	.190	
85.00	.057	.012	.052	.049	.105	.168	.190	
90.00	.024	.063	.093	.093	.135	.217	.186	
95.00	.006	.146	.107	.138	.180	.220	.197	
GAP		.746	.695	.393	.513	.274		
M = 0.90 α = 15.6°								
UPPER SURFACE	.00	.128	1.094	1.035	.851	.533	.489	.535
	1.25	.503	1.502	.986	.741	.527	.489	.475
	2.50	.722	1.473	.925	.738	.522	.489	.473
	5.00	.895	1.490	.931	.730	.522	.481	.475
	7.50	.931	1.501	.950	.724	.516	.492	.481
	10.00	.918	1.487	.931	.718	.510	.492	.473
	15.00	.898	1.409	.950	.695	.500	.493	.467
	20.00	.829	1.357	.937	.684	.504	.493	.466
	25.00	.773	1.269	.920	.687	.506	.484	.467
	30.00	.734	1.197	.911	.670	.506	.489	.466
	35.00	.697	1.133	.899	.653	.509	.486	.471
	40.00	.671	1.066	.879	.638	.509	.470	.466
	45.00	.621	.850	.861	.624	.503	.458	.461
	50.00	.546	.629	.808	.611	.507	.470	.449
	55.00	.462	.544	.793	.601	.509	.470	.449
	60.00	.367	.528	.756	.531	.517	.472	.437
LOWER SURFACE	65.00	.267	.659	.785	.598	.509	.479	.443
	70.00	.055	.467	.688	.553	.494	.477	.438
	75.00	1.160	.929	.620	.566	.492	.426	.417
	80.00	.821	.940	.669	.573	.495	.442	.408
	85.00	.691	.884	.687	.569	.496	.446	.390
	90.00	.699	.748	.641	.559	.495	.438	.373
	95.00	.569	.536	.631	.550	.487	.421	.351
	GAP							
	1.25	.359	.610	.530	.573	.489	.465	.439
	2.50	.508	.632	.570	.536	.512	.494	.439
	5.00	.666	.605	.541	.535	.479	.446	.392
	7.50	.709	.562	.507	.487	.440	.404	.329
	10.00	.665	.533	.478	.446	.401	.384	.274
	15.00	.618	.464	.418	.379	.349	.306	.166
	20.00	.528	.415	.363	.324	.290	.244	.005
	25.00	.485	.364	.313	.280	.243	.190	.005
30.00	.430	.323	.273	.244	.197	.138	.073	
35.00	.378	.283	.234	.199	.154	.083	.131	
40.00	.333	.244	.194	.159	.112	.034	.168	
45.00	.300	.200	.158	.115	.066	.022	.191	
50.00	.238	.136	.119	.074	.027	.076	.204	
55.00	.230	.119	.085	.034	.016	.130	.214	
60.00	.161	.070	.046	.010	.083	.170	.231	
65.00	.140	.009	.016	.074	.119	.186	.260	
70.00	.048						.268	
75.00	.012	.238	.194	.114	.039	.111	.267	
80.00	.168	.117	.085	.039	.078	.145	.291	
85.00	.117	.049	.008	.075	.148	.218	.288	
90.00	.068	.006	.076	.144	.211	.303	.280	
95.00	.036	.067	.137	.213	.289	.385	.289	
GAP		.793	.741	.540	.602	.422		

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M = 0.90 α = 19.8°								
	.00	.019	1.189	.942	.806	.647	.588	.634	
	1.25	.608	1.111	.884	.709	.640	.582	.607	
	2.50	.814	1.106	.854	.708	.640	.579	.601	
	5.00	1.130	1.094	.872	.698	.639	.571	.601	
	7.50	1.126	1.086	.861	.696	.636	.572	.600	
	10.00	1.079	1.066	.857	.698	.636	.572	.591	
	15.00	1.059	1.035	.859	.681	.636	.580	.586	
	20.00	.985	1.028	.835	.678	.633	.580	.584	
	25.00	.917	1.008	.821	.678	.628	.570	.583	
	30.00	.848	.993	.808	.668	.627	.576	.585	
	35.00	.782	.977	.796	.666	.620	.569	.586	
	40.00	.752	.953	.781	.663	.613	.562	.584	
	45.00	.730	.927	.767	.658	.610	.556	.582	
	50.00	.733	.903	.751	.653	.602	.559	.579	
	55.00	.663	.877	.739	.648	.593	.570	.573	
	60.00	.597	.877	.723	.674	.598	.589	.561	
LOWER SURFACE	65.00	.508	.818	.728	.640	.596	.613	.567	
	70.00	.317	.837	.691	.618	.657	.604	.560	
	75.00	.987	.784	.685	.617	.659	.678	.566	
	80.00	.810	.785	.685	.617	.659	.678	.566	
	85.00	.656	.729	.691	.615	.671	.696	.513	
	90.00	.626	.696	.683	.607	.667	.686	.498	
	95.00	.511	.680	.674	.605	.655	.669	.482	
	GAP								
	1.25	.463	.604	.498	.578	.442	.416	.423	
	2.50	.479	.682	.596	.585	.521	.502	.477	
	5.00	.739	.690	.605	.585	.526	.490	.463	
	7.50	.815	.662	.583	.549	.500	.459	.411	
	10.00	.770	.633	.560	.518	.467	.446	.363	
	15.00	.727	.575	.512	.463	.423	.376	.285	
	20.00	.647	.530	.487	.418	.366	.319	.134	
	25.00	.601	.478	.411	.366	.328	.282	.061	
	30.00	.550	.438	.371	.329	.287	.242	.020	
35.00	.485	.394	.329	.281	.239	.190	.085		
40.00	.450	.354	.289	.237	.190	.143	.131		
45.00	.416	.308	.249	.190	.143	.093	.170		
50.00	.382	.261	.207	.143	.097	.007	.194		
55.00	.341	.222	.164	.099	.049	.058	.220		
60.00	.291	.173	.117	.049	.004	.118	.248		
65.00	.247	.105	.044	.019	.067	.139	.287		
70.00	.140						.310		
75.00	.092	.385	.232	.166	.095	.042	.326		
80.00	.246	.184	.085	.018	.038	.124	.359		
85.00	.182	.107	.014	.040	.108	.213	.377		
90.00	.127	.036	.063	.117	.175	.295	.385		
95.00	.086	.060	.134	.190	.242	.338	.401		
M = 0.90 α = 26.1°									
UPPER SURFACE	.00	.173	.805	.885	.780	.743	.961	.628	
	1.25	.465	.798	.808	.752	.738	.681	.609	
	2.50	.841	.793	.786	.758	.731	.669	.603	
	5.00	.836	.799	.805	.751	.733	.636	.583	
	7.50	.825	.801	.796	.750	.733	.643	.589	
	10.00	.815	.807	.801	.743	.733	.644	.594	
	15.00	.801	.801	.797	.750	.738	.644	.588	
	20.00	.790	.812	.796	.747	.736	.648	.587	
	25.00	.790	.817	.795	.753	.748	.636	.593	
	30.00	.789	.817	.794	.746	.744	.648	.591	
	35.00	.793	.828	.786	.747	.746	.643	.597	
	40.00	.797	.832	.779	.746	.746	.648	.598	
	45.00	.803	.832	.772	.746	.746	.646	.603	
	50.00	.799	.826	.768	.747	.749	.672	.601	
	55.00	.794	.825	.764	.747	.749	.693	.608	
	60.00	.763	.800	.741	.733	.736	.724	.590	
	65.00	.672	.798	.743	.736	.719	.752	.576	
70.00	.826	.775	.735	.704	.657	.682	.571		
75.00	.786	.779	.743	.705	.669	.693	.560		
80.00	.772	.772	.740	.705	.669	.694	.544		
85.00	.762	.762	.737	.708	.664	.686	.533		
90.00	.741	.762	.729	.698	.651	.674	.514		
95.00									
LOWER SURFACE	1.25	.088	.548	.419	.483	.338	.291	.308	
	2.50	.370	.705	.624	.578	.428	.382	.428	
	5.00	.803	.765	.682	.636	.483	.428	.466	
	7.50	.866	.761	.671	.613	.463	.423	.445	
	10.00	.846	.751	.656	.608	.450	.428	.416	
	15.00	.840	.705	.626	.570	.416	.470	.340	
	20.00	.780	.665	.583	.527	.473	.424	.342	
	25.00	.737	.620	.543	.486	.431	.371	.177	
	30.00	.691	.576	.502	.447	.390	.324	.104	
	35.00	.637	.532	.462	.408	.347	.274	.080	
	40.00	.594	.495	.419	.358	.305	.221	.009	
	45.00	.558	.448	.375	.314	.256	.170	.046	
	50.00	.501	.400	.332	.266	.211	.109	.081	
	55.00	.480	.359	.290	.219	.161	.084	.118	
	60.00	.430	.307	.233	.163	.109	.040	.145	
	65.00	.388	.234	.187	.093	.037		.175	
	70.00	.267						.214	
75.00	.233	.415	.315	.256	.188	.108	.266		
80.00	.241	.270	.166	.100	.083	.105	.264		
85.00	.275	.176	.094	.044	.027	.145	.309		
90.00	.206	.097	.007	.042	.109	.223	.340		
95.00	.141	.019	.075	.129	.186	.296	.371		
GAP									

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TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

	PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M = 0.94 α = -0.3°							
	.00	.282	.539	.409	.790	.462	.462	.321
	1.25	.122	.012	.023	.048	.155	.284	.201
	2.50	.114	.010	.012	.037	.073	.128	.130
	3.75	.102	.021	.017	.014	.053	.098	.070
	5.00	.083	.042	.016	.028	.027	.061	.035
	7.50	.048	.054	.031	.028	.004	.050	.021
	10.00	.014	.072	.023	.004	.009	.022	.084
	15.00	.018	.072	.008	.010	.009	.030	.144
	20.00	.045	.085	.008	.013	.022	.030	.093
	25.00	.059	.072	.024	.024	.035	.030	.078
	30.00	.059	.044	.045	.048	.057	.039	.066
	35.00	.074	.058	.077	.069	.088	.056	.088
	40.00	.072	.118	.120	.114	.131	.088	.072
	45.00	.050	.205	.190	.181	.193	.143	.098
	50.00	.036	.315	.272	.269	.274	.209	.098
	55.00	.148	.367	.304	.359	.312	.231	.080
	60.00	.231	.360	.267	.347	.309	.206	.081
	65.00	.358	.296	.286	.368	.324	.203	.093
	70.00	.409	.761	.443	.318	.276	.197	.099
LOWER SURFACE	75.00	.458	.790	.433	.318	.277	.191	.111
	80.00	.569	.732	.486	.318	.277	.184	.128
	85.00	.431	.677	.425	.318	.277	.181	.130
	90.00	.355	.601	.418	.318	.277	.181	.133
	1.25	.217	.092	.127	.221	.371	.484	.633
	2.50	.158	.082	.095	.172	.277	.383	.510
	3.75	.110	.043	.096	.100	.250	.320	.430
	5.00	.059	.067	.108	.134	.287	.307	.370
	7.50	.029	.060	.120	.157	.263	.265	.388
	10.00	.005	.100	.145	.210	.267	.319	.405
	15.00	.052	.120	.170	.236	.289	.355	.414
	20.00	.044	.141	.192	.255	.328	.377	.416
	25.00	.092	.160	.220	.263	.346	.396	.400
	30.00	.112	.183	.242	.298	.372	.425	.360
	35.00	.123	.200	.269	.328	.406	.440	.284
	40.00	.143	.230	.294	.371	.447	.450	.216
	45.00	.195	.257	.331	.419	.478	.416	.186
	50.00	.193	.254	.367	.448	.459	.387	.180
	55.00	.211	.296	.411	.448	.351	.313	.163
	60.00	.235	.365	.449	.349	.306	.211	.144
65.00	.258	.435	.498	.270	.296	.160	.144	
70.00	.257	.435	.498	.270	.296	.160	.130	
75.00	.257	.435	.498	.270	.296	.160	.117	
80.00	.257	.435	.498	.270	.296	.160	.109	
85.00	.257	.435	.498	.270	.296	.160	.109	
90.00	.257	.435	.498	.270	.296	.160	.109	
GAP	.261	.207	.286	.231	.240	.137	.094	
		.826	.512	.328	.297	.215	.108	
UPPER SURFACE	M = 0.94 α = 3.9°							
	.00	.282	.243	.130	.397	.024	.019	.117
	1.25	.140	.798	.979	.865	.872	.917	.907
	2.50	.130	.781	.881	.770	.809	.849	.835
	3.75	.133	.340	.528	.548	.477	.481	.489
	5.00	.138	.380	.387	.348	.382	.388	.384
	7.50	.136	.315	.394	.277	.316	.308	.323
	10.00	.174	.295	.378	.248	.249	.248	.293
	15.00	.187	.292	.318	.211	.217	.203	.188
	20.00	.217	.293	.298	.165	.177	.161	.107
	25.00	.283	.289	.084	.117	.127	.118	.089
	30.00	.219	.302	.063	.076	.078	.069	.077
	35.00	.241	.072	.088	.026	.020	.022	.065
	40.00	.256	.065	.089	.034	.041	.040	.068
	45.00	.266	.176	.090	.093	.096	.093	.077
	50.00	.080	.217	.140	.138	.132	.126	.089
	55.00	.056	.177	.184	.190	.148	.136	.085
	60.00	.170	.045	.148	.173	.143	.130	.090
	65.00	.291	.085	.168	.184	.170	.120	.089
	70.00	.451	.793	.483	.335	.278	.197	.091
75.00	.502	.803	.456	.335	.278	.190	.087	
80.00	.585	.746	.449	.331	.276	.187	.093	
85.00	.506	.678	.441	.330	.276	.182	.096	
90.00	.431	.583	.427	.328	.272	.178	.090	
LOWER SURFACE	1.25	.358	.335	.341	.338	.315	.325	.266
	2.50	.333	.264	.260	.246	.240	.248	.184
	3.75	.289	.204	.184	.208	.180	.139	.110
	5.00	.243	.156	.135	.149	.096	.086	.038
	7.50	.205	.142	.107	.114	.062	.076	.024
	10.00	.169	.077	.082	.028	.020	.028	.141
	15.00	.111	.042	.008	.016	.035	.054	.207
	20.00	.102	.007	.032	.054	.078	.093	.290
	25.00	.053	.022	.068	.074	.108	.131	.304
	30.00	.080	.052	.096	.099	.125	.172	.278
	35.00	.000	.075	.139	.138	.167	.212	.265
	40.00	.082	.110	.183	.151	.200	.256	.248
	45.00	.084	.141	.274	.188	.232	.297	.214
	50.00	.077	.145	.274	.188	.232	.297	.188
	55.00	.100	.222	.185	.193	.221	.340	.176
	60.00	.136	.319	.229	.234	.221	.281	.138
	65.00	.164						.138
	70.00	.353	.082	.003	.113	.201	.283	.109
	75.00	.136	.038	.100	.187	.212	.193	.108
	80.00	.131	.067	.142	.200	.225	.168	.093
85.00	.108	.105	.184	.221	.235	.148	.071	
90.00	.138	.144	.222	.223	.261	.141	.065	
GAP		.815	.617	.336	.311	.215		

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT.						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M = 0.94 α = 5.8°								
	.00	.442	.011	.083	.273	.361	.398	.228	
	1.25	.362	.042	.187	.150	.204	.063	.937	
	2.50	.313	.061	.144	.1201	.1209	.001	.854	
	5.00	.286	.075	.1092	.1126	.1144	.893	.755	
	7.50	.267	.087	.089	.1126	.1071	.800	.682	
	10.00	.271	.097	.063	.1016	.885	.669	.602	
	15.00	.275	.119	.072	.652	.415	.494	.486	
	20.00	.278	.136	.052	.244	.296	.396	.372	
	25.00	.298	.157	.041	.181	.234	.308	.261	
	30.00	.301	.165	.173	.121	.188	.247	.197	
	35.00	.299	.193	.100	.085	.136	.180	.153	
	40.00	.312	.155	.047	.044	.082	.120	.124	
	45.00	.328	.053	.003	.006	.026	.058	.116	
	50.00	.367	.088	.049	.066	.028	.002	.113	
	55.00	.226	.140	.085	.120	.074	.042	.121	
	60.00	.011	.108	.104	.183	.112	.067	.120	
	65.00	.138	.105	.097	.167	.119	.071	.128	
	70.00	.251	.030	.120	.179	.142	.075	.128	
75.00	.491	.779	.478	.348	.281	.210	.127		
80.00	.534	.773	.484	.346	.277	.209	.121		
85.00	.581	.789	.460	.343	.277	.207	.117		
90.00	.513	.670	.444	.341	.277	.203	.108		
95.00	.457	.592	.433	.338	.271	.193	.095		
LOWER SURFACE	1.25	.411	.449	.437	.407	.418	.431	.378	
	2.50	.409	.371	.361	.321	.349	.350	.303	
	5.00	.377	.303	.278	.285	.259	.247	.217	
	7.50	.335	.248	.221	.231	.202	.191	.138	
	10.00	.291	.229	.189	.194	.161	.173	.076	
	15.00	.255	.156	.135	.108	.107	.092	.032	
	20.00	.191	.117	.080	.065	.030	.030	.108	
	25.00	.177	.079	.037	.027	.010	.020	.227	
	30.00	.189	.047	.001	.001	.025	.058	.278	
	35.00	.087	.012	.036	.029	.062	.103	.302	
	40.00	.063	.013	.074	.065	.099	.141	.288	
	45.00	.038	.049	.094	.095	.127	.189	.256	
	50.00	.021	.087	.115	.128	.147	.249	.212	
	55.00	.020	.107	.127	.148	.192	.274	.170	
	60.00	.049	.185	.151	.147	.219	.303	.160	
	65.00	.088	.269	.217	.219	.195	.290	.132	
	70.00	.128	.128	.082	.044	.128	.330	.102	
	75.00	.281	.102	.081	.152	.175	.221	.096	
	80.00	.064	.016	.119	.164	.204	.194	.092	
85.00	.057	.048	.175	.194	.218	.139	.065		
90.00	.057	.093	.211	.206	.245	.139	.057		
95.00	.104	.146	.247	.337	.321	.220	.057		
GAP		.785	.785	.647	.647	.647	.647	.647	
UPPER SURFACE	M = 0.94 α = 7.8°								
	.00	.288	.207	.261	.148	.581	.644	.415	
	1.25	.530	1.074	1.273	1.248	1.290	.786	.573	
	2.50	.492	1.063	1.215	1.288	1.287	.769	.559	
	5.00	.449	1.051	1.265	1.232	1.242	.737	.550	
	7.50	.408	.903	1.132	1.182	1.196	.732	.543	
	10.00	.373	.818	1.077	1.126	1.135	.719	.533	
	15.00	.374	.643	.842	1.071	.830	.702	.517	
	20.00	.363	.502	.607	.829	.655	.603	.499	
	25.00	.373	.438	.482	.447	.553	.630	.483	
	30.00	.369	.423	.345	.378	.474	.600	.466	
	35.00	.368	.433	.377	.302	.403	.595	.448	
	40.00	.370	.433	.198	.160	.328	.564	.429	
	45.00	.385	.425	.132	.055	.251	.532	.415	
	50.00	.425	.038	.090	.009	.174	.471	.396	
	55.00	.322	.093	.061	.062	.108	.421	.379	
	60.00	.061	.089	.053	.136	.057	.376	.352	
	65.00	.113	.120	.058	.138	.039	.320	.316	
	70.00	.245	.059	.018	.160	.007	.255	.306	
75.00	.540	.787	.531	.379	.295	.302	.295		
80.00	.571	.774	.524	.375	.305	.323	.271		
85.00	.593	.736	.486	.373	.302	.289	.247		
90.00	.519	.687	.477	.370	.297	.249	.222		
95.00	.503	.619	.459	.361	.285	.203	.199		
LOWER SURFACE	1.25	.443	.523	.492	.464	.454	.463	.420	
	2.50	.473	.455	.425	.383	.394	.404	.349	
	5.00	.451	.384	.345	.340	.310	.308	.267	
	7.50	.427	.331	.292	.283	.256	.249	.190	
	10.00	.377	.303	.262	.247	.213	.230	.130	
	15.00	.340	.229	.198	.170	.155	.152	.019	
	20.00	.271	.183	.144	.126	.099	.088	.053	
	25.00	.249	.143	.099	.081	.054	.035	.146	
	30.00	.198	.107	.089	.049	.017	.009	.198	
	35.00	.152	.071	.084	.012	.020	.053	.229	
	40.00	.123	.042	.009	.026	.056	.097	.243	
	45.00	.096	.004	.036	.063	.089	.150	.258	
	50.00	.038	.036	.066	.098	.115	.189	.255	
	55.00	.038	.061	.088	.120	.164	.221	.226	
	60.00	.008	.138	.112	.134	.168	.264	.175	
	65.00	.038	.158	.142	.206	.267	.264	.119	
	70.00	.076	.138	.050	.015	.100	.246	.104	
	75.00	.216	.012	.085	.139	.149	.193	.104	
	80.00	.013	.031	.189	.146	.188	.160	.100	
85.00	.004	.077	.267	.177	.200	.128	.086		
90.00	.017	.146	.205	.194	.230	.118	.090		
95.00	.061	.783	.685	.361	.340	.289	.090		

CONFIDENTIAL

NACA RM L54C17a

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

TABLE II		PRESSURE COEFFICIENT, P, AT:						
PERCENT CHORD								
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
M = 0.94 $\alpha = 11.4^\circ$								
UPPER SURFACE	.00	.252	-.589	-.628	-.157	-.832	-.593	-.529
	1.25	.260	-.589	-.628	-.157	-.832	-.593	-.529
	2.50	.416	-.589	-.628	-.157	-.832	-.593	-.529
	3.75	.534	-.589	-.628	-.157	-.832	-.593	-.529
	5.00	.577	-.589	-.628	-.157	-.832	-.593	-.529
	7.50	.574	-.589	-.628	-.157	-.832	-.593	-.529
	10.00	.571	-.589	-.628	-.157	-.832	-.593	-.529
	15.00	.537	-.589	-.628	-.157	-.832	-.593	-.529
	20.00	.519	-.589	-.628	-.157	-.832	-.593	-.529
	25.00	.509	-.589	-.628	-.157	-.832	-.593	-.529
	30.00	.494	-.589	-.628	-.157	-.832	-.593	-.529
	35.00	.493	-.589	-.628	-.157	-.832	-.593	-.529
	40.00	.501	-.589	-.628	-.157	-.832	-.593	-.529
	45.00	.533	-.589	-.628	-.157	-.832	-.593	-.529
	50.00	.536	-.589	-.628	-.157	-.832	-.593	-.529
	55.00	.526	-.589	-.628	-.157	-.832	-.593	-.529
	60.00	.502	-.589	-.628	-.157	-.832	-.593	-.529
	65.00	.483	-.589	-.628	-.157	-.832	-.593	-.529
	70.00	.483	-.589	-.628	-.157	-.832	-.593	-.529
LOWER SURFACE	.00	.439	.603	.545	.538	.500	.491	.483
	1.25	.527	.570	.515	.487	.475	.468	.437
	2.50	.636	.510	.452	.444	.415	.393	.359
	3.75	.577	.457	.404	.391	.364	.342	.284
	5.00	.534	.424	.367	.351	.320	.321	.219
	7.50	.486	.356	.308	.276	.262	.236	.098
	10.00	.403	.305	.254	.229	.202	.175	.003
	15.00	.319	.222	.204	.183	.160	.122	.080
	20.00	.267	.182	.168	.146	.114	.074	.150
	25.00	.228	.148	.138	.104	.073	.021	.202
	30.00	.200	.105	.091	.068	.039	.022	.256
	35.00	.141	.063	.035	.024	.002	.072	.263
	40.00	.135	.026	.002	.014	.041	.118	.256
	45.00	.093	.025	.007	.050	.078	.220	.244
	50.00	.055	.071	.085	.081	.140	.231	.270
	55.00	.008	.204	.115	.058	.035	.131	.280
	60.00	.128	.069	.019	.089	.061	.134	.282
	65.00	.077	.005	.073	.097	.127	.199	.302
	70.00	.061	.046	.141	.135	.165	.263	.286
75.00	.032	.117	.176	.173	.191	.286	.288	
80.00	.011	.776	.736	.455	.616	.325		
GAP								
M = 0.94 $\alpha = 15.7^\circ$								
UPPER SURFACE	.00	.156	-.968	-.1.047	-.679	-.576	-.521	-.609
	1.25	.414	-.968	-.1.047	-.679	-.576	-.521	-.609
	2.50	.629	-.968	-.1.047	-.679	-.576	-.521	-.609
	3.75	.799	-.968	-.1.047	-.679	-.576	-.521	-.609
	5.00	.841	-.968	-.1.047	-.679	-.576	-.521	-.609
	7.50	.811	-.968	-.1.047	-.679	-.576	-.521	-.609
	10.00	.757	-.968	-.1.047	-.679	-.576	-.521	-.609
	15.00	.643	-.968	-.1.047	-.679	-.576	-.521	-.609
	20.00	.634	-.968	-.1.047	-.679	-.576	-.521	-.609
	25.00	.633	-.968	-.1.047	-.679	-.576	-.521	-.609
	30.00	.655	-.968	-.1.047	-.679	-.576	-.521	-.609
	35.00	.536	-.968	-.1.047	-.679	-.576	-.521	-.609
	40.00	.420	-.968	-.1.047	-.679	-.576	-.521	-.609
	45.00	.290	-.968	-.1.047	-.679	-.576	-.521	-.609
	50.00	.188	-.968	-.1.047	-.679	-.576	-.521	-.609
	55.00	.088	-.968	-.1.047	-.679	-.576	-.521	-.609
	60.00	.028	-.968	-.1.047	-.679	-.576	-.521	-.609
	65.00	.024	-.968	-.1.047	-.679	-.576	-.521	-.609
	70.00	.028	-.968	-.1.047	-.679	-.576	-.521	-.609
LOWER SURFACE	.00	.365	.633	.551	.584	.500	.480	.463
	1.25	.534	.651	.583	.571	.522	.508	.487
	2.50	.699	.620	.552	.546	.491	.462	.422
	3.75	.729	.575	.519	.497	.452	.421	.359
	5.00	.685	.547	.488	.458	.411	.398	.304
	7.50	.628	.478	.430	.390	.359	.324	.275
	10.00	.540	.375	.338	.344	.303	.268	.210
	15.00	.499	.288	.253	.293	.256	.210	.151
	20.00	.447	.239	.206	.255	.215	.164	.106
	25.00	.391	.197	.164	.207	.173	.110	.057
	30.00	.348	.154	.121	.170	.139	.079	.029
	35.00	.315	.115	.082	.130	.104	.049	.001
	40.00	.282	.078	.045	.096	.064	.014	-.036
	45.00	.248	.041	.008	.044	.000	.164	.263
	50.00	.215	.004	-.026	.001	.033	.201	.307
	55.00	.182	-.027	-.058	-.068	.094		.340
	60.00	.148	-.058	-.088	-.098			.353
	65.00	.115	-.088	-.118	-.128			.366
	70.00	.082	-.118	-.148	-.158			.379
	75.00	.049	-.148	-.178	-.188			.392
	80.00	.016	-.178	-.208	-.218			.405
	85.00	-.016	-.208	-.238	-.248			.418
	90.00	-.049	-.238	-.268	-.278			.431
	95.00	-.082	-.268	-.298	-.308			.444
GAP								

CONFIDENTIAL

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:					
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2
UPPER SURFACE	M = 0.94 α = 20.0°							
	0.00	-.046	-1.192	-.977	-.847	-.654	-.632	-.659
	1.25	-.522	-1.288	-.969	-.757	-.654	-.629	-.643
	2.50	-.793	-1.260	-.942	-.752	-.656	-.628	-.642
	5.00	-1.047	-1.261	-.978	-.750	-.656	-.616	-.639
	7.50	-1.070	-1.245	-.968	-.748	-.656	-.616	-.641
	10.00	-1.060	-1.227	-.970	-.764	-.660	-.625	-.633
	15.00	-1.050	-1.180	-.960	-.724	-.667	-.622	-.631
	20.00	-.924	-1.172	-.932	-.705	-.671	-.616	-.631
	25.00	-.807	-1.144	-.911	-.707	-.671	-.605	-.633
	30.00	-.727	-1.125	-.894	-.706	-.675	-.611	-.636
	35.00	-.694	-1.108	-.881	-.706	-.675	-.611	-.641
	40.00	-.707	-1.080	-.858	-.699	-.672	-.607	-.640
	45.00	-.718	1.041	-.838	-.694	-.666	-.602	-.639
	50.00	-.742	.979	-.814	-.688	-.661	-.604	-.635
	55.00	-.682	.944	-.792	-.683	-.653	-.615	-.624
	60.00	-.590	.975	-.766	-.680	-.634	-.630	-.634
	65.00	-.470	.859	-.730	-.648	-.671	-.633	-.627
	70.00	-.337	.797	-.742	-.671	-.613	-.559	-.605
	75.00	1.045	.912	-.754	-.677	-.621	-.569	-.589
	80.00	.877	.929	-.748	-.673	-.621	-.575	-.579
85.00	.695	.795	-.737	-.668	-.621	-.570	-.565	
90.00	.700	.713	-.737	-.668	-.621	-.570	-.565	
95.00	.610	.663	-.733	-.662	-.616	-.556	-.548	
LOWER SURFACE	1.25	.277	.627	.518	.574	.452	.425	.419
	2.50	.507	.702	.611	.596	.515	.504	.477
	5.00	.776	.706	.617	.602	.535	.475	.470
	7.50	.837	.679	.593	.568	.513	.475	.420
	10.00	.792	.654	.575	.555	.480	.464	.378
	15.00	.743	.594	.529	.513	.437	.397	.273
	20.00	.686	.546	.475	.436	.382	.340	.183
	25.00	.620	.499	.438	.398	.337	.288	.082
	30.00	.570	.457	.390	.352	.293	.239	.001
	35.00	.515	.416	.350	.307	.251	.186	.070
	40.00	.469	.377	.310	.263	.207	.133	.084
	45.00	.437	.331	.278	.221	.161	.079	.167
	50.00	.374	.284	.229	.177	.114	.028	.125
	55.00	.365	.248	.195	.132	.071	.038	.125
	60.00	.315	.198	.147	.088	.025	.016	.125
	65.00	.271	.138	.075	.013	-.042	-.116	.125
	70.00	.184	-.063	-.210	-.051	-.185	-.007	.331
	75.00	.102	.354	.263	.051	-.003	-.096	.368
	80.00	.269	.214	.122	-.004	-.085	-.192	.403
	85.00	.205	.074	.083	-.079	-.162	-.271	.440
	90.00	.146	-.013	-.087	-.156	-.219	-.333	.468
95.00	.107	-.690	-.716	-.584	-.693	-.493	.468	
GAP								
UPPER SURFACE	M = 0.94 α = 24.2°							
	0.00	.094	-.944	-.942	-.838	-.760	-.891	-.749
	1.25	.415	-.931	-.896	-.796	-.756	-.755	-.731
	2.50	.864	-.924	-.867	-.794	-.751	-.731	-.724
	5.00	.933	-.930	-.888	-.795	-.751	-.711	-.704
	7.50	.919	-.927	-.881	-.790	-.751	-.691	-.710
	10.00	.916	-.931	-.888	-.795	-.746	-.692	-.708
	15.00	.919	-.923	-.888	-.795	-.746	-.692	-.704
	20.00	.912	-.934	-.878	-.793	-.747	-.686	-.706
	25.00	.912	-.934	-.869	-.786	-.747	-.694	-.704
	30.00	.899	-.928	-.860	-.779	-.745	-.692	-.706
	35.00	.877	-.929	-.846	-.778	-.743	-.691	-.710
	40.00	.854	-.926	-.837	-.774	-.741	-.695	-.713
	45.00	.833	-.919	-.829	-.766	-.737	-.700	-.713
	50.00	.832	-.909	-.821	-.757	-.735	-.719	-.714
	55.00	.793	-.903	-.815	-.750	-.735	-.719	-.703
	60.00	.765	-.907	-.815	-.740	-.735	-.714	-.696
	65.00	.667	-.879	-.796	-.734	-.720	-.711	-.690
	70.00	.535	-.885	-.789	-.730	-.710	-.645	-.680
	75.00	.917	-.840	-.795	-.737	-.710	-.645	-.680
	80.00	.822	-.832	-.795	-.734	-.713	-.648	-.647
85.00	.768	-.797	-.787	-.734	-.709	-.635	-.649	
90.00	.716	-.787	-.781	-.728	-.697	-.624	-.634	
LOWER SURFACE	1.25	.170	.596	.465	.508	.389	.348	.353
	2.50	.439	.725	.618	.590	.525	.503	.465
	5.00	.828	.765	.660	.637	.567	.521	.496
	7.50	.887	.752	.657	.608	.529	.514	.466
	10.00	.854	.737	.644	.598	.504	.459	.434
	15.00	.829	.688	.602	.550	.458	.410	.338
	20.00	.761	.645	.519	.461	.414	.360	.161
	25.00	.718	.657	.512	.424	.373	.310	.078
	30.00	.672	.618	.437	.381	.330	.260	.016
	35.00	.618	.572	.395	.339	.286	.208	.039
	40.00	.535	.432	.356	.293	.242	.152	.087
	45.00	.475	.384	.312	.248	.196	.098	.118
	50.00	.463	.345	.269	.200	.147	.042	.150
	55.00	.410	.292	.220	.144	.101	.014	.181
	60.00	.370	.224	.145	.077	.029	.033	.221
	65.00	.254	-.417	.310	.258	.191	.076	.298
	70.00	.208	.275	.283	.206	.088	.101	.318
	75.00	.338	.190	.093	.039	-.088	.162	.366
	80.00	.269	.124	.014	-.039	-.102	.219	.416
	85.00	.206	.009	-.087	-.121	-.172	.293	.469
	90.00	.148	-.614	-.684	-.581	-.691	-.470	.469
GAP								

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	$M = 0.98 \quad \alpha = 0.0^\circ$								
	.00	.303	.562	.503	.833	.507	.478	.414	
	1.25	.073	.088	.070	.019	.111	.201	.122	
	2.50	.085	.016	.039	.018	.039	.049	.075	
	5.00	.089	.017	.086	.005	.030	.046	.045	
	7.50	.081	.033	.080	.024	.007	.019	.030	
	10.00	.073	.047	.031	.022	.014	.010	.081	
	15.00	.032	.065	.017	.005	.006	.010	.165	
	20.00	.003	.081	.005	.013	.008	.006	.164	
	25.00	.033	.083	.082	.013	.008	.002	.109	
	30.00	.049	.077	.083	.025	.011	.016	.083	
	35.00	.082	.053	.059	.038	.030	.029	.082	
	40.00	.068	.053	.087	.062	.062	.056	.097	
	45.00	.078	.148	.137	.104	.105	.109	.136	
	50.00	.046	.240	.203	.167	.170	.183	.192	
	55.00	.054	.348	.381	.249	.258	.210	.189	
	60.00	.146	.376	.393	.322	.311	.198	.154	
	65.00	.231	.363	.270	.328	.316	.189	.158	
	70.00	.373	.299	.286	.342	.338	.290	.179	
	LOWER SURFACE	.00	.624	.819	.583	.424	.351	.285	.187
1.25		.648	.866	.584	.424	.353	.285	.179	
2.50		.681	.779	.519	.423	.349	.274	.187	
5.00		.554	.705	.514	.423	.344	.252	.198	
7.50		.416	.602	.507	.400	.339	.252	.198	
1.25		.258	.007	.034	.105	.223	.284	.392	
2.50		.205	.006	.025	.085	.160	.225	.296	
5.00		.157	.007	.030	.038	.163	.229	.235	
7.50		.104	.012	.053	.068	.179	.219	.265	
10.00		.079	.023	.061	.085	.192	.215	.273	
15.00		.046	.055	.093	.154	.203	.240	.257	
20.00		.003	.072	.116	.173	.225	.272	.315	
25.00		.005	.094	.147	.206	.264	.312	.344	
30.00		.040	.111	.162	.203	.274	.322	.344	
35.00		.063	.134	.186	.233	.293	.343	.390	
40.00		.074	.149	.211	.261	.314	.364	.377	
45.00		.097	.179	.236	.286	.344	.394	.381	
50.00		.096	.208	.268	.315	.374	.429	.402	
55.00		.144	.205	.272	.321	.380	.431	.378	
60.00		.164	.238	.310	.352	.410	.461	.312	
UPPER SURFACE	.00	.314	.292	.178	.449	.048	.057	.152	
	1.25	.079	.707	.138	.800	.736	.749	.770	
	2.50	.028	.636	.073	.818	.682	.746	.713	
	5.00	.046	.878	.074	.729	.485	.576	.586	
	7.50	.083	.863	.086	.570	.344	.303	.420	
	10.00	.094	.257	.341	.333	.292	.263	.291	
	15.00	.123	.348	.333	.150	.237	.237	.258	
	20.00	.144	.247	.319	.118	.202	.190	.253	
	25.00	.169	.253	.288	.101	.165	.147	.167	
	30.00	.179	.254	.063	.081	.116	.117	.116	
	35.00	.177	.269	.084	.059	.069	.073	.090	
	40.00	.194	.204	.037	.016	.019	.030	.104	
	45.00	.216	.079	.063	.044	.036	.019	.128	
	50.00	.253	.200	.113	.108	.041	.071	.173	
	55.00	.123	.229	.150	.141	.141	.107	.163	
	60.00	.119	.161	.160	.188	.164	.109	.183	
	65.00	.283	.103	.157	.165	.172	.105	.181	
	70.00	.349	.049	.180	.178	.188	.091	.175	
	75.00	.551	.815	.531	.436	.353	.287	.178	
	80.00	.551	.840	.547	.434	.353	.282	.168	
LOWER SURFACE	.00	.536	.670	.586	.431	.345	.262	.189	
	1.25	.507	.567	.519	.424	.340	.251	.191	
	1.25	.385	.360	.384	.335	.315	.323	.277	
	2.50	.364	.294	.276	.250	.242	.248	.199	
	5.00	.330	.233	.201	.213	.149	.144	.128	
	7.50	.275	.184	.154	.158	.093	.098	.055	
	10.00	.237	.168	.128	.124	.058	.088	.006	
	15.00	.204	.109	.074	.019	.009	.014	.110	
	20.00	.144	.075	.088	.012	.042	.032	.183	
	25.00	.140	.042	.007	.050	.048	.047	.258	
	30.00	.090	.013	.043	.090	.088	.110	.296	
	35.00	.052	.017	.076	.133	.117	.154	.306	
	40.00	.036	.042	.114	.166	.166	.177	.315	
	45.00	.009	.073	.183	.217	.186	.208	.323	
	50.00	.032	.103	.284	.189	.189	.272	.325	
	55.00	.043	.112	.246	.181	.232	.314	.336	
	60.00	.049	.166	.288	.217	.288	.355	.303	
	65.00	.105	.254	.266	.216	.266	.356	.284	
	70.00	.119	.004	.019	.041	.144	.361	.291	
	75.00	.332	.073	.077	.148	.200	.380	.300	
80.00	.117	.105	.109	.164	.289	.380	.266		
85.00	.132	.118	.145	.204	.286	.325	.210		
90.00	.153	.117	.185	.296	.296	.311	.150		
95.00	.158	.042	.678	.414	.398	.305			

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

		PRESSURE COEFFICIENT, P, AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
PERCENT CHORD	M = 0.98 α = 5.9°							
UPPER SURFACE	0.00	.309	.067	.030	.338	.273	.333	.194
	1.25	.258	.088	1.032	.994	1.009	1.021	1.043
	2.50	.230	.094	.989	1.031	1.014	1.049	1.019
	5.00	.217	.043	.941	.971	.950	.957	.889
	7.50	.213	.499	.846	.908	.891	.897	.873
	10.00	.213	.442	.799	.836	.862	.860	.851
	15.00	.235	.376	.458	.783	.751	.579	.450
	20.00	.242	.362	.410	.774	.780	.522	.272
	25.00	.262	.353	.416	.793	.785	.260	.183
	30.00	.265	.348	.389	.741	.745	.213	.152
	35.00	.263	.358	.113	.094	.101	.159	.137
	40.00	.274	.365	.054	.028	.021	.105	.126
	45.00	.294	.070	.031	.021	.008	.048	.129
	50.00	.334	.100	.042	.078	.047	.010	.145
	55.00	.240	.143	.086	.126	.101	.059	.183
	60.00	.004	.130	.104	.181	.133	.091	.194
	65.00	.130	.160	.100	.164	.147	.097	.212
	70.00	.220	.018	.187	.175	.169	.097	.211
	LOWER SURFACE	75.00	.537	.834	.569	.477	.386	.312
80.00		.553	.841	.588	.471	.386	.307	.208
85.00		.567	.774	.567	.469	.383	.298	.214
90.00		.608	.698	.547	.471	.380	.286	.211
95.00		.538	.604	.532	.461	.375	.276	.207
GAP								
1.25		.430	.468	.445	.414	.417	.429	.395
2.50		.431	.392	.388	.325	.343	.354	.314
5.00		.400	.323	.384	.289	.254	.250	.234
7.50		.360	.271	.234	.240	.199	.195	.156
10.00		.315	.249	.203	.203	.166	.184	.100
15.00		.280	.180	.148	.110	.110	.101	.020
20.00		.255	.140	.092	.066	.055	.040	.113
25.00		.201	.103	.051	.027	.013	.000	.207
30.00		.152	.070	.010	.006	.028	.036	.267
35.00		.110	.036	.026	.027	.055	.083	.305
40.00		.085	.012	.070	.063	.087	.128	.334
45.00		.061	.025	.115	.086	.116	.180	.361
50.00		.004	.062	.162	.111	.160	.216	.361
55.00	.002	.078	.115	.163	.173	.278	.339	
60.00	.025	.148	.135	.171	.195	.311	.300	
65.00	.066	.246	.190	.184	.285	.349	.282	
70.00	.087						.280	
75.00	.296	.101	.051	.000	.097	.334	.291	
80.00	.080	.018	.062	.126	.168	.330	.292	
85.00	.086	.030	.112	.145	.209	.335	.270	
90.00	.089	.068	.149	.193	.248	.360	.243	
95.00	.097	.116	.191	.230	.292	.343	.178	
GAP		.826	.702	.433	.446	.336		
M = 0.98 α = 7.9°								
UPPER SURFACE	0.00	.313	.139	.184	.226	.484	.561	.458
	1.25	.419	1.044	1.149	1.121	1.151	1.087	.911
	2.50	.386	1.029	1.105	1.157	1.157	1.098	.905
	5.00	.347	.960	1.038	1.108	1.110	1.022	.862
	7.50	.318	.762	1.036	1.063	1.071	.980	.827
	10.00	.309	.662	1.010	1.016	1.047	.889	.797
	15.00	.327	.538	.920	.988	.972	.670	.691
	20.00	.321	.473	.608	.913	.701	.560	.641
	25.00	.310	.436	.493	.607	.533	.504	.601
	30.00	.331	.419	.418	.408	.415	.481	.558
	35.00	.334	.428	.214	.324	.352	.455	.518
	40.00	.348	.222	.119	.183	.222	.435	.478
	45.00	.389	.015	.056	.037	.238	.408	.438
	50.00	.311	.081	.011	.053	.179	.375	.394
	55.00	.311	.040	.082	.118	.184	.334	.374
	60.00	.145	.162	.064	.182	.065	.300	.357
	65.00	.207	.038	.096	.194	.042	.259	.349
	70.00	.676	.838	.606	.507	.434	.391	.334
	LOWER SURFACE	75.00	.614	.834	.527	.501	.446	.395
80.00		.554	.783	.598	.499	.439	.380	.295
85.00		.526	.722	.570	.502	.427	.359	.283
90.00		.519	.637	.560	.491	.412	.333	.261
GAP								
1.25		.462	.540	.511	.477	.468	.467	.432
2.50		.493	.473	.445	.398	.407	.408	.363
5.00		.469	.404	.361	.353	.324	.317	.207
7.50		.446	.350	.310	.299	.269	.242	.152
10.00		.398	.322	.278	.266	.231	.159	.039
15.00		.362	.250	.220	.184	.115	.099	.050
20.00		.292	.204	.165	.140	.071	.049	.142
25.00		.273	.164	.129	.099	.037	.007	.209
30.00		.219	.128	.044	.031	.002	.046	.265
35.00		.171	.084	.005	.000	.039	.079	.296
40.00		.144	.026	.020	.038	.085	.120	.321
45.00		.118	.012	.047	.083	.099	.179	.332
50.00		.056	.038	.057	.102	.119	.243	.360
55.00		.086	.114	.093	.124	.182	.266	.332
60.00	.015	.209	.136	.165	.187	.331	.296	
65.00	.048						.297	
70.00	.331	.164	.076	.034	.059	.261	.302	
75.00	.016	.036	.030	.111	.140	.254	.301	
80.00	.000	.018	.079	.121	.183	.304	.294	
85.00	.003	.050	.126	.224	.231	.347	.274	
90.00	.034	.113	.170	.224	.273	.361	.215	
95.00		.785	.709	.472	.489	.390		

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M = 0.98 α = 11.4°								
	.00	.295	-.467	-.490	-.013	-.722	-.763	-.717	
	1.25	.172	1.106	1.092	1.065	.826	.711	.643	
	2.50	.323	1.087	1.038	1.059	.820	.686	.633	
	3.75	.441	1.120	.984	1.045	.818	.664	.623	
	5.00	.483	1.080	.966	1.030	.817	.674	.624	
	7.50	.483	1.030	.868	.999	.817	.674	.613	
	10.00	.486	.844	.824	.941	.817	.670	.597	
	12.50	.455	.898	.757	.871	.818	.661	.581	
	15.00	.442	.830	.672	.834	.814	.646	.567	
	17.50	.432	.723	.591	.798	.805	.651	.557	
	20.00	.420	.565	.513	.767	.793	.643	.549	
	22.50	.419	.430	.448	.736	.775	.631	.541	
	25.00	.427	.448	.457	.698	.760	.612	.534	
	27.50	.460	.004	.518	.648	.744	.592	.530	
	30.00	.385	.103	.536	.576	.727	.565	.522	
	32.50	.235	.103	.533	.421	.699	.540	.512	
	35.00	.060	.100	.513	.418	.653	.510	.511	
	37.50	.230	.023	.468	.385	.712	.488	.510	
	40.00	1.021	.851	.661	.600	.653	.490	.504	
	42.50	.649	.853	.697	.603	.701	.498	.497	
45.00	.537	.803	.665	.571	.696	.500	.494		
47.50	.539	.744	.621	.543	.662	.488	.486		
50.00	.456	.666	.589	.507	.613	.474	.483		
LOWER SURFACE	1.25	.467	.638	.575	.569	.525	.508	.365	
	2.50	.567	.598	.541	.509	.495	.483	.324	
	3.75	.708	.540	.477	.468	.431	.408	.259	
	5.00	.613	.487	.430	.418	.381	.355	.193	
	7.50	.571	.458	.394	.377	.343	.337	.178	
	10.00	.520	.384	.339	.307	.284	.255	.032	
	12.50	.439	.336	.282	.258	.226	.198	.075	
	15.00	.403	.291	.237	.211	.180	.143	.133	
	17.50	.355	.253	.198	.177	.142	.100	.221	
	20.00	.302	.216	.159	.138	.103	.055	.268	
	22.50	.265	.182	.123	.097	.065	.007	.300	
	25.00	.236	.142	.095	.060	.025	.050	.329	
	27.50	.182	.098	.063	.020	.010	.111	.393	
	30.00	.173	.060	.037	.012	.061	.151	.350	
	32.50	.130	.001	.085	.034	.111	.188	.347	
	35.00	.099	.034	.040	.121	.101	.274	.373	
	37.50	.028		.169	.135	.053	.116	.392	
	40.00	.130	.242	.034	.040	.060	.141	.412	
	42.50	.111	.112	.071	.077	.116	.206	.432	
	45.00	.093	.049	.076	.126	.172	.281	.447	
	47.50	.057	.004	.122	.176	.215	.319	.468	
50.00	.042	.067	.124	.176	.215	.319	.468		
GAP		.724	.724	.544	.731	.472			
UPPER SURFACE	M = 0.98 α = 15.8°								
	.00	.187	-.849	-.921	-.706	-.563	-.630	-.658	
	1.25	.319	1.287	1.178	.849	.532	.660	.589	
	2.50	.534	1.234	1.111	.842	.534	.628	.588	
	3.75	.701	1.254	1.143	.829	.563	.611	.590	
	5.00	.742	1.260	1.124	.822	.560	.626	.592	
	7.50	.736	1.243	1.143	.841	.597	.628	.590	
	10.00	.726	1.186	1.146	.812	.640	.631	.590	
	12.50	.670	1.126	1.115	.808	.679	.633	.588	
	15.00	.627	1.083	1.083	.836	.699	.621	.585	
	17.50	.593	1.158	1.047	.847	.711	.626	.587	
	20.00	.564	1.147	1.024	.896	.711	.619	.584	
	22.50	.555	1.063	1.016	.867	.708	.614	.580	
	25.00	.557	.695	1.025	.819	.702	.609	.575	
	27.50	.579	.266	1.040	.799	.683	.600	.571	
	30.00	.493	.094	1.146	.780	.678	.589	.566	
	32.50	.392	.093	1.110	.670	.665	.580	.563	
	35.00	.290	.358	.900	.754	.648	.572	.566	
	37.50	.080	.118	.855	.727	.630	.563	.572	
	40.00	.998	.780	.804	.643	.616	.563	.569	
	42.50	.709	.771	.939	.729	.648	.570	.564	
45.00	.602	.748	.877	.700	.628	.565	.559		
47.50	.619	.705	.637	.690	.613	.561	.530		
50.00	.524	.640	.578	.690	.613	.561	.530		
LOWER SURFACE	1.25	.398	.669	.581	.593	.524	.494	.479	
	2.50	.575	.683	.609	.586	.542	.526	.489	
	3.75	.748	.652	.578	.565	.509	.483	.448	
	5.00	.722	.608	.541	.520	.471	.440	.385	
	7.50	.721	.577	.510	.480	.434	.421	.339	
	10.00	.668	.511	.457	.418	.381	.349	.238	
	12.50	.580	.463	.404	.378	.328	.294	.125	
	15.00	.536	.416	.357	.320	.288	.239	.051	
	17.50	.485	.375	.315	.284	.250	.194	.038	
	20.00	.429	.333	.278	.240	.205	.144	.097	
	22.50	.385	.297	.239	.206	.161	.098	.128	
	25.00	.383	.284	.206	.157	.120	.048	.184	
	27.50	.296	.208	.161	.115	.081	.007	.210	
	30.00	.240	.172	.127	.074	.043	.069	.240	
	32.50	.203	.125	.094	.031	.011	.121	.208	
	35.00	.103	.070	.033	.034	.047	.176	.246	
	37.50	.070						.283	
	40.00	.070	.308	.289	.210	.136	.053	.311	
	42.50	.216	.173	.090	.040	.022	.116	.335	
	45.00	.159	.104	.089	.009	.063	.187	.364	
	47.50	.103	.043	.036	.046	.129	.262	.394	
50.00	.070	.036	.088	.106	.189	.312	.443		
GAP		.654	.727	.638	.662	.516			

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:						
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
$M = 0.98 \quad \alpha = 20.2^\circ$							
UPPER SURFACE							
0.00	.053	-1.079	-1.144	-.736	-.717	-.682	-.698
1.25	.409	-1.307	-1.171	-.673	-.717	-.703	-.688
2.50	.698	-1.283	-1.111	-.673	-.720	-.682	-.688
5.00	.941	-1.301	-1.146	-.688	-.720	-.673	-.676
7.50	.968	-1.306	-1.127	-.690	-.725	-.685	-.685
10.00	.951	-1.309	-1.110	-.699	-.727	-.686	-.686
15.00	.931	-1.264	-1.074	-.723	-.732	-.687	-.686
20.00	.853	-1.275	-1.053	-.762	-.736	-.685	-.685
25.00	.773	-1.279	-1.053	-.796	-.737	-.672	-.684
30.00	.716	-1.236	-1.066	-.807	-.736	-.683	-.683
35.00	.683	-1.282	-1.044	-.807	-.732	-.683	-.683
40.00	.669	-1.292	-1.010	-.804	-.730	-.683	-.681
45.00	.667	-1.215	-.984	-.795	-.726	-.678	-.679
50.00	.680	-.906	-.961	-.784	-.721	-.672	-.680
55.00	.603	-.530	-.924	-.772	-.714	-.669	-.680
60.00	.514	-.369	-.888	-.677	-.707	-.662	-.675
65.00	.474	-.396	-.874	-.759	-.697	-.666	-.673
70.00	.379	-.269	-.849	-.728	-.723	-.669	-.682
75.00	.435	-.015	-.741	-.751	-.679	-.646	-.672
80.00	.786	-1.082	-.771	-.771	-.686	-.649	-.670
85.00	.651	-1.019	-.825	-.757	-.688	-.651	-.661
90.00	.633	-.872	-.818	-.745	-.682	-.648	-.650
95.00	.580	-.656	-.814	-.723	-.673	-.641	-.635
1.25	.300	.660	.549	.564	.478	.447	.432
2.50	.545	.732	.645	.607	.556	.536	.494
5.00	.819	.732	.644	.622	.560	.530	.490
7.50	.870	.705	.628	.590	.539	.503	.445
10.00	.823	.679	.603	.564	.508	.492	.402
15.00	.777	.622	.559	.510	.468	.427	.312
20.00	.696	.575	.505	.467	.417	.375	.203
25.00	.652	.527	.465	.420	.373	.324	.129
30.00	.603	.486	.425	.381	.328	.276	.048
35.00	.549	.443	.388	.336	.288	.234	.018
40.00	.504	.408	.357	.295	.248	.179	.079
45.00	.471	.363	.306	.253	.206	.124	.110
50.00	.411	.318	.266	.212	.162	.066	.161
55.00	.400	.282	.229	.170	.118	.007	.139
60.00	.351	.231	.182	.122	.080	-.048	.163
65.00	.311	.171	.116	.057	.009	-.087	.213
70.00	.202						.271
75.00	.128	.384	.300	.256	.182	.034	.283
80.00	.300	.247	.164	.096	.059	-.098	.307
85.00	.233	.180	.105	.042	.025	-.158	.347
90.00	.171	.107	.034	-.021	.096	-.215	.389
95.00	.122	.015	-.023	-.099	.157	-.280	.434
GAP		.619	-.687	-.572	-.680	-.500	
$M = 1.00 \quad \alpha = -0.1^\circ$							
UPPER SURFACE							
0.00	.311	.573	.509	.830	.510	.483	.401
1.25	.135	.001	.045	.051	.136	.225	.147
2.50	.188	.011	.024	.049	.063	.083	.098
5.00	.118	.005	.019	.040	.056	.067	.073
7.50	.109	.013	.014	.054	.029	.036	.054
10.00	.091	.028	.015	.049	.008	.028	.010
15.00	.055	.047	.004	.030	.012	.005	.150
20.00	.020	.066	.086	.034	.005	.004	.169
25.00	.010	.082	.048	.036	.014	.011	.131
30.00	.029	.075	.061	.047	.031	.014	.078
35.00	.033	.027	.080	.061	.046	.029	.073
40.00	.054	.090	.109	.084	.079	.037	.091
45.00	.070	.167	.159	.124	.117	.060	.091
50.00	.039	.257	.224	.185	.178	.105	.139
55.00	.073	.362	.304	.269	.265	.179	.225
60.00	.132	.404	.324	.365	.321	.224	.259
65.00	.328	.391	.288	.350	.329	.228	.256
70.00	.425	.326	.306	.362	.329	.216	.264
75.00	.549	.830	.585	.453	.422	.346	.269
80.00	.568	.887	.531	.462	.430	.342	.265
85.00	.602	.798	.528	.458	.429	.334	.252
90.00	.473	.720	.527	.460	.427	.317	.265
95.00	.393	.601	.520	.466	.428	.311	.265
1.25	.269	.008	.035	.107	.230	.299	.402
2.50	.217	.010	.018	.003	.163	.233	.309
5.00	.172	.022	.027	.030	.158	.223	.241
7.50	.128	.002	.048	.068	.172	.218	.264
10.00	.025	.003	.064	.085	.187	.182	.272
15.00	.059	.042	.087	.143	.198	.236	.259
20.00	.010	.060	.113	.161	.218	.268	.309
25.00	.015	.083	.141	.196	.256	.303	.364
30.00	.032	.102	.155	.195	.262	.311	.345
35.00	.054	.124	.181	.225	.281	.336	.306
40.00	.083	.143	.204	.250	.312	.362	.332
45.00	.085	.170	.231	.289	.349	.397	.388
50.00	.105	.199	.257	.325	.379	.427	.403
55.00	.134	.193	.281	.354	.414	.475	.432
60.00	.153	.225	.319	.376	.424	.493	.438
65.00	.153	.256	.329	.355	.368	.400	.359
70.00	.198						.320
75.00	.368	.101	.141	.200	.226	.286	.275
80.00	.177	.182	.191	.221	.260	.331	.220
85.00	.195	.182	.181	.218	.287	.383	.230
90.00	.207	.187	.194	.244	.290	.343	.243
95.00	.211	.148	.285	.269	.297	.273	.234
GAP		.037	.597	.474	.450	.367	
LOWER SURFACE							

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M = 1.00 α = 3.8°								
	.00	.385	.328	.223	.487	.083	.018	.206	
	1.25	.148	.657	.780	.755	.677	.711	.719	
	2.50	.095	.585	.780	.777	.627	.698	.666	
	5.00	.074	.247	.307	.686	.446	.547	.481	
	7.50	.071	.237	.302	.524	.327	.381	.371	
	10.00	.074	.230	.306	.321	.265	.250	.255	
	15.00	.099	.222	.302	.145	.207	.223	.220	
	20.00	.119	.223	.296	.108	.183	.175	.232	
	25.00	.146	.228	.233	.086	.150	.133	.156	
	30.00	.156	.233	.044	.056	.108	.102	.104	
	35.00	.154	.249	.005	.031	.060	.061	.084	
	40.00	.172	.205	.033	.009	.009	.018	.076	
	45.00	.194	.090	.081	.065	.050	.028	.092	
	50.00	.432	.211	.134	.120	.110	.083	.130	
	55.00	.107	.238	.174	.163	.157	.120	.168	
	60.00	.108	.174	.183	.209	.181	.128	.230	
	65.00	.288	.099	.183	.186	.189	.121	.253	
	70.00	.355	.070	.204	.199	.203	.108	.244	
	75.00	.517	.602	.537	.471	.414	.335	.335	
	80.00	.554	.849	.535	.471	.418	.330	.222	
	85.00	.628	.763	.550	.471	.418	.330	.229	
	90.00	.566	.676	.540	.471	.418	.323	.231	
95.00	.481	.566	.529	.474	.417	.311	.232		
LOWER SURFACE	1.25	.392	.363	.356	.328	.318	.301	.266	
	2.50	.373	.299	.284	.249	.246	.209	.189	
	5.00	.355	.240	.210	.211	.156	.129	.124	
	7.50	.285	.193	.163	.151	.100	.081	.055	
	10.00	.246	.177	.133	.116	.061	.078	.012	
	15.00	.213	.119	.088	.039	.020	.011	.098	
	20.00	.155	.082	.041	.001	.040	.044	.153	
	25.00	.148	.051	.005	.040	.086	.084	.245	
	30.00	.100	.021	.027	.075	.118	.121	.302	
	35.00	.063	.008	.055	.119	.136	.160	.309	
	40.00	.044	.030	.094	.156	.150	.175	.317	
	45.00	.020	.085	.131	.201	.179	.223	.331	
	50.00	.044	.096	.120	.241	.214	.252	.331	
	55.00	.034	.097	.220	.209	.215	.316	.331	
	60.00	.057	.150	.263	.218	.240	.345	.323	
	65.00	.096	.236	.281	.248	.262	.379	.280	
	70.00	.104						.279	
	75.00	.314	.000	.012	.030	.105	.309	.278	
	80.00	.108	.075	.072	.137	.170	.280	.289	
	85.00	.123	.100	.107	.148	.207	.296	.290	
	90.00	.120	.129	.138	.188	.240	.317	.293	
	95.00	.157	.124	.178	.227	.274	.326	.272	
	GAP			.826	.679	.443	.474	.348	
UPPER SURFACE	M = 1.00 α = 5.8°								
	.00	.327	.127	.035	.378	.179	.245	.115	
	1.25	.401	.819	.966	.934	.940	.966	.974	
	2.50	.279	.828	.985	.974	.943	.976	.955	
	5.00	.226	.549	.874	.917	.885	.889	.825	
	7.50	.201	.365	.737	.850	.836	.835	.812	
	10.00	.180	.333	.397	.786	.805	.803	.796	
	15.00	.194	.333	.362	.738	.707	.570	.436	
	20.00	.200	.318	.354	.354	.334	.246	.252	
	25.00	.221	.313	.372	.167	.153	.190	.159	
	30.00	.227	.310	.236	.118	.117	.168	.133	
	35.00	.226	.321	.076	.070	.078	.128	.114	
	40.00	.239	.328	.019	.019	.031	.076	.102	
	45.00	.259	.047	.032	.039	.003	.028	.112	
	50.00	.298	.130	.080	.095	.066	.036	.143	
	55.00	.207	.176	.113	.148	.117	.085	.200	
	60.00	.006	.129	.129	.203	.153	.113	.235	
	65.00	.140	.124	.128	.181	.161	.101	.262	
	70.00	.247	.016	.058	.193	.181	.117	.258	
	75.00	.580	.818	.581	.493	.433	.347	.257	
	80.00	.588	.838	.581	.493	.441	.345	.249	
	85.00	.599	.755	.560	.494	.441	.341	.255	
	90.00	.573	.677	.540	.494	.440	.329	.254	
95.00	.527	.582	.589	.494	.435	.316	.248		
LOWER SURFACE	1.25	.439	.474	.489	.413	.392	.424	.391	
	2.50	.442	.401	.381	.320	.308	.350	.311	
	5.00	.416	.332	.289	.287	.248	.250	.233	
	7.50	.370	.282	.238	.238	.194	.195	.162	
	10.00	.327	.259	.216	.199	.153	.179	.109	
	15.00	.293	.192	.164	.111	.113	.072	.004	
	20.00	.293	.183	.123	.065	.060	.029	.097	
	25.00	.217	.118	.069	.028	.000	.003	.186	
	30.00	.164	.083	.031	.017	.022	.038	.254	
	35.00	.124	.080	.000	.055	.057	.074	.300	
	40.00	.078	.086	.041	.058	.089	.098	.381	
	45.00	.011	.011	.089	.078	.109	.157	.335	
	50.00	.011	.043	.143	.092	.421	.217	.383	
	55.00	.031	.084	.183	.139	.184	.245	.338	
	60.00	.009	.120	.164	.169	.183	.308	.381	
	65.00	.044	.215	.176	.180	.238	.339	.263	
	70.00	.063						.268	
	75.00	.277	.068	.090	.007	.059	.278	.277	
	80.00	.069	.008	.028	.113	.141	.281	.287	
	85.00	.063	.049	.077	.126	.158	.318	.295	
	90.00	.080	.075	.114	.167	.208	.339	.304	
	95.00	.118	.092	.161	.208	.261	.339	.283	
	GAP			.819	.674	.441	.500	.361	

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

PERCENT CHORD		PRESSURE COEFFICIENT, P , AT:												
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2						
UPPER SURFACE	$M = 1.00 \quad \alpha = 7.8^\circ$													
	1.00	.331	..	.069	..	.126	..	.469	..	.404	..	.483	..	.383
	1.25	.500	..	.091	..	1.071	..	1.046	..	1.064	..	1.069	..	.984
	2.50	.416	..	.065	..	1.030	..	1.079	..	1.072	..	1.073	..	.978
	5.00	.344	..	.090	..	1.016	..	1.033	..	1.029	..	.994	..	.895
	7.50	.299	..	.087	..	.956	..	.988	..	.990	..	.965	..	.860
	10.00	.278	..	.057	..	.939	..	.931	..	.976	..	.895	..	.821
	15.00	.286	..	.446	..	.826	..	.915	..	.903	..	.649	..	.646
	20.00	.284	..	.410	..	.564	..	.823	..	.652	..	.490	..	.547
	25.00	.294	..	.391	..	.450	..	.564	..	.476	..	.427	..	.505
	30.00	.292	..	.378	..	.399	..	.362	..	.358	..	.406	..	.462
	35.00	.287	..	.382	..	.188	..	.285	..	.296	..	.383	..	.434
	40.00	.298	..	.393	..	.088	..	.146	..	.249	..	.370	..	.408
	45.00	.314	..	.184	..	.026	..	.005	..	.194	..	.348	..	.391
	50.00	.354	..	.042	..	.087	..	.075	..	.142	..	.318	..	.377
	55.00	.279	..	.106	..	.067	..	.134	..	.094	..	.302	..	.378
	60.00	.109	..	.064	..	.097	..	.211	..	.053	..	.293	..	.372
	65.00	.057	..	.141	..	.167	..	.196	..	.033	..	.252	..	.366
	70.00	.169	..	.032	..	.134	..	.213	..	.033	..	.230	..	.363
	75.00	.623	..	.829	..	.591	..	.515	..	.449	..	.433	..	.366
80.00	.627	..	.839	..	.617	..	.516	..	.464	..	.435	..	.349	
85.00	.567	..	.767	..	.581	..	.514	..	.461	..	.427	..	.332	
90.00	.503	..	.698	..	.558	..	.514	..	.452	..	.405	..	.323	
95.00	.517	..	.608	..	.546	..	.510	..	.436	..	.388	..	.309	
LOWER SURFACE	1.25	.473	..	.546	..	.517	..	.480	..	.471	..	.469	..	.435
	2.50	.506	..	.478	..	.450	..	.391	..	.390	..	.387	..	.367
	5.00	.482	..	.410	..	.364	..	.350	..	.324	..	.312	..	.286
	7.50	.456	..	.355	..	.315	..	.299	..	.271	..	.255	..	.218
	10.00	.408	..	.329	..	.282	..	.258	..	.228	..	.237	..	.162
	15.00	.370	..	.259	..	.224	..	.183	..	.176	..	.161	..	.053
	20.00	.302	..	.212	..	.171	..	.139	..	.122	..	.103	..	.006
	25.00	.283	..	.176	..	.127	..	.101	..	.061	..	.059	..	.138
	30.00	.228	..	.137	..	.082	..	.069	..	.040	..	.019	..	.206
	35.00	.184	..	.106	..	.048	..	.033	..	.010	..	.029	..	.260
	40.00	.157	..	.075	..	.083	..	.004	..	.025	..	.075	..	.295
	45.00	.130	..	.038	..	.034	..	.023	..	.068	..	.122	..	.324
	50.00	.063	..	.003	..	.043	..	.070	..	.112	..	.157	..	.338
	55.00	.071	..	.018	..	.050	..	.109	..	.115	..	.220	..	.353
	60.00	.039	..	.093	..	.080	..	.119	..	.149	..	.259	..	.354
	65.00	.001	..	.194	..	.127	..	.143	..	.186	..	.317	..	.286
	70.00	.029291
	75.00	.239	..	.181	..	.099	..	.054	..	.032	..	.230	..	.299
	80.00	.025	..	.056	..	.019	..	.094	..	.121	..	.231	..	.303
	85.00	.009	..	.015	..	.063	..	.105	..	.166	..	.277	..	.314
90.00	.005	..	.033	..	.104	..	.152	..	.212	..	.338	..	.319	
95.00	.010	..	.091	..	.147	..	.199	..	.246	..	.354	..	.308	
GAP		..	.778	..	.690	..	.468	..	.508	..	.423	
UPPER SURFACE	$M = 1.00 \quad \alpha = 11.4^\circ$													
	.00	.302	..	.441	..	.459	..	.008	..	.692	..	.750	..	.608
	1.25	.153	..	1.080	..	1.057	..	1.004	..	.787	..	.699	..	.546
	2.50	.302	..	1.063	..	1.018	..	.992	..	.781	..	.676	..	.538
	5.00	.418	..	1.094	..	.977	..	.978	..	.779	..	.655	..	.525
	7.50	.463	..	1.049	..	.896	..	.967	..	.778	..	.666	..	.523
	10.00	.444	..	1.000	..	.848	..	.945	..	.778	..	.666	..	.510
	15.00	.468	..	.915	..	.803	..	.899	..	.781	..	.663	..	.497
	20.00	.438	..	.867	..	.739	..	.842	..	.784	..	.656	..	.482
	25.00	.426	..	.802	..	.683	..	.813	..	.780	..	.643	..	.468
	30.00	.417	..	.699	..	.584	..	.781	..	.772	..	.648	..	.457
	35.00	.404	..	.551	..	.509	..	.750	..	.762	..	.640	..	.448
	40.00	.402	..	.417	..	.455	..	.722	..	.749	..	.631	..	.443
	45.00	.411	..	.406	..	.464	..	.689	..	.736	..	.615	..	.436
	50.00	.440	..	.002	..	.520	..	.647	..	.722	..	.596	..	.430
	55.00	.372	..	.113	..	.529	..	.596	..	.706	..	.575	..	.424
	60.00	.223	..	.108	..	.520	..	.459	..	.684	..	.550	..	.414
	65.00	.054	..	.110	..	.493	..	.460	..	.638	..	.523	..	.414
	70.00	.253	..	.018	..	.436	..	.429	..	.715	..	.503	..	.415
	75.00	.989	..	.830	..	.644	..	.612	..	.646	..	.504	..	.406
80.00	.641	..	.831	..	.670	..	.618	..	.689	..	.512	..	.403	
85.00	.513	..	.783	..	.643	..	.586	..	.694	..	.512	..	.398	
90.00	.527	..	.722	..	.601	..	.555	..	.663	..	.506	..	.393	
95.00	.444	..	.645	..	.576	..	.521	..	.620	..	.492	..	.389	
LOWER SURFACE	1.25	.472	..	.647	..	.588	..	.579	..	.531	..	.517	..	.481
	2.50	.577	..	.606	..	.554	..	.521	..	.500	..	.493	..	.442
	5.00	.692	..	.546	..	.489	..	.481	..	.438	..	.417	..	.374
	7.50	.620	..	.494	..	.445	..	.428	..	.388	..	.367	..	.309
	10.00	.578	..	.466	..	.406	..	.428	..	.388	..	.348	..	.255
	15.00	.528	..	.398	..	.349	..	.315	..	.291	..	.267	..	.151
	20.00	.446	..	.344	..	.293	..	.270	..	.235	..	.210	..	.048
	25.00	.412	..	.301	..	.248	..	.221	..	.188	..	.158	..	.102
	30.00	.368	..	.262	..	.209	..	.185	..	.149	..	.126	..	.153
	35.00	.310	..	.225	..	.175	..	.146	..	.110	..	.071	..	.188
	40.00	.271	..	.190	..	.139	..	.109	..	.074	..	.034	..	.214
	45.00	.243	..	.151	..	.107	..	.069	..	.036	..	.002	..	.259
	50.00	.190	..	.108	..	.072	..	.038	..	.006	..	.097	..	.285
	55.00	.181	..	.070	..	.048	..	.002	..	.049	..	.135	..	.247
	60.00	.138	..	.006	..	.019	..	.016	..	.107	..	.169	..	.235
	65.00	.108	..	.034	..	.083	..	.103	..	.099	..	.256	..	.258
	70.00	.034281
	75.00	.153	..	.253	..	.182	..	.152	..	.071	..	.101	..	.302
	80.00	.125	..	.123	..	.047	..	.018	..	.049	..	.140	..	.321
	85.00	.094	..	.072	..	.003	..	.064	..	.103	..	.206	..	.342
90.00	.050	..	.017	..	.087	..	.110	..	.157	..	.268	..	.342	
95.00	.033	..	.032	..	.189	..	.164	..	.202	..	.305	..	.365	
GAP		..	.706	..	.704	..	.544	..	.726	..	.481	

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:						
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
M = 1.00 $\alpha = 15.9^\circ$							
UPPER SURFACE							
.00	.207	-.786	-.865	-.653	-1.149	-.712	-.627
1.25	.270	-1.163	-1.104	-.829	-.552	-.719	-.575
2.50	.482	-1.146	-1.142	-.821	-.548	-.684	-.576
5.00	.646	-1.159	-1.068	-.814	-.548	-.649	-.555
7.50	.690	-1.164	-1.053	-.811	-.548	-.648	-.567
10.00	.684	-1.156	-1.047	-.812	-.553	-.640	-.576
15.00	.676	-1.105	-1.064	-.837	-.583	-.627	-.573
20.00	.623	-1.076	-1.049	-.871	-.607	-.631	-.574
25.00	.579	-1.056	-1.033	-.937	-.630	-.631	-.567
30.00	.547	-1.036	-1.017	-.959	-.655	-.633	-.564
35.00	.542	-1.007	-1.013	-.931	-.697	-.636	-.559
40.00	.511	-.998	-.961	-.881	-.695	-.631	-.552
45.00	.514	-.965	-.974	-.850	-.693	-.612	-.547
50.00	.536	-.954	-.945	-.822	-.689	-.580	-.541
55.00	.458	-.954	1.045	-.691	-.678	-.580	-.533
60.00	.462	-.969	1.059	-.691	-.648	-.556	-.528
65.00	.243	-.988	1.000	-.739	-.773	-.556	-.530
70.00	.031	-.984	-.924	-.772	-.597	-.564	-.528
75.00	-.654	-.788	-.854	-.841	-.630	-.564	-.524
80.00	-.684	-.783	-.947	-.877	-.644	-.556	-.525
85.00	-.622	-.755	-.768	-.842	-.636	-.547	-.520
90.00	-.600	-.710	-.664	-.771	-.624		
95.00	-.511	-.641	-.616				
GAP							
LOWER SURFACE							
.00	.418	-.692	-.598	-.594	-.538	-.509	-.476
1.25	.599	-.703	-.624	-.596	-.558	-.539	-.482
2.50	.777	-.627	-.587	-.576	-.527	-.499	-.444
5.00	.782	-.596	-.553	-.530	-.487	-.456	-.390
7.50	.736	-.531	-.520	-.494	-.451	-.441	-.343
10.00	.682	-.484	-.468	-.428	-.399	-.367	-.246
15.00	.600	-.437	-.414	-.381	-.344	-.309	-.148
20.00	.554	-.395	-.371	-.332	-.297	-.260	-.065
25.00	.504	-.353	-.330	-.299	-.256	-.212	-.009
30.00	.448	-.319	-.297	-.253	-.213	-.166	-.062
35.00	.405	-.275	-.257	-.213	-.175	-.118	-.154
40.00	.376	-.239	-.221	-.173	-.136	-.072	-.187
45.00	.319	-.193	-.186	-.129	-.096	-.039	-.211
50.00	.308	-.147	-.113	-.092	-.053	-.113	-.171
55.00	.263	-.095	-.051	-.047	-.015	-.145	-.203
60.00	.227						-.253
65.00	.127		-.248	-.230	-.149	-.014	-.269
70.00	.017	.330	-.106	-.055	-.040	-.125	-.271
75.00	.235	.138	-.050	-.009	-.045	-.169	-.300
80.00	.176	.071	-.013	-.045	-.100	-.208	-.338
85.00	.118	-.009	-.067	-.104	-.153	-.271	-.377
90.00	.079		-.709	-.663	-.628	-.483	
GAP							
M = 1.03 $\alpha = 0.3^\circ$							
UPPER SURFACE							
.00	.252	-.573	-.507	-.846	-.544	-.525	-.449
1.25	.074	-.001	-.051	-.054	-.180	-.304	-.189
2.50	.078	-.009	-.030	-.059	-.113	-.130	-.142
5.00	.074	-.013	-.034	-.054	-.101	-.116	-.110
7.50	.070	-.028	-.044	-.079	-.075	-.084	-.023
10.00	.068	-.043	-.067	-.074	-.056	-.078	-.015
15.00	.042	-.062	-.080	-.068	-.057	-.042	-.098
20.00	.012	-.080	-.087	-.076	-.051	-.053	-.113
25.00	.020	-.093	-.095	-.078	-.056	-.089	-.081
30.00	.031	-.096	-.114	-.092	-.073	-.079	-.028
35.00	.055	-.049	-.145	-.105	-.087	-.087	-.033
40.00	.083	-.238	-.124	-.124	-.122	-.111	-.039
45.00	.102	-.297	-.159	-.166	-.163	-.158	-.086
50.00	.015	-.394	-.239	-.223	-.220	-.235	-.174
55.00	.023	-.430	-.362	-.301	-.306	-.280	-.213
60.00	.342	-.418	-.386	-.378	-.361	-.272	-.213
65.00	.472	-.350	-.346	-.371	-.394	-.262	-.218
70.00	.507	-.350	-.475	-.386	-.374	-.301	-.212
75.00	.513	-.771	-.484	-.408	-.380	-.297	-.212
80.00	.465	-.825	-.467	-.413	-.380	-.290	-.232
85.00	.421	-.735	-.473	-.421	-.380	-.275	-.232
90.00	.324	-.656	-.464	-.428	-.381	-.268	-.226
GAP							
LOWER SURFACE							
.00	.226	-.012	-.036	-.067	-.193	-.257	-.358
1.25	.186	-.005	-.018	-.051	-.121	-.185	-.260
2.50	.139	-.014	-.032	-.000	-.114	-.174	-.254
5.00	.091	-.005	-.044	-.037	-.129	-.166	-.214
7.50	.070	-.005	-.057	-.053	-.141	-.130	-.220
10.00	.044	-.036	-.088	-.114	-.152	-.185	-.248
15.00	.001	-.054	-.130	-.131	-.172	-.214	-.302
20.00	.006	-.075	-.145	-.163	-.207	-.252	-.288
25.00	.025	-.090	-.162	-.191	-.218	-.276	-.254
30.00	.053	-.112	-.185	-.215	-.256	-.298	-.284
35.00	.058	-.131	-.208	-.245	-.291	-.333	-.333
40.00	.073	-.161	-.235	-.278	-.317	-.361	-.349
45.00	.092	-.189	-.245	-.300	-.352	-.409	-.379
50.00	.125	-.209	-.273	-.318	-.365	-.437	-.399
55.00	.144	-.223	-.281	-.308	-.317	-.363	-.334
60.00	.144						-.286
65.00	.184		-.131	-.193	-.200	-.253	-.335
70.00	.160	-.040	-.168	-.174	-.212	-.291	-.176
75.00	.167	-.154	-.148	-.163	-.232	-.243	-.185
80.00	.178	-.154	-.159	-.198	-.232	-.258	-.193
85.00	.190	-.146	-.178	-.216	-.243	-.286	-.189
90.00	.186	-.106	-.157	-.425	-.398	-.323	
GAP							

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M = 1.03 α = 3.8°								
	.00	.459	.357	.254	.529	.110	.150	.239	
	1.25	.114	.637	.788	.727	.685	.656	.673	
	2.50	.088	.579	.677	.754	.681	.644	.626	
	5.00	.088	.239	.551	.663	.609	.469	.472	
	7.50	.088	.230	.284	.559	.516	.268	.342	
	10.00	.086	.219	.287	.348	.304	.222	.218	
	15.00	.112	.214	.289	.278	.082	.191	.178	
	20.00	.121	.214	.284	.097	.082	.144	.181	
	25.00	.145	.219	.284	.067	.073	.101	.103	
	30.00	.152	.222	.057	.032	.046	.067	.056	
	35.00	.151	.236	.016	.003	.011	.022	.040	
	40.00	.166	.244	.055	.049	.039	.019	.030	
	45.00	.168	.091	.097	.108	.090	.066	.044	
	50.00	.221	.219	.141	.160	.152	.117	.082	
	55.00	.160	.236	.171	.196	.193	.161	.146	
	60.00	.072	.144	.189	.240	.219	.174	.191	
	65.00	.217	.108	.186	.240	.227	.165	.221	
	70.00	.337	.099	.217	.232	.241	.157	.217	
	75.00	.542	.737	.485	.425	.381	.299	.213	
80.00	.532	.759	.501	.425	.404	.295	.197		
85.00	.521	.683	.497	.424	.390	.294	.195		
90.00	.501	.606	.482	.431	.392	.288	.201		
95.00	.448	.510	.469	.433	.393	.281	.199		
LOWER SURFACE	1.25	.329	.353	.357	.341	.348	.353	.308	
	2.50	.333	.290	.289	.260	.277	.278	.234	
	5.00	.304	.235	.215	.225	.191	.178	.169	
	7.50	.266	.189	.171	.178	.136	.126	.103	
	10.00	.227	.177	.144	.135	.099	.119	.062	
	15.00	.201	.118	.097	.105	.061	.055	.046	
	20.00	.149	.086	.053	.029	.008	.005	.057	
	25.00	.146	.056	.022	.007	.039	.026	.192	
	30.00	.101	.027	.012	.039	.073	.063	.242	
	35.00	.063	.000	.036	.078	.107	.091	.263	
	40.00	.047	.019	.068	.116	.131	.122	.269	
	45.00	.027	.019	.084	.159	.130	.173	.269	
	50.00	.033	.079	.139	.197	.155	.200	.287	
	55.00	.023	.077	.177	.213	.162	.259	.286	
	60.00	.045	.122	.216	.179	.187	.297	.284	
	65.00	.076	.198	.267	.219	.223	.327	.241	
	70.00	.086						.237	
	75.00	.305	.015	.040	.010	.050	.252	.231	
	80.00	.091	.056	.041	.092	.117	.241	.241	
	85.00	.089	.080	.073	.103	.153	.265	.251	
90.00	.100	.090	.099	.140	.192	.269	.253		
95.00	.102	.095	.135	.174	.219	.278	.243		
GAP			.771	.615	.395	.443	.513		
UPPER SURFACE	M = 1.03 α = 5.8°								
	.00	.278	.171	.076	.423	.140	.197	.241	
	1.25	.308	.844	.904	.879	.891	.905	.669	
	2.50	.244	.795	.867	.913	.910	.918	.626	
	5.00	.204	.675	.830	.857	.864	.831	.466	
	7.50	.182	.388	.751	.806	.794	.794	.342	
	10.00	.192	.315	.721	.742	.780	.763	.219	
	15.00	.187	.294	.452	.717	.688	.617	.177	
	20.00	.187	.286	.332	.542	.487	.276	.181	
	25.00	.205	.284	.307	.176	.162	.231	.102	
	30.00	.206	.280	.290	.086	.073	.163	.059	
	35.00	.204	.280	.061	.040	.035	.096	.039	
	40.00	.214	.304	.008	.020	.010	.048	.030	
	45.00	.213	.031	.057	.076	.051	.003	.044	
	50.00	.267	.147	.104	.135	.098	.054	.082	
	55.00	.230	.182	.134	.182	.148	.094	.147	
	60.00	.013	.106	.153	.237	.179	.118	.192	
	65.00	.206	.117	.153	.226	.189	.117	.220	
	70.00	.294	.047	.181	.238	.244	.116	.214	
	75.00	.604	.741	.508	.449	.401	.318	.195	
80.00	.665	.749	.532	.449	.417	.310	.195		
85.00	.507	.675	.584	.451	.417	.301	.197		
90.00	.470	.605	.482	.455	.415	.291	.197		
95.00	.444	.522	.469	.452					
LOWER SURFACE	1.25	.390	.485	.478	.443	.439	.456	.309	
	2.50	.428	.414	.404	.357	.369	.385	.237	
	5.00	.409	.347	.326	.227	.281	.286	.170	
	7.50	.384	.299	.277	.274	.227	.231	.104	
	10.00	.342	.277	.246	.231	.192	.216	.060	
	15.00	.311	.213	.196	.155	.149	.147	.045	
	20.00	.248	.175	.145	.111	.099	.089	.056	
	25.00	.237	.141	.108	.065	.062	.042	.193	
	30.00	.190	.110	.073	.029	.023	.041	.264	
	35.00	.149	.080	.039	.015	.013	.061	.268	
	40.00	.127	.059	.002	.035	.066	.106	.268	
	45.00	.108	.024	.040	.042	.084	.166	.266	
	50.00	.047	.004	.093	.080	.134	.208	.285	
	55.00	.055	.012	.132	.113	.136	.254	.282	
	60.00	.087	.073	.137	.138	.184	.289	.242	
	65.00	.001	.163	.139				.235	
	70.00	.026						.230	
	75.00	.028	.094	.132	.048	.019	.222	.242	
	80.00	.029	.025	.089	.065	.095	.236	.250	
	85.00	.029	.073	.084	.080	.135	.263	.253	
90.00	.047	.066	.061	.117	.175	.296	.253		
95.00	.038	.035	.111	.158	.213	.303	.242		
GAP			.756	.617	.401	.468	.532		

TABLE III

WING WITH UPPER SURFACE SPOILER (WITH GAP)

PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:						
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
$M = 1.03 \quad \alpha = 7.8^\circ$							
UPPER SURFACE							
0.00	.484	-.024	-.085	-.308	-.352	-.423	-.063
1.25	.521	-.963	1.008	-.960	1.002	-.969	-.921
2.50	.430	-.920	-.967	1.017	1.012	-.969	-.900
5.00	.359	-.889	-.964	-.968	-.975	-.897	-.783
7.50	.304	-.763	-.903	-.925	-.935	-.862	-.774
10.00	.260	-.548	-.881	-.880	-.913	-.788	-.756
15.00	.267	-.390	-.807	-.865	-.839	-.587	-.541
20.00	.263	-.360	-.640	-.812	-.644	-.464	-.289
25.00	.273	-.348	-.439	-.578	-.428	-.411	-.133
30.00	.268	-.342	-.385	-.368	-.329	-.390	-.111
35.00	.265	-.346	-.195	-.315	-.283	-.369	-.102
40.00	.271	-.357	-.076	-.217	-.255	-.361	-.088
45.00	.289	-.172	-.081	-.024	-.224	-.364	-.093
50.00	.319	-.067	-.088	-.123	-.197	-.324	-.119
55.00	.476	-.124	-.099	-.164	-.161	-.317	-.169
60.00	.044	-.070	-.133	-.267	-.167	-.312	-.198
65.00	.158	-.114	-.144	-.274	-.142	-.293	-.230
70.00	.242	-.000	-.170	-.295	-.214	-.247	-.231
75.00	.661	-.753	-.543	-.462	-.425	-.430	-.233
80.00	.634	-.756	-.565	-.457	-.448	-.435	-.224
85.00	.511	-.693	-.526	-.457	-.425	-.437	-.224
90.00	.443	-.627	-.496	-.460	-.402	-.394	-.226
95.00	.445	-.545	-.486	-.457	-.384	-.375	-.218
LOWER SURFACE							
1.25	.422	.571	.541	.513	.504	.225	.420
2.50	.499	.502	.476	.430	.444	.442	.344
5.00	.494	.435	.394	.367	.365	.332	.269
7.50	.484	.384	.349	.336	.309	.294	.198
10.00	.435	.358	.313	.293	.273	.281	.146
15.00	.401	.289	.260	.223	.220	.204	.042
20.00	.329	.246	.208	.160	.162	.124	.075
25.00	.308	.209	.166	.144	.118	.064	.133
30.00	.258	.174	.125	.114	.081	.066	.201
35.00	.211	.143	.090	.081	.054	.016	.252
40.00	.186	.116	.044	.047	.018	.025	.276
45.00	.163	.080	.083	.028	.013	.073	.295
50.00	.103	.047	.024	.018	.043	.107	.310
55.00	.104	.028	.064	.052	.078	.166	.310
60.00	.074	.044	.032	.074	.097	.221	.284
65.00	.041	.140	.077	.091	.136	.261	.226
70.00	.006	.238	.096	.096	.081	.165	.230
75.00	.013	.114	.088	.044	.070	.187	.244
80.00	.013	.101	.025	.062	.110	.237	.253
85.00	.006	.068	.058	.103	.152	.285	.267
90.00	.023	.041	.103	.149	.194	.306	.258
GAP		.722	.631	.425	.472	.419	
$M = 1.03 \quad \alpha = 11.4^\circ$							
UPPER SURFACE							
0.00	.305	-.382	-.396	.054	-.680	-.695	-.479
1.25	.101	1.004	.986	.930	-.730	-.664	.634
2.50	.241	.988	.945	.918	-.725	-.623	.624
5.00	.365	1.014	.900	.901	-.723	-.634	.604
7.50	.413	.972	.788	.861	-.723	-.635	.593
10.00	.417	.925	.742	.806	-.725	-.635	.577
15.00	.421	.846	.681	.750	-.728	-.630	.562
20.00	.388	.743	.607	.728	-.725	-.620	.546
25.00	.372	.648	.530	.695	-.717	-.630	.532
30.00	.351	.511	.463	.676	-.704	-.623	.523
35.00	.344	.375	.411	.647	-.689	-.615	.516
40.00	.358	.347	.421	.618	-.679	-.601	.509
45.00	.378	.050	.477	.588	-.664	-.589	.504
50.00	.334	.161	.488	.540	-.648	-.567	.496
55.00	.172	.151	.483	.416	-.629	-.549	.488
60.00	.004	.077	.447	.424	-.608	-.512	.483
65.00	.302	.024	.415	.392	-.633	-.495	.476
70.00	.917	.769	.593	.581	-.606	-.516	.473
75.00	.577	.771	.599	.569	-.647	-.518	.467
80.00	.444	.722	.557	.532	-.626	-.508	.463
85.00	.476	.584	.589	.501	-.689	-.493	.460
90.00	.393						
LOWER SURFACE							
1.25	.469	.674	.616	.597	.555	.539	.465
2.50	.594	.635	.581	.545	.526	.514	.487
5.00	.735	.578	.517	.507	.462	.439	.358
7.50	.651	.526	.472	.458	.414	.390	.290
10.00	.609	.498	.435	.418	.376	.376	.258
15.00	.563	.427	.381	.348	.317	.298	.133
20.00	.479	.377	.325	.296	.263	.236	.028
25.00	.444	.335	.278	.253	.219	.185	.057
30.00	.395	.294	.237	.217	.181	.144	.132
35.00	.343	.259	.206	.181	.141	.099	.183
40.00	.304	.226	.167	.140	.104	.062	.223
45.00	.279	.185	.139	.104	.067	.004	.248
50.00	.226	.142	.107	.067	.045	.058	.290
55.00	.218	.108	.084	.033	.007	.102	.331
60.00	.174	.039	.088	.016	.063	.128	.291
65.00	.145	.018	.013	.059	.071	.210	.268
70.00	.088						.290
75.00	.148	.213	.213	.189	.114	.038	.314
80.00	.161	.138	.088	.021	.007	.133	.331
85.00	.123	.105	.030	.020	.088	.153	.354
90.00	.078	.062	.012	.071	.113	.287	.377
95.00	.086	.005	.063	.128	.189	.268	.405
GAP		.654	.656	.521	.681	.473	

TABLE IV

WING WITH LOWER SURFACE SPOILER (WITH GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P , AT						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
		M = 0.60	$\alpha = 0.0^\circ$						
UPPER SURFACE		0.00	.017	.465	.434	.447	.406	.399	.271
		1.25	.106	.160	.242	.300	.378	.496	.456
		2.50	.039	.123	.185	.236	.284	.317	.338
		5.00	.013	.113	.178	.164	.231	.256	.261
		7.50	.031	.126	.166	.183	.225	.239	.260
		10.00	.051	.133	.173	.188	.219	.230	.245
		15.00	.082	.154	.184	.213	.206	.217	.210
		20.00	.117	.182	.213	.216	.219	.226	.206
		25.00	.128	.190	.217	.225	.225	.227	.188
		30.00	.145	.202	.227	.234	.231	.227	.183
		35.00	.154	.217	.235	.241	.234	.226	.176
		40.00	.183	.232	.242	.249	.243	.225	.180
		45.00	.199	.247	.248	.254	.241	.224	.173
		50.00	.210	.247	.244	.250	.241	.219	.162
		55.00	.218	.254	.250	.247	.233	.203	.141
		60.00	.239	.269	.249	.248	.221	.193	.128
	LOWER SURFACE		70.00	.282	.750	.490	.303	.269	.180
		75.00	.219	.003	.103	.156	.138	.150	.108
		80.00	.088	.057	.153	.171	.158	.129	.092
		85.00	.106	.080	.156	.165	.159	.125	.084
		90.00	.099	.090	.181	.178	.166	.125	.074
		95.00	.089	.115	.184	.191	.172	.117	.070
		GAP							
		1.25	.159	.040	.088	.036	.206	.221	.173
		2.50	.091	.036	.001	.005	.102	.098	.122
		5.00	.030	.042	.015	.029	.084	.103	.052
		7.50	.004	.056	.081	.055	.053	.069	.017
		10.00	.009	.066	.092	.011	.034	.055	.006
		15.00	.033	.069	.090	.002	.029	.026	.023
		20.00	.047	.070	.093	.000	.025	.028	.041
		25.00	.062	.068	.093	.010	.030	.031	.034
		30.00	.074	.066	.093	.019	.039	.031	.033
		35.00	.087	.060	.093	.039	.057	.041	.032
	40.00	.087	.056	.086	.061	.086	.061	.034	
	45.00	.082	.056	.086	.099	.122	.088	.033	
	50.00	.051	.118	.141	.151	.176	.133	.037	
	55.00	.004	.234	.235	.227	.253	.196	.044	
	60.00	.096	.291	.309	.312	.321	.253	.032	
	65.00	.296	.278	.302	.308	.320	.233	.035	
	70.00	.503						.038	
	75.00	1.095	.672	.340	.295	.229	.164	.047	
	80.00	.523	.735	.351	.296	.235	.157	.059	
	85.00	.328	.704	.367	.304	.235	.147	.079	
	90.00	.281	.623	.374	.314	.239	.136	.090	
	95.00	.118	.510	.374	.322	.246	.133	.085	
	GAP		.229	.315	.320	.332	.222		
		M = 0.60	$\alpha = 3.8^\circ$						
UPPER SURFACE		0.00	.021	.200	.483	.224	.454	.354	.577
		1.25	.157	1.054	1.383	1.274	.995	.931	.975
		2.50	.231	.581	.714	.878	.856	.803	.825
		5.00	.256	.436	.547	.610	.753	.742	.696
		7.50	.242	.385	.484	.533	.673	.674	.618
		10.00	.257	.354	.443	.490	.611	.583	.541
		15.00	.252	.340	.406	.453	.510	.534	.407
		20.00	.274	.320	.382	.405	.457	.467	.358
		25.00	.275	.320	.368	.388	.424	.421	.310
		30.00	.282	.315	.352	.356	.389	.379	.259
		35.00	.271	.317	.345	.349	.367	.344	.258
		40.00	.268	.315	.337	.336	.345	.322	.245
		45.00	.269	.326	.330	.338	.337	.280	.237
		50.00	.291	.330	.320	.319	.315	.257	.223
		55.00	.295	.320	.304	.303	.301	.233	.204
		60.00	.297	.317	.279	.266	.276	.206	.192
	LOWER SURFACE		65.00	.309	.383	.273	.266	.248	.177
		70.00	.333	.635	.361	.252	.202	.200	.178
		75.00	.271	.031	.152	.196	.156	.166	.164
		80.00	.132	.073	.174	.182	.182	.160	.167
		85.00	.138	.072	.170	.180	.175	.142	.160
		90.00	.116	.078	.180	.179	.171	.132	.150
		95.00	.100	.078	.178				
		GAP							
		1.25	.309	.373	.408	.465	.466	.479	.398
		2.50	.283	.294	.339	.361	.384	.355	.341
		5.00	.245	.216	.255	.260	.319	.311	.239
		7.50	.204	.169	.236	.213	.265	.258	.168
		10.00	.178	.139	.179	.206	.226	.224	.124
		15.00	.137	.107	.146	.173	.194	.166	.087
		20.00	.107	.085	.126	.149	.166	.143	.002
		25.00	.079	.067	.112	.136	.150	.127	.010
		30.00	.087	.064	.100	.127	.143	.113	.007
	35.00	.083	.070	.111	.133	.143	.105	.027	
	40.00	.044	.083	.124	.142	.147	.107	.036	
	45.00	.037	.121	.153	.164	.173	.123	.047	
	50.00	.045	.173	.192	.202	.211	.152	.054	
	55.00	.069	.267	.266	.265	.268	.200	.061	
	60.00	.165	.321	.310	.338	.316	.230	.051	
	65.00	.367	.296	.311	.322	.317	.209	.060	
	70.00	.579						.073	
	75.00	1.002	.586	.287	.251	.209	.143	.081	
	80.00	.671	.465	.295	.259	.214	.136	.094	
	85.00	.220	.638	.310	.269	.213	.128	.107	
	90.00	.142	.544	.332	.281	.221	.120	.109	
	95.00	.047	.414	.327	.286	.222	.129	.106	
	GAP		.255	.315	.330	.317	.195		

TABLE IV

WING WITH LOWER SURFACE SPOILER (WITH GAP)

	PERCENT CHORD	PRESSURE COEFFICIENT, P_i AT						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M 0.60 α 6.0°							
	0.00	.008	-.677	-.782	-.431	-.830	-.741	-.770
	1.25	.397	1.146	-.945	-.970	-.981	1.003	-.912
	2.50	.468	1.043	-.914	-.947	-.965	.962	-.892
	5.00	.457	.878	-.877	-.917	-.928	.954	-.851
	7.50	.410	.757	-.826	-.884	-.890	.915	-.838
	10.00	.426	.672	-.763	-.842	-.847	.891	-.806
	15.00	.394	.569	-.683	-.728	-.744	.824	-.729
	20.00	.407	.499	-.604	-.637	-.663	.779	-.666
	25.00	.403	.456	-.558	-.565	-.598	.723	-.589
	30.00	.397	.430	-.511	-.507	-.547	.645	-.531
	35.00	.370	.418	-.472	-.468	-.497	.558	-.482
	40.00	.359	.403	-.442	-.427	-.451	.475	-.457
	45.00	.362	.403	-.414	-.396	-.414	.396	-.425
	50.00	.374	.397	-.386	-.367	-.374	.340	-.403
	55.00	.365	.380	-.358	-.341	-.341	.299	-.388
	60.00	.361	.368	-.323	-.318	-.308	.262	-.366
	65.00	.370	.359	-.306	-.299	-.269	.227	-.359
	70.00	.377	.619	-.302	-.292	-.230	.185	-.354
	75.00	.314	.101	-.205	-.225	-.213	.177	-.327
	80.00	.171	.111	-.189	-.199	-.201	.169	-.324
	85.00	.167	.101	-.189	-.185	-.198	.146	-.294
	90.00	.137	.083	-.168	-.168	-.169	.129	-.257
	95.00	.105	.071	-.163				
LOWER SURFACE	M 0.60 α 8.1°							
	1.25	.345	.462	.461	.512	.483	.490	.419
	2.50	.353	.408	.422	.428	.450	.426	.395
	5.00	.348	.328	.351	.351	.402	.390	.307
	7.50	.313	.275	.339	.310	.342	.341	.229
	10.00	.285	.240	.269	.290	.306	.305	.179
	15.00	.235	.196	.228	.246	.258	.235	.119
	20.00	.198	.160	.196	.214	.224	.204	.050
	25.00	.161	.136	.177	.197	.203	.176	.018
	30.00	.130	.124	.158	.183	.187	.156	.000
	35.00	.121	.124	.161	.182	.184	.144	.017
	40.00	.107	.113	.158	.182	.187	.141	.028
	45.00	.092	.110	.166	.200	.199	.146	.040
	50.00	.098	.210	.216	.229	.229	.164	.049
	55.00	.114	.285	.289	.282	.283	.206	.060
	60.00	.200	.344	.332	.355	.333	.241	.057
	65.00	.397	.324	.334	.349	.339	.223	.062
	70.00	.636						.072
	75.00	.997	.568	.282	.243	.206	.125	.081
	80.00	.429	.684	.292	.252	.211	.119	.095
	85.00	.181	.639	.307	.264	.218	.115	.106
	90.00	.099	.527	.323	.279	.219	.119	.106
	95.00	.005	.367	.308	.266	.219	.131	.112
	GAP		.283	.337	.355	.336	.204	
UPPER SURFACE	M 0.60 α 8.1°							
	0.00	.002	.990	1.143	.910	1.024	.692	.548
	1.25	.646	1.162	1.038	1.306	.993	.802	.579
	2.50	.703	1.124	1.024	1.300	.983	.771	.569
	5.00	.665	1.063	1.006	1.284	.965	.785	.541
	7.50	.589	.981	.933	1.263	.970	.775	.547
	10.00	.529	.912	.861	1.300	.958	.774	.535
	15.00	.514	.855	.791	1.278	.925	.743	.515
	20.00	.514	.785	.791	1.234	.922	.713	.476
	25.00	.501	.633	.748	1.114	.879	.687	.445
	30.00	.488	.541	.682	.897	.847	.666	.423
	35.00	.482	.487	.610	.678	.808	.640	.403
	40.00	.439	.455	.548	.471	.763	.610	.391
	45.00	.423	.443	.488	.336	.711	.570	.361
	50.00	.426	.433	.431	.261	.645	.534	.344
	55.00	.412	.410	.387	.232	.585	.495	.332
	60.00	.399	.386	.338	.213	.515	.460	.317
	65.00	.404	.367	.321	.206	.486	.409	.313
	70.00	.409	.600	.351	.230	.358	.372	.311
	75.00	.342	.152	.351	.161	.314	.356	.287
	80.00	.200	.137	.284	.138	.314	.345	.283
	85.00	.189	.131	.181	.125	.278	.330	.279
	90.00	.151	.081	.177	.126	.258	.286	.259
	95.00	.101	.050	.145	.129	.199	.263	.251
LOWER SURFACE	M 0.60 α 8.1°							
	1.25	.349	.488	.471	.520	.490	.487	.422
	2.50	.384	.469	.467	.482	.489	.445	.408
	5.00	.438	.399	.410	.417	.452	.417	.325
	7.50	.404	.383	.407	.373	.399	.371	.245
	10.00	.380	.313	.338	.359	.362	.333	.205
	15.00	.326	.264	.293	.304	.308	.266	.141
	20.00	.280	.224	.254	.278	.270	.231	.099
	25.00	.237	.196	.230	.246	.242	.204	.018
	30.00	.199	.180	.207	.227	.219	.174	.002
	35.00	.185	.173	.205	.219	.212	.155	.019
	40.00	.169	.176	.220	.228	.219	.147	.027
	45.00	.147	.196	.246	.252	.243	.166	.036
	50.00	.148	.235	.304	.299	.287	.203	.055
	55.00	.158	.308	.385	.372	.338	.244	.052
	60.00	.235	.350	.360	.367	.344	.229	.064
	65.00	.488						.087
	70.00	.673						.094
	75.00	1.003	.585	.278	.247	.251	.178	.123
	80.00	.355	.707	.288	.253	.256	.179	.123
	85.00	.189	.644	.302	.266	.263	.179	.134
	90.00	.061	.502	.319	.288	.288	.200	.134
	95.00	.046	.303	.295	.268	.268	.224	.148
	GAP		.311	.362	.369	.345	.211	

TABLE IV

WING WITH LOWER SURFACE SPOILER (WITH GAP)

TABLE IV								
PRESSURE COEFFICIENT, P, AT.								
PERCENT CHORD	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
M = 0.60 $\alpha = 12.1^\circ$								
UPPER SURFACE	0.00	-.004	-.1.854	-1.556	-1.046	-.775	-.488	-.411
	1.25	-.1.253	-.1.714	-1.175	-.911	-.624	-.442	-.356
	2.50	-.1.377	-.1.759	-1.190	-.911	-.624	-.442	-.349
	5.00	-.1.241	-.1.842	-1.221	-.907	-.618	-.432	-.334
	7.50	-.1.059	-.1.966	-1.236	-.907	-.611	-.429	-.342
	10.00	-.1.053	-.2.120	-1.260	-.908	-.608	-.427	-.338
	15.00	-.855	-.2.115	-1.304	-.890	-.592	-.415	-.337
	20.00	-.752	-.2.115	-1.375	-.878	-.576	-.404	-.317
	25.00	-.691	-.1.963	-1.394	-.861	-.566	-.393	-.313
	30.00	-.644	-.947	-1.351	-.858	-.556	-.382	-.304
	35.00	-.577	-.940	-1.303	-.857	-.541	-.375	-.304
	40.00	-.538	-.940	-1.216	-.840	-.530	-.361	-.284
	45.00	-.516	-.981	-1.111	-.827	-.521	-.362	-.274
	50.00	-.500	-.903	-.836	-.803	-.517	-.352	-.266
	55.00	-.475	-.988	-.691	-.782	-.514	-.348	-.255
	60.00	-.451	-.969	-.545	-.747	-.509	-.338	-.251
	65.00	-.443	-.932	-.454	-.718	-.492	-.333	-.251
	70.00	-.427	-.935	-.270	-.712	-.489	-.319	-.238
	75.00	-.357	-.917	-.168	-.674	-.485	-.301	-.235
	80.00	-.226	-.087	-.143	-.638	-.485	-.295	-.234
	85.00	-.192	-.046	-.109	-.586	-.499	-.287	-.222
	90.00	-.149			-.559	-.512	-.278	-.212
	95.00	-.114	-.011		-.518			
LOWER SURFACE	0.00							
	1.25	-.280	-.467	-.438	-.503	-.473	-.471	-.408
	2.50	-.375	-.542	-.509	-.479	-.504	-.444	-.405
	5.00	-.549	-.515	-.502	-.479	-.483	-.429	-.336
	7.50	-.536	-.478	-.437	-.433	-.435	-.384	-.274
	10.00	-.539	-.439	-.386	-.419	-.396	-.346	-.223
	15.00	-.481	-.386	-.343	-.366	-.342	-.278	-.158
	20.00	-.422	-.340	-.309	-.327	-.292	-.239	-.084
	25.00	-.367	-.281	-.267	-.295	-.260	-.201	-.014
	30.00	-.322	-.260	-.257	-.267	-.229	-.161	-.001
	35.00	-.298	-.260	-.230	-.246	-.205	-.137	-.038
	40.00	-.273	-.260	-.231	-.230	-.191	-.120	-.065
	45.00	-.247	-.272	-.231	-.217	-.167	-.117	-.092
	50.00	-.244	-.351	-.322	-.239	-.196	-.125	-.124
	55.00	-.238	-.412	-.371	-.270	-.230	-.162	-.127
	60.00	-.204	-.393	-.373	-.350	-.298	-.213	-.144
	65.00	-.189			-.351	-.318	-.199	-.146
	70.00	-.097			-.497	-.654	-.385	-.156
	75.00	-.050	-.556	-.366	-.577	-.694	-.358	-.152
	80.00	-.032	-.787	-.392	-.646	-.728	-.329	-.152
	85.00	-.059	-.673	-.392	-.616	-.737	-.307	-.157
	90.00	-.110	-.406	-.349	-.501	-.718	-.291	-.170
	95.00	-.123	-.095	-.374	-.352	-.314	-.167	
GAP								
M = 0.85 $\alpha = 0.0^\circ$								
UPPER SURFACE	0.00	-.023	-.509	-.480	-.715	-.460	-.453	-.307
	1.25	-.189	-.093	-.147	-.223	-.311	-.426	-.458
	2.50	-.104	-.074	-.119	-.188	-.244	-.299	-.370
	5.00	-.035	-.074	-.123	-.166	-.213	-.251	-.284
	7.50	-.012	-.092	-.131	-.156	-.218	-.244	-.308
	10.00	-.007	-.101	-.142	-.170	-.224	-.238	-.303
	15.00	-.041	-.133	-.167	-.215	-.235	-.238	-.253
	20.00	-.092	-.151	-.190	-.216	-.237	-.260	-.248
	25.00	-.114	-.174	-.217	-.237	-.255	-.271	-.237
	30.00	-.128	-.191	-.229	-.249	-.267	-.276	-.210
	35.00	-.143	-.207	-.245	-.257	-.278	-.270	-.212
	40.00	-.151	-.225	-.269	-.272	-.291	-.264	-.202
	45.00	-.178	-.256	-.272	-.276	-.291	-.253	-.188
	50.00	-.195	-.285	-.287	-.271	-.276	-.233	-.164
	55.00	-.221	-.299	-.289	-.266	-.252	-.198	-.139
	60.00	-.239	-.318	-.284	-.303	-.278	-.162	-.120
	65.00	-.259	-.320	-.293	-.293	-.287	-.143	-.099
	70.00	-.272	-.332	-.293	-.293	-.287	-.137	-.096
	75.00	-.281	-.346	-.293	-.293	-.287	-.128	-.081
	80.00	-.293	-.353	-.293	-.293	-.287	-.127	-.081
	85.00	-.303	-.364	-.293	-.293	-.287	-.127	-.081
	90.00	-.314	-.374	-.293	-.293	-.287	-.127	-.081
	95.00	-.323	-.383	-.293	-.293	-.287	-.127	-.081
LOWER SURFACE	0.00							
	1.25	-.203	-.067	-.044	-.007	-.164	-.207	-.153
	2.50	-.130	-.054	-.035	-.011	-.064	-.073	-.106
	5.00	-.062	-.055	-.035	-.041	-.060	-.097	-.036
	7.50	-.024	-.066	-.088	-.043	-.030	-.066	-.001
	10.00	-.007	-.072	-.041	-.028	-.015	-.053	-.026
	15.00	-.027	-.082	-.043	-.010	-.012	-.021	-.051
	20.00	-.045	-.087	-.089	-.008	-.014	-.028	-.063
	25.00	-.067	-.084	-.017	-.007	-.028	-.032	-.047
	30.00	-.083	-.073	-.005	-.021	-.041	-.043	-.036
	35.00	-.083	-.046	-.024	-.048	-.059	-.063	-.039
	40.00	-.081	-.000	-.054	-.070	-.082	-.097	-.049
	45.00	-.076	-.072	-.168	-.114	-.132	-.150	-.058
	50.00	-.058	-.167	-.254	-.176	-.192	-.223	-.068
	55.00	-.010	-.271	-.292	-.255	-.270	-.282	-.055
	60.00	-.136	-.264	-.283	-.330	-.315	-.218	-.054
	65.00	-.333			-.315	-.304		-.070
	70.00	-.533			-.298	-.251	-.182	-.072
	75.00	-.998	-.684	-.364	-.298	-.249	-.174	-.085
	80.00	-.543	-.703	-.377	-.301	-.249	-.164	-.100
	85.00	-.378	-.617	-.388	-.308	-.249	-.154	-.109
	90.00	-.280	-.536	-.383	-.310	-.246	-.150	-.108
	95.00	-.191	-.195	-.305	-.333	-.319	-.217	
GAP								

TABLE IV

WING WITH LOWER SURFACE SPOILER (WITH GAP)

PERCENT CHORD		PRESSURE COEFFICIENT, C_p , AT							
		0.135b/2	0.25b/2	0.46b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2	
UPPER SURFACE	M = 0.85	$\alpha = 4.0^\circ$							
	.00	.027	.123	-	.017	.231	-.117	-.129	-.381
	1.25	.051	.917	-	1.124	-	1.245	-	1.253
	2.50	.156	.731	-	1.126	-	1.192	-	1.202
	5.00	.207	.481	-	.803	-	1.029	-	1.048
	7.50	.203	.405	-	.886	-	.899	-	.981
	10.00	.236	.379	-	.824	-	.805	-	.902
	15.00	.243	.362	-	.561	-	.643	-	.697
	20.00	.281	.355	-	.527	-	.558	-	.569
	25.00	.225	.358	-	.510	-	.508	-	.455
	30.00	.308	.365	-	.479	-	.468	-	.358
	35.00	.296	.376	-	.451	-	.447	-	.304
	40.00	.298	.384	-	.415	-	.432	-	.294
	45.00	.323	.414	-	.398	-	.418	-	.282
	50.00	.347	.432	-	.381	-	.388	-	.280
	55.00	.360	.436	-	.360	-	.362	-	.270
	60.00	.373	.475	-	.334	-	.324	-	.261
	65.00	.398	.522	-	.306	-	.277	-	.250
LOWER SURFACE	70.00	.424	.681	-	.270	-	.249	-	.253
	75.00	.476	.084	-	.296	-	.265	-	.249
	80.00	.282	.117	-	.258	-	.239	-	.241
	85.00	.238	.025	-	.246	-	.217	-	.243
	90.00	.184	-.098	-	.216	-	.161	-	.241
	95.00	.124	-.106	-	.211	-	.202	-	.222
	GAP								
	1.25	.341	.375	.404	.470	.478	.505	.411	
	2.50	.327	.299	.331	.345	.369	.371	.358	
	5.00	.282	.225	.258	.261	.320	.329	.261	
	7.50	.240	.179	.287	.205	.267	.275	.192	
	10.00	.215	.148	.190	.212	.233	.243	.139	
	15.00	.165	.114	.159	.179	.199	.186	.071	
	20.00	.133	.089	.137	.160	.171	.165	.003	
	25.00	.098	.073	.185	.146	.156	.144	.027	
	30.00	.073	.071	.117	.142	.150	.124	.038	
	35.00	.065	.081	.127	.147	.149	.117	.047	
	40.00	.055	.106	.143	.158	.162	.117	.054	
45.00	.047	.154	.171	.185	.187	.130	.069		
50.00	.059	.225	.224	.230	.228	.161	.080		
55.00	.091	.313	.295	.291	.287	.212	.089		
60.00	.211	.321	.331	.355	.324	.233	.080		
65.00	.420	.281	.326	.341	.320	.211	.085		
70.00	.650						.095		
75.00	.950	.619	.319	.263	.228	.160	.100		
80.00	.508	.676	.325	.268	.232	.152	.115		
85.00	.275	.636	.332	.276	.236	.147	.120		
90.00	.201	.562	.344	.284	.241	.147	.119		
95.00	.110	.452	.339	.263	.243	.151	.121		
GAP		.238	.333	.347	.327	.197			
M = 0.85		$\alpha = 6.0^\circ$							
UPPER SURFACE	.00	.013	.114	-.237	.015	.378	.324	.466	
	1.25	.184	1.148	1.304	1.331	1.397	1.100	.632	
	2.50	.308	1.114	1.201	1.241	1.380	1.040	.819	
	5.00	.351	.951	.951	1.048	1.309	1.029	.764	
	7.50	.338	.758	.821	.963	1.245	1.001	.761	
	10.00	.384	.570	.758	.903	1.195	.958	.736	
	15.00	.360	.465	.595	.824	1.055	.883	.679	
	20.00	.391	.448	.595	.760	.945	.826	.633	
	25.00	.406	.450	.576	.716	.878	.801	.581	
	30.00	.411	.455	.564	.641	.746	.736	.526	
	35.00	.388	.464	.561	.585	.637	.667	.464	
	40.00	.381	.475	.569	.513	.531	.642	.448	
	45.00	.409	.498	.573	.453	.456	.569	.431	
	50.00	.427	.516	.558	.408	.388	.543	.415	
	55.00	.446	.519	.472	.375	.341	.498	.396	
	60.00	.461	.545	.372	.347	.308	.447	.390	
	65.00	.469	.590	.328	.315	.269	.379	.367	
	LOWER SURFACE	70.00	.497	.692	.377	.277	.260	.356	.366
75.00		.563	.154	.210	.301	.246	.347	.351	
80.00		.378	.141	.218	.270	.227	.326	.328	
85.00		.306	.118	.226	.240	.209	.296	.327	
90.00		.221	.087	.226	.226	.203	.240	.309	
95.00		.140	.074	.217	.210	.196	.220	.286	
GAP									
1.25		.373	.464	.471	.463	.517	.524	.427	
2.50		.395	.395	.412	.410	.449	.416	.395	
5.00		.375	.313	.335	.334	.398	.387	.303	
7.50		.331	.262	.301	.285	.337	.336	.240	
10.00		.303	.232	.260	.281	.302	.298	.180	
15.00		.244	.185	.223	.239	.253	.235	.112	
20.00		.203	.154	.192	.214	.222	.206	.027	
25.00		.164	.131	.175	.194	.202	.180	.007	
30.00		.133	.123	.162	.188	.191	.155	.031	
35.00		.123	.128	.166	.182	.186	.144	.045	
40.00		.104	.145	.174	.189	.193	.138	.054	
45.00	.092	.183	.199	.210	.209	.145	.069		
50.00	.103	.243	.241	.246	.242	.166	.078		
55.00	.123	.326	.310	.303	.298	.210	.086		
60.00	.238	.342	.342	.371	.339	.238	.080		
65.00	.703	.300	.346	.359	.340	.220	.088		
70.00	.962						.106		
75.00	.982	.610	.318	.265	.233	.174	.111		
80.00	.492	.697	.384	.271	.238	.158	.131		
85.00	.234	.655	.331	.279	.250	.162	.139		
90.00	.170	.560	.340	.289	.257	.172	.145		
95.00	.090	.416	.330	.284	.255	.189	.163		
GAP		.256	.353	.364	.343	.202			

TABLE IV

WING WITH LOWER SURFACE SPOILER (WITH GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P, AT											
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2					
UPPER SURFACE	M = 0.85 $\alpha = 8.0^\circ$													
	0.00	.012	-	.384	-	.433	-	.188	-	.595	-	.402	-	.434
	1.25	.303	-	1.218	-	1.384	-	1.464	-	1.177	-	.731	-	.544
	2.50	.449	-	1.153	-	1.204	-	1.423	-	1.168	-	.680	-	.537
	5.00	.495	-	1.035	-	1.081	-	1.312	-	1.138	-	.699	-	.510
	7.50	.484	-	.969	-	1.024	-	1.234	-	1.110	-	.697	-	.522
	10.00	.528	-	.912	-	.985	-	1.181	-	1.091	-	.690	-	.518
	15.00	.486	-	.825	-	.931	-	1.096	-	1.039	-	.660	-	.516
	20.00	.495	-	.714	-	.860	-	1.027	-	1.002	-	.631	-	.493
	25.00	.495	-	.634	-	.825	-	.973	-	.966	-	.597	-	.464
	30.00	.491	-	.573	-	.771	-	.865	-	.908	-	.584	-	.458
	35.00	.454	-	.552	-	.739	-	.784	-	.861	-	.567	-	.438
	40.00	.460	-	.547	-	.714	-	.698	-	.806	-	.543	-	.433
	45.00	.484	-	.562	-	.677	-	.609	-	.753	-	.516	-	.407
	50.00	.504	-	.580	-	.614	-	.525	-	.702	-	.500	-	.390
	55.00	.519	-	.586	-	.508	-	.457	-	.655	-	.479	-	.375
	60.00	.532	-	.606	-	.415	-	.397	-	.601	-	.464	-	.354
65.00	.532	-	.644	-	.354	-	.334	-	.500	-	.446	-	.340	
70.00	.556	-	.688	-	.261	-	.284	-	.531	-	.433	-	.329	
75.00	.626	-	.735	-	.256	-	.300	-	.459	-	.385	-	.304	
80.00	.450	-	.146	-	.244	-	.267	-	.432	-	.384	-	.302	
85.00	.360	-	.104	-	.230	-	.229	-	.395	-	.373	-	.302	
90.00	.232	-	.066	-	.226	-	.216	-	.364	-	.342	-	.283	
95.00	.128	-	.024	-	.201	-	.203	-	.310	-	.342	-	.274	
LOWER SURFACE	1.25	.391	-	.528	-	.513	-	.465	-	.528	-	.523	-	.439
	2.50	.456	-	.476	-	.473	-	.450	-	.480	-	.447	-	.420
	5.00	.470	-	.399	-	.407	-	.396	-	.447	-	.424	-	.332
	7.50	.432	-	.349	-	.370	-	.360	-	.390	-	.374	-	.279
	10.00	.401	-	.312	-	.328	-	.342	-	.357	-	.339	-	.213
	15.00	.359	-	.263	-	.288	-	.296	-	.301	-	.271	-	.143
	20.00	.291	-	.229	-	.250	-	.266	-	.270	-	.239	-	.054
	25.00	.266	-	.202	-	.229	-	.244	-	.245	-	.205	-	.013
	30.00	.209	-	.187	-	.209	-	.227	-	.225	-	.178	-	.017
	35.00	.193	-	.186	-	.207	-	.222	-	.214	-	.156	-	.041
	40.00	.174	-	.196	-	.211	-	.221	-	.216	-	.143	-	.059
	45.00	.154	-	.225	-	.230	-	.235	-	.226	-	.144	-	.081
	50.00	.160	-	.273	-	.264	-	.265	-	.252	-	.160	-	.104
	55.00	.172	-	.348	-	.328	-	.317	-	.301	-	.199	-	.117
	60.00	.276	-	.373	-	.370	-	.388	-	.348	-	.292	-	.111
	65.00	.483	-	.330	-	.368	-	.377	-	.349	-	.216	-	.129
	70.00	.741	-	-	-	-	-	-	-	-	-	-	-	.148
75.00	.958	-	.593	-	.319	-	.275	-	.324	-	.269	-	.157	
80.00	.421	-	.720	-	.328	-	.292	-	.331	-	.257	-	.172	
85.00	.188	-	.664	-	.366	-	.308	-	.383	-	.273	-	.183	
90.00	.107	-	.538	-	.346	-	.317	-	.362	-	.296	-	.189	
95.00	.036	-	.384	-	.322	-	.297	-	.345	-	.322	-	.197	
GAP			.291	-	.373	-	.380	-	.350	-	.196	-		
M = 0.85 $\alpha = 12.1^\circ$														
UPPER SURFACE	0.00	.002	-	.752	-	.838	-	.515	-	.729	-	.456	-	.372
	1.25	.092	-	1.461	-	1.165	-	.843	-	.715	-	.487	-	.341
	2.50	.730	-	1.454	-	1.118	-	.835	-	.702	-	.458	-	.335
	5.00	.799	-	1.424	-	1.103	-	.826	-	.695	-	.475	-	.324
	7.50	.794	-	1.419	-	1.083	-	.832	-	.686	-	.473	-	.321
	10.00	.828	-	1.390	-	1.063	-	.835	-	.681	-	.452	-	.317
	15.00	.783	-	1.375	-	1.039	-	.825	-	.663	-	.434	-	.303
	20.00	.752	-	1.355	-	1.029	-	.814	-	.646	-	.420	-	.281
	25.00	.726	-	1.357	-	1.026	-	.799	-	.627	-	.412	-	.280
	30.00	.693	-	1.358	-	.980	-	.774	-	.620	-	.408	-	.263
	35.00	.644	-	1.357	-	.945	-	.745	-	.595	-	.401	-	.276
	40.00	.627	-	1.356	-	.918	-	.728	-	.587	-	.391	-	.260
	45.00	.636	-	1.309	-	.889	-	.710	-	.576	-	.391	-	.258
	50.00	.635	-	1.209	-	.850	-	.690	-	.564	-	.386	-	.250
	55.00	.619	-	1.151	-	.825	-	.665	-	.554	-	.383	-	.239
	60.00	.581	-	1.097	-	.767	-	.630	-	.521	-	.388	-	.267
	65.00	.510	-	.997	-	.737	-	.620	-	.529	-	.385	-	.265
70.00	.497	-	.906	-	.709	-	.585	-	.513	-	.360	-	.269	
75.00	.557	-	.898	-	.675	-	.603	-	.501	-	.338	-	.273	
80.00	.444	-	.847	-	.681	-	.573	-	.486	-	.340	-	.269	
85.00	.425	-	.847	-	.558	-	.563	-	.457	-	.325	-	.270	
90.00	.336	-	.847	-	.470	-	.554	-		-		-		
95.00	.233	-	.098	-		-		-		-		-		
LOWER SURFACE	1.25	.351	-	.574	-	.526	-	.495	-	.509	-	.504	-	.441
	2.50	.507	-	.576	-	.503	-	.499	-	.512	-	.464	-	.442
	5.00	.620	-	.523	-	.503	-	.471	-	.484	-	.453	-	.371
	7.50	.595	-	.479	-	.477	-	.429	-	.435	-	.408	-	.314
	10.00	.571	-	.438	-	.432	-	.417	-	.401	-	.371	-	.259
	15.00	.438	-	.388	-	.383	-	.364	-	.343	-	.302	-	.190
	20.00	.434	-	.343	-	.340	-	.325	-	.297	-	.266	-	.104
	25.00	.379	-	.309	-	.310	-	.294	-	.265	-	.230	-	.056
	30.00	.332	-	.288	-	.281	-	.266	-	.237	-	.192	-	.002
	35.00	.313	-	.277	-	.265	-	.246	-	.214	-	.166	-	.008
	40.00	.282	-	.275	-	.254	-	.232	-	.204	-	.148	-	
	45.00	.257	-	.289	-	.259	-	.233	-	.203	-	.142	-	.077
	50.00	.253	-	.384	-	.276	-	.249	-	.221	-	.154	-	.119
	55.00	.284	-	.380	-	.284	-	.289	-	.281	-	.194	-	.162
	60.00	.339	-	.412	-	.374	-	.373	-	.358	-	.242	-	.188
	65.00	.534	-	.374	-	.378	-	.369	-	.342	-	.221	-	.191
	70.00	.777	-	-	-	-	-	-	-	-	-	.390	-	.191
75.00	1.103	-	.702	-	.684	-	.613	-	.636	-	.381	-	.183	
80.00	.281	-	.028	-	.634	-	.708	-	.638	-	.368	-	.187	
85.00	.218	-	.357	-	.603	-	.693	-	.685	-	.350	-	.201	
90.00	.029	-	.088	-	.606	-	.627	-	.603	-	.332	-	.228	
95.00	.069	-	.353	-	.351	-	.371	-	.342	-	.192	-		

TABLE IV

WING WITH LOWER SURFACE SPOILER (WITH GAP)

		PERCENT CHORD	PRESSURE COEFFICIENT, P , AT										
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2				
UPPER SURFACE	$M = 0.90 \quad \alpha = 0.1^\circ$												
	.00	.034	.526	-	.486	-	.730	-	.464	-	.460	-	.320
	1.25	.195	.075	-	.128	-	.402	-	.323	-	.422	-	.480
	2.50	.132	.053	-	.103	-	.175	-	.254	-	.314	-	.389
	5.00	.064	.054	-	.104	-	.117	-	.225	-	.257	-	.291
	7.50	.038	.070	-	.121	-	.150	-	.232	-	.260	-	.326
	10.00	.017	.086	-	.131	-	.167	-	.239	-	.260	-	.330
	15.00	.022	.117	-	.158	-	.217	-	.222	-	.254	-	.350
	20.00	.075	.132	-	.184	-	.232	-	.242	-	.279	-	.303
	25.00	.093	.155	-	.216	-	.261	-	.265	-	.303	-	.258
	30.00	.111	.174	-	.230	-	.260	-	.268	-	.310	-	.239
	35.00	.122	.194	-	.255	-	.263	-	.267	-	.323	-	.232
	40.00	.134	.212	-	.284	-	.269	-	.268	-	.332	-	.233
	45.00	.143	.245	-	.319	-	.264	-	.264	-	.337	-	.227
	50.00	.163	.271	-	.354	-	.276	-	.267	-	.306	-	.215
	55.00	.208	.287	-	.334	-	.288	-	.267	-	.259	-	.201
	60.00	.226	.328	-	.257	-	.277	-	.263	-	.239	-	.175
65.00	.244	.398	-	.244	-	.284	-	.277	-	.210	-	.148	
70.00	.295	.756	-	.520	-	.315	-	.283	-	.205	-	.145	
75.00	.378	.046	-	.085	-	.214	-	.236	-	.182	-	.124	
80.00	.206	.080	-	.165	-	.230	-	.222	-	.152	-	.106	
85.00	.202	.095	-	.183	-	.212	-	.208	-	.151	-	.106	
90.00	.166	.108	-	.220	-	.218	-	.207	-	.137	-	.089	
95.00	.162	.142	-	.230	-	.221	-	.199	-	.136	-	.090	
LOWER SURFACE	1.25	.230	.046	-	.046	-	.009	-	.163	-	.212	-	.157
	2.50	.160	.033	-	.034	-	.012	-	.065	-	.070	-	.110
	5.00	.090	.037	-	.035	-	.038	-	.058	-	.098	-	.039
	7.50	.051	.051	-	.029	-	.042	-	.030	-	.065	-	.004
	10.00	.029	.058	-	.038	-	.020	-	.014	-	.049	-	.030
	15.00	.006	.072	-	.038	-	.008	-	.013	-	.018	-	.069
	20.00	.030	.078	-	.024	-	.003	-	.013	-	.024	-	.081
	25.00	.053	.081	-	.021	-	.009	-	.029	-	.025	-	.060
	30.00	.070	.069	-	.000	-	.023	-	.041	-	.025	-	.047
	35.00	.069	.036	-	.025	-	.046	-	.058	-	.037	-	.042
	40.00	.074	.016	-	.057	-	.074	-	.089	-	.058	-	.042
	45.00	.071	.097	-	.104	-	.120	-	.130	-	.094	-	.054
	50.00	.051	.194	-	.172	-	.181	-	.193	-	.148	-	.082
	55.00	.025	.288	-	.257	-	.263	-	.273	-	.218	-	.082
	60.00	.143	.271	-	.293	-	.337	-	.315	-	.242	-	.066
	65.00	.355	.236	-	.289	-	.324	-	.304	-	.213	-	.066
	70.00	.552	-	-	-	-	-	-	-	-	-	-	.075
75.00	.498	.688	-	.377	-	.304	-	.261	-	.193	-	.082	
80.00	.512	.705	-	.384	-	.305	-	.262	-	.184	-	.096	
85.00	.378	.678	-	.391	-	.305	-	.259	-	.174	-	.110	
90.00	.307	.622	-	.401	-	.309	-	.262	-	.168	-	.115	
95.00	.195	.539	-	.395	-	.314	-	.260	-	.164	-	.115	
GAP			.193		.312		.341		.318		.210		
$M = 0.90 \quad \alpha = 4.0^\circ$													
UPPER SURFACE	.00	.028	.214	-	.090	-	.341	-	.004	-	.022	-	.260
	1.25	.002	.806	-	.981	-	1.151	-	1.221	-	1.281	-	1.299
	2.50	.102	.708	-	.910	-	1.121	-	1.209	-	1.217	-	1.280
	5.00	.163	.403	-	.866	-	.954	-	1.077	-	1.189	-	1.162
	7.50	.166	.351	-	.475	-	.665	-	1.016	-	1.100	-	1.146
	10.00	.200	.320	-	.441	-	.494	-	.961	-	1.048	-	1.089
	15.00	.210	.338	-	.418	-	.481	-	.855	-	.980	-	.918
	20.00	.258	.327	-	.414	-	.489	-	.826	-	.849	-	.648
	25.00	.274	.327	-	.424	-	.495	-	.800	-	.857	-	.502
	30.00	.277	.331	-	.431	-	.490	-	.800	-	.857	-	.336
	35.00	.273	.349	-	.444	-	.511	-	.802	-	.879	-	.223
	40.00	.283	.365	-	.455	-	.510	-	.804	-	.890	-	.196
	45.00	.298	.391	-	.483	-	.558	-	.814	-	.895	-	.184
	50.00	.324	.414	-	.505	-	.583	-	.851	-	.881	-	.215
	55.00	.344	.422	-	.525	-	.584	-	.801	-	.858	-	.235
	60.00	.363	.447	-	.554	-	.451	-	.333	-	.169	-	.235
	65.00	.375	.502	-	.517	-	.328	-	.288	-	.177	-	.230
70.00	.403	.739	-	.375	-	.278	-	.248	-	.190	-	.232	
75.00	.514	.325	-	.292	-	.311	-	.253	-	.163	-	.235	
80.00	.353	.338	-	.256	-	.273	-	.211	-	.147	-	.227	
85.00	.178	.224	-	.243	-	.241	-	.193	-	.153	-	.222	
90.00	.166	.150	-	.249	-	.229	-	.183	-	.157	-	.218	
95.00	.266	.139	-	.247	-	.219	-	.182	-	.154	-	.205	
LOWER SURFACE	1.25	.385	.370	-	.388	-	.459	-	.466	-	.490	-	.405
	2.50	.340	.295	-	.316	-	.328	-	.369	-	.355	-	.348
	5.00	.294	.224	-	.243	-	.236	-	.304	-	.315	-	.252
	7.50	.246	.181	-	.213	-	.184	-	.253	-	.262	-	.188
	10.00	.220	.147	-	.180	-	.194	-	.223	-	.231	-	.131
	15.00	.170	.115	-	.147	-	.163	-	.189	-	.176	-	.059
	20.00	.135	.088	-	.128	-	.145	-	.162	-	.156	-	.034
	25.00	.103	.073	-	.117	-	.114	-	.147	-	.135	-	.073
	30.00	.075	.071	-	.110	-	.109	-	.143	-	.116	-	.069
	35.00	.066	.082	-	.117	-	.136	-	.144	-	.109	-	.096
	40.00	.055	.112	-	.133	-	.150	-	.160	-	.109	-	.092
	45.00	.046	.165	-	.163	-	.179	-	.184	-	.121	-	.091
	50.00	.039	.239	-	.212	-	.227	-	.229	-	.153	-	.092
	55.00	.034	.325	-	.287	-	.291	-	.288	-	.201	-	.099
	60.00	.225	.317	-	.383	-	.353	-	.383	-	.220	-	.092
	65.00	.433	.263	-	.380	-	.337	-	.318	-	.193	-	.097
	70.00	.667	-	-	-	-	-	-	-	-	-	-	.109
75.00	.988	.657	-	.382	-	.276	-	.236	-	.173	-	.116	
80.00	.471	.695	-	.384	-	.279	-	.236	-	.168	-	.127	
85.00	.295	.663	-	.388	-	.284	-	.244	-	.168	-	.127	
90.00	.199	.578	-	.389	-	.290	-	.260	-	.168	-	.124	
95.00	.145	.463	-	.385	-	.287	-	.253	-	.168	-	.126	
GAP			.218		.332		.346		.324		.179		

TABLE IV

WING WITH LOWER SURFACE SPOILER (WITH GAP)

PERCENT CHORD	PRESSURE COEFFICIENT, P , AT						
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
$M = 0.90 \quad \alpha = 6.00^\circ$							
UPPER SURFACE	1.00	.0241	.0114	.103	.142	.214	.470
	1.25	.104	.062	.197	.352	.391	.660
	2.50	.232	.106	.373	.535	.585	.985
	5.00	.481	.241	.653	.841	.894	1.201
	7.50	.681	.381	.888	1.069	1.160	1.605
	10.00	.829	.496	1.093	1.266	1.304	1.796
	15.00	.918	.543	1.242	1.458	1.472	1.926
	20.00	.943	.566	1.331	1.555	1.555	2.038
	25.00	.957	.575	1.391	1.608	1.608	2.119
	30.00	.960	.579	1.431	1.633	1.633	2.178
	35.00	.961	.580	1.451	1.644	1.644	2.218
	40.00	.961	.580	1.451	1.644	1.644	2.218
	45.00	.961	.580	1.451	1.644	1.644	2.218
	50.00	.961	.580	1.451	1.644	1.644	2.218
	55.00	.961	.580	1.451	1.644	1.644	2.218
	60.00	.961	.580	1.451	1.644	1.644	2.218
	65.00	.961	.580	1.451	1.644	1.644	2.218
	70.00	.961	.580	1.451	1.644	1.644	2.218
	75.00	.961	.580	1.451	1.644	1.644	2.218
	80.00	.961	.580	1.451	1.644	1.644	2.218
	85.00	.961	.580	1.451	1.644	1.644	2.218
	90.00	.961	.580	1.451	1.644	1.644	2.218
	95.00	.961	.580	1.451	1.644	1.644	2.218
LOWER SURFACE	1.00	.393	.470	.467	.471	.506	.426
	1.25	.416	.470	.467	.471	.506	.426
	2.50	.416	.470	.467	.471	.506	.426
	5.00	.416	.470	.467	.471	.506	.426
	7.50	.416	.470	.467	.471	.506	.426
	10.00	.416	.470	.467	.471	.506	.426
	15.00	.416	.470	.467	.471	.506	.426
	20.00	.416	.470	.467	.471	.506	.426
	25.00	.416	.470	.467	.471	.506	.426
	30.00	.416	.470	.467	.471	.506	.426
	35.00	.416	.470	.467	.471	.506	.426
	40.00	.416	.470	.467	.471	.506	.426
	45.00	.416	.470	.467	.471	.506	.426
	50.00	.416	.470	.467	.471	.506	.426
	55.00	.416	.470	.467	.471	.506	.426
	60.00	.416	.470	.467	.471	.506	.426
	65.00	.416	.470	.467	.471	.506	.426
	70.00	.416	.470	.467	.471	.506	.426
	75.00	.416	.470	.467	.471	.506	.426
	80.00	.416	.470	.467	.471	.506	.426
	85.00	.416	.470	.467	.471	.506	.426
	90.00	.416	.470	.467	.471	.506	.426
	95.00	.416	.470	.467	.471	.506	.426
$M = 0.90 \quad \alpha = 8.00^\circ$							
UPPER SURFACE	1.00	.009	.177	.277	.038	.425	.352
	1.25	.012	.181	.281	.041	.428	.355
	2.50	.015	.185	.285	.044	.431	.358
	5.00	.018	.188	.288	.047	.434	.361
	7.50	.021	.191	.291	.050	.437	.364
	10.00	.024	.194	.294	.053	.440	.367
	15.00	.027	.197	.297	.056	.443	.370
	20.00	.030	.200	.300	.059	.446	.373
	25.00	.033	.203	.303	.062	.449	.376
	30.00	.036	.206	.306	.065	.452	.379
	35.00	.039	.209	.309	.068	.455	.382
	40.00	.042	.212	.312	.071	.458	.385
	45.00	.045	.215	.315	.074	.461	.388
	50.00	.048	.218	.318	.077	.464	.391
	55.00	.051	.221	.321	.080	.467	.394
	60.00	.054	.224	.324	.083	.470	.397
	65.00	.057	.227	.327	.086	.473	.400
	70.00	.060	.230	.330	.089	.476	.403
	75.00	.063	.233	.333	.092	.479	.406
	80.00	.066	.236	.336	.095	.482	.409
	85.00	.069	.239	.339	.098	.485	.412
	90.00	.072	.242	.342	.101	.488	.415
	95.00	.075	.245	.345	.104	.491	.418
LOWER SURFACE	1.00	.406	.527	.511	.476	.523	.435
	1.25	.406	.527	.511	.476	.523	.435
	2.50	.406	.527	.511	.476	.523	.435
	5.00	.406	.527	.511	.476	.523	.435
	7.50	.406	.527	.511	.476	.523	.435
	10.00	.406	.527	.511	.476	.523	.435
	15.00	.406	.527	.511	.476	.523	.435
	20.00	.406	.527	.511	.476	.523	.435
	25.00	.406	.527	.511	.476	.523	.435
	30.00	.406	.527	.511	.476	.523	.435
	35.00	.406	.527	.511	.476	.523	.435
	40.00	.406	.527	.511	.476	.523	.435
	45.00	.406	.527	.511	.476	.523	.435
	50.00	.406	.527	.511	.476	.523	.435
	55.00	.406	.527	.511	.476	.523	.435
	60.00	.406	.527	.511	.476	.523	.435
	65.00	.406	.527	.511	.476	.523	.435
	70.00	.406	.527	.511	.476	.523	.435
	75.00	.406	.527	.511	.476	.523	.435
	80.00	.406	.527	.511	.476	.523	.435
	85.00	.406	.527	.511	.476	.523	.435
	90.00	.406	.527	.511	.476	.523	.435
	95.00	.406	.527	.511	.476	.523	.435

TABLE IV

WING WITH LOWER SURFACE SPOILER (WITH GAP)

TABLE IV															
WING WITH LOWER SURFACE SPOILER (WITH CAT)															
PRESSURE COEFFICIENT, P, AT:															
PERCENT CHORD	0.135b/2		0.25b/2		0.40b/2		0.55b/2		0.70b/2		0.85b/2		0.95b/2		
M = 0.90 $\alpha = 12.1^\circ$															
UPPER SURFACE	0.00	-	.001	-	.578	-	.660	-	.452	-	.673	-	.482	-	.423
	1.25	-	.426	-	.354	-	.233	-	.865	-	.700	-	.524	-	.384
	2.50	-	.611	-	.364	-	.164	-	.872	-	.694	-	.493	-	.381
	5.00	-	.690	-	.352	-	.150	-	.881	-	.682	-	.507	-	.359
	7.50	-	.693	-	.345	-	.137	-	.892	-	.674	-	.505	-	.372
	10.00	-	.732	-	.318	-	.130	-	.902	-	.674	-	.504	-	.366
	15.00	-	.701	-	.287	-	.102	-	.853	-	.654	-	.473	-	.350
	20.00	-	.684	-	.252	-	.094	-	.835	-	.633	-	.462	-	.336
	25.00	-	.661	-	.241	-	.067	-	.809	-	.631	-	.453	-	.323
	30.00	-	.639	-	.213	-	.041	-	.782	-	.622	-	.445	-	.322
	35.00	-	.595	-	.111	-	.993	-	.751	-	.613	-	.439	-	.311
	40.00	-	.579	-	.489	-	.941	-	.735	-	.602	-	.431	-	.309
	45.00	-	.591	-	.557	-	.901	-	.716	-	.580	-	.428	-	.305
	50.00	-	.609	-	.597	-	.859	-	.701	-	.570	-	.422	-	.304
LOWER SURFACE	1.25	-	.378	-	.594	-	.546	-	.498	-	.515	-	.498	-	.435
	2.50	-	.534	-	.585	-	.545	-	.497	-	.511	-	.457	-	.442
	5.00	-	.615	-	.481	-	.477	-	.467	-	.485	-	.450	-	.374
	7.50	-	.585	-	.443	-	.435	-	.428	-	.435	-	.404	-	.323
	10.00	-	.507	-	.390	-	.386	-	.417	-	.401	-	.369	-	.262
	15.00	-	.444	-	.348	-	.346	-	.434	-	.344	-	.302	-	.190
	20.00	-	.387	-	.316	-	.314	-	.429	-	.306	-	.266	-	.097
	25.00	-	.342	-	.284	-	.286	-	.428	-	.270	-	.227	-	.043
	30.00	-	.320	-	.283	-	.271	-	.436	-	.243	-	.191	-	.005
	35.00	-	.293	-	.264	-	.263	-	.438	-	.221	-	.167	-	.027
	40.00	-	.265	-	.301	-	.266	-	.436	-	.210	-	.148	-	.062
	45.00	-	.261	-	.334	-	.266	-	.437	-	.211	-	.150	-	.163
	50.00	-	.261	-	.393	-	.286	-	.437	-	.227	-	.150	-	.212
	55.00	-	.355	-	.419	-	.384	-	.437	-	.333	-	.229	-	.216
60.00	-	.354	-	.374	-	.384	-	.371	-	.344	-	.216	-	.224	
65.00	-	.798	-	-	-	.629	-	-	-	.639	-	.429	-	.215	
70.00	-	.798	-	.727	-	.644	-	.627	-	.639	-	.418	-	.208	
75.00	-	.377	-	.056	-	.644	-	.693	-	.633	-	.403	-	.215	
80.00	-	.243	-	.083	-	.613	-	.719	-	.625	-	.382	-	.230	
85.00	-	.046	-	.320	-	.553	-	.703	-	.611	-	.361	-	.253	
90.00	-	.059	-	.059	-	.553	-	.652	-	.585	-	.361	-	-	
95.00	-	.053	-	.333	-	.386	-	.374	-	.344	-	.184	-	-	
GAP															
M = 0.94 $\alpha = 0.0^\circ$															
UPPER SURFACE	0.00	-	.023	-	.533	-	.496	-	.739	-	.476	-	.468	-	.335
	1.25	-	.205	-	.072	-	.108	-	.181	-	.299	-	.332	-	.532
	2.50	-	.147	-	.049	-	.085	-	.155	-	.233	-	.328	-	.428
	5.00	-	.073	-	.051	-	.085	-	.136	-	.226	-	.289	-	.322
	7.50	-	.046	-	.067	-	.118	-	.167	-	.240	-	.291	-	.352
	10.00	-	.028	-	.084	-	.147	-	.205	-	.256	-	.300	-	.358
	15.00	-	.013	-	.111	-	.172	-	.234	-	.288	-	.305	-	.369
	20.00	-	.066	-	.128	-	.195	-	.268	-	.321	-	.341	-	.382
	25.00	-	.088	-	.150	-	.220	-	.302	-	.357	-	.368	-	.354
	30.00	-	.107	-	.171	-	.248	-	.332	-	.385	-	.377	-	.305
	35.00	-	.117	-	.191	-	.271	-	.352	-	.394	-	.392	-	.287
	40.00	-	.127	-	.209	-	.299	-	.376	-	.437	-	.385	-	.258
	45.00	-	.158	-	.266	-	.338	-	.441	-	.432	-	.378	-	.232
	50.00	-	.184	-	.279	-	.370	-	.469	-	.437	-	.383	-	.197
55.00	-	.205	-	.309	-	.411	-	.523	-	.432	-	.401	-	.167	
60.00	-	.223	-	.370	-	.425	-	.547	-	.432	-	.344	-	.148	
65.00	-	.239	-	.434	-	.428	-	.577	-	.432	-	.239	-	.144	
70.00	-	.277	-	.468	-	.477	-	.602	-	.432	-	.223	-	.129	
75.00	-	.339	-	.486	-	.501	-	.623	-	.432	-	.190	-	.112	
80.00	-	.238	-	.486	-	.501	-	.623	-	.432	-	.140	-	.112	
85.00	-	.267	-	.486	-	.501	-	.623	-	.432	-	.140	-	.096	
90.00	-	.258	-	.486	-	.501	-	.623	-	.432	-	.143	-	.102	
95.00	-	.281	-	.486	-	.501	-	.623	-	.432	-	.143	-	-	
GAP															
LOWER SURFACE	1.25	-	.240	-	.040	-	.053	-	.015	-	.129	-	.191	-	.160
	2.50	-	.174	-	.033	-	.037	-	.006	-	.036	-	.047	-	.113
	5.00	-	.101	-	.037	-	.038	-	.006	-	.031	-	.083	-	.043
	7.50	-	.060	-	.053	-	.031	-	.006	-	.003	-	.051	-	.002
	10.00	-	.039	-	.064	-	.041	-	.006	-	.016	-	.036	-	.030
	15.00	-	.001	-	.079	-	.021	-	.006	-	.013	-	.003	-	.096
	20.00	-	.028	-	.090	-	.010	-	.006	-	.013	-	.013	-	.138
	25.00	-	.054	-	.096	-	.000	-	.006	-	.003	-	.017	-	.092
	30.00	-	.074	-	.081	-	.028	-	.006	-	.003	-	.016	-	.062
	35.00	-	.077	-	.044	-	.081	-	.006	-	.003	-	.006	-	.056
	40.00	-	.085	-	.106	-	.096	-	.006	-	.003	-	.006	-	.073
	45.00	-	.084	-	.209	-	.168	-	.006	-	.003	-	.006	-	.094
	50.00	-	.063	-	.297	-	.244	-	.006	-	.003	-	.006	-	.108
	55.00	-	.021	-	.268	-	.280	-	.006	-	.003	-	.006	-	.088
60.00	-	.174	-	.216	-	.280	-	.006	-	.003	-	.006	-	.084	
65.00	-	.360	-	-	-	.280	-	.006	-	.003	-	.006	-	.093	
70.00	-	.359	-	-	-	.280	-	.006	-	.003	-	.006	-	.101	
75.00	-	.040	-	.759	-	.444	-	.330	-	.286	-	.209	-	.118	
80.00	-	.441	-	.775	-	.433	-	.329	-	.283	-	.204	-	.131	
85.00	-	.368	-	.728	-	.433	-	.330	-	.283	-	.196	-	.134	
90.00	-	.368	-	.672	-	.433	-	.330	-	.283	-	.196	-	.134	
95.00	-	.278	-	.599	-	.433	-	.330	-	.283	-	.196	-	.134	
GAP															

TABLE IV

WING WITH LOWER SURFACE SPOILER (WITH GAP)

	PERCENT CHORD	PRESSURE COEFFICIENT, P, AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M = 0.94 α = 4.0°							
	0.00	.024	.273	.171	.430	.115	.093	.151
	1.25	.043	.716	.095	1.013	1.061	1.102	1.130
	2.50	.061	.646	.046	.997	1.059	1.062	1.123
	3.75	.123	.366	.593	.867	.936	1.036	1.016
	5.00	.134	.331	.408	.620	.889	.961	1.021
	7.50	.169	.303	.386	.435	.654	.929	.968
	10.00	.183	.293	.372	.431	.559	.882	.923
	12.50	.224	.298	.372	.447	.474	.790	.828
	15.00	.244	.303	.382	.457	.509	.592	.826
	17.50	.254	.305	.391	.451	.520	.553	.646
	20.00	.241	.323	.404	.472	.541	.580	.509
	22.50	.249	.336	.421	.495	.556	.605	.523
	25.00	.273	.365	.448	.521	.589	.638	.526
	27.50	.295	.389	.473	.558	.615	.673	.508
	30.00	.315	.397	.486	.568	.646	.715	.516
	32.50	.333	.412	.506	.578	.641	.574	.310
	35.00	.345	.451	.534	.507	.450	.280	.156
	37.50	.369	.4814	.454	.331	.321	.220	.145
	40.00	.360	.306	.431	.529	.605	.255	.153
	42.50	.331	.342	.388	.451	.489	.184	.155
	45.00	.355	.308	.330	.406	.401	.171	.162
47.50	.358	.237	.346	.383	.347	.137	.165	
50.00	.341	.173	.356	.346	.267	.142	.173	
LOWER SURFACE	1.25	.366	.361	.377	.442	.444	.468	.390
	2.50	.350	.290	.306	.304	.338	.323	.329
	3.75	.302	.221	.237	.218	.276	.265	.236
	5.00	.258	.173	.206	.171	.224	.237	.166
	7.50	.231	.145	.176	.179	.193	.204	.120
	10.00	.179	.107	.144	.144	.160	.148	.048
	12.50	.144	.083	.125	.127	.135	.134	.052
	15.00	.106	.066	.116	.111	.124	.116	.098
	17.50	.077	.065	.105	.107	.119	.101	.117
	20.00	.069	.080	.112	.111	.125	.094	.123
	22.50	.054	.116	.125	.125	.141	.110	.142
	25.00	.043	.174	.154	.155	.171	.142	.163
	27.50	.060	.251	.201	.208	.218	.192	.172
	30.00	.098	.334	.278	.278	.280	.208	.155
	32.50	.239	.315	.309	.339	.312	.208	.149
	35.00	.445	.850	.305	.322	.305	.183	.148
	37.50	.680						.143
	40.00	1.057	.763	.438	.332	.278	.184	.147
	42.50	.450	.817	.434	.335	.276	.182	.147
	45.00	.396	.755	.434	.336	.278	.184	.147
	47.50	.317	.673	.438	.337	.278	.189	.148
	50.00	.240	.554	.428	.331	.268	.192	.147
GAP		.204	.315	.330	.316	.171		
UPPER SURFACE	M = 0.94 α = 6.0°							
	0.00	.019	.093	.006	.247	.075	.075	.343
	1.25	.051	.094	.1147	1.250	1.281	1.306	1.326
	2.50	.177	.042	1.147	1.231	1.277	1.214	1.311
	3.75	.339	.775	1.076	1.155	1.190	1.236	1.185
	5.00	.250	.576	.971	1.096	1.152	1.192	1.218
	7.50	.289	.479	.757	1.051	1.122	1.154	1.194
	10.00	.290	.416	.506	.925	1.067	1.105	1.093
	12.50	.310	.397	.461	.610	1.031	1.058	1.072
	15.00	.332	.390	.456	.535	.977	1.055	1.046
	17.50	.340	.387	.465	.516	.742	1.026	1.044
	20.00	.318	.398	.474	.537	.629	1.014	.934
	22.50	.319	.405	.495	.558	.607	.974	.856
	25.00	.339	.430	.514	.584	.628	.966	.758
	27.50	.361	.453	.540	.612	.653	.965	.701
	30.00	.373	.488	.566	.633	.679	.936	.768
	32.50	.395	.476	.580	.633	.679	.936	.742
	35.00	.412	.507	.580	.633	.679	.936	.664
	37.50	.419	.518	.480	.553	.472	.441	.550
	40.00	.457	.394	.510	.627	.710	.389	.438
	42.50	.389	.411	.421	.521	.535	.291	.349
	45.00	.408	.339	.380	.445	.437	.225	.313
47.50	.418	.255	.382	.405	.375	.147	.278	
50.00	.389	.126	.368	.357	.282	.133	.264	
LOWER SURFACE	1.25	.406	.470	.469	.493	.490	.507	.424
	2.50	.428	.399	.402	.379	.411	.390	.383
	3.75	.397	.380	.387	.304	.346	.322	.293
	5.00	.358	.264	.293	.251	.293	.299	.238
	7.50	.328	.236	.253	.254	.258	.266	.173
	10.00	.270	.192	.218	.217	.207	.207	.101
	12.50	.228	.188	.190	.187	.187	.184	.003
	15.00	.186	.136	.156	.155	.170	.164	.058
	17.50	.150	.138	.157	.154	.160	.131	.107
	20.00	.138	.162	.163	.162	.171	.128	.120
	22.50	.104	.208	.186	.186	.192	.135	.137
	25.00	.116	.274	.285	.289	.235	.162	.156
	27.50	.142	.352	.296	.291	.294	.208	.168
	30.00	.269	.334	.334	.358	.331	.250	.154
	32.50	.482	.273	.332	.344	.331	.204	.156
	35.00	.747						.164
	37.50	1.074	.780	.446	.339	.280	.177	.168
	40.00	.466	.777	.445	.342	.277	.165	.168
	42.50	.401	.797	.449	.344	.277	.170	.166
	45.00	.324	.690	.481	.343	.271	.176	.166
	47.50	.221	.511	.438	.338	.261	.181	.168
	50.00	.228	.342	.342	.352	.336	.189	
GAP								

TABLE IV

WING WITH LOWER SURFACE SPOILER (WITH GAP)

PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:						
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
M = 0.94 $\alpha = 8.0^\circ$							
UPPER SURFACE	0.00	.015	-.066	-.164	-.079	-.270	-.244
	1.25	.139	-.1086	-.1163	-.1324	-.1103	-.468
	2.50	.279	-.1109	-.1138	-.1308	-.1090	-.725
	5.00	.338	-.0976	-.1058	-.1249	-.1064	-.713
	7.50	.348	-.0866	-.1035	-.1210	-.1045	-.660
	10.00	.381	-.0799	-.1026	-.1171	-.1035	-.673
	15.00	.394	-.0693	-.0970	-.1075	-.1008	-.671
	20.00	.394	-.0567	-.0900	-.1000	-.983	-.706
	25.00	.405	-.0477	-.0835	-.872	-.947	-.657
	30.00	.407	-.0429	-.0766	-.738	-.881	-.604
	35.00	.378	-.0428	-.0699	-.698	-.829	-.616
	40.00	.378	-.0439	-.0598	-.654	-.783	-.625
	45.00	.396	-.0465	-.0600	-.618	-.743	-.571
	50.00	.414	-.0493	-.0613	-.605	-.702	-.516
	55.00	.433	-.0498	-.0618	-.599	-.667	-.490
	60.00	.446	-.0514	-.0617	-.578	-.623	-.447
	65.00	.455	-.0534	-.0589	-.503	-.548	-.449
	70.00	.469	-.0506	-.0502	-.373	-.477	-.422
LOWER SURFACE	0.00	.015	-.066	-.164	-.079	-.270	-.244
	1.25	.139	-.1086	-.1163	-.1324	-.1103	-.468
	2.50	.279	-.1109	-.1138	-.1308	-.1090	-.725
	5.00	.338	-.0976	-.1058	-.1249	-.1064	-.713
	7.50	.348	-.0866	-.1035	-.1210	-.1045	-.660
	10.00	.381	-.0799	-.1026	-.1171	-.1035	-.673
	15.00	.394	-.0693	-.0970	-.1075	-.1008	-.671
	20.00	.394	-.0567	-.0900	-.1000	-.983	-.706
	25.00	.405	-.0477	-.0835	-.872	-.947	-.657
	30.00	.407	-.0429	-.0766	-.738	-.881	-.604
	35.00	.378	-.0428	-.0699	-.698	-.829	-.616
	40.00	.378	-.0439	-.0598	-.654	-.783	-.625
	45.00	.396	-.0465	-.0600	-.618	-.743	-.571
	50.00	.414	-.0493	-.0613	-.605	-.702	-.516
	55.00	.433	-.0498	-.0618	-.599	-.667	-.490
	60.00	.446	-.0514	-.0617	-.578	-.623	-.447
	65.00	.455	-.0534	-.0589	-.503	-.548	-.449
	70.00	.469	-.0506	-.0502	-.373	-.477	-.422
UPPER SURFACE	0.00	.015	-.066	-.164	-.079	-.270	-.244
	1.25	.139	-.1086	-.1163	-.1324	-.1103	-.468
	2.50	.279	-.1109	-.1138	-.1308	-.1090	-.725
	5.00	.338	-.0976	-.1058	-.1249	-.1064	-.713
	7.50	.348	-.0866	-.1035	-.1210	-.1045	-.660
	10.00	.381	-.0799	-.1026	-.1171	-.1035	-.673
	15.00	.394	-.0693	-.0970	-.1075	-.1008	-.671
	20.00	.394	-.0567	-.0900	-.1000	-.983	-.706
	25.00	.405	-.0477	-.0835	-.872	-.947	-.657
	30.00	.407	-.0429	-.0766	-.738	-.881	-.604
	35.00	.378	-.0428	-.0699	-.698	-.829	-.616
	40.00	.378	-.0439	-.0598	-.654	-.783	-.625
	45.00	.396	-.0465	-.0600	-.618	-.743	-.571
	50.00	.414	-.0493	-.0613	-.605	-.702	-.516
	55.00	.433	-.0498	-.0618	-.599	-.667	-.490
	60.00	.446	-.0514	-.0617	-.578	-.623	-.447
	65.00	.455	-.0534	-.0589	-.503	-.548	-.449
	70.00	.469	-.0506	-.0502	-.373	-.477	-.422
LOWER SURFACE	0.00	.015	-.066	-.164	-.079	-.270	-.244
	1.25	.139	-.1086	-.1163	-.1324	-.1103	-.468
	2.50	.279	-.1109	-.1138	-.1308	-.1090	-.725
	5.00	.338	-.0976	-.1058	-.1249	-.1064	-.713
	7.50	.348	-.0866	-.1035	-.1210	-.1045	-.660
	10.00	.381	-.0799	-.1026	-.1171	-.1035	-.673
	15.00	.394	-.0693	-.0970	-.1075	-.1008	-.671
	20.00	.394	-.0567	-.0900	-.1000	-.983	-.706
	25.00	.405	-.0477	-.0835	-.872	-.947	-.657
	30.00	.407	-.0429	-.0766	-.738	-.881	-.604
	35.00	.378	-.0428	-.0699	-.698	-.829	-.616
	40.00	.378	-.0439	-.0598	-.654	-.783	-.625
	45.00	.396	-.0465	-.0600	-.618	-.743	-.571
	50.00	.414	-.0493	-.0613	-.605	-.702	-.516
	55.00	.433	-.0498	-.0618	-.599	-.667	-.490
	60.00	.446	-.0514	-.0617	-.578	-.623	-.447
	65.00	.455	-.0534	-.0589	-.503	-.548	-.449
	70.00	.469	-.0506	-.0502	-.373	-.477	-.422

TABLE IV

WING WITH LOWER SURFACE SPOILER (WITH GAP)

PERCENT CHORD	PRESSURE COEFFICIENT, P_i AT:						
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
$M = 0.98 \quad \alpha = 4.0^\circ$							
UPPER SURFACE	0.00	.045	.332	.235	.494	.192	.177
	1.25	.093	.617	.777	.863	.905	.930
	2.50	.007	.574	.738	.846	.902	.945
	5.00	.074	.288	.484	.729	.787	.854
	7.50	.087	.260	.388	.567	.750	.801
	10.00	.119	.247	.313	.573	.720	.791
	12.50	.137	.249	.313	.562	.701	.742
	15.00	.174	.251	.311	.565	.676	.700
	20.00	.194	.257	.327	.539	.653	.720
	30.00	.212	.257	.333	.594	.655	.753
	35.00	.203	.273	.350	.613	.680	.763
	40.00	.204	.287	.370	.636	.696	.777
	45.00	.228	.314	.393	.655	.729	.801
	50.00	.249	.339	.418	.692	.755	.827
	55.00	.276	.345	.430	.711	.782	.855
	60.00	.288	.364	.446	.716	.795	.866
	65.00	.297	.390	.471	.714	.794	.865
	70.00	.323	.847	.534	.710	.791	.865
LOWER SURFACE	75.00	.450	.264	.398	.500	.574	.596
	80.00	.490	.300	.416	.416	.518	.593
	85.00	.314	.287	.297	.372	.470	.495
	90.00	.321	.256	.301	.375	.460	.407
	95.00	.314	.136	.325	.389	.431	.348
	GAP						
	1.25	.387	.374	.384	.435	.438	.452
	2.50	.372	.307	.316	.312	.326	.296
	5.00	.326	.237	.248	.228	.266	.223
	7.50	.280	.193	.220	.190	.216	.211
	10.00	.255	.166	.190	.192	.186	.184
	15.00	.202	.130	.165	.157	.149	.130
	20.00	.168	.104	.146	.138	.122	.114
	25.00	.130	.087	.139	.122	.110	.097
	30.00	.100	.087	.129	.115	.103	.080
	35.00	.091	.107	.133	.117	.105	.073
	40.00	.076	.143	.145	.127	.121	.075
	45.00	.065	.208	.171	.155	.151	.088
	50.00	.084	.286	.219	.202	.201	.122
	55.00	.127	.368	.294	.268	.263	.176
	60.00	.172	.338	.324	.330	.296	.195
	65.00	.172	.264	.317	.312	.289	.169
	70.00	.705					
	75.00	1.029	.820	.512	.435	.396	.306
	80.00	.513	.905	.519	.442	.399	.297
	85.00	.410	.831	.518	.449	.399	.289
	90.00	.350	.728	.520	.459	.403	.282
	95.00	.244	.557	.514	.459	.398	.275
	GAP		.219	.328	.321	.299	.153
$M = 0.98 \quad \alpha = 5.9^\circ$							
UPPER SURFACE	0.00	.040	.174	.079	.324	.020	.031
	1.25	.013	.848	.972	1.038	1.059	1.071
	2.50	.105	.864	.962	1.026	1.064	1.024
	5.00	.167	.597	.858	.951	.980	1.023
	7.50	.180	.438	.779	.873	.947	.980
	10.00	.218	.379	.705	.799	.924	.961
	12.50	.225	.344	.621	.724	.840	.920
	15.00	.252	.333	.574	.637	.799	.873
	20.00	.268	.334	.566	.667	.799	.860
	30.00	.281	.331	.595	.648	.798	.850
	35.00	.264	.340	.608	.668	.792	.846
	40.00	.267	.346	.629	.687	.780	.832
	45.00	.286	.372	.649	.715	.780	.832
	50.00	.304	.394	.676	.745	.780	.832
	55.00	.323	.401	.686	.752	.780	.832
	60.00	.349	.414	.697	.760	.780	.832
	65.00	.372	.433	.713	.770	.780	.832
	70.00	.498	.336	.687	.751	.780	.832
	75.00	.339	.360	.606	.677	.780	.832
LOWER SURFACE	80.00	.361	.336	.643	.707	.780	.832
	85.00	.372	.288	.637	.693	.780	.832
	90.00	.364	.118	.635	.687	.780	.832
	95.00			.635	.687	.780	.832
	GAP						
	1.25	.422	.479	.475	.516	.491	.501
	2.50	.442	.410	.409	.389	.401	.370
	5.00	.418	.332	.336	.308	.339	.285
	7.50	.373	.283	.303	.262	.285	.224
	10.00	.341	.252	.267	.258	.250	.173
	15.00	.287	.209	.234	.217	.209	.192
	20.00	.242	.176	.205	.198	.176	.171
	25.00	.203	.156	.192	.173	.159	.146
	30.00	.167	.148	.176	.159	.146	.127
	35.00	.155	.157	.173	.155	.146	.127
	40.00	.139	.187	.180	.160	.146	.127
	45.00	.122	.208	.201	.180	.146	.127
	50.00	.164	.308	.242	.224	.177	.116
	55.00	.299	.383	.310	.285	.281	.188
	60.00	.506	.365	.346	.353	.318	.214
	65.00	.767	.291	.342	.334	.314	.190
	70.00						
	75.00	1.024	.808	.504	.433	.411	.326
	80.00	.469	.929	.510	.445	.418	.312
	85.00	.388	.853	.508	.453	.427	.307
	90.00	.305	.732	.513	.462	.430	.306
	95.00	.230	.496	.496	.462	.430	.308
	GAP		.245	.353	.341	.321	.169

TABLE IV

WING WITH LOWER SURFACE SPOILER (WITH GAP)

PERCENT CHORD		PRESSURE COEFFICIENT, P, AT:						
		0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	M = 0.98	$\alpha = 7.9^\circ$						
	0.00	.029	.015	.079	.159	.167	.155	.398
	1.25	.073	1.026	1.120	1.188	1.189	1.178	.929
	2.50	.211	1.058	1.109	1.168	1.191	1.097	.918
	5.00	.274	.901	1.068	1.118	1.124	1.133	.847
	7.50	.287	.722	1.017	1.070	1.093	1.100	.878
	10.00	.328	.616	.974	1.035	1.079	1.076	.869
	15.00	.326	.538	.869	.956	1.033	1.033	.836
	20.00	.340	.483	.804	.901	1.007	.984	.829
	25.00	.349	.445	.722	.894	1.003	.962	.807
	30.00	.358	.416	.687	.701	.934	.902	.800
	35.00	.334	.411	.685	.647	.902	.864	.759
	40.00	.335	.414	.659	.617	.884	.826	.751
	45.00	.349	.434	.619	.601	.823	.777	.722
	50.00	.366	.456	.583	.601	.739	.724	.706
	55.00	.385	.461	.553	.611	.699	.685	.685
	60.00	.401	.473	.544	.619	.665	.676	.657
	65.00	.403	.487	.573	.589	.567	.641	.634
	70.00	.420	.803	.513	.422	.547	.626	.620
LOWER SURFACE	75.00	.547	.430	.562	.658	.717	.653	.599
	80.00	.392	.429	.466	.580	.610	.587	.591
	85.00	.414	.391	.393	.491	.560	.576	.588
	90.00	.428	.326	.364	.453	.514	.541	.566
	95.00	.408	.123	.394	.427	.460	.513	.544
	1.25	.450	.555	.534	.506	.536	.523	.451
	2.50	.509	.494	.478	.447	.465	.418	.426
	5.00	.508	.418	.408	.375	.413	.395	.338
	7.50	.471	.370	.374	.334	.353	.344	.270
	10.00	.440	.335	.333	.323	.316	.310	.224
	15.00	.376	.286	.295	.276	.267	.244	.149
	20.00	.327	.246	.260	.245	.232	.218	.039
	25.00	.280	.224	.240	.228	.211	.192	.029
	30.00	.249	.214	.221	.205	.196	.164	.069
	35.00	.226	.217	.215	.196	.189	.148	.099
	40.00	.204	.237	.235	.195	.192	.135	.117
	45.00	.183	.277	.230	.209	.210	.135	.150
	50.00	.195	.333	.263	.246	.243	.150	.204
	55.00	.210	.403	.388	.303	.299	.190	.279
60.00	.332	.395	.367	.373	.344	.221	.296	
65.00	.539	.329	.366	.360	.344	.202	.320	
UPPER SURFACE	70.00	.806						.339
	75.00	1.016	.788	.506	.426	.424	.424	.342
	80.00	.425	.561	.514	.441	.427	.405	.354
	85.00	.354	.869	.511	.450	.437	.422	.366
	90.00	.270	.716	.511	.458	.440	.444	.370
	95.00	.194	.316	.480	.449	.436	.468	.376
	GAP		.284	.375	.364	.347	.175	
	M = 1.00	$\alpha = 0.0^\circ$						
	0.00	.089	.376	.583	.772	.510	.501	.407
	1.25	.280	.001	.028	.084	.194	.284	.365
	2.50	.208	.017	.013	.075	.143	.203	.284
	5.00	.138	.035	.043	.056	.107	.202	.256
	7.50	.112	.007	.044	.078	.166	.209	.269
	10.00	.092	.020	.058	.088	.183	.222	.275
	15.00	.049	.050	.085	.137	.195	.232	.289
	20.00	.002	.066	.109	.164	.216	.264	.332
	25.00	.024	.089	.136	.192	.252	.292	.369
	30.00	.044	.110	.154	.200	.269	.309	.374
	35.00	.057	.134	.178	.232	.292	.337	.370
40.00	.063	.150	.203	.259	.321	.364	.388	
45.00	.094	.176	.230	.294	.360	.399	.413	
50.00	.125	.204	.260	.332	.393	.432	.425	
55.00	.148	.217	.285	.364	.426	.480	.449	
60.00	.168	.231	.317	.385	.437	.504	.448	
65.00	.179	.280	.356	.388	.406	.459	.406	
70.00	.207	.307	.373	.470	.467	.368	.341	
75.00	.331	.116	.189	.170	.198	.318	.284	
80.00	.180	.160	.180	.215	.255	.288	.292	
85.00	.108	.172	.168	.227	.275	.297	.291	
90.00	.209	.161	.199	.252	.299	.301	.285	
95.00	.239	.131	.227	.278	.312	.306	.279	
UPPER SURFACE	1.25	.289	.000	.049	.009	.133	.157	.121
	2.50	.226	.010	.028	.001	.045	.019	.089
	5.00	.158	.006	.024	.020	.047	.072	.036
	7.50	.117	.009	.008	.014	.022	.039	.002
	10.00	.097	.022	.010	.006	.027	.027	.015
	15.00	.056	.040	.004	.012	.002	.004	.074
	20.00	.027	.055	.030	.017	.001	.009	.152
	25.00	.005	.077	.046	.023	.008	.015	.162
	30.00	.031	.078	.086	.034	.027	.020	.119
	35.00	.036	.073	.081	.031	.030	.046	.075
	40.00	.081	.078	.098	.077	.074	.041	.073
	45.00	.065	.168	.139	.118	.117	.067	.088
	50.00	.035	.271	.200	.181	.181	.118	.140
	55.00	.073	.355	.277	.263	.270	.200	.220
	60.00	.242	.320	.308	.341	.329	.240	.254
	65.00	.413	.264	.313	.333	.330	.217	.273
	70.00	.589						.272
	75.00	.986	.835	.586	.456	.424	.337	.258
	80.00	.582	.884	.534	.458	.427	.331	.261
85.00	.408	.802	.531	.459	.430	.329	.278	
90.00	.338	.714	.530	.468	.431	.320	.284	
95.00	.268	.608	.527	.476	.438	.313	.279	
LOWER SURFACE	GAP		.238	.332	.347	.341	.213	

TABLE IV

WING WITH LOWER SURFACE SPOILER (WITH GAP)

PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:						
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
M = 1.00 $\alpha = 4.0^\circ$							
UPPER SURFACE							
1.00	.083	.389	.262	.511	.214	.203	.013
1.25	.113	.601	.770	.858	.895	.908	.939
2.50	.088	.588	.788	.843	.892	.886	.948
5.00	.046	.265	.483	.733	.786	.863	.846
7.50	.061	.240	.289	.579	.748	.798	.847
10.00	.092	.223	.297	.360	.719	.777	.805
15.00	.112	.225	.292	.340	.513	.739	.685
20.00	.122	.230	.294	.363	.391	.681	.702
25.00	.170	.234	.303	.377	.427	.540	.728
30.00	.186	.236	.312	.374	.440	.461	.587
35.00	.179	.252	.347	.393	.456	.484	.469
40.00	.183	.261	.347	.413	.474	.508	.469
45.00	.209	.291	.369	.439	.501	.540	.471
50.00	.230	.315	.396	.467	.527	.568	.505
55.00	.249	.321	.408	.486	.557	.603	.547
60.00	.266	.337	.421	.491	.569	.637	.570
65.00	.274	.362	.446	.491	.539	.598	.580
70.00	.293	.380	.455	.428	.489	.401	.574
75.00	.424	.243	.390	.503	.563	.555	.554
80.00	.273	.284	.341	.413	.522	.560	.538
85.00	.295	.274	.288	.357	.464	.509	.556
90.00	.302	.254	.282	.355	.450	.443	.555
95.00	.292	.120	.384	.369	.427	.402	.495
1.25	.401	.394	.480	.455	.453	.465	.389
2.50	.390	.326	.333	.330	.347	.314	.328
5.00	.344	.257	.266	.248	.287	.281	.239
7.50	.301	.216	.238	.200	.235	.231	.190
10.00	.276	.187	.208	.207	.206	.204	.133
15.00	.223	.150	.184	.178	.171	.148	.061
20.00	.187	.124	.158	.159	.142	.131	.041
25.00	.150	.106	.159	.144	.129	.114	.102
30.00	.119	.105	.151	.136	.122	.099	.127
35.00	.109	.126	.155	.137	.123	.091	.138
40.00	.095	.164	.144	.146	.139	.091	.136
45.00	.082	.230	.192	.174	.167	.101	.154
50.00	.102	.314	.238	.221	.215	.131	.202
55.00	.145	.387	.309	.286	.277	.183	.278
60.00	.292	.360	.338	.346	.310	.203	.299
65.00	.490	.281	.334	.328	.303	.181	.313
70.00	.719						.314
75.00	.996	.796	.498	.437	.420	.356	.315
80.00	.490	.883	.507	.445	.426	.348	.324
85.00	.392	.814	.506	.434	.432	.343	.328
90.00	.322	.708	.507	.444	.440	.342	.322
95.00	.233	.526	.505	.468	.445	.345	.318
GAP		.240	.345	.334	.311	.162	
M = 1.00 $\alpha = 5.0^\circ$							
UPPER SURFACE							
1.00	.033	.206	.103	.346	.047	.056	.181
1.25	.037	.844	.959	1.023	1.049	1.062	1.078
2.50	.082	.866	.948	1.013	1.053	1.005	1.072
5.00	.149	.601	.866	.940	.974	1.011	.970
7.50	.157	.421	.797	.879	.941	.972	.992
10.00	.198	.362	.731	.819	.918	.950	.969
15.00	.225	.326	.412	.748	.854	.914	.864
20.00	.229	.311	.382	.548	.821	.869	.879
25.00	.254	.311	.363	.448	.815	.866	.886
30.00	.258	.310	.375	.425	.673	.815	.808
35.00	.244	.317	.385	.444	.545	.808	.726
40.00	.248	.326	.406	.463	.517	.798	.757
45.00	.263	.351	.429	.490	.535	.759	.787
50.00	.284	.372	.457	.520	.559	.692	.701
55.00	.306	.377	.464	.543	.593	.665	.804
60.00	.322	.394	.474	.550	.613	.659	.784
65.00	.331	.410	.489	.541	.587	.574	.688
70.00	.347	.478	.501	.418	.433	.444	.686
75.00	.477	.320	.477	.506	.639	.598	.671
80.00	.319	.342	.407	.506	.608	.611	.647
85.00	.340	.321	.332	.429	.569	.570	.652
90.00	.354	.285	.317	.408	.533	.519	.652
95.00	.347	.103	.338	.406	.474	.468	.574
1.25	.438	.498	.488	.531	.500	.511	.434
2.50	.462	.425	.422	.404	.413	.380	.387
5.00	.434	.352	.351	.326	.349	.342	.299
7.50	.393	.303	.320	.277	.297	.291	.242
10.00	.365	.270	.284	.278	.265	.258	.188
15.00	.308	.225	.251	.237	.228	.201	.119
20.00	.265	.196	.224	.213	.193	.181	.008
25.00	.224	.175	.209	.192	.173	.157	.057
30.00	.189	.170	.195	.179	.161	.136	.089
35.00	.175	.180	.192	.174	.158	.121	.110
40.00	.157	.212	.197	.178	.158	.127	.122
45.00	.140	.263	.218	.199	.189	.150	.147
50.00	.155	.331	.260	.238	.230	.198	.198
55.00	.183	.408	.360	.299	.290	.222	.275
60.00	.280	.384	.369	.366	.328	.222	.297
65.00	.784	.399	.389	.348	.323	.200	.316
70.00	.989						.326
75.00	.440	.780	.490	.427	.417	.362	.327
80.00	.367	.907	.498	.439	.424	.350	.338
85.00	.286	.830	.496	.451	.435	.350	.345
90.00	.218	.708	.498	.463	.445	.352	.342
95.00		.467	.483	.460	.447	.363	.342
GAP		.265	.367	.353	.332	.179	

TABLE IV

WING WITH LOWER SURFACE SPOILER (WITH GAP)

PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:						
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
$M = 1.00 \quad \alpha = 7.9^\circ$							
UPPER SURFACE	.00	.004	.060	.030	.201	.115	.097
	1.25	.029	.956	1.048	1.107	1.132	1.133
	2.50	.162	.983	1.038	1.095	1.124	1.127
	5.00	.230	.824	.994	1.034	1.056	1.084
	7.50	.244	.630	.944	.991	1.004	1.049
	10.00	.280	.518	.897	.953	1.004	1.030
	15.00	.282	.451	.766	.883	.962	.992
	20.00	.298	.415	.548	.802	.936	.951
	25.00	.307	.398	.469	.787	.931	.925
	30.00	.314	.376	.434	.636	.854	.925
	35.00	.293	.372	.436	.579	.800	.908
	40.00	.294	.374	.431	.561	.780	.885
	45.00	.311	.393	.473	.554	.753	.851
	50.00	.323	.417	.499	.560	.696	.711
	55.00	.345	.430	.517	.573	.667	.658
	60.00	.364	.443	.525	.580	.646	.629
	65.00	.369	.443	.525	.583	.583	.623
	70.00	.389	.443	.497	.414	.533	.603
	75.00	.311	.390	.531	.621	.639	.679
	80.00	.354	.390	.441	.556	.617	.642
	85.00	.377	.311	.363	.468	.574	.580
	90.00	.391	.311	.348	.439	.503	.543
	95.00	.378	.103	.362	.420		
LOWER SURFACE	.00						
	1.25	.487	.572	.544	.532	.541	.544
	2.50	.525	.509	.488	.460	.466	.422
	5.00	.526	.430	.418	.389	.416	.399
	7.50	.487	.382	.384	.344	.356	.346
	10.00	.457	.349	.347	.336	.322	.312
	15.00	.393	.299	.307	.288	.272	.250
	20.00	.344	.267	.277	.261	.239	.225
	25.00	.299	.242	.238	.236	.214	.199
	30.00	.259	.232	.231	.220	.202	.175
	35.00	.243	.235	.231	.211	.193	.159
	40.00	.219	.254	.234	.211	.198	.149
	45.00	.200	.279	.247	.226	.216	.166
	50.00	.212	.355	.344	.261	.305	.208
	55.00	.227	.424	.344	.316	.348	.239
	60.00	.352	.417	.381	.385	.349	.218
	65.00	.358	.347	.381	.370		
	70.00	.833		.494	.421	.435	.395
	75.00	.873	.756	.499	.438	.443	.378
	80.00	.402	.835	.497	.450	.455	.391
	85.00	.339	.692	.497	.459	.460	.412
	90.00	.179	.319	.469	.452	.456	.436
	95.00		.293	.388	.375	.351	.193
$M = 1.03 \quad \alpha = 0.0^\circ$							
UPPER SURFACE	.00	.055	.572	.518	.789	.536	.532
	1.25	.205	.025	.046	.063	.184	.273
	2.50	.164	.004	.028	.059	.125	.183
	5.00	.095	.005	.037	.028	.103	.170
	7.50	.069	.024	.052	.079	.141	.180
	10.00	.056	.034	.068	.123	.155	.194
	15.00	.025	.063	.094	.146	.167	.196
	20.00	.024	.077	.120	.171	.186	.222
	25.00	.038	.095	.138	.182	.220	.251
	30.00	.048	.111	.157	.211	.234	.268
	35.00	.072	.137	.174	.235	.255	.295
	40.00	.077	.155	.198	.264	.279	.322
	45.00	.099	.183	.230	.295	.316	.357
	50.00	.129	.208	.262	.318	.343	.389
	55.00	.154	.216	.283	.340	.375	.436
	60.00	.168	.232	.321	.357	.391	.471
	65.00	.184	.262	.359	.357	.422	.453
	70.00	.208	.831	.559	.433	.482	.337
	75.00	.339	.134	.166	.203	.210	.321
	80.00	.188	.169	.183	.186	.223	.286
	85.00	.206	.181	.155	.189	.235	.284
	90.00	.206	.162	.172	.211	.258	.268
	95.00	.194	.110	.189	.236	.268	.269
LOWER SURFACE	.00						
	1.25	.242	.006	.059	.022	.170	.232
	2.50	.184	.001	.041	.014	.089	.067
	5.00	.119	.005	.020	.003	.086	.118
	7.50	.081	.020	.024	.021	.068	.084
	10.00	.065	.032	.074	.038	.042	.071
	15.00	.032	.047	.078	.037	.043	.039
	20.00	.011	.064	.079	.047	.049	.055
	25.00	.022	.084	.081	.069	.066	.071
	30.00	.043	.102	.099	.085	.084	.084
	35.00	.047	.124	.123	.111	.113	.105
	40.00	.062	.139	.133	.154	.154	.153
	45.00	.087	.159	.158	.216	.216	.233
	50.00	.108	.185	.196	.293	.293	.270
	55.00	.008	.330	.331	.358	.358	.244
	60.00	.244	.839	.347	.358		
	65.00	.408					
	70.00	.393	.797	.489	.416	.389	.318
	75.00	.529	.842	.493	.420	.394	.307
	80.00	.308	.761	.487	.424	.397	.306
	85.00	.338	.666	.481	.434	.397	.291
	90.00	.271	.884	.481	.444	.403	.291
	95.00		.240	.368	.368	.378	.243

TABLE IV

WING WITH LOWER SURFACE SPOILER (WITH GAF)

		PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:						
			0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
UPPER SURFACE	$M = 1.03 \quad \alpha = 4.0^\circ$								
	.00	.069	.392	.301	.546	.253	.245	.035	
	1.25	.098	.603	.762	.838	.877	.883	.906	
	2.50	.009	.568	.732	.834	.884	.855	.913	
	5.00	.064	.276	.576	.748	.791	.848	.809	
	7.50	.087	.237	.271	.634	.753	.796	.622	
	10.00	.109	.224	.261	.409	.781	.771	.796	
	15.00	.129	.223	.267	.310	.570	.733	.696	
	20.00	.159	.217	.273	.327	.356	.684	.701	
	25.00	.171	.222	.279	.341	.377	.567	.703	
	30.00	.175	.236	.288	.340	.392	.423	.597	
	35.00	.180	.238	.298	.359	.410	.429	.470	
	40.00	.183	.245	.319	.379	.427	.453	.442	
	45.00	.197	.249	.342	.404	.455	.484	.419	
	50.00	.218	.249	.367	.432	.477	.512	.448	
	55.00	.238	.250	.378	.445	.508	.549	.486	
	60.00	.253	.253	.386	.455	.523	.580	.510	
	65.00	.265	.255	.402	.454	.499	.538	.531	
	70.00	.277	.266	.474	.384	.383	.362	.528	
75.00	.406	.229	.382	.489	.536	.553	.518		
80.00	.253	.259	.321	.410	.509	.531	.520		
85.00	.271	.269	.267	.334	.461	.502	.542		
90.00	.282	.242	.243	.317	.432	.461	.555		
95.00	.279	.091	.259	.325	.388	.403	.518		
LOWER SURFACE	1.25	.344	.376	.402	.482	.486	.510	.428	
	2.50	.352	.352	.356	.358	.385	.351	.371	
	5.00	.317	.247	.275	.274	.322	.321	.285	
	7.50	.277	.208	.232	.241	.273	.271	.237	
	10.00	.256	.180	.223	.247	.242	.244	.180	
	15.00	.209	.147	.207	.213	.211	.191	.109	
	20.00	.174	.122	.153	.194	.184	.173	.012	
	25.00	.141	.110	.139	.180	.168	.157	.050	
	30.00	.113	.116	.124	.173	.163	.140	.076	
	35.00	.103	.143	.187	.177	.165	.134	.087	
	40.00	.095	.192	.198	.186	.181	.135	.087	
	45.00	.088	.261	.285	.213	.207	.135	.104	
	50.00	.115	.345	.274	.258	.253	.173	.151	
	55.00	.161	.420	.343	.328	.314	.221	.232	
	60.00	.320	.389	.371	.328	.347	.244	.258	
	65.00	.512	.302	.369	.364	.342	.220	.276	
	70.00	.747						.278	
	75.00	.747		.449	.389	.373	.322	.280	
	80.00	.442	.828	.457	.398	.381	.310	.286	
85.00	.364	.756	.456	.408	.389	.308	.291		
90.00	.301	.646	.455	.417	.395	.307	.287		
95.00	.237	.464	.450	.421	.402	.313	.284		
GAP		.266	.380	.371	.350	.203			
UPPER SURFACE	$M = 1.03 \quad \alpha = 5.0^\circ$								
	.00	.084	.255	.147	.368	.094	.103	.126	
	1.25	.052	.814	.884	.946	.975	.984	1.000	
	2.50	.032	.795	.874	.932	.977	.930	.997	
	5.00	.131	.623	.804	.870	.903	.937	.894	
	7.50	.147	.379	.721	.813	.868	.899	.916	
	10.00	.177	.305	.676	.744	.846	.879	.896	
	15.00	.191	.289	.485	.679	.787	.842	.801	
	20.00	.214	.277	.315	.564	.733	.802	.813	
	25.00	.223	.277	.328	.454	.653	.733	.815	
	30.00	.226	.275	.335	.378	.528	.722	.741	
	35.00	.220	.283	.346	.393	.469	.719	.660	
	40.00	.220	.291	.364	.409	.476	.706	.676	
	45.00	.236	.312	.384	.433	.496	.666	.689	
	50.00	.252	.322	.419	.463	.527	.620	.700	
	55.00	.258	.340	.429	.477	.547	.600	.715	
	60.00	.268	.351	.426	.494	.547	.600	.719	
	65.00	.294	.365	.437	.485	.526	.527	.654	
	70.00	.310	.737	.452	.370	.381	.400	.637	
75.00	.436	.284	.446	.530	.579	.541	.620		
80.00	.284	.298	.370	.472	.559	.561	.604		
85.00	.304	.288	.299	.391	.529	.536	.620		
90.00	.313	.247	.273	.359	.494	.498	.630		
95.00	.305	.067	.285	.354	.428	.449	.564		
LOWER SURFACE	1.25	.386	.503	.505	.566	.532	.545	.465	
	2.50	.439	.435	.441	.440	.449	.410	.416	
	5.00	.431	.362	.373	.361	.385	.378	.330	
	7.50	.399	.316	.343	.315	.335	.327	.279	
	10.00	.372	.287	.309	.316	.300	.295	.223	
	15.00	.318	.245	.280	.273	.262	.236	.159	
	20.00	.276	.217	.257	.251	.231	.218	.048	
	25.00	.237	.199	.243	.238	.212	.196	.016	
	30.00	.205	.195	.230	.219	.200	.175	.050	
	35.00	.191	.208	.230	.216	.197	.163	.070	
	40.00	.177	.243	.235	.219	.207	.158	.079	
	45.00	.162	.298	.254	.241	.228	.168	.102	
	50.00	.183	.368	.295	.280	.268	.187	.153	
	55.00	.211	.440	.361	.337	.327	.257	.231	
	60.00	.352	.418	.394	.401	.364	.257	.257	
	65.00	.549	.341	.393	.385	.359	.235	.279	
	70.00	.816						.289	
	75.00	.921	.722	.441	.376	.367	.325	.293	
	80.00	.403	.835	.447	.388	.377	.313	.303	
85.00	.328	.759	.447	.399	.385	.313	.312		
90.00	.256	.639	.446	.408	.390	.320	.311		
95.00	.194	.407	.433	.408	.399	.333	.312		
GAP		.229	.401	.389	.364	.216			

TABLE IV

WING WITH LOWER SURFACE SPOILER (WITH GAP)

PERCENT CHORD	PRESSURE COEFFICIENT, P , AT:						
	0.135b/2	0.25b/2	0.40b/2	0.55b/2	0.70b/2	0.85b/2	0.95b/2
$M = 1.03 \quad \alpha = 7.9^\circ$							
UPPER SURFACE	0.00	.110	.097	.005	.240	-.075	-.053
	1.25	.005	.952	.991	1.045	1.062	1.074
	2.50	.128	.945	.980	1.032	1.064	1.077
	5.00	.208	.885	.946	.981	.998	1.000
	7.50	.258	.677	.900	.939	.969	1.028
	10.00	.288	.490	.868	.902	.953	1.094
	15.00	.267	.366	.790	.838	.910	.974
	20.00	.264	.344	.618	.798	.882	.937
	25.00	.286	.355	.484	.788	.880	.897
	30.00	.284	.344	.401	.774	.835	.896
	35.00	.273	.345	.393	.569	.799	.866
	40.00	.269	.347	.402	.536	.779	.861
	45.00	.283	.365	.425	.520	.761	.851
	50.00	.297	.385	.450	.517	.699	.854
	55.00	.318	.386	.439	.516	.628	.733
	60.00	.332	.396	.468	.527	.599	.641
	65.00	.337	.408	.476	.508	.537	.618
LOWER SURFACE	70.00	.354	.719	.444	.569	.511	.609
	75.00	.476	.350	.488	.565	.511	.650
	80.00	.328	.152	.401	.515	.597	.627
	85.00	.340	.118	.316	.431	.585	.598
	90.00	.353	.267	.298	.395	.536	.549
	95.00	.334	.069	.308	.376	.480	.505
	1.25	.432	.594	.572	.564	.572	.577
	2.50	.523	.533	.519	.497	.500	.448
	5.00	.548	.459	.450	.421	.450	.426
	7.50	.517	.412	.419	.383	.392	.377
	10.00	.488	.376	.381	.372	.357	.345
	15.00	.423	.332	.345	.326	.308	.283
	20.00	.371	.296	.313	.299	.275	.259
	25.00	.325	.274	.294	.276	.252	.233
	30.00	.289	.262	.275	.248	.238	.210
	35.00	.269	.267	.270	.247	.229	.193
	40.00	.251	.330	.285	.263	.232	.183
	45.00	.245	.390	.317	.295	.282	.185
	50.00	.255	.458	.379	.349	.337	.199
	55.00	.366	.447	.415	.418	.376	.239
	60.00	.585	.382	.415	.403	.380	.270
	65.00	.659					.251
	70.00	.913	.698	.440	.373	.398	.375
	75.00	.400	.848	.446	.391	.405	.355
	80.00	.297	.763	.442	.403	.418	.368
	85.00	.194	.627	.443	.412	.422	.391
	90.00	.301	.415	.415	.405	.422	.414
	95.00	.072	.332	.424	.407	.383	.226
	GAP						

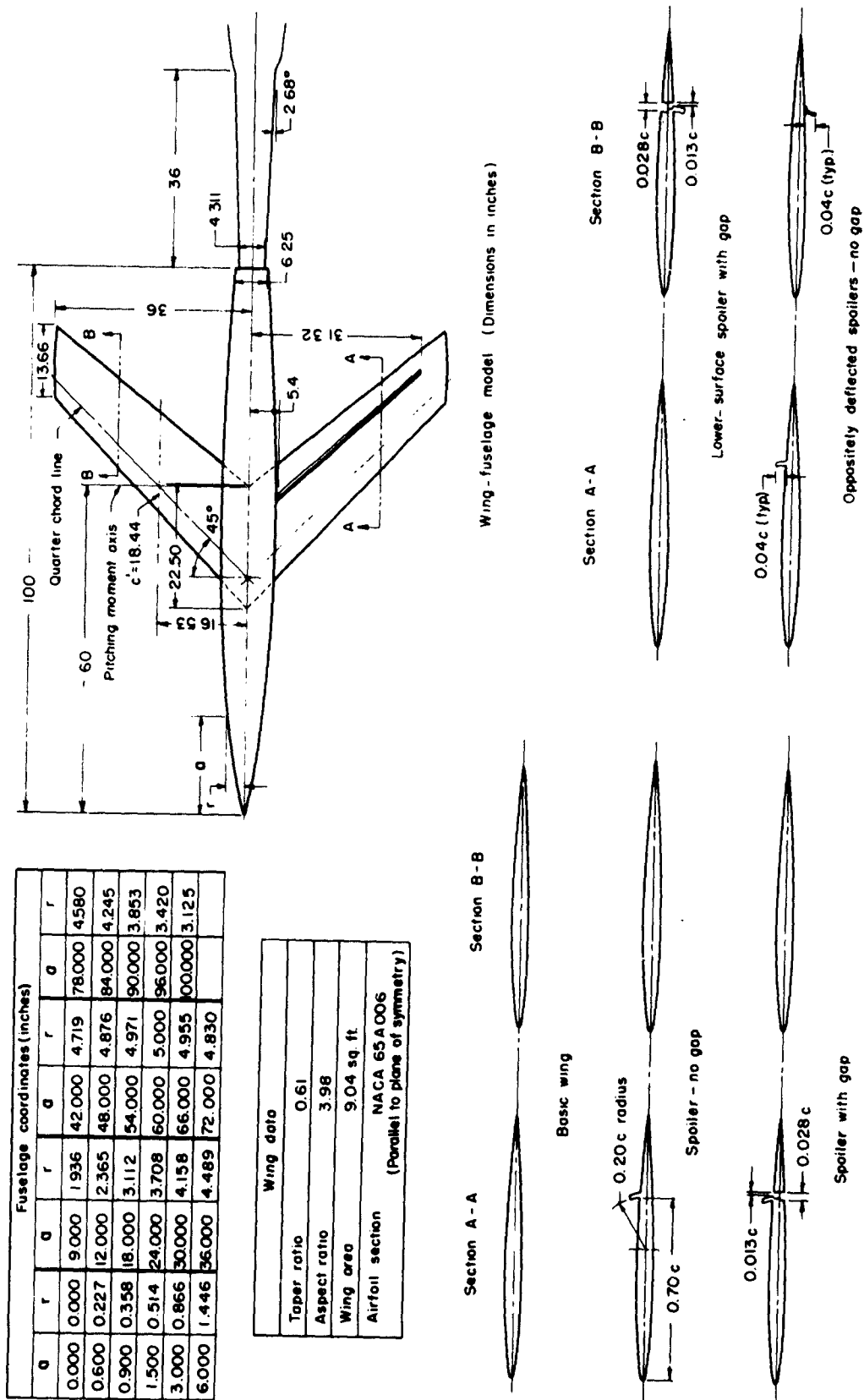


Figure 1.- Geometry of test configurations.

Chordwise locations of wing orifices at each station are given with the pressure coefficient data in Tables I-IV.

	Spoiler-orifice locations, z/h				
	$y/b/2 = 0.25$	$y/b/2 = 0.40$	$y/b/2 = 0.55$	$y/b/2 = 0.70$	$y/b/2 = 0.85$
Front	0.201	0.207	0.177	0.188	0.159
Front	.455	.448	.422	.435	.405
Front	.704	.720	.668	.687	.643
Front	.917	.898	.866	.880	.835
Back	.060	.035	.030	.091	.178
Back	.448	.445	.428	.413	.463
Back	.865	.841	.808	.810	.801

Typical section showing details in vicinity of spoiler

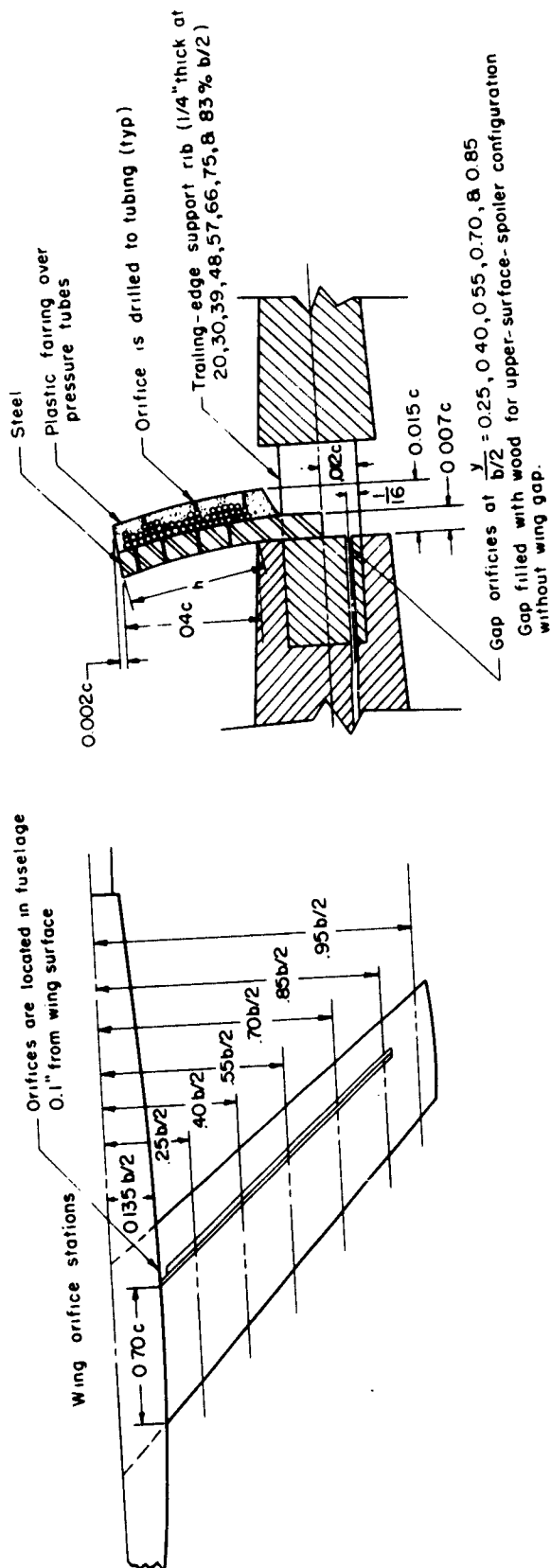


Figure 2.- Pressure-orifice locations and details of spoiler and wing gap.

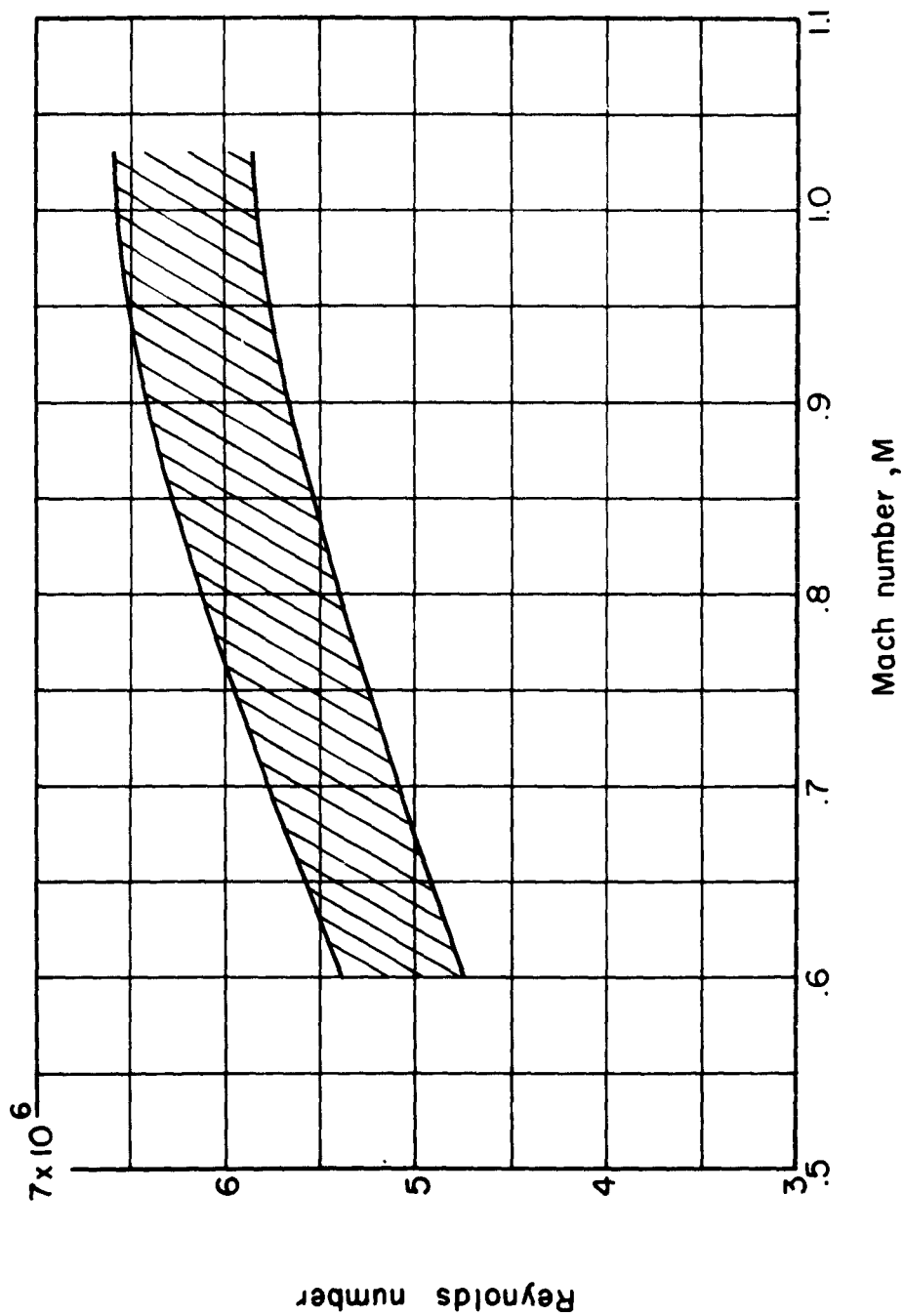


Figure 3.- Variation of Reynolds number based on mean aerodynamic chord with Mach number.

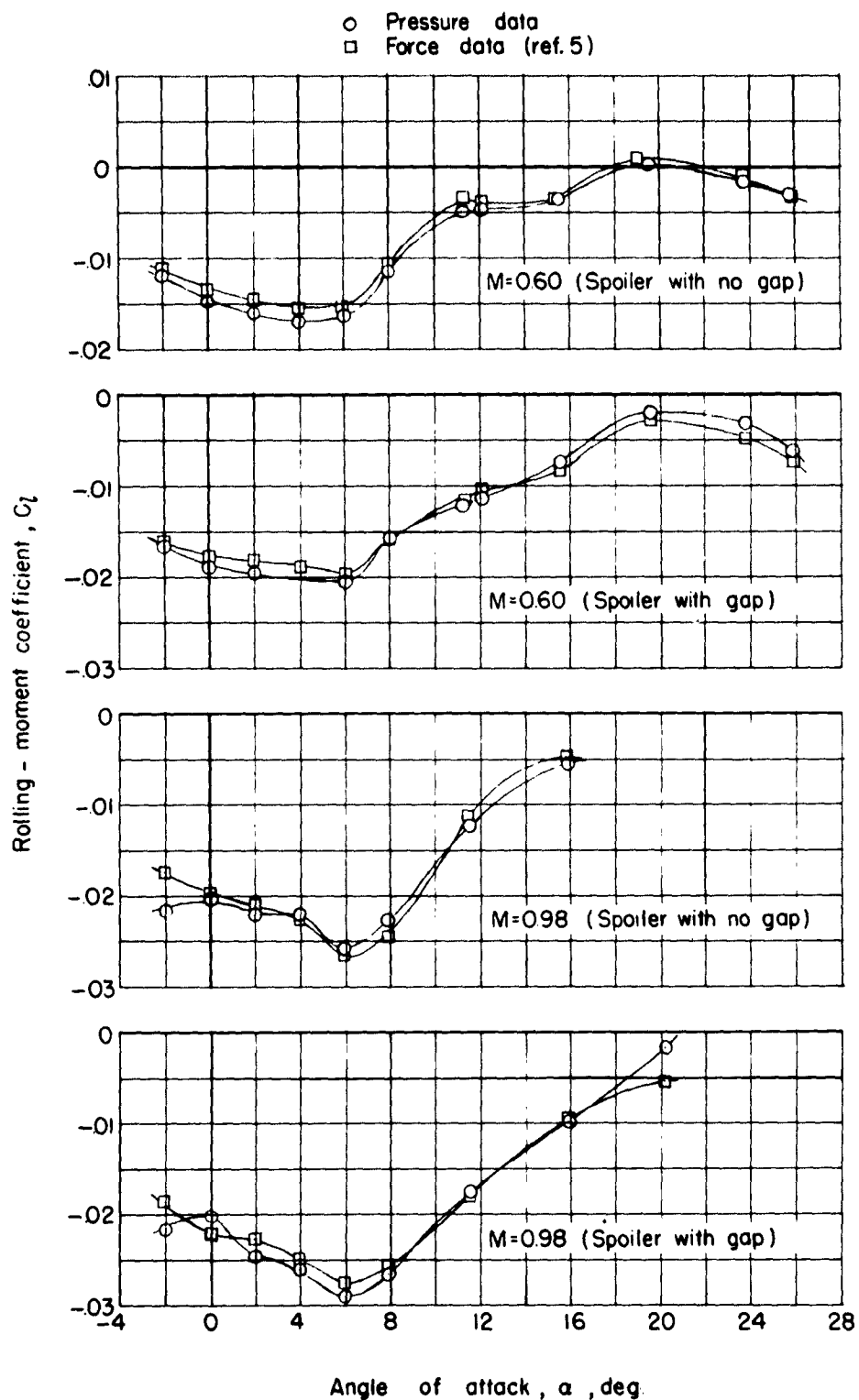


Figure 4.- Comparison of rolling-moment characteristics as obtained from pressure data and from force data for the model with upper-surface spoilers.

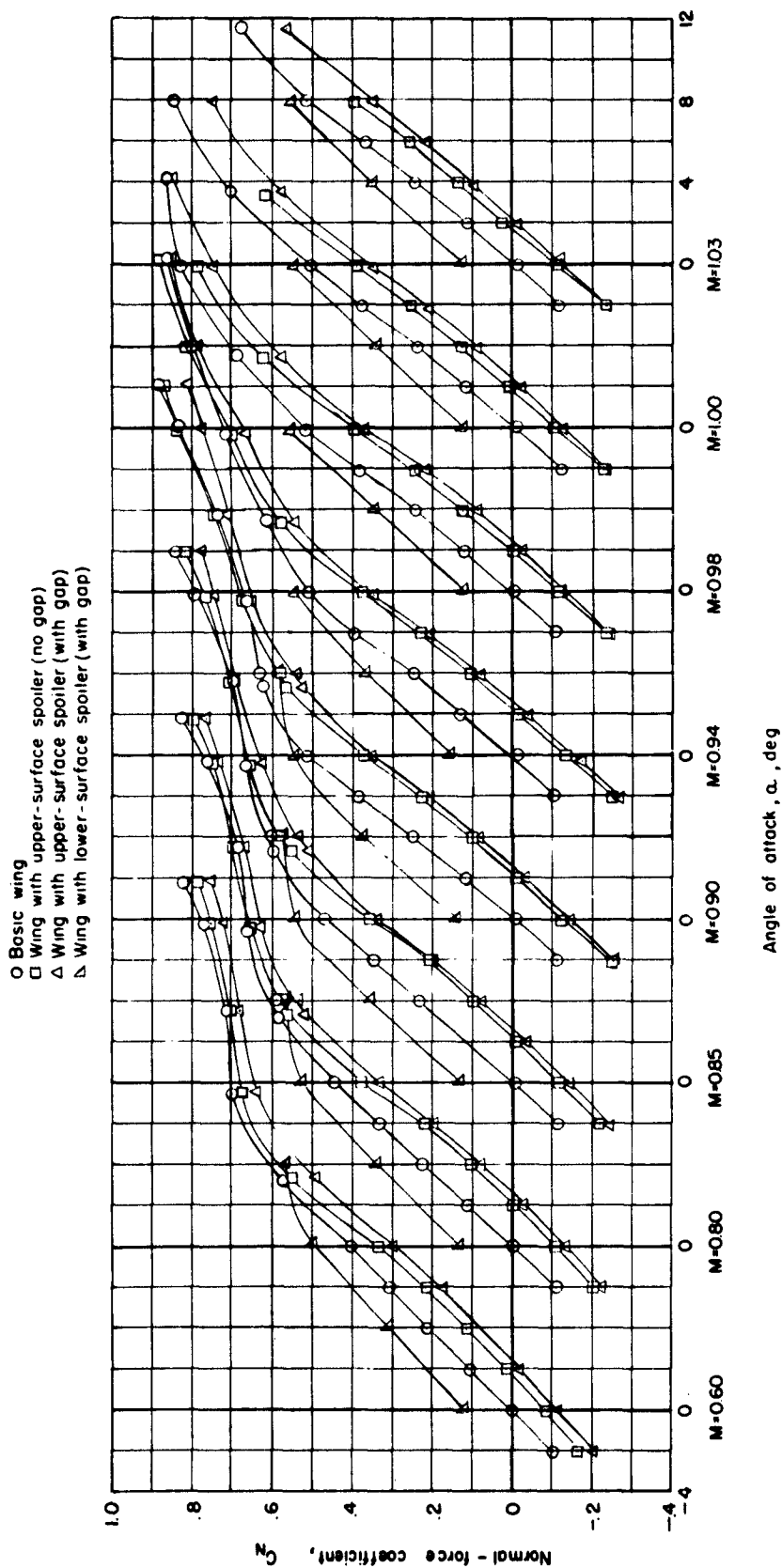
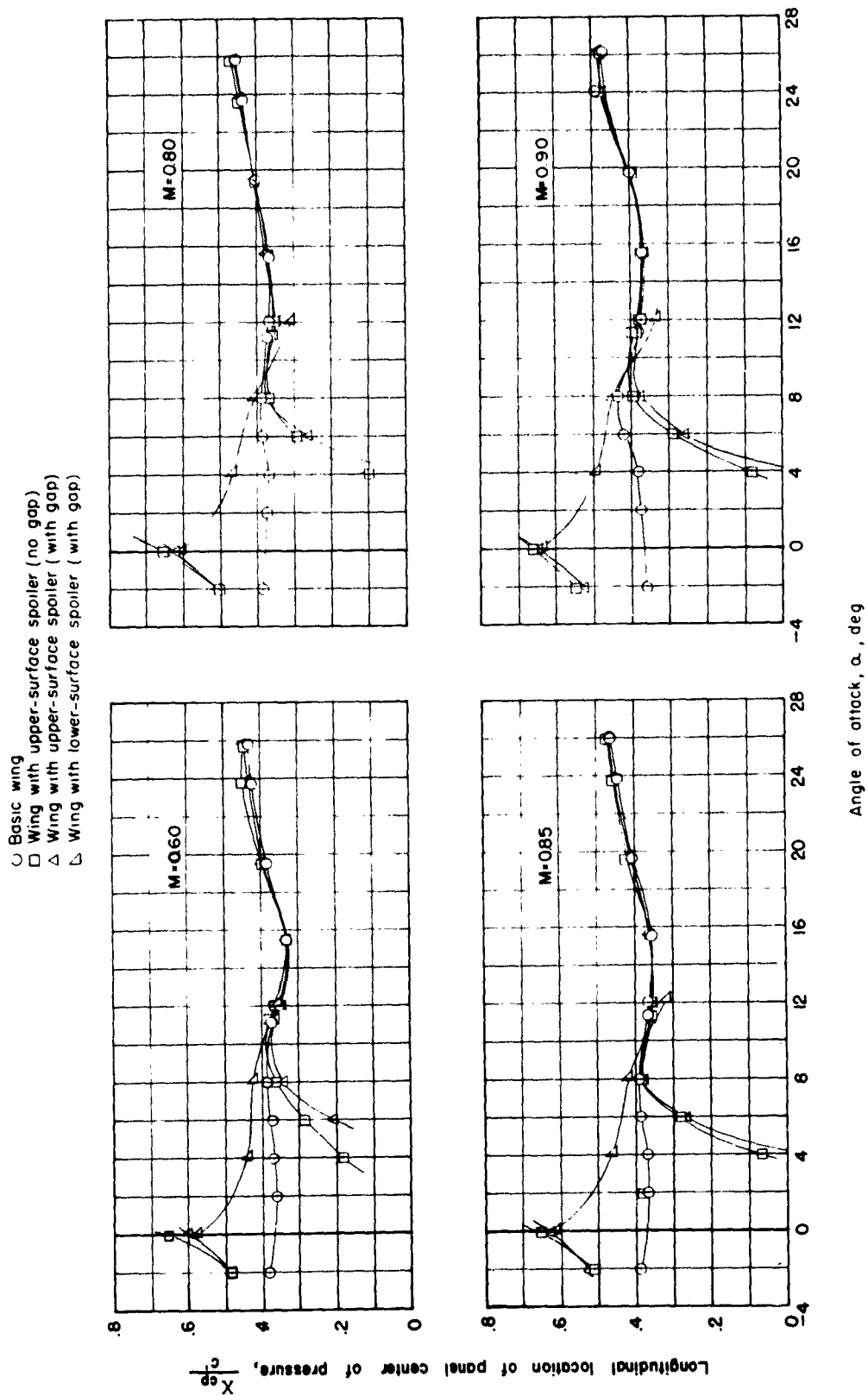


Figure 5.- Wing normal-force characteristics for the basic wing and various spoiler configurations. $M = 0.60$ to $M = 1.03$.



(a) $M = 0.60, 0.80, 0.85, 0.90$.

Figure 6.- Longitudinal position of wing center of pressure for the basic wing and various spoiler configurations.

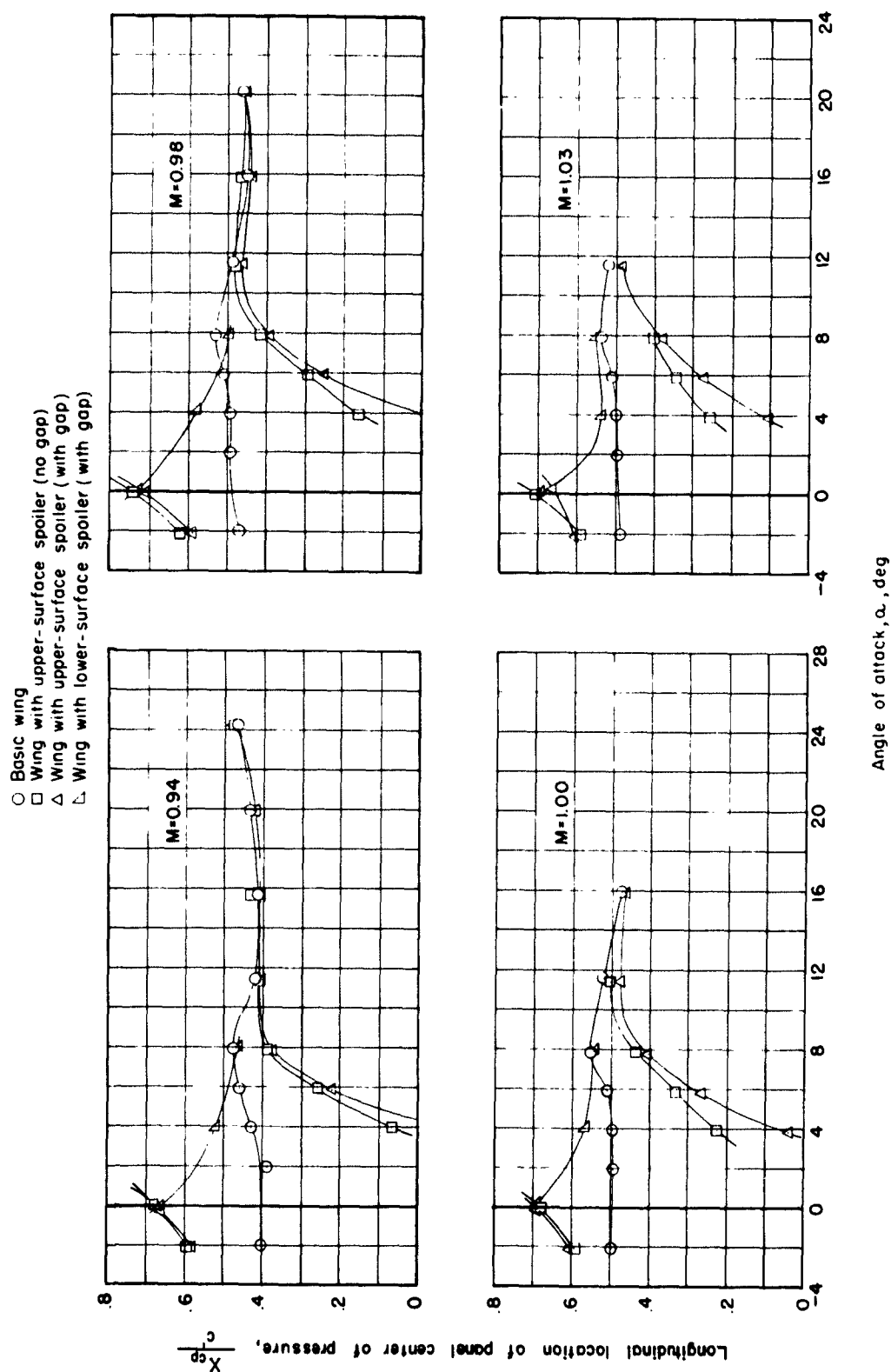
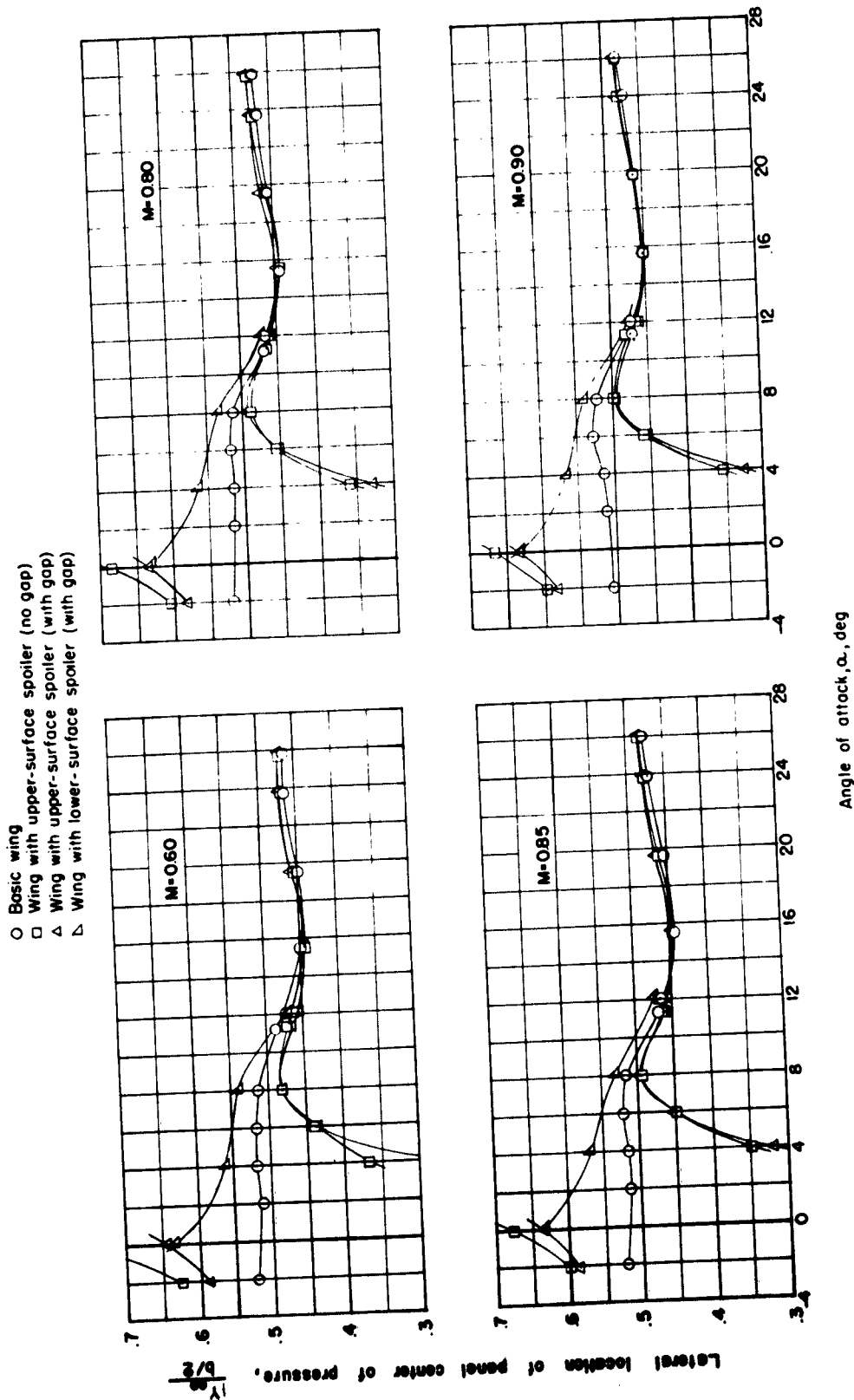
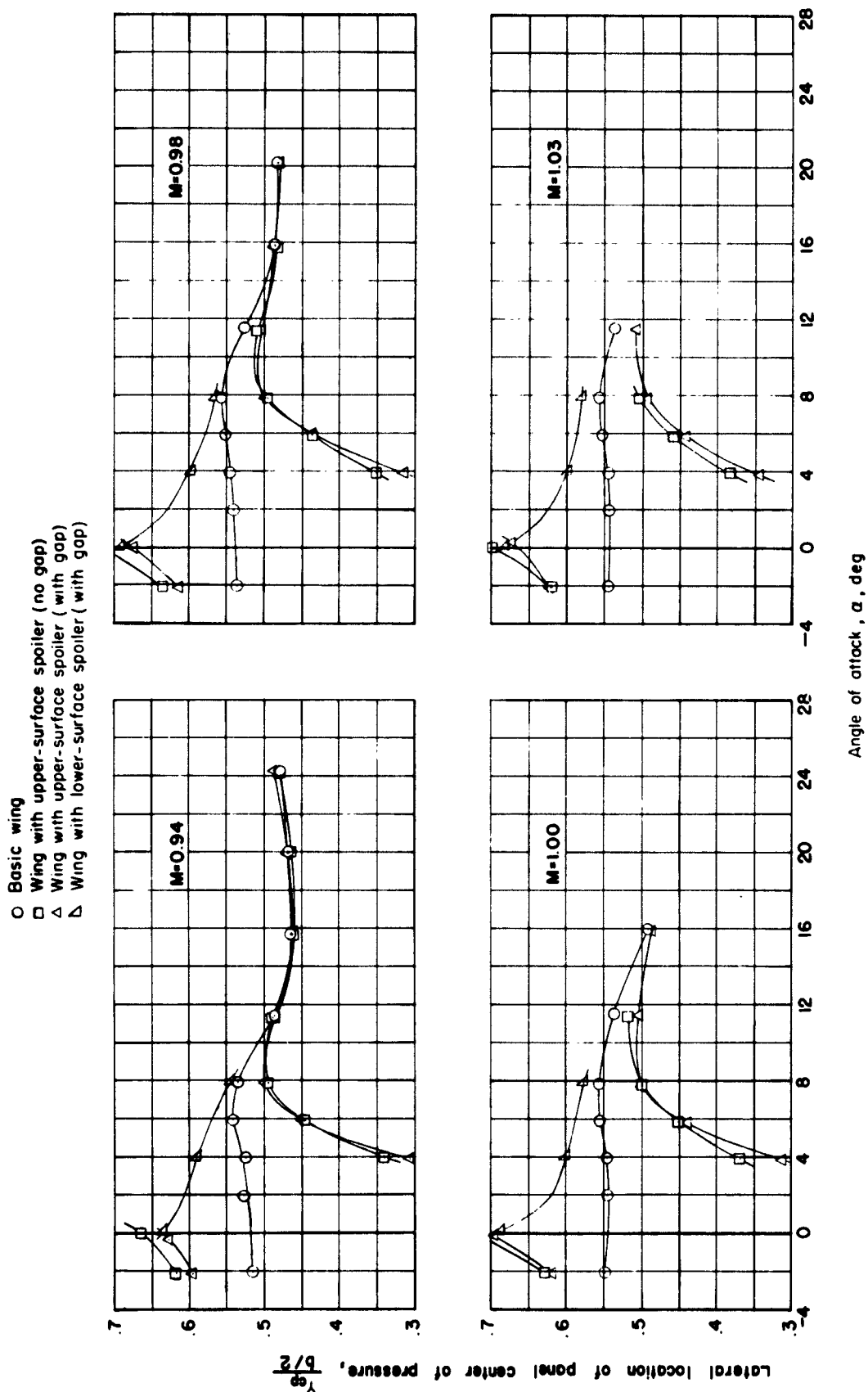
(b) $M = 0.94, 0.98, 1.00, 1.03$.

Figure 6.- Concluded.



(a) $M = 0.60, 0.80, 0.85, 0.90$.

Figure 7.- Lateral position of wing center of pressure for the basic and various spoiler configurations.



(b) $M = 0.94, 0.98, 1.00, 1.03$.

Figure 7.- Concluded.

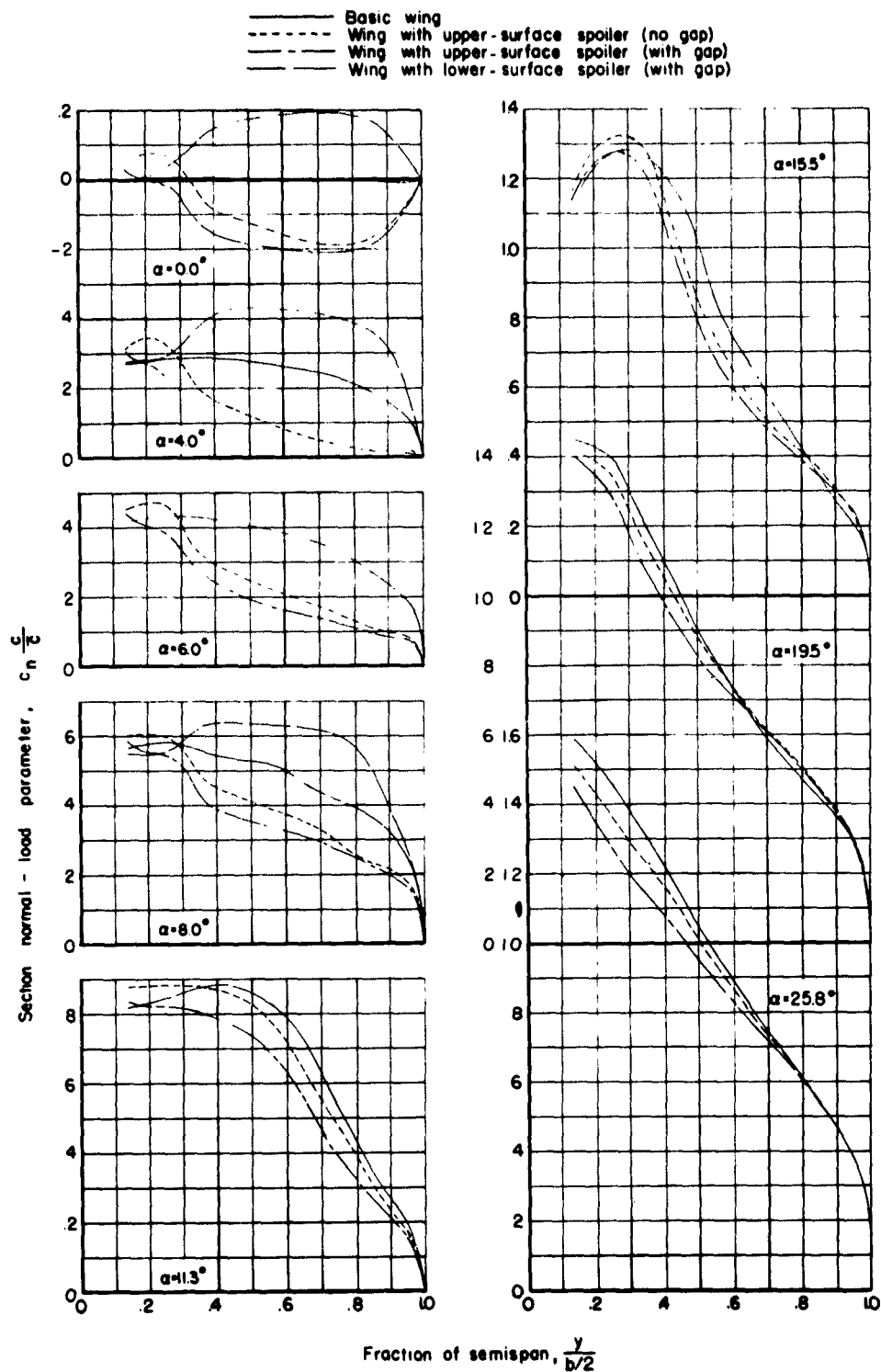
(a) $M = 0.60$.

Figure 8.- Wing semispan load distributions for the basic wing and various spoiler configurations.

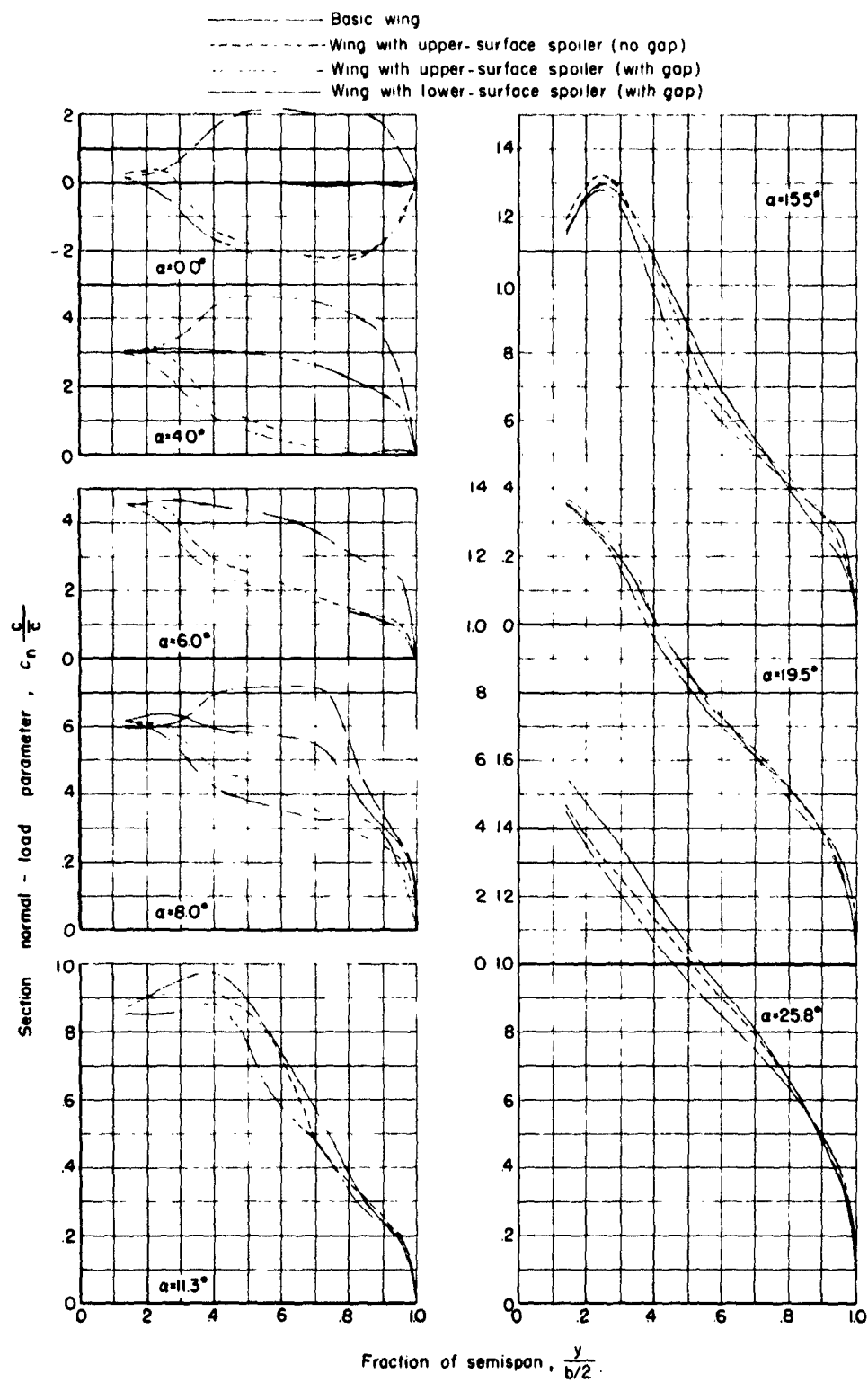
(b) $M = 0.80$.

Figure 8.- Continued.

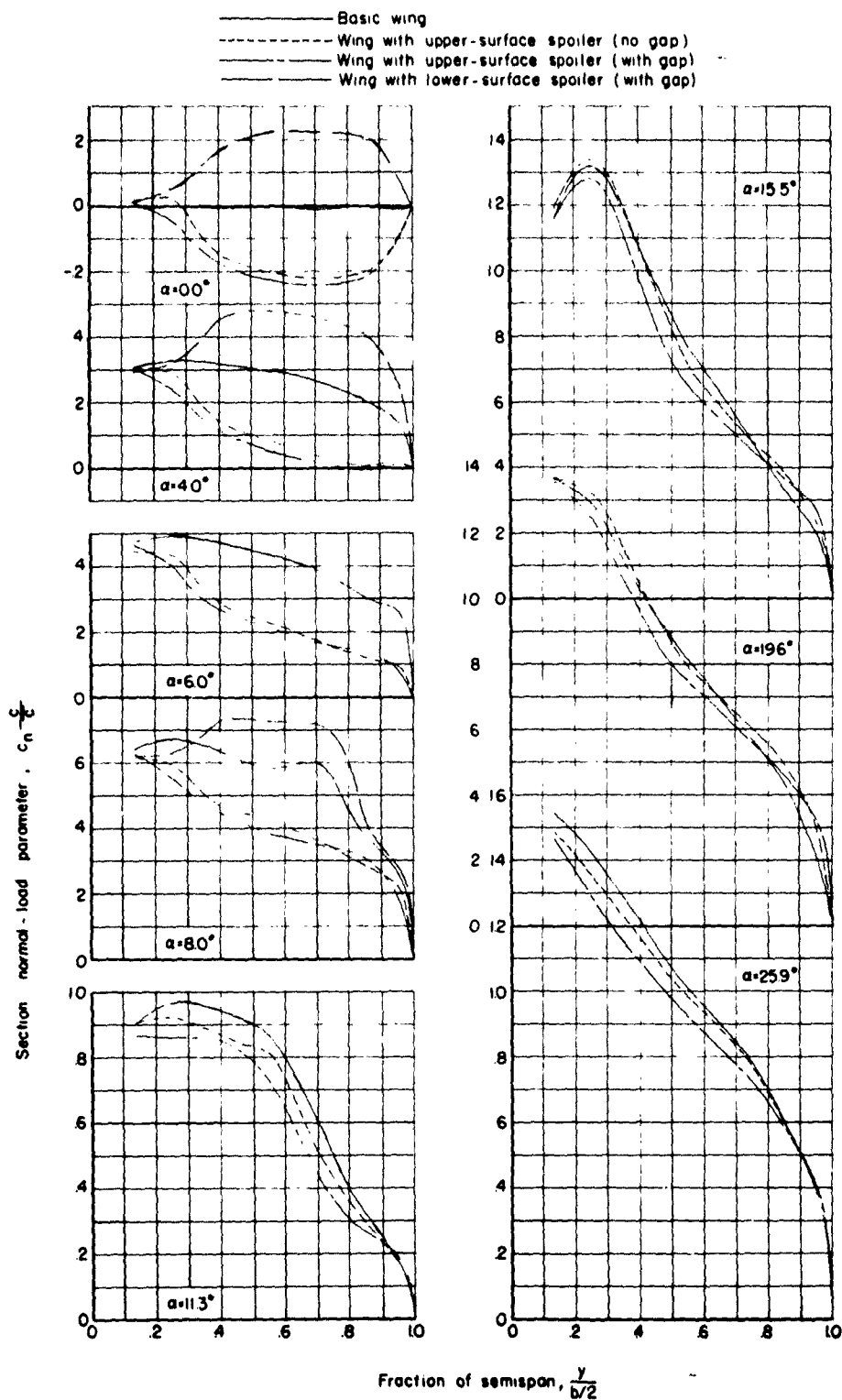
(c) $M_\infty = 0.85$.

Figure 8.- Continued.

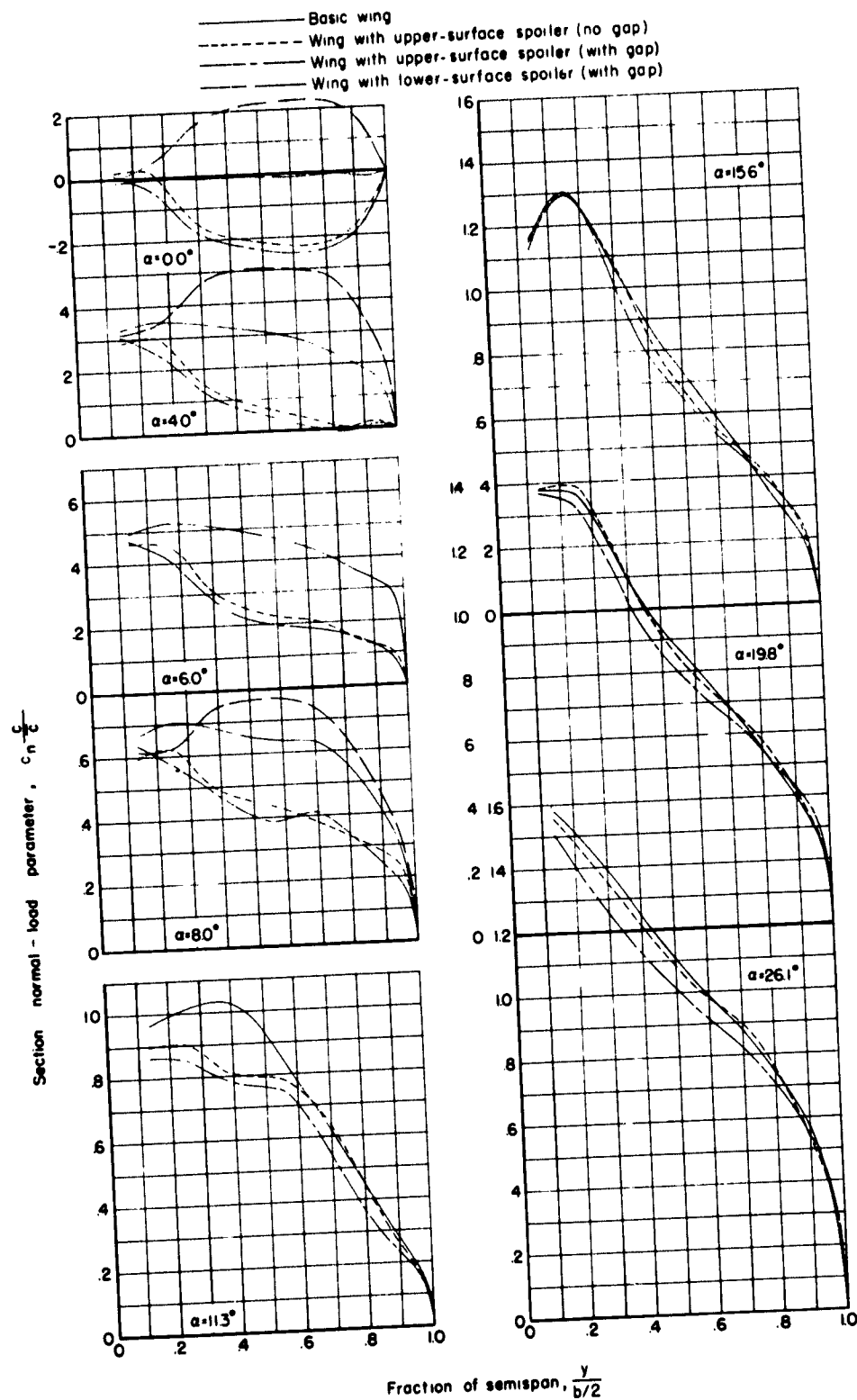
(d) $M = 0.90$.

Figure 8.- Continued.

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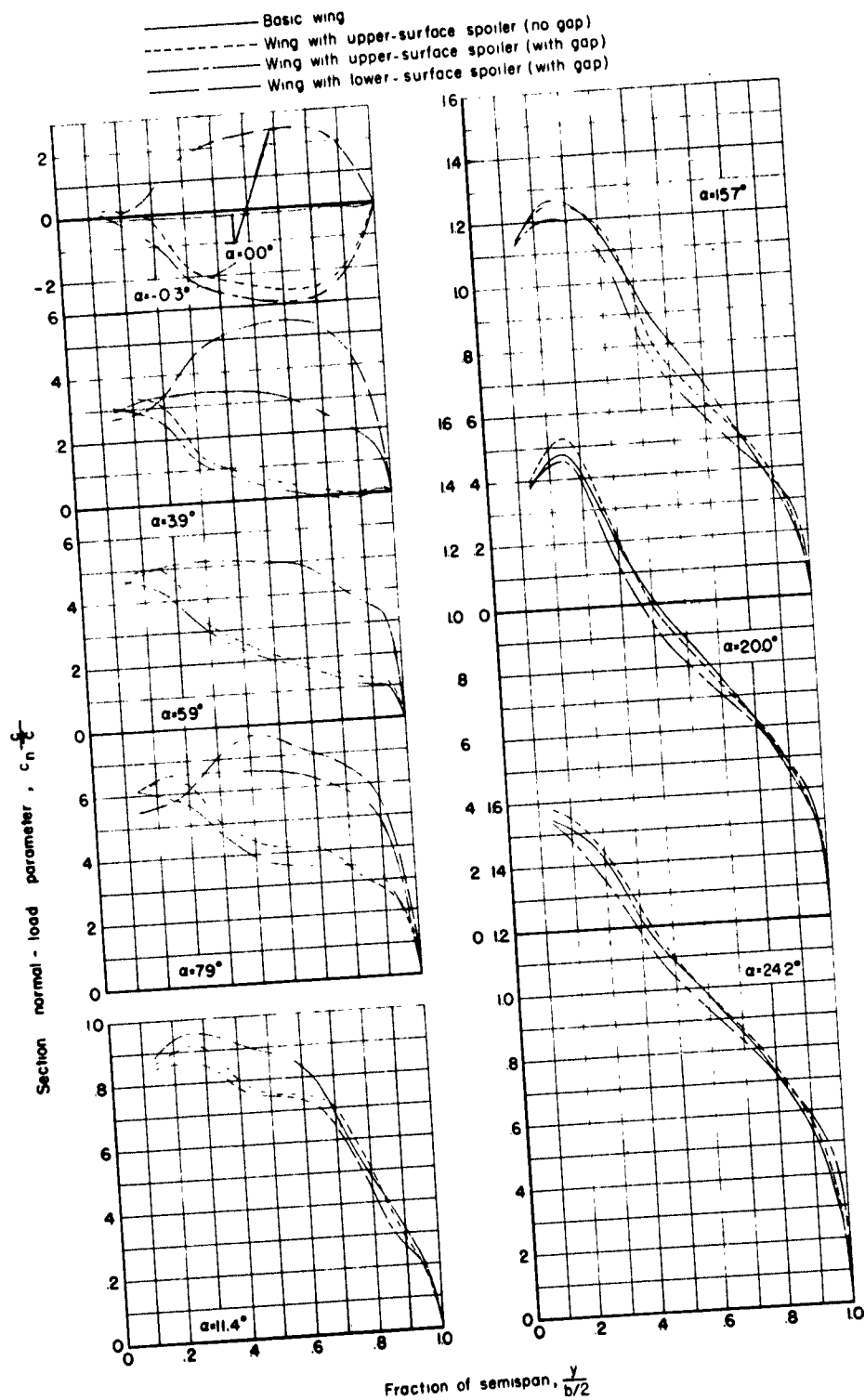
(e) $M = 0.94$.

Figure 8.- Continued.

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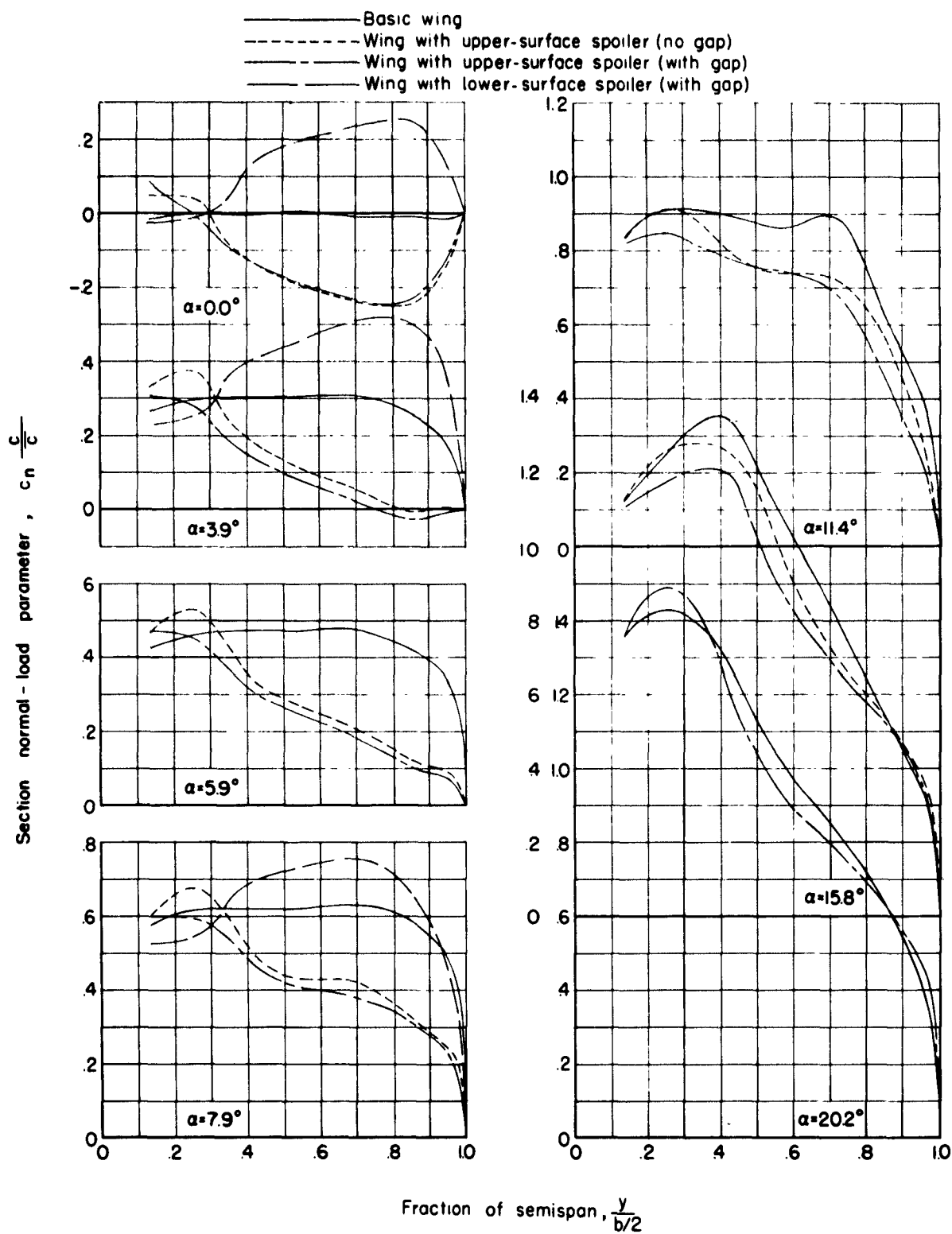
(f) $M = 0.98$.

Figure 8.- Continued.

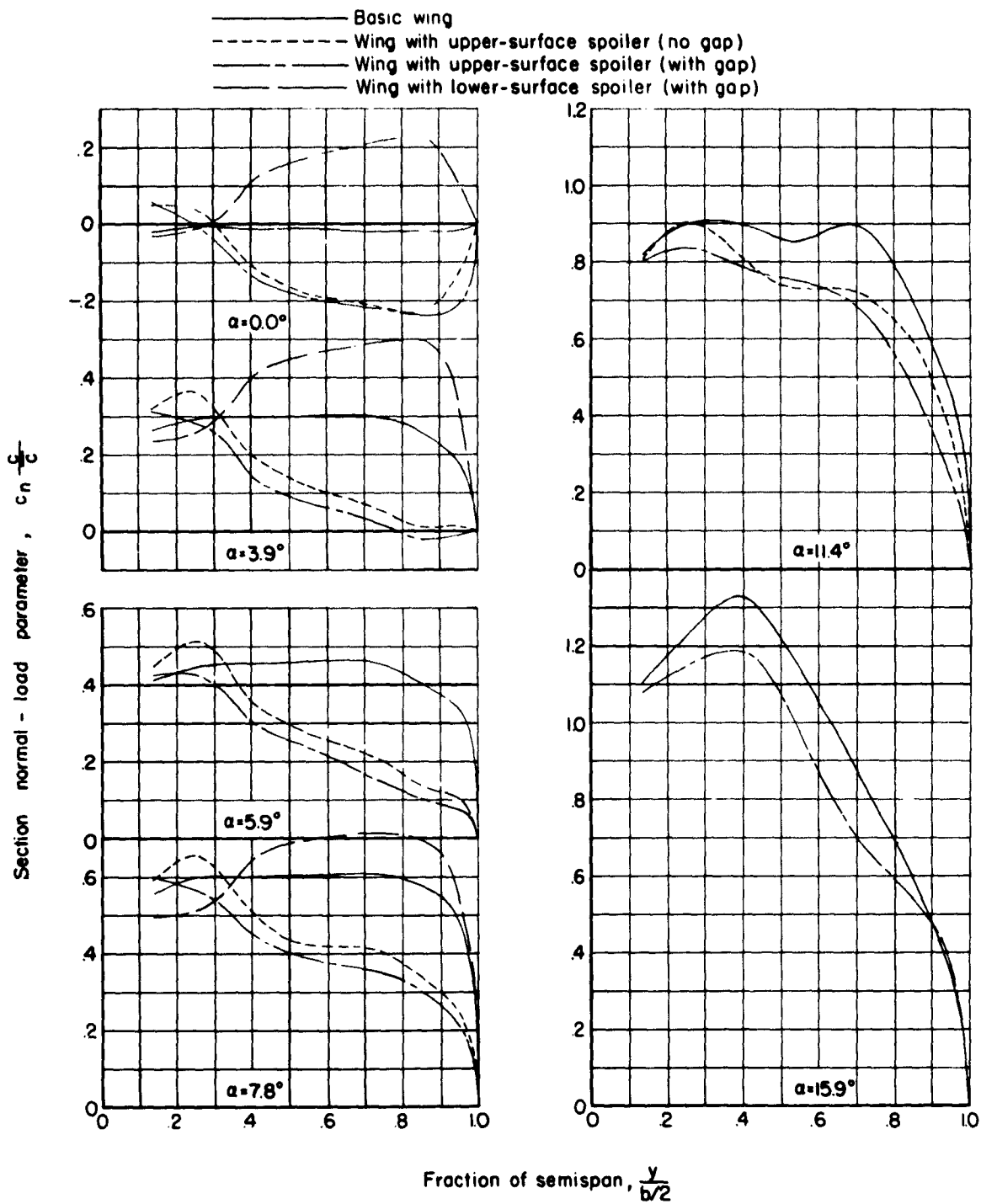
(g) $M = 1.00$.(g) $M = 1.00$.

Figure 8.- Continued.

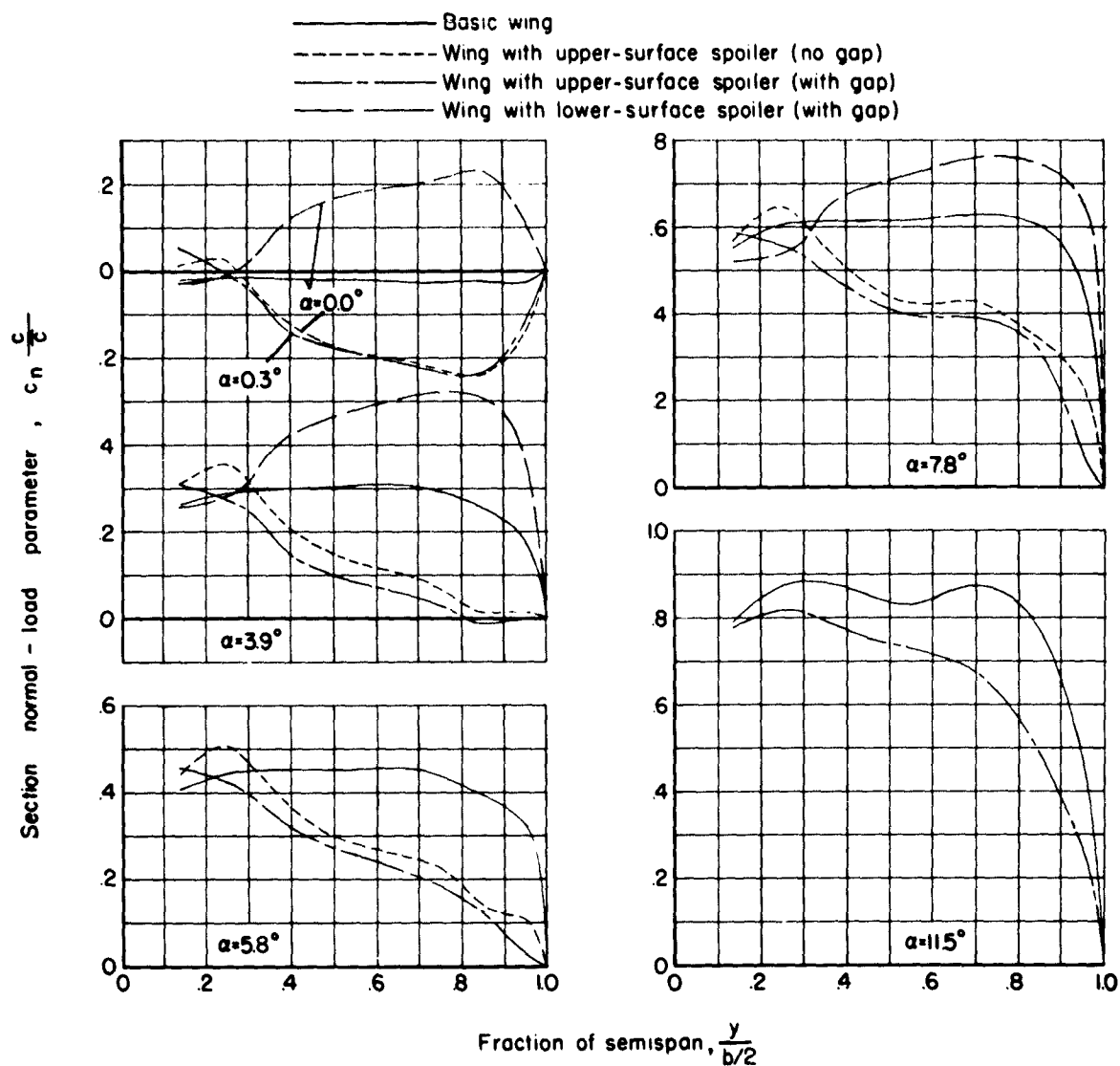
(h) $M = 1.03$.

Figure 8.- Concluded.

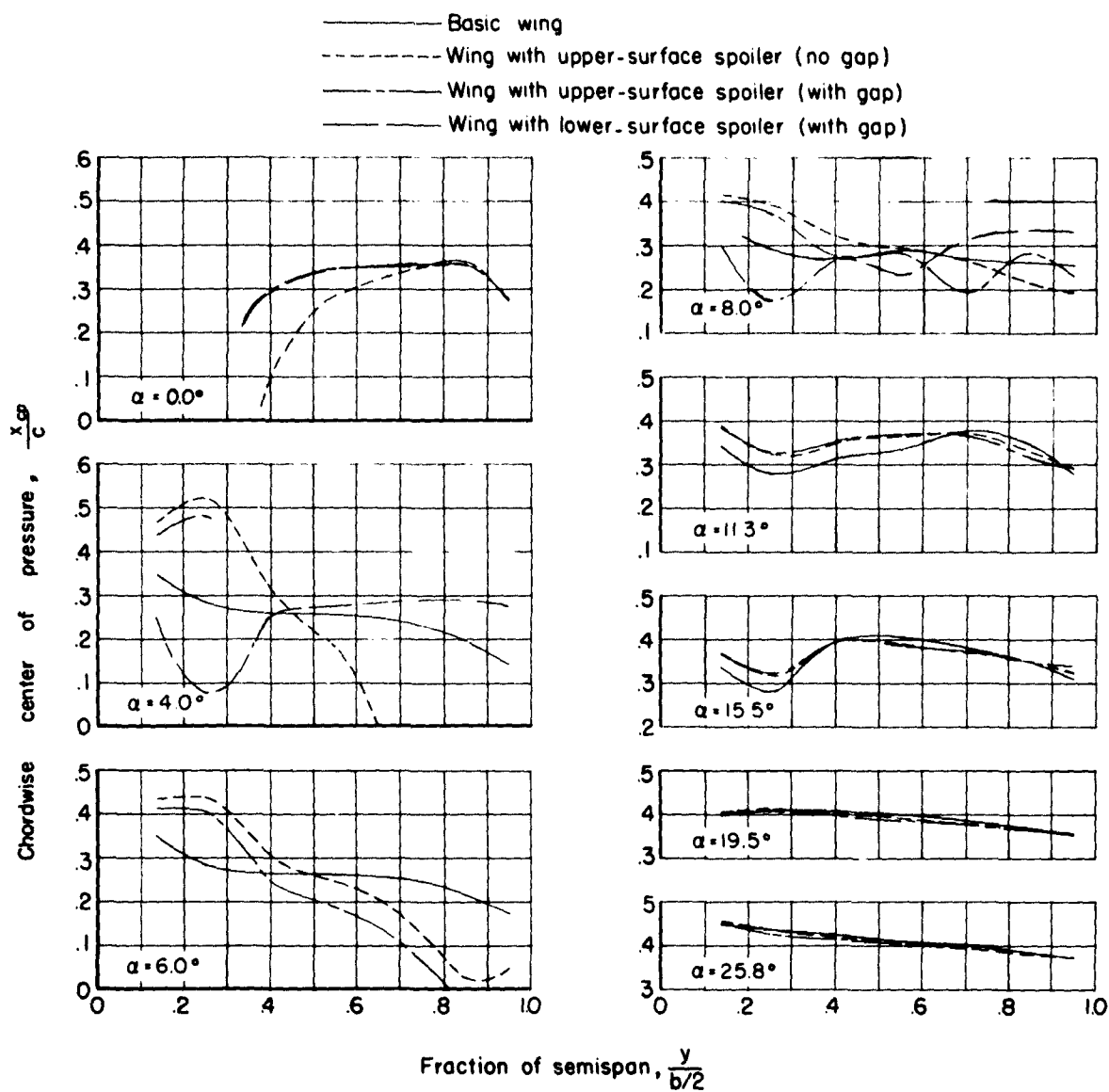
(a) $M = 0.60$.

Figure 9.- Wing-section center of pressure across the semispan for the basic wing and various spoiler configurations.

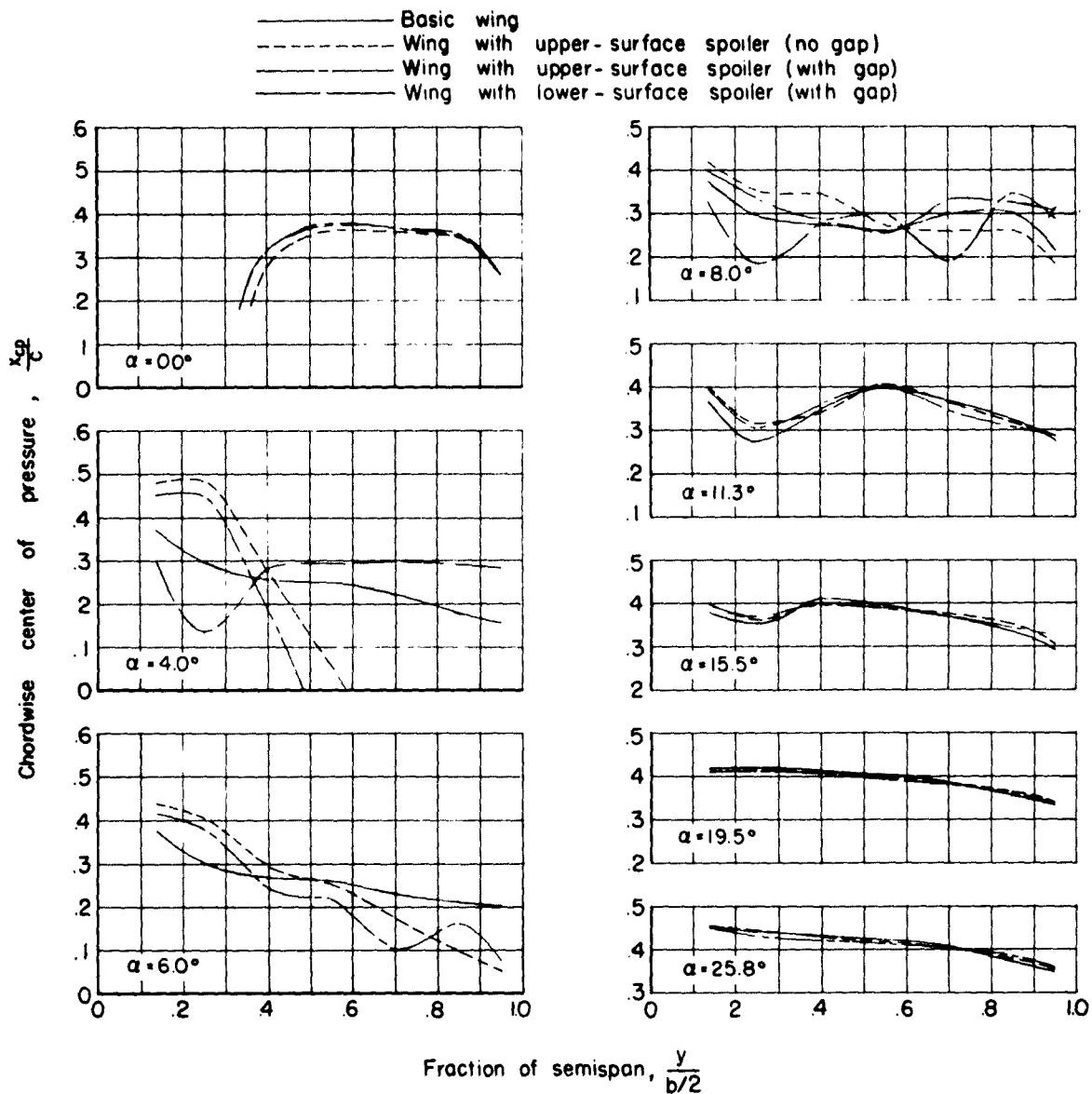
(b) $M = 0.80$.

Figure 9.- Continued.

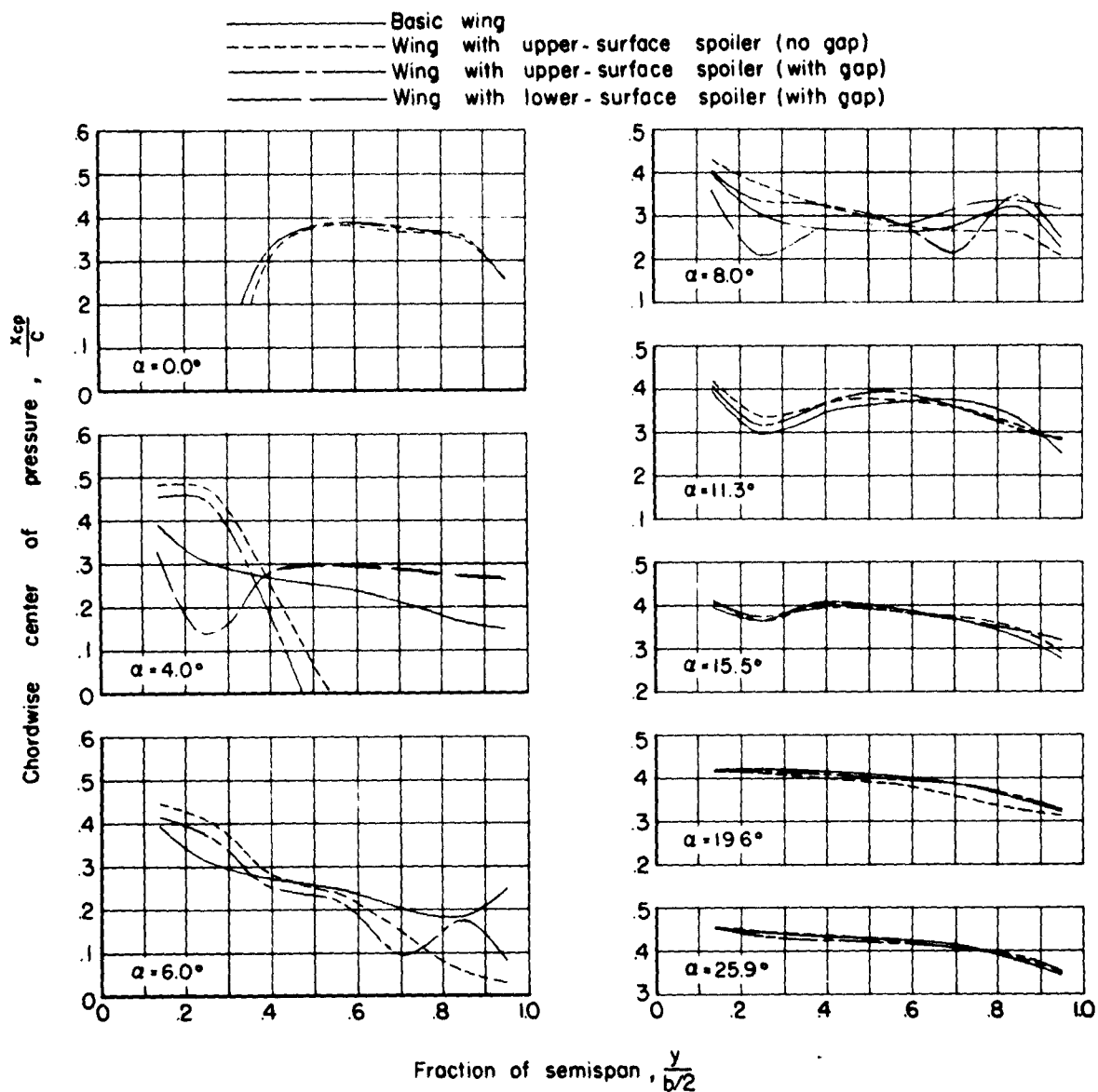
(c) $M = 0.85$.

Figure 9.- Continued.

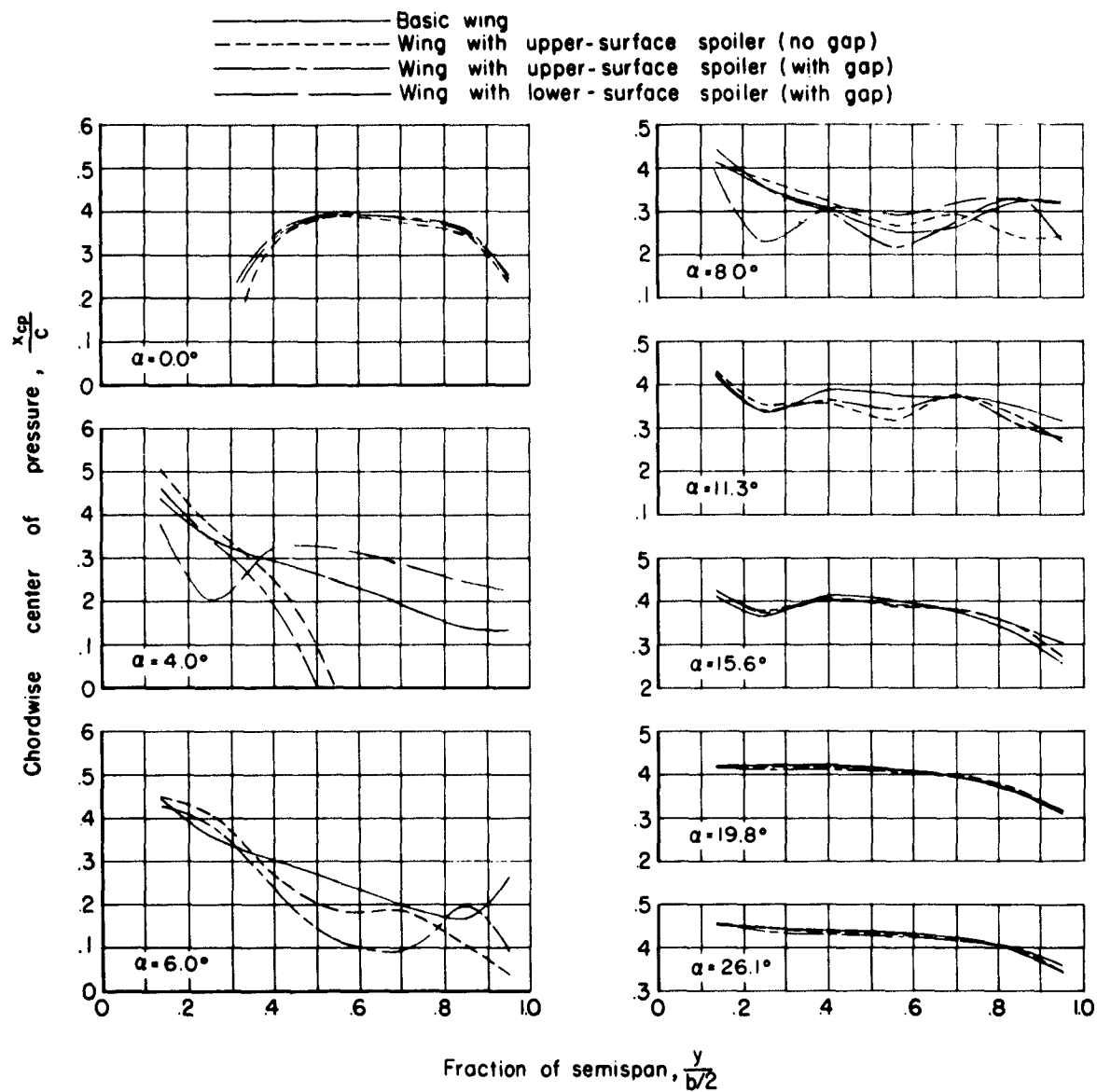
(d) $M = 0.90$.

Figure 9.- Continued.

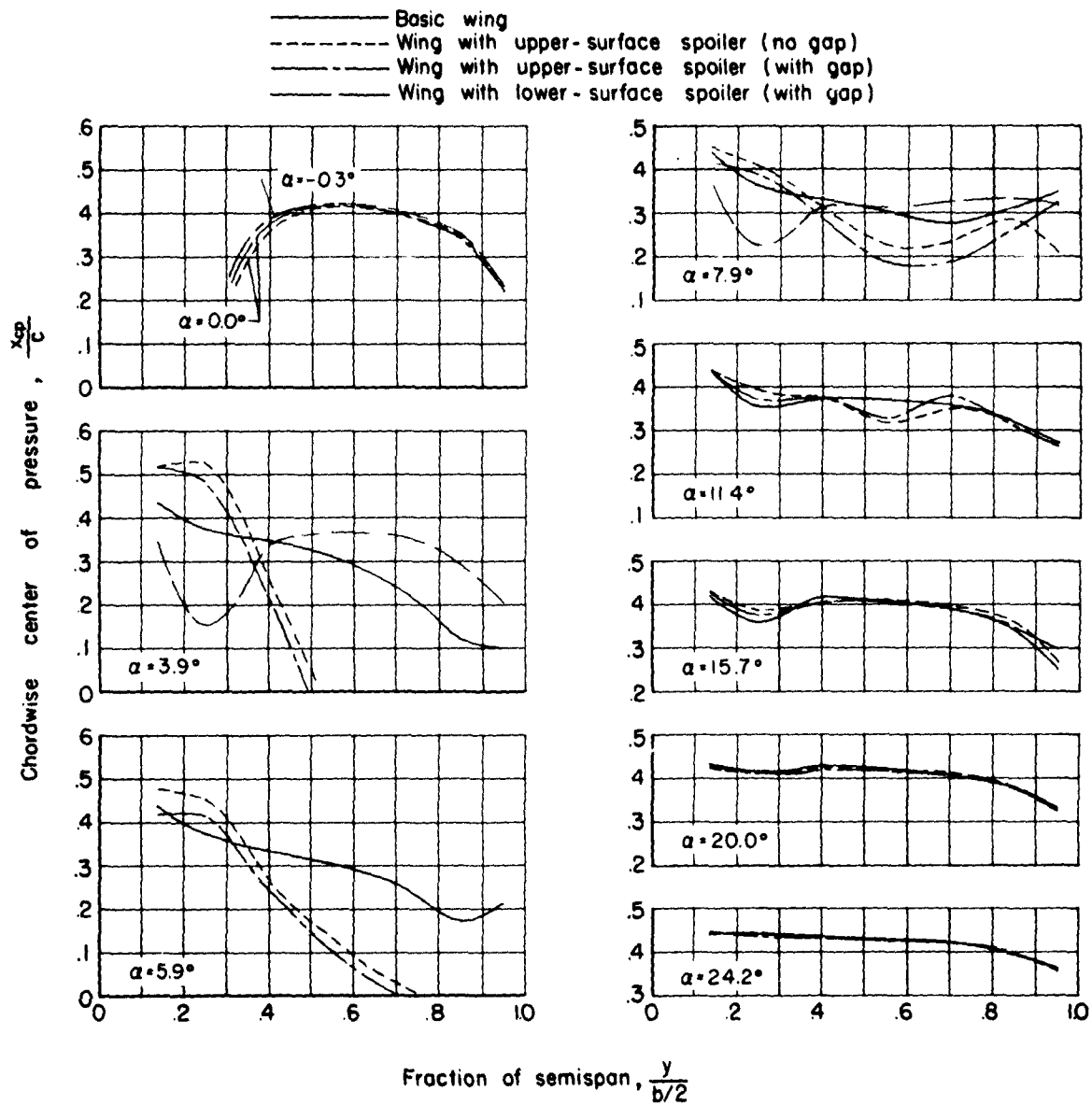
(e) $M = 0.94$.

Figure 9.- Continued.

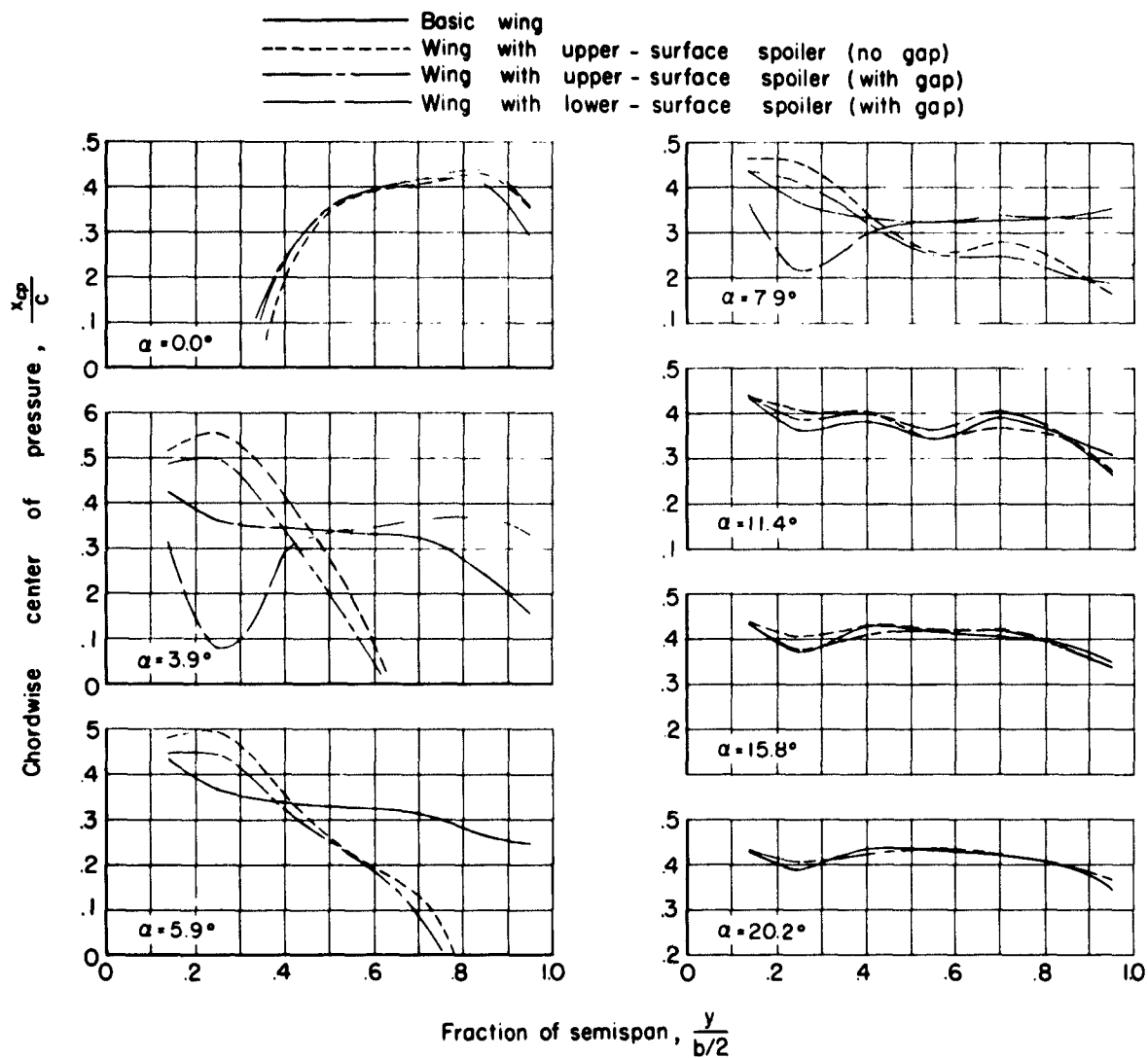
(f) $M = 0.98$.

Figure 9.- Continued.

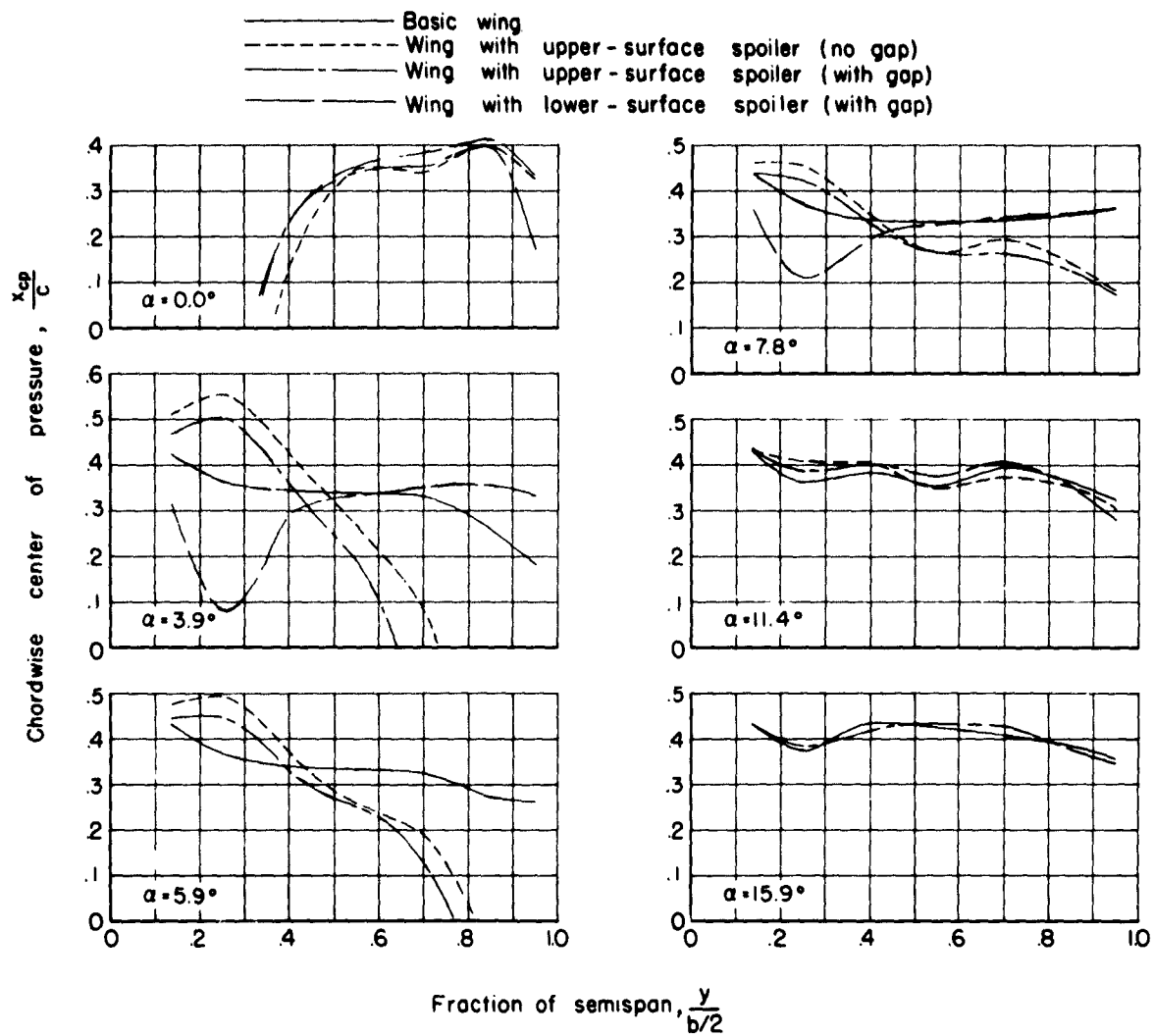
(g) $M = 1.00$.

Figure 9.- Continued.

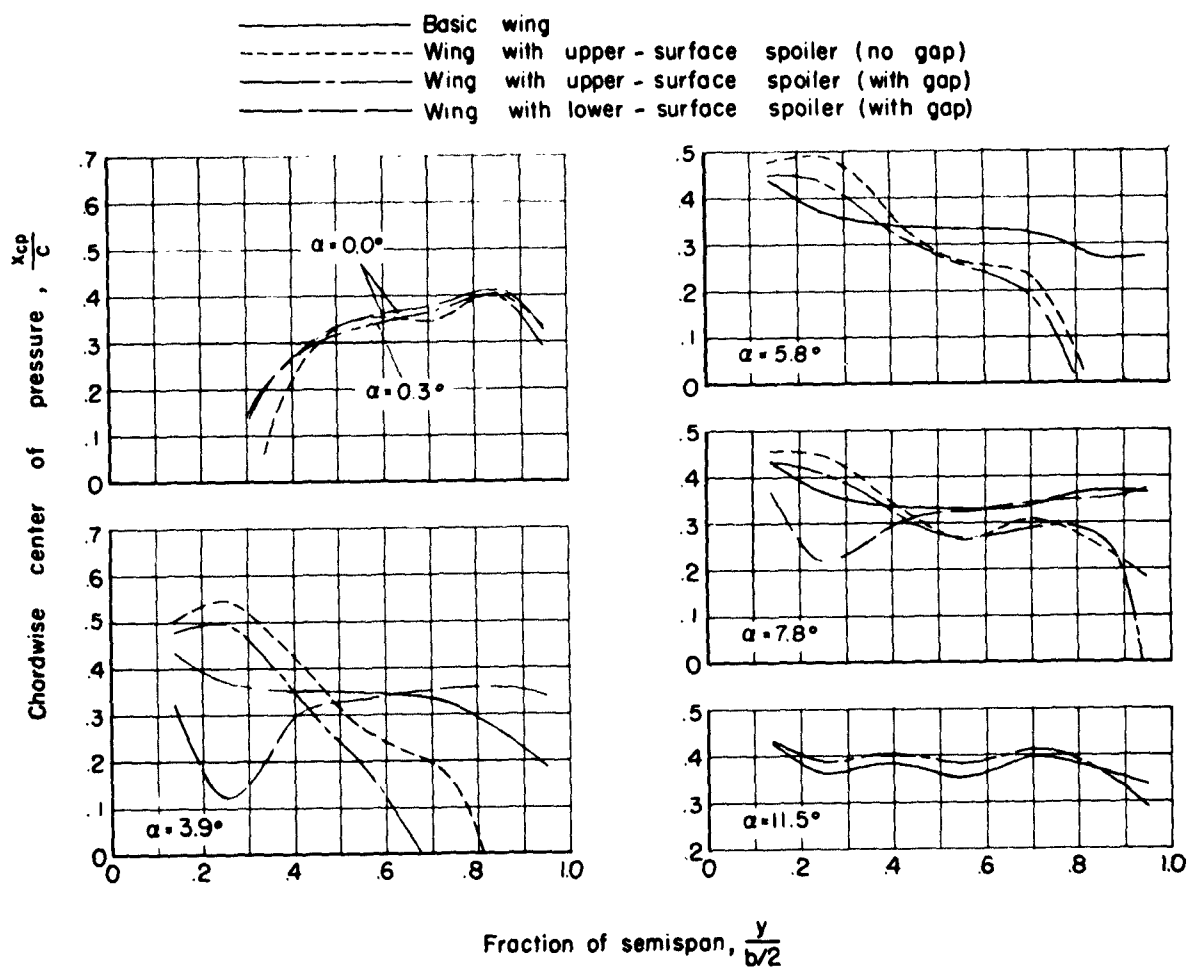
(h) $M = 1.03$.

Figure 9.- Concluded.

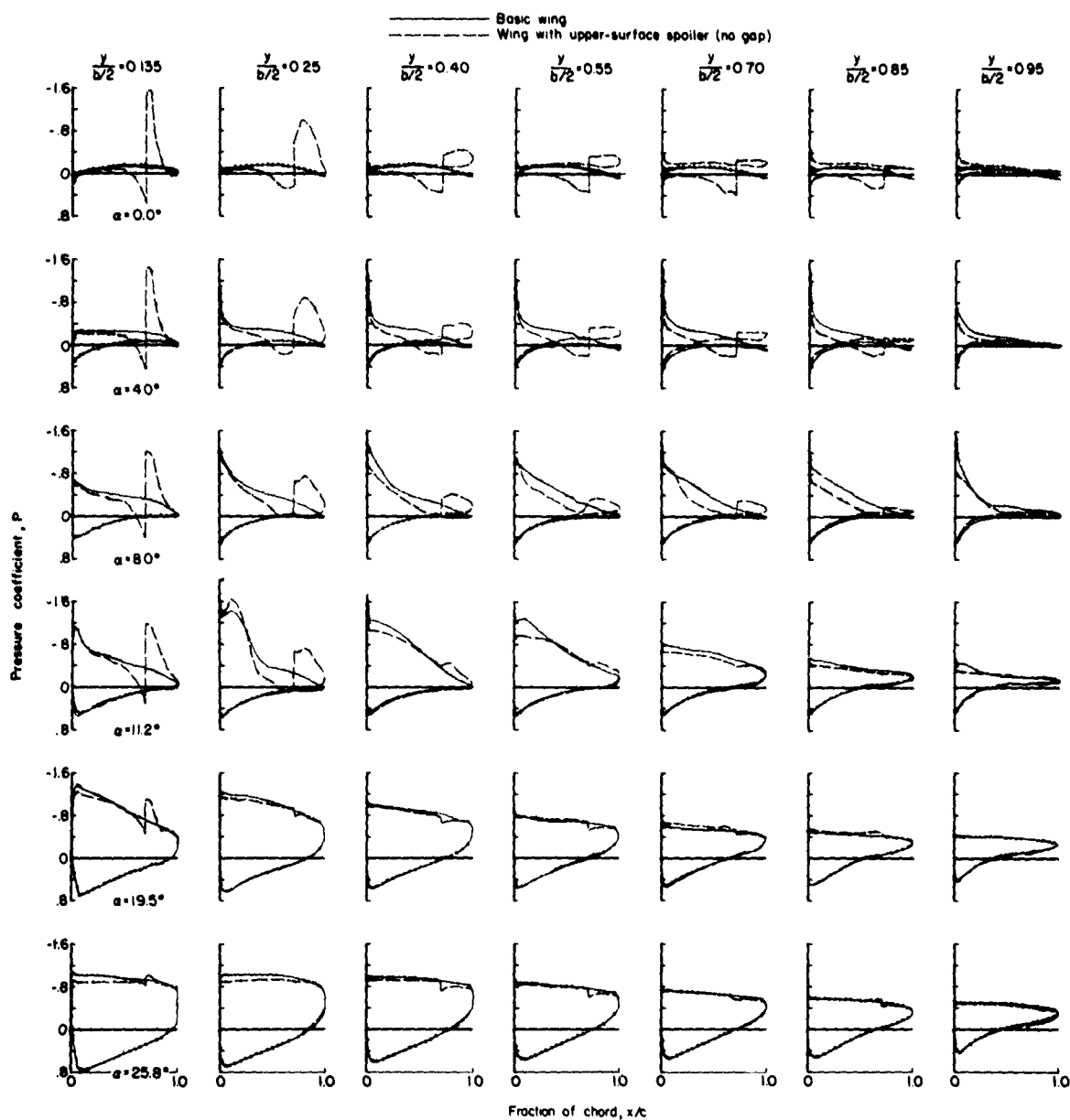
(a) $M = 0.60$.

Figure 10.- Chordwise pressure distributions on the wing; basic wing compared with the upper-surface spoiler (no gap) configuration.

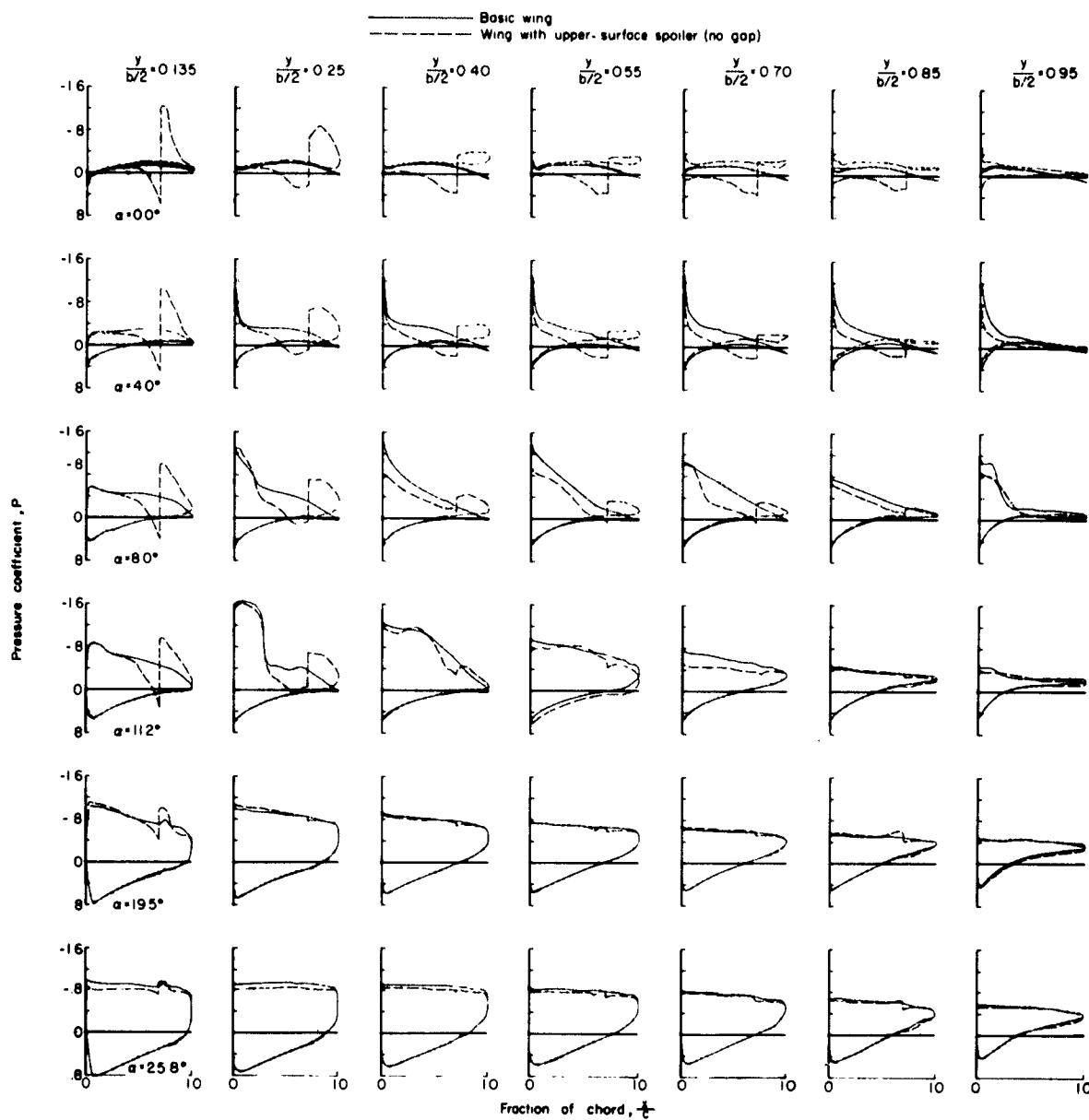
(b) $M = 0.80$.

Figure 10.- Continued.

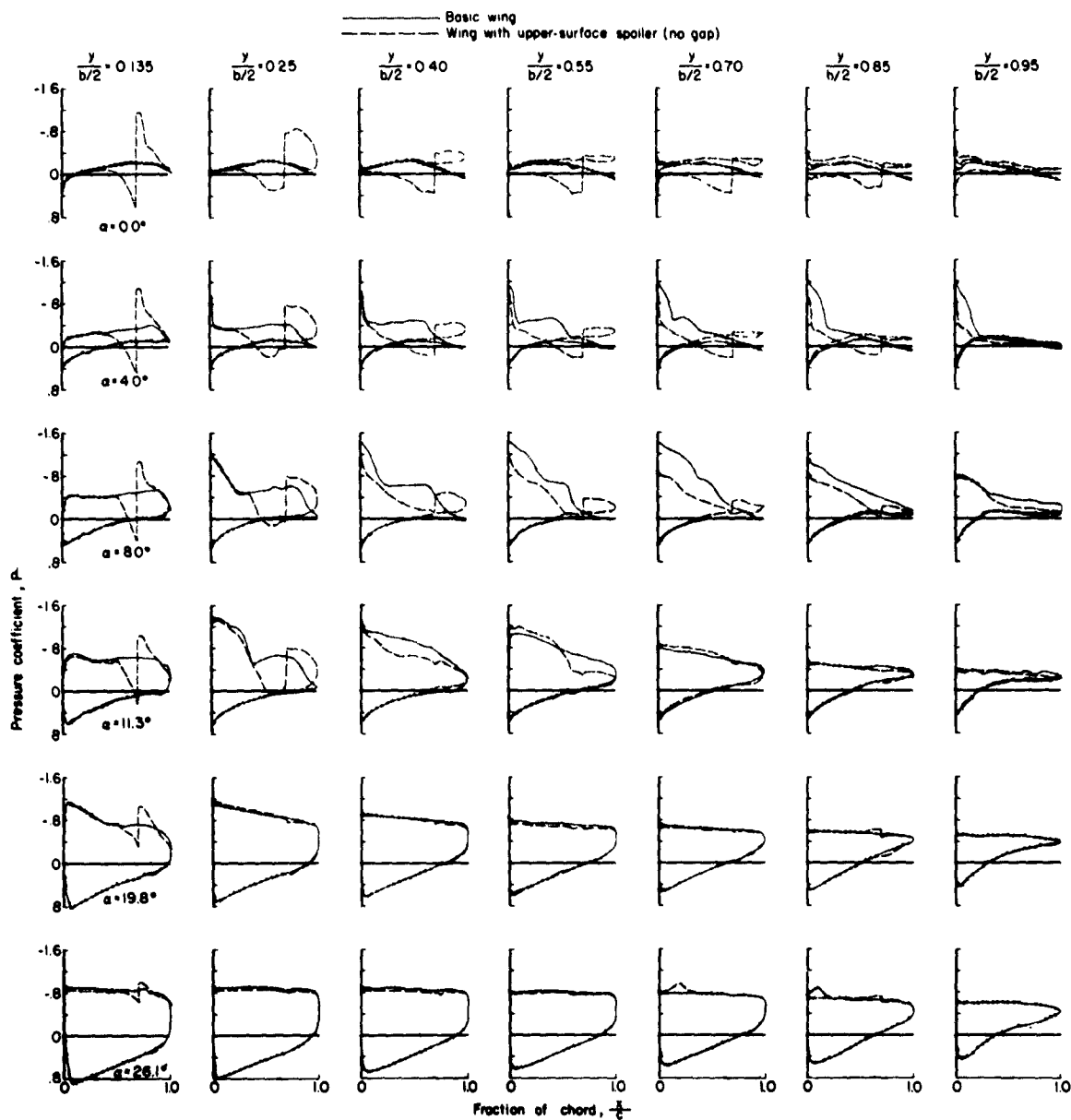
(c) $M = 0.90$.

Figure 10.- Continued.

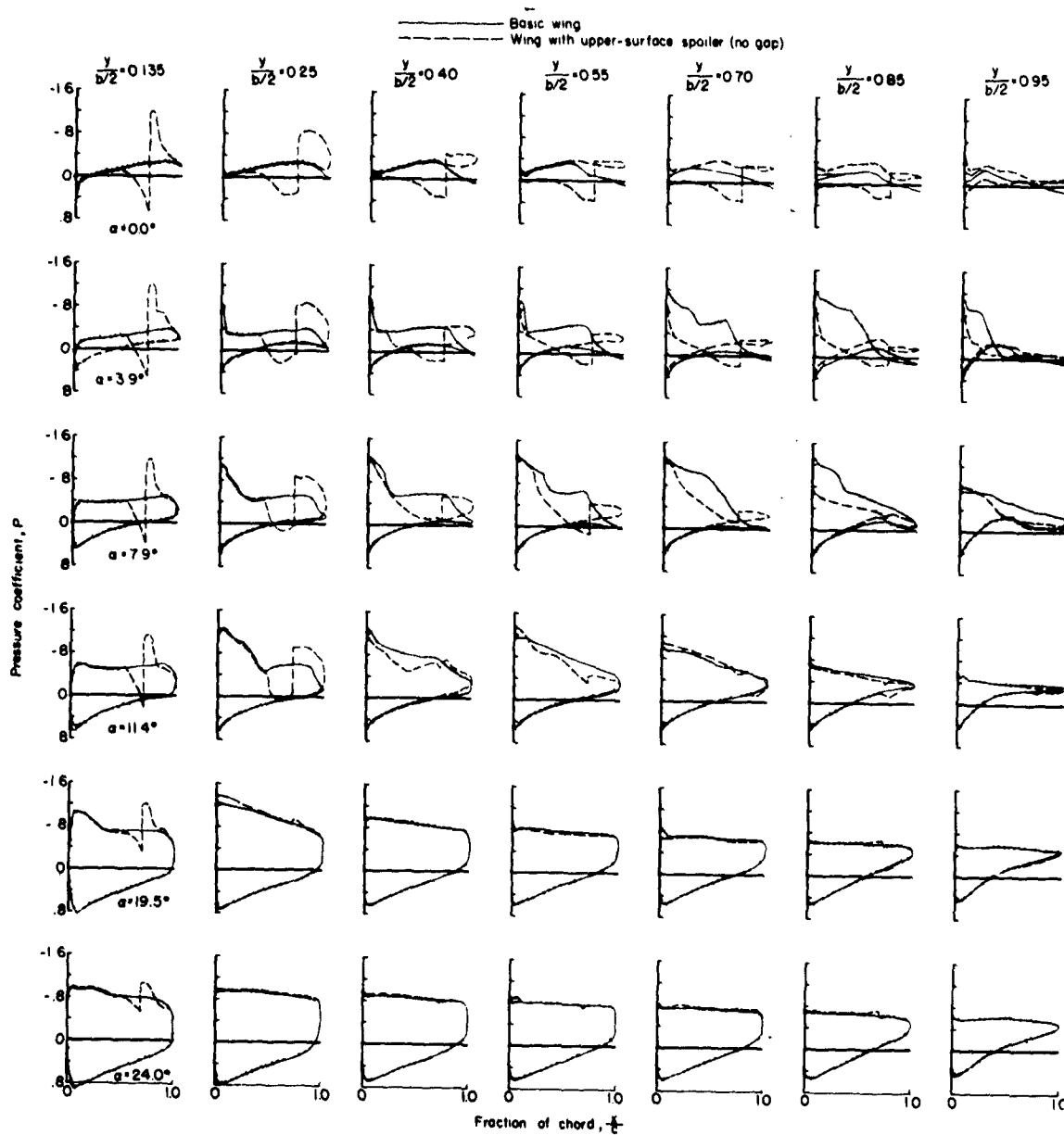
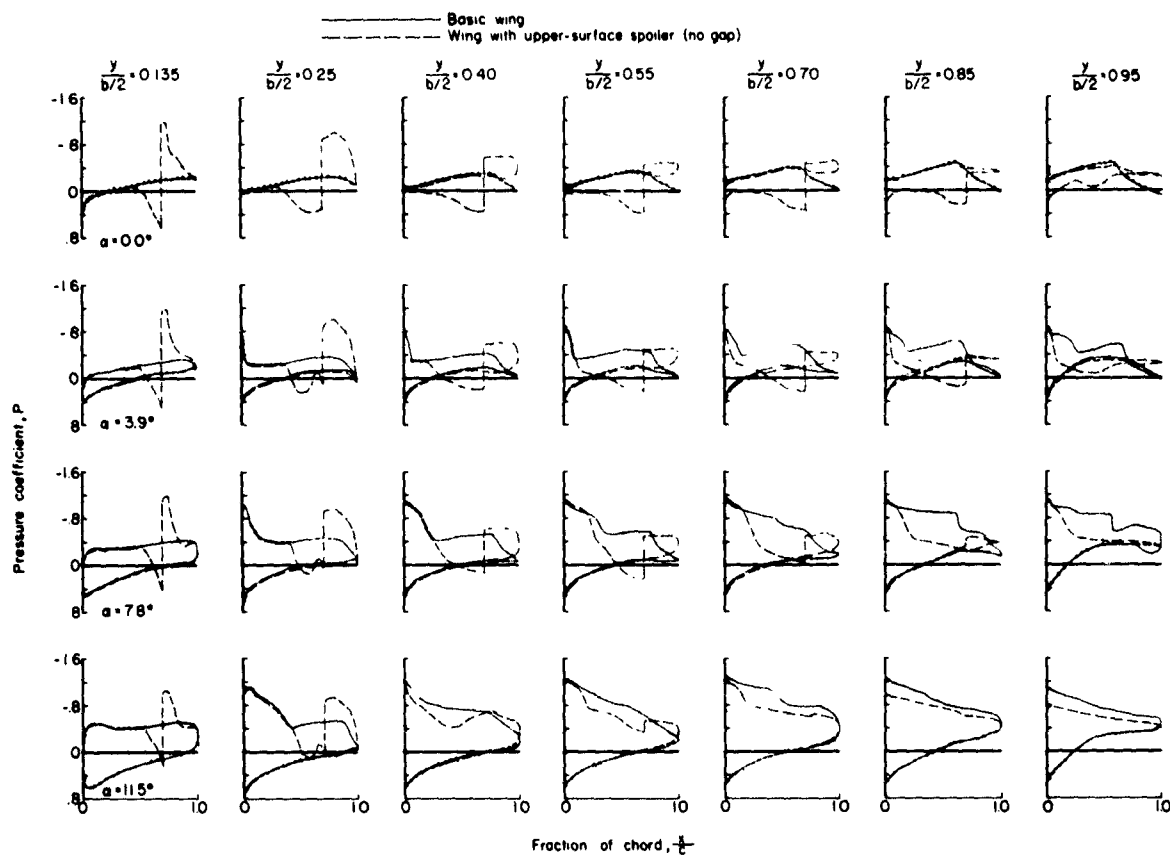
(d) $M = 0.94$.

Figure 10.- Continued.



(e) $M = 1.00$.

Figure 10.- Concluded.

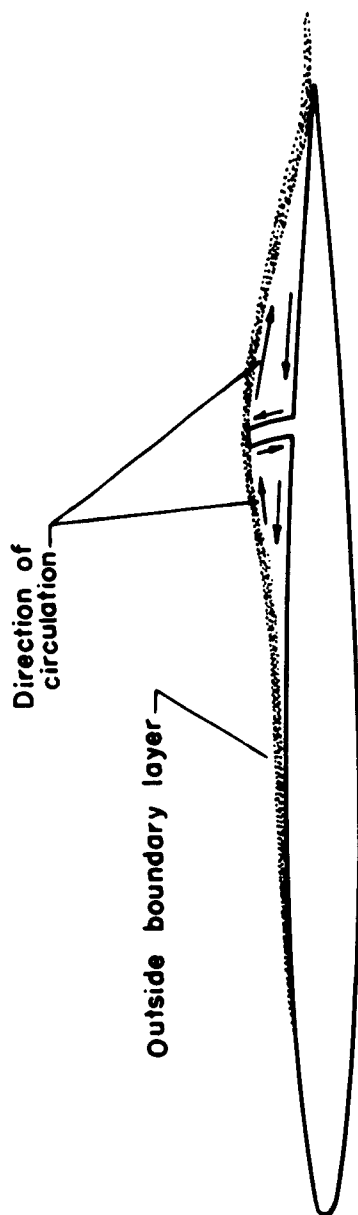


Figure 11.- Concept of flow in the boundary layer at a typical wing-spoiler section for a Mach number of 0.60.

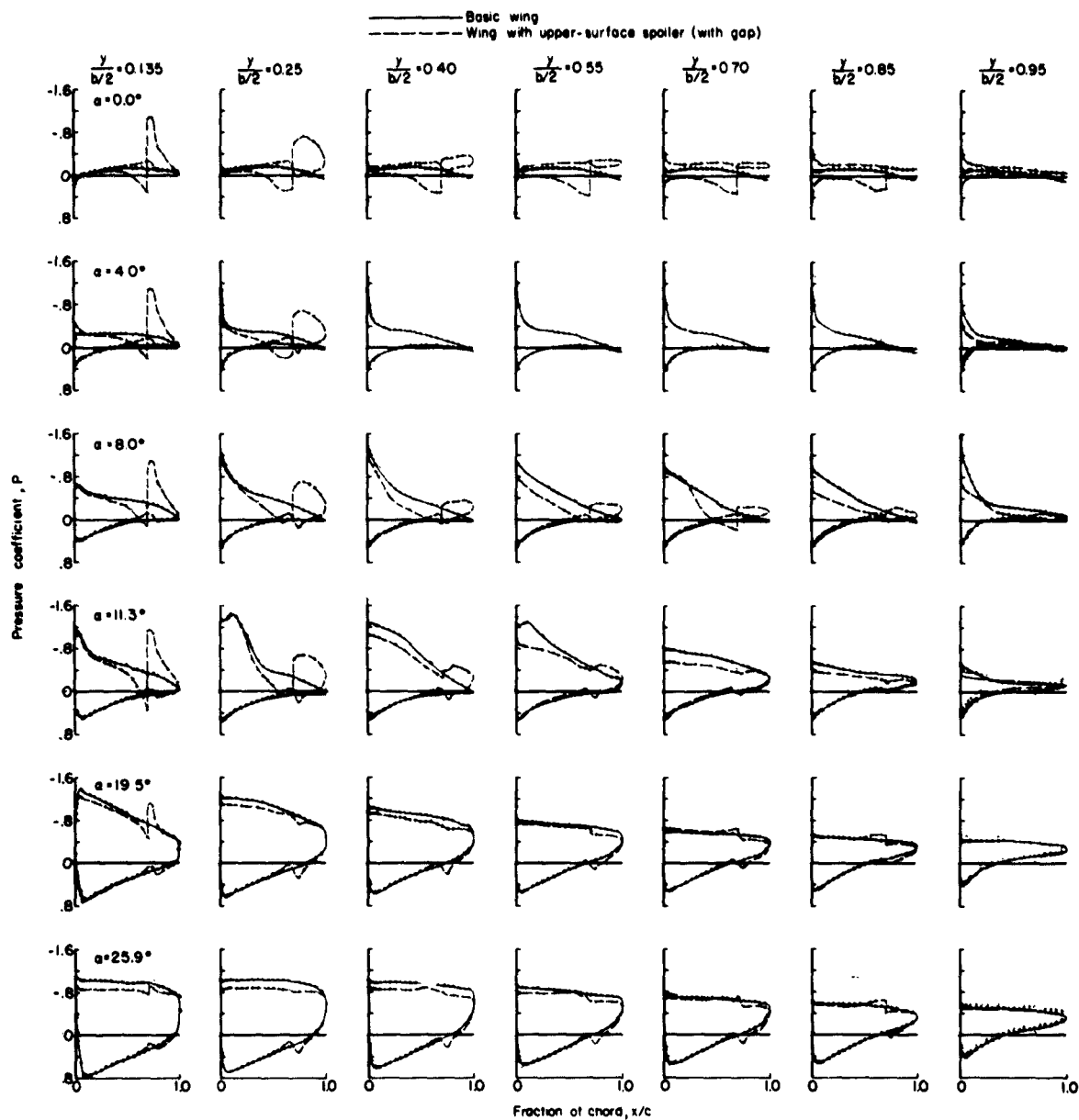
(a) $M = 0.60$.

Figure 12.- Chordwise pressure distributions on the wing; basic wing compared with the upper-surface spoiler (with gap) configuration.

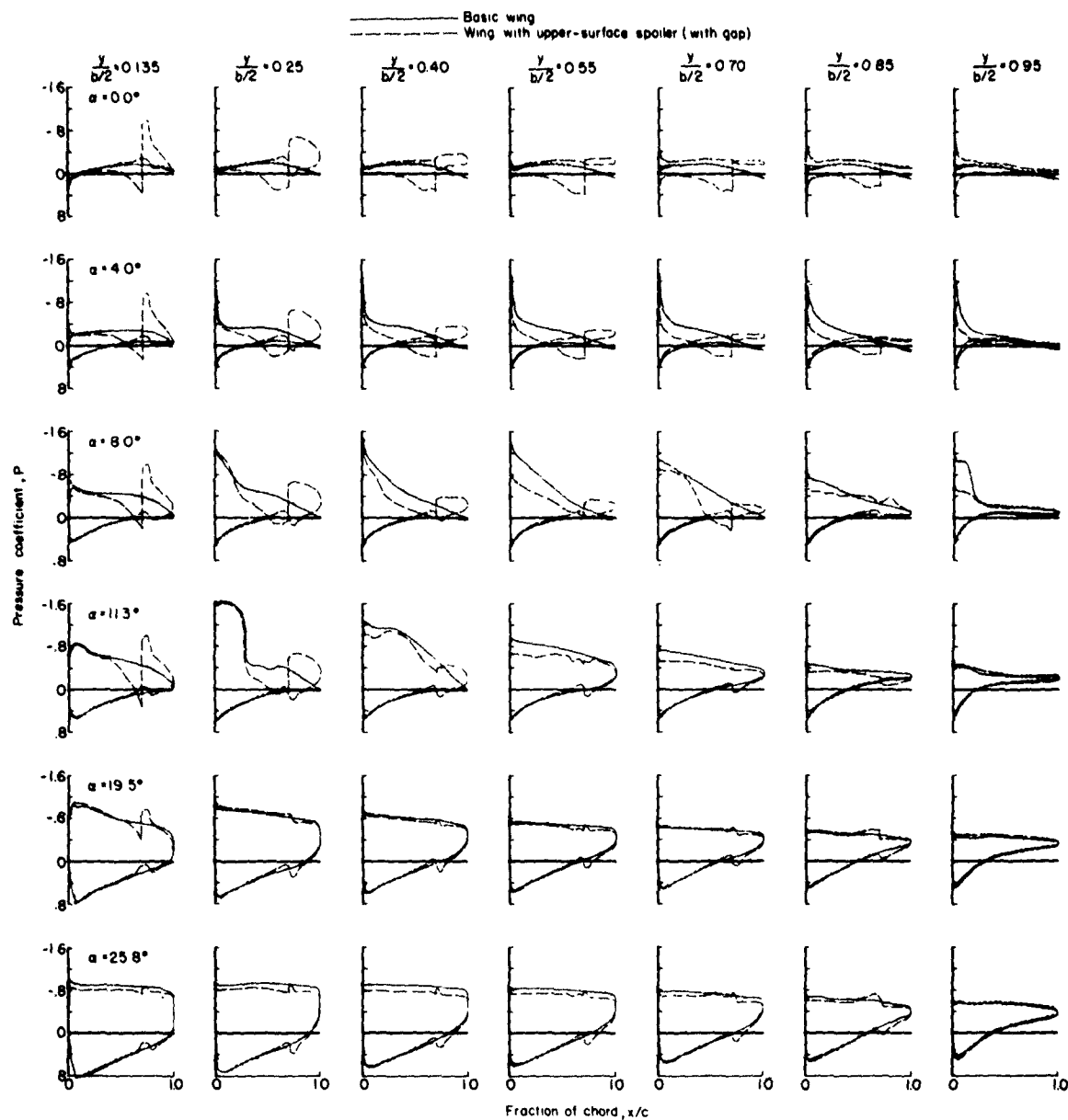
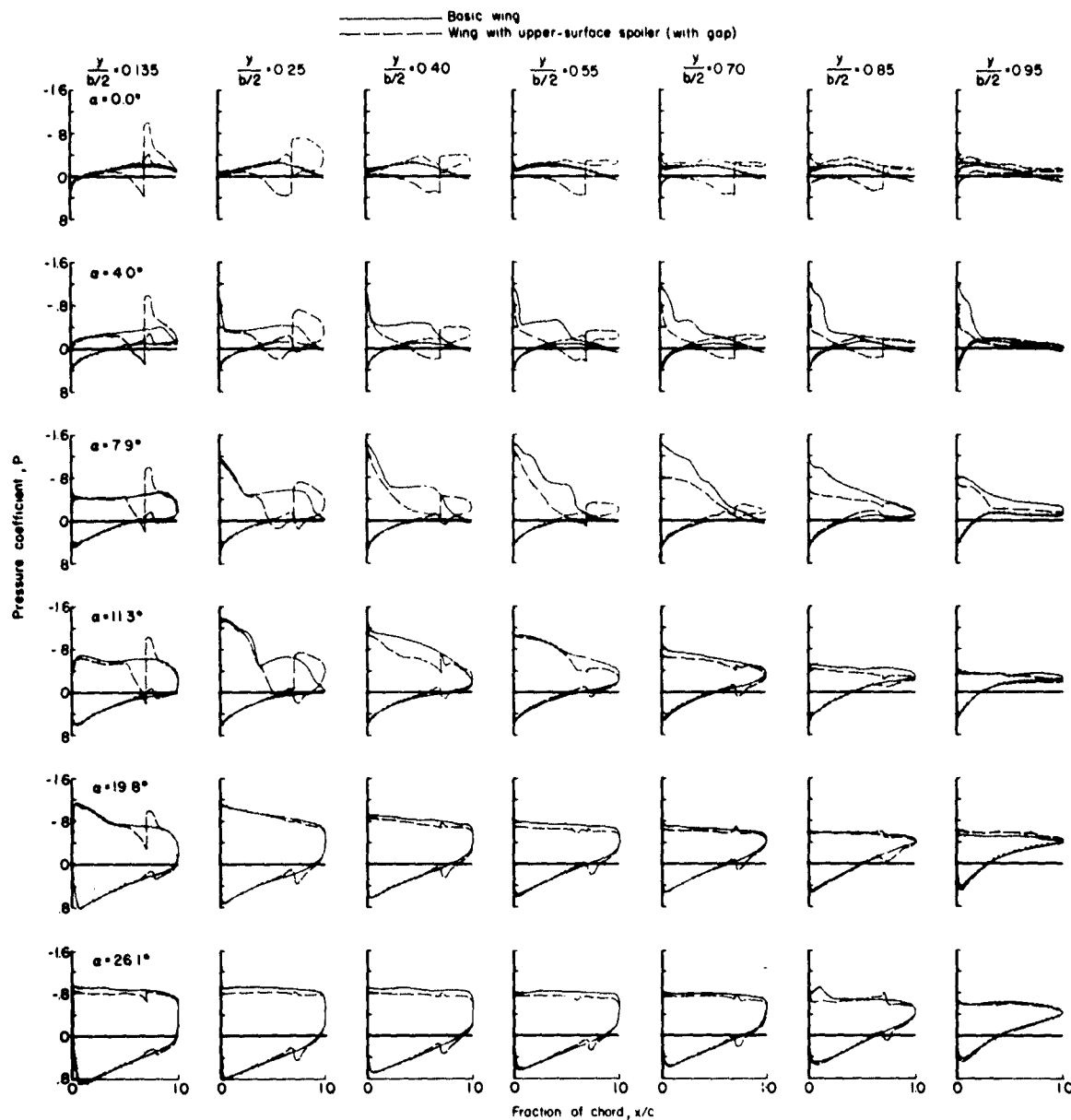
(b) $M = 0.80$.

Figure 12.- Continued.



(c) $M = 0.90$.

Figure 12.- Continued.

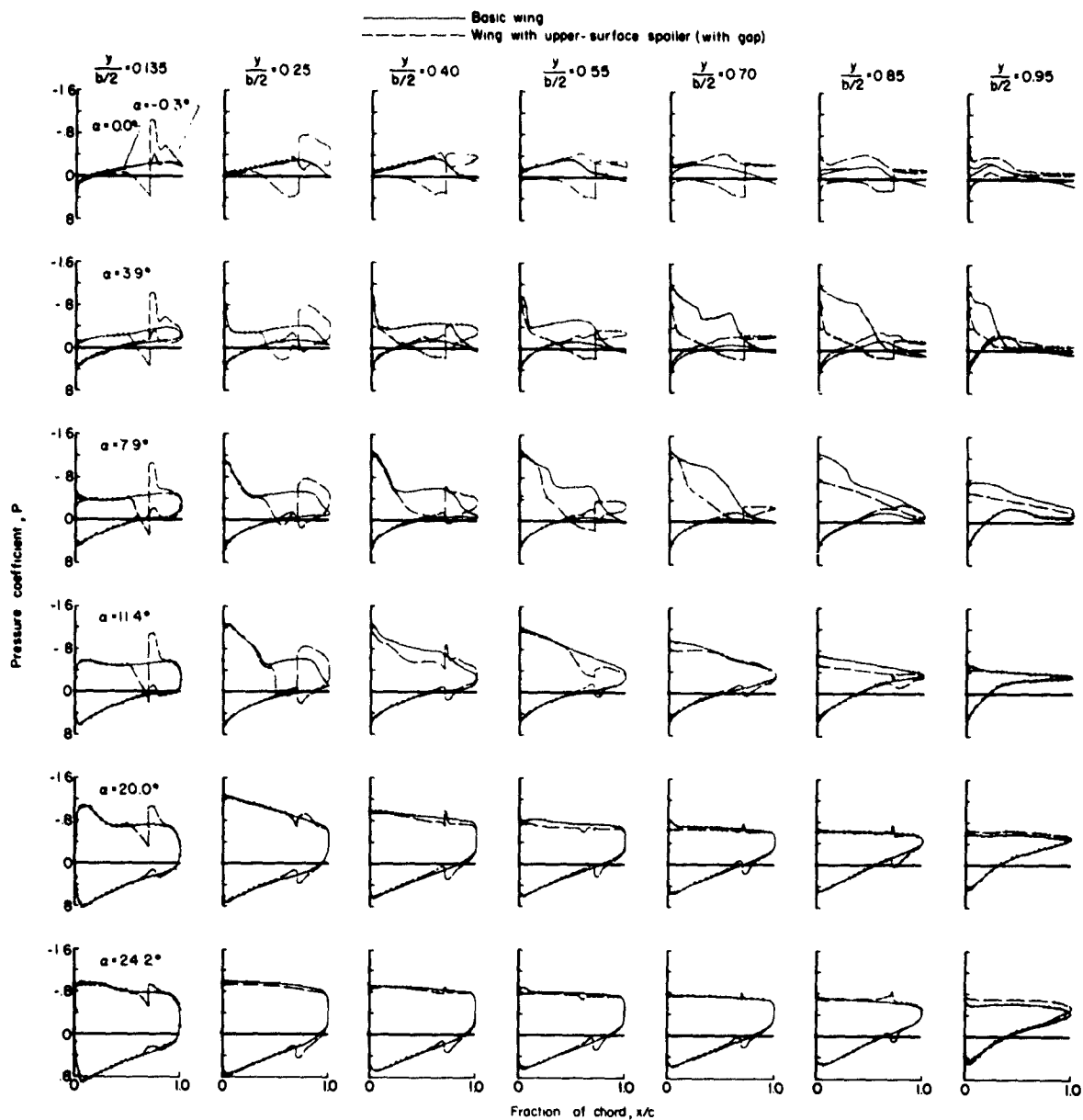
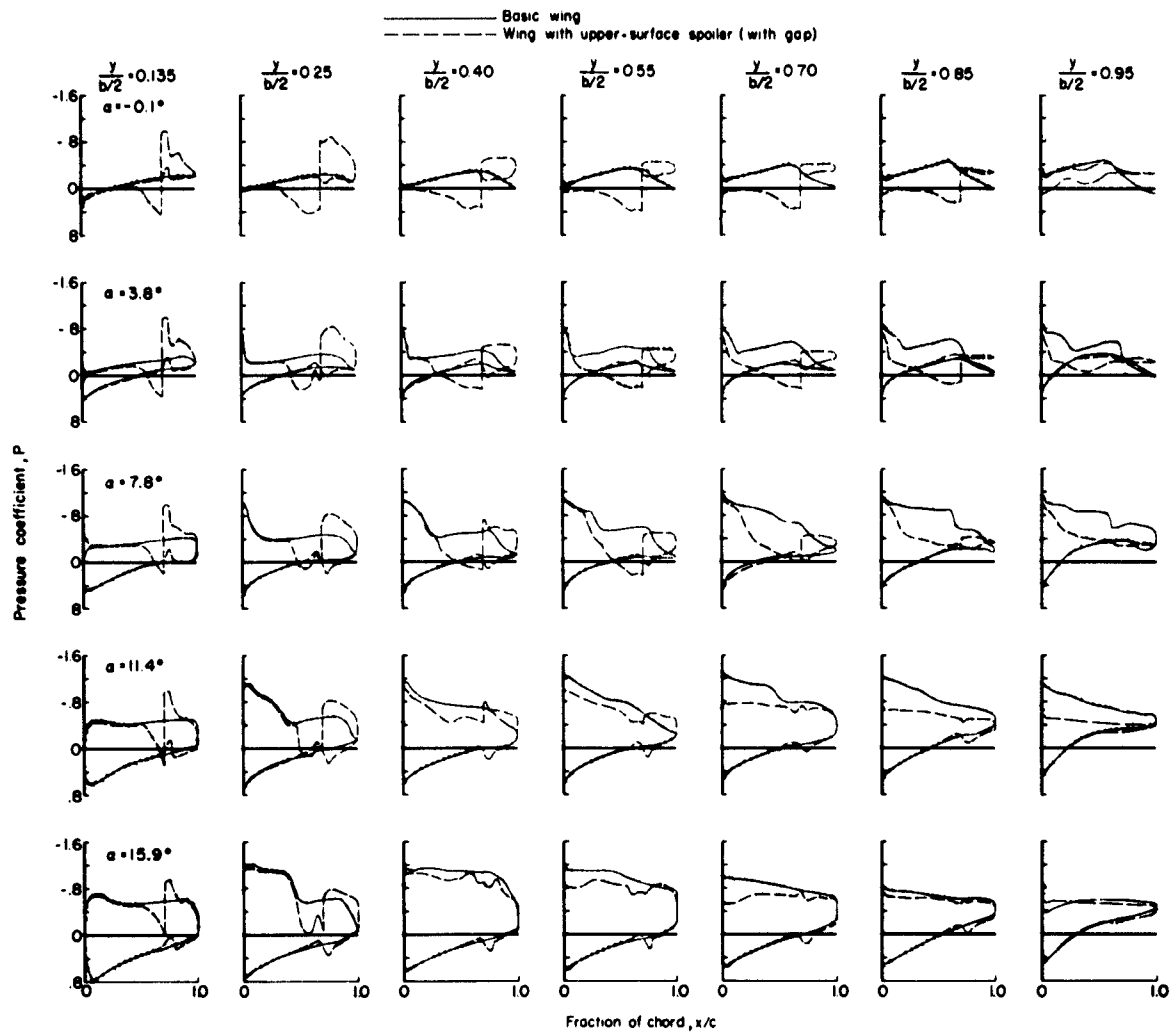
(d) $M = 0.94$.

Figure 12.- Continued.



(e) $M = 1.00$.

Figure 12.- Concluded.

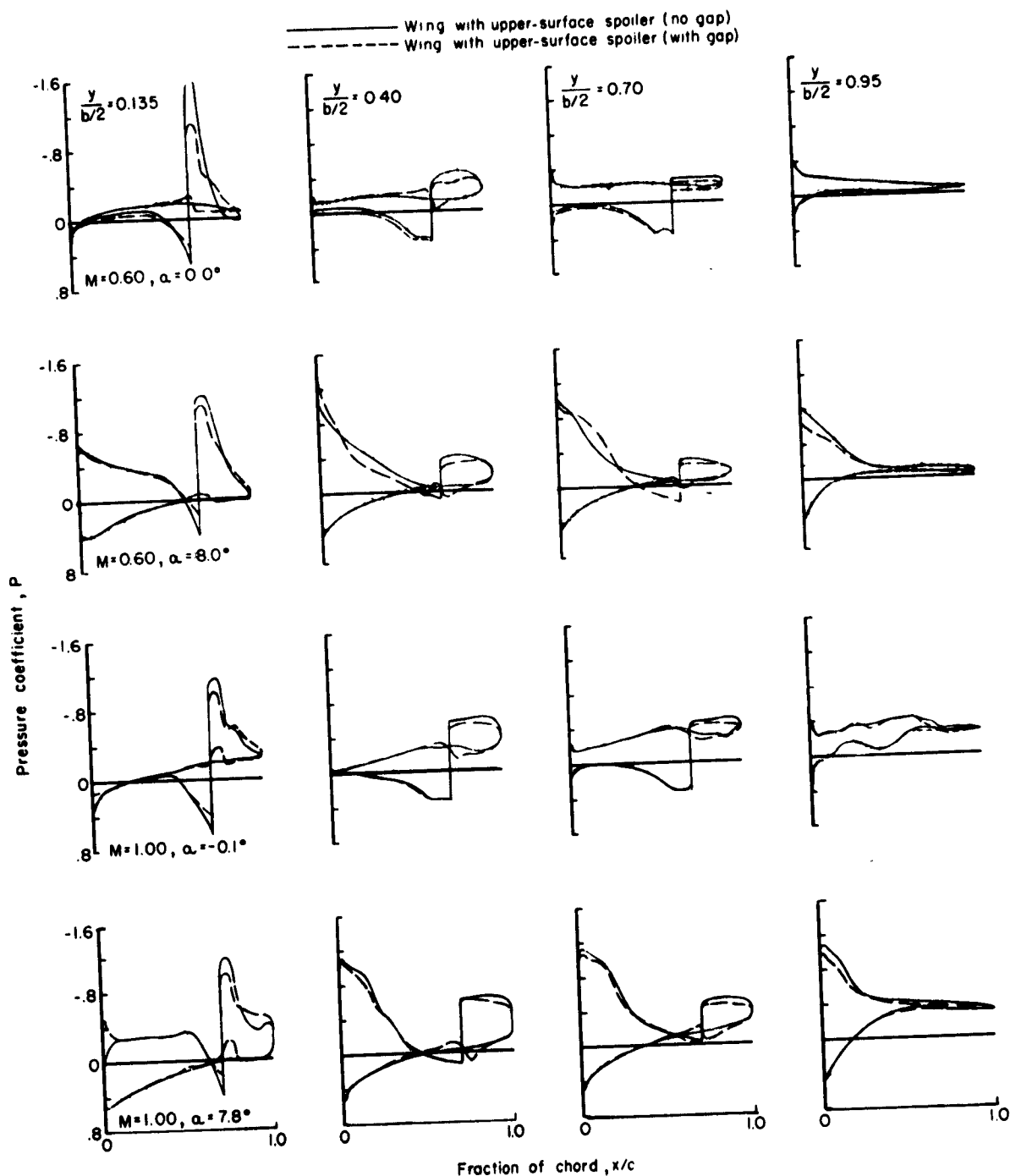


Figure 13.- Effect of wing gap behind an upper-surface spoiler on wing chordwise pressure distributions.

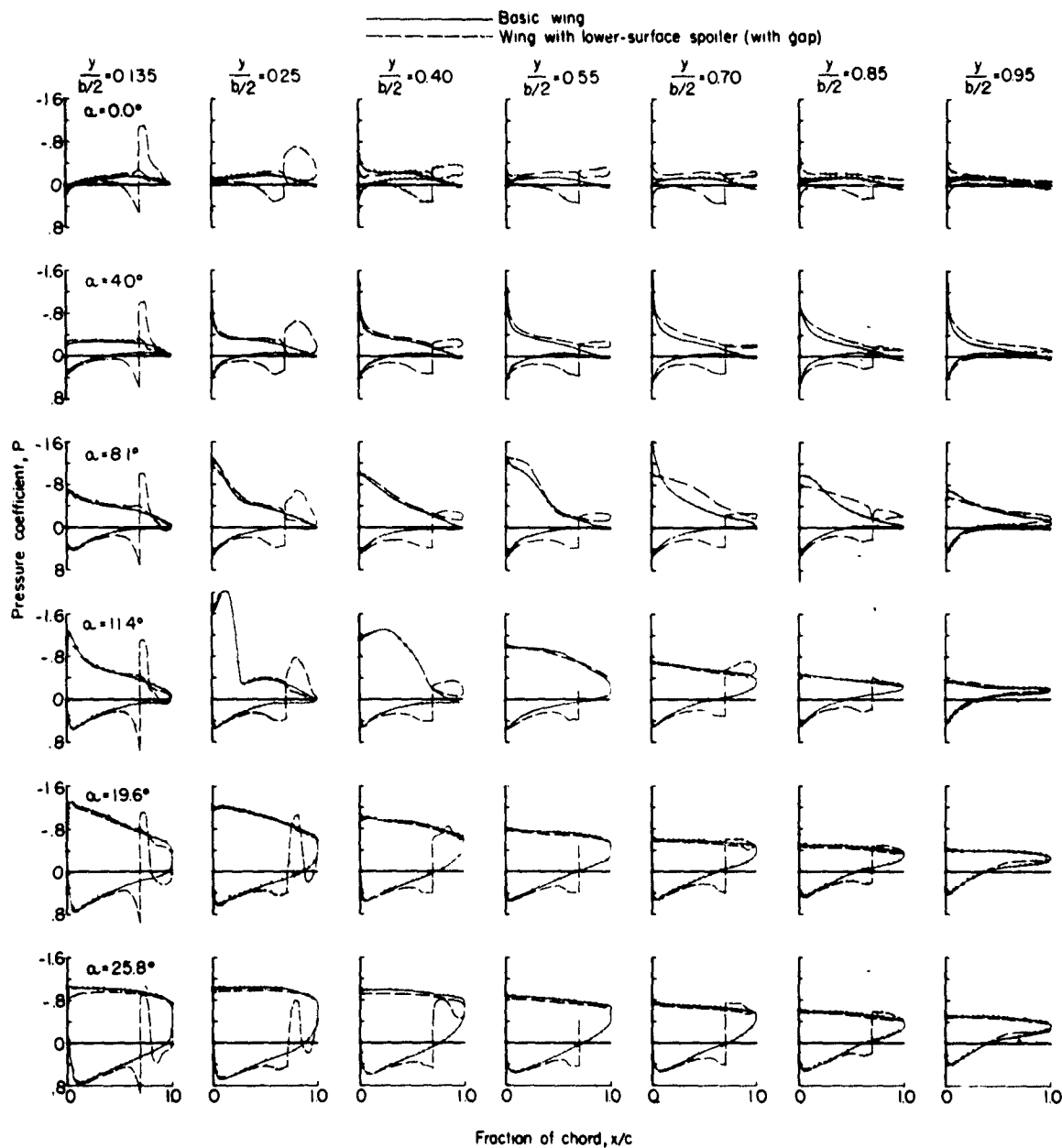
(a) $M = 0.60$.

Figure 14.- Chordwise pressure distributions on the wing; basic wing compared with the lower-surface spoiler (with gap) configuration.

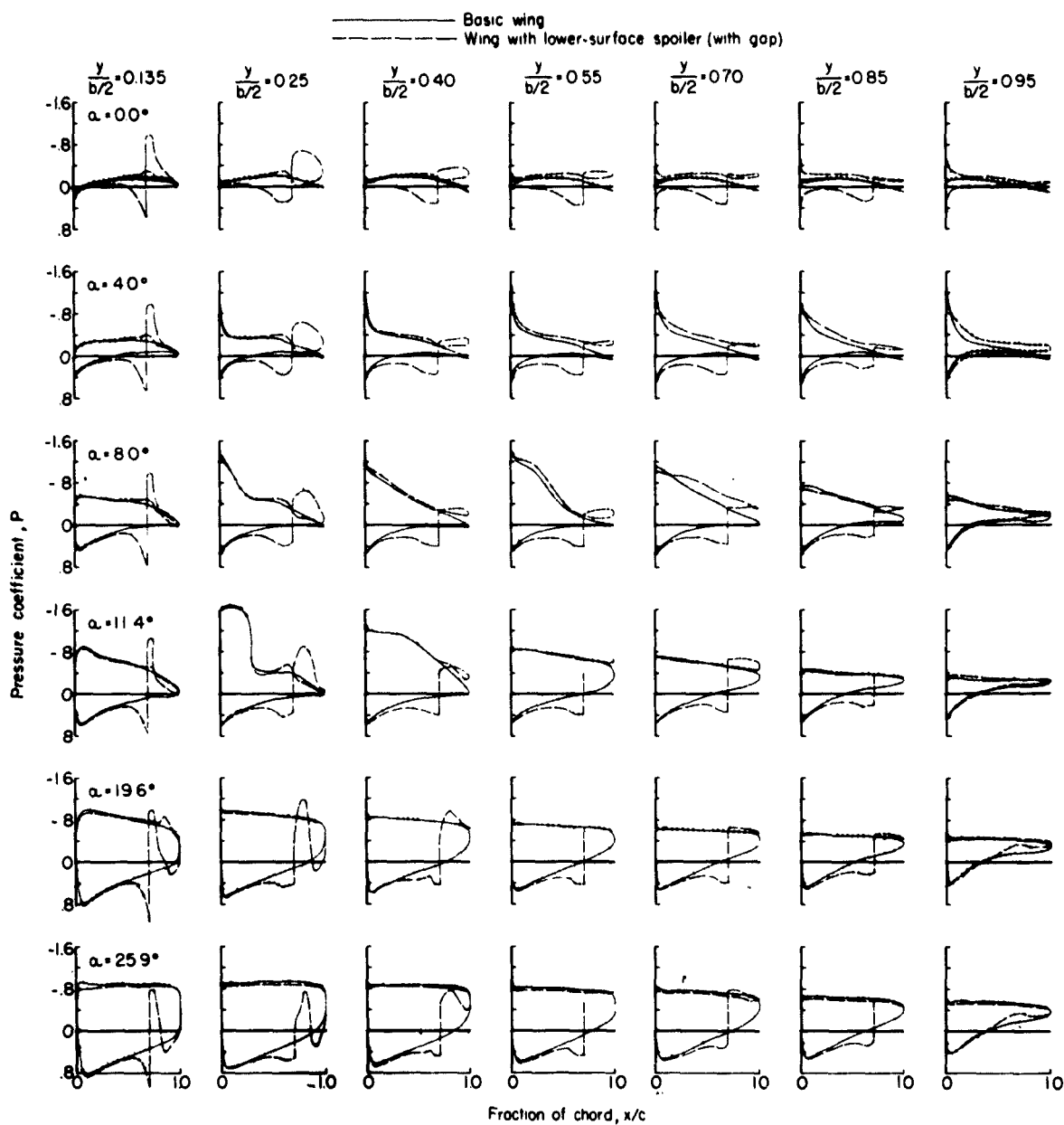
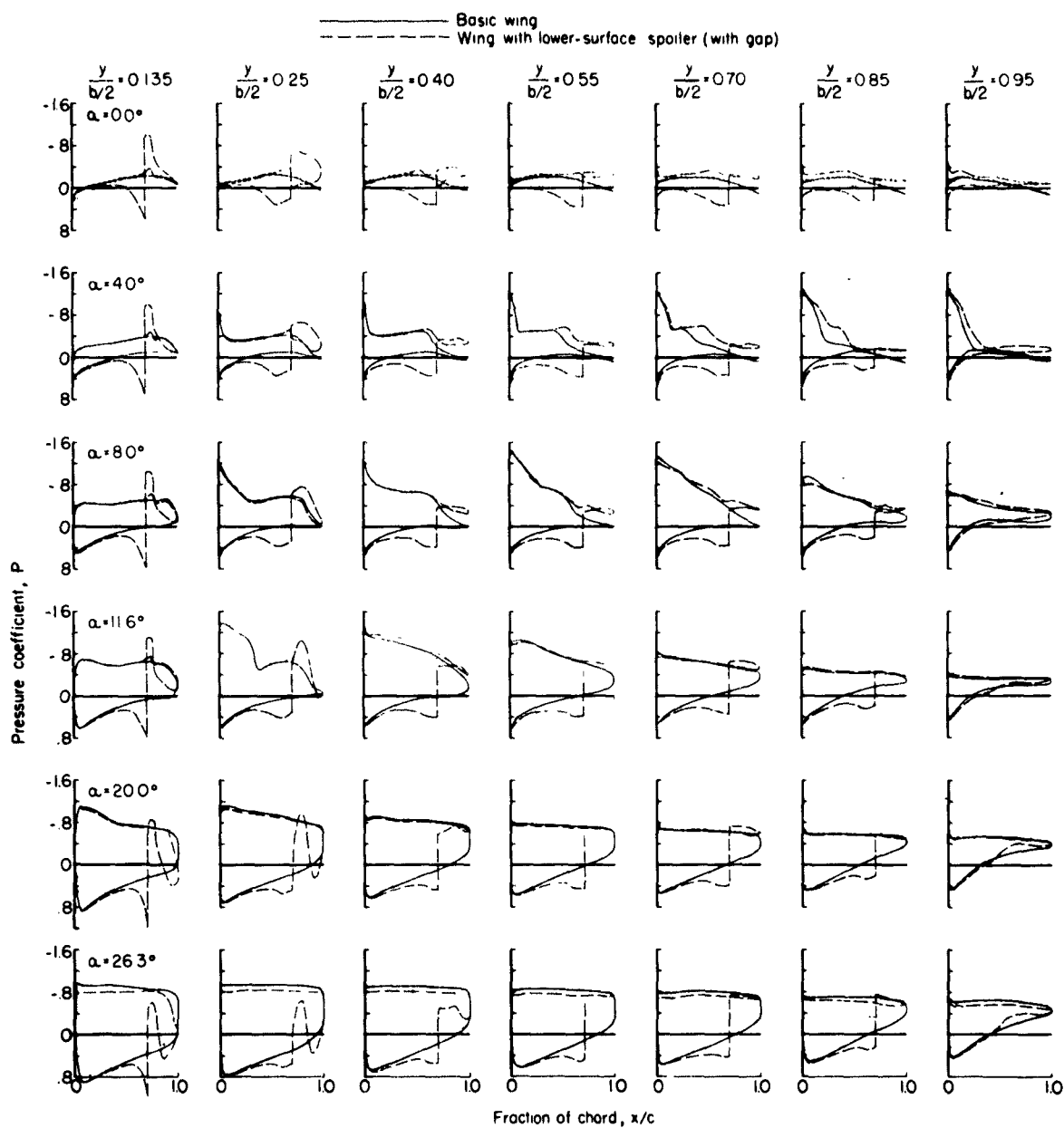
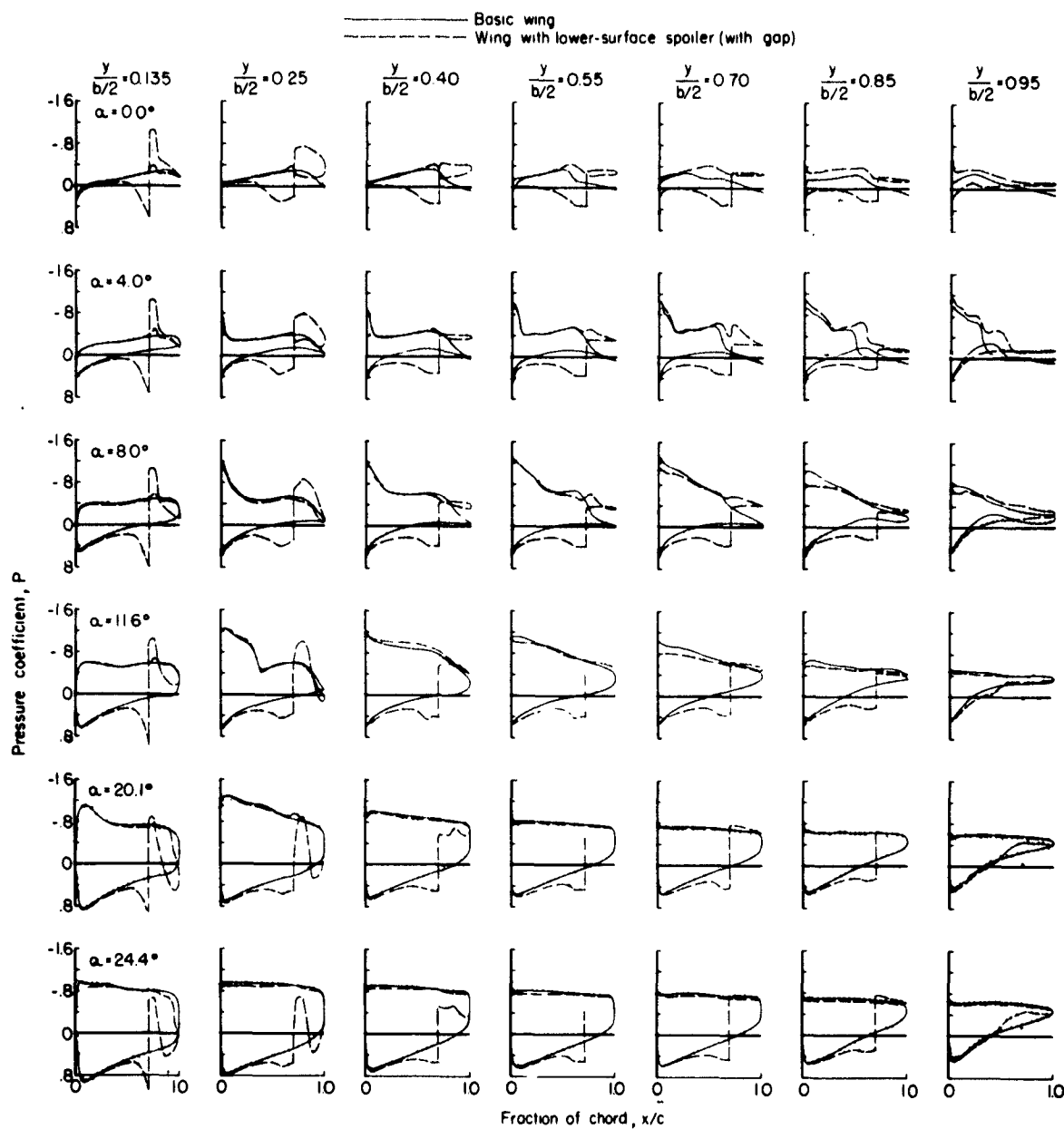
(b) $M = 0.80$.

Figure 14.- Continued.



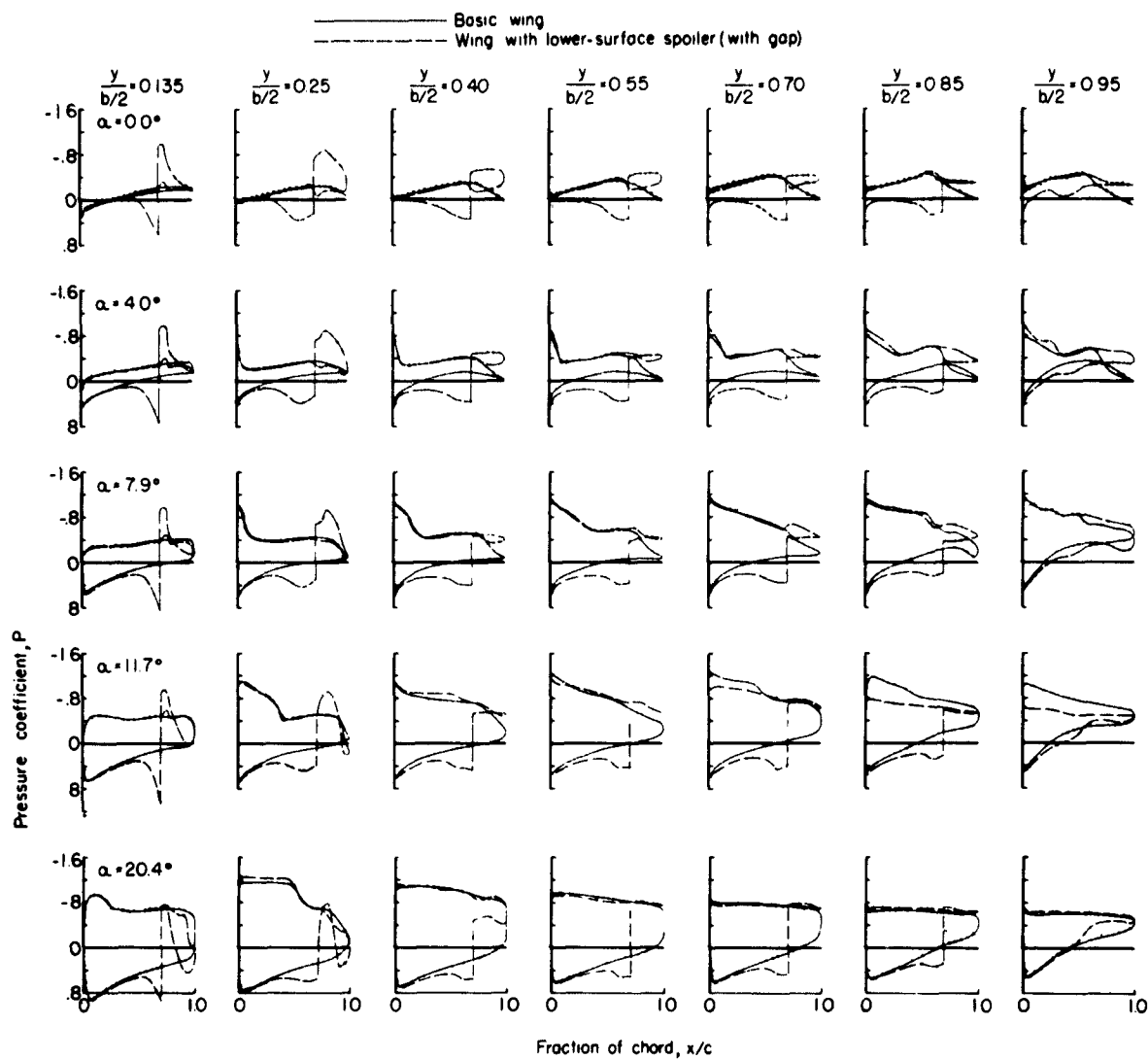
(c) $M = 0.90$.

Figure 14.- Continued.



(d) $M = 0.94$.

Figure 14.- Continued.



(e) $M = 1.00$.

Figure 14.- Concluded.

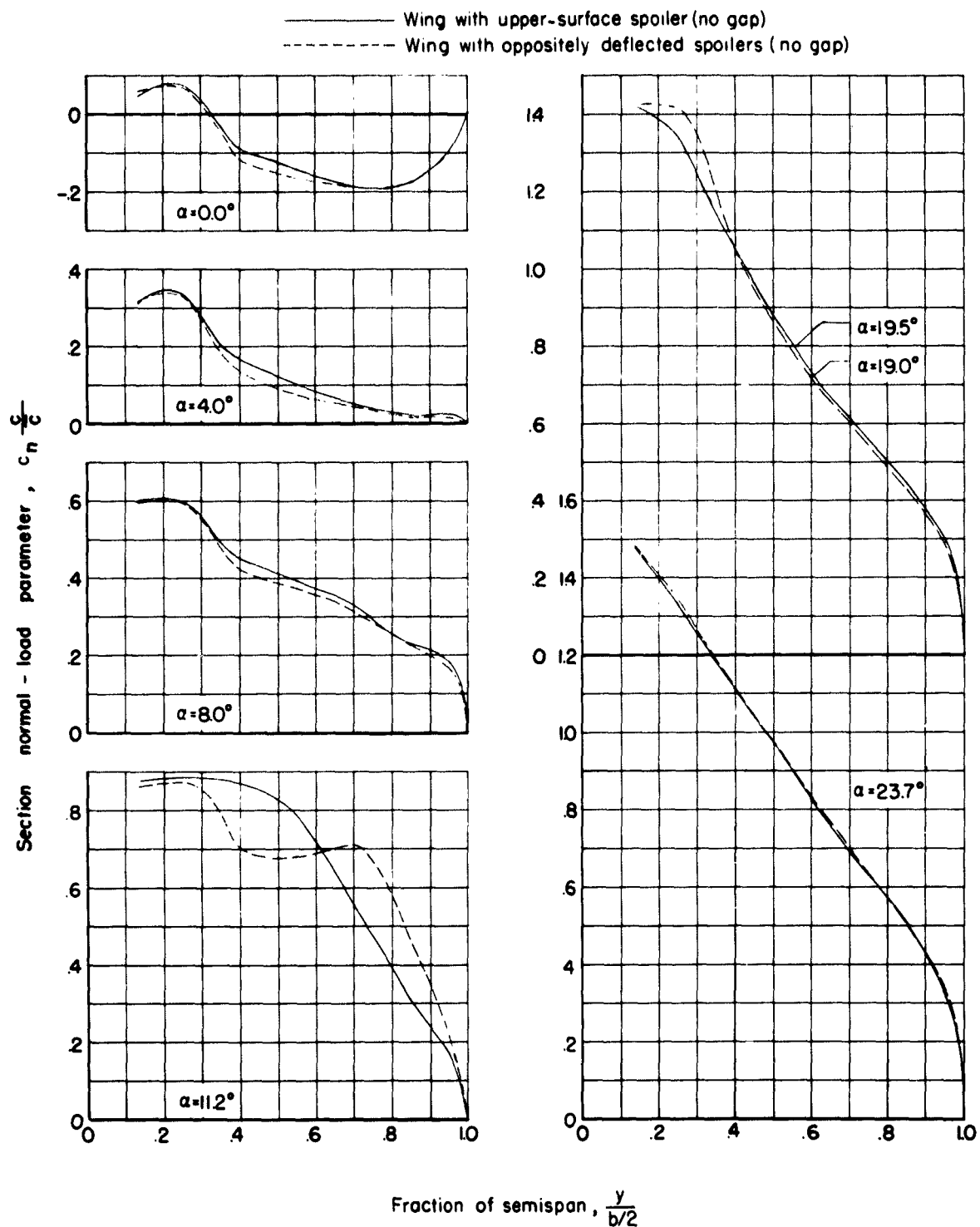


Figure 15.- Effect of a lower-surface spoiler on the opposite wing semi-span load distribution for the upper-surface spoiler (no gap) configuration.

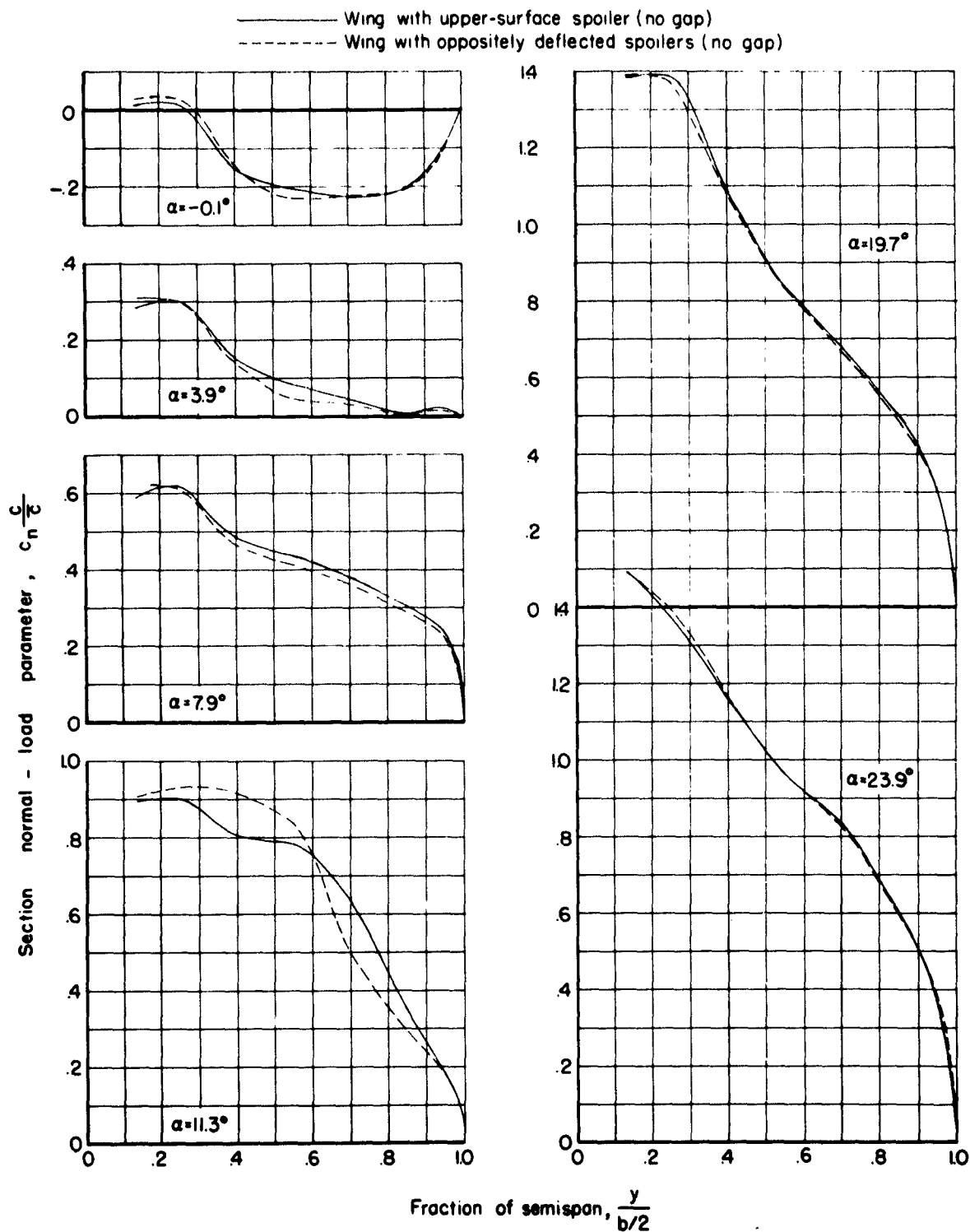
(b) $M = 0.90$.

Figure 15.- Continued.

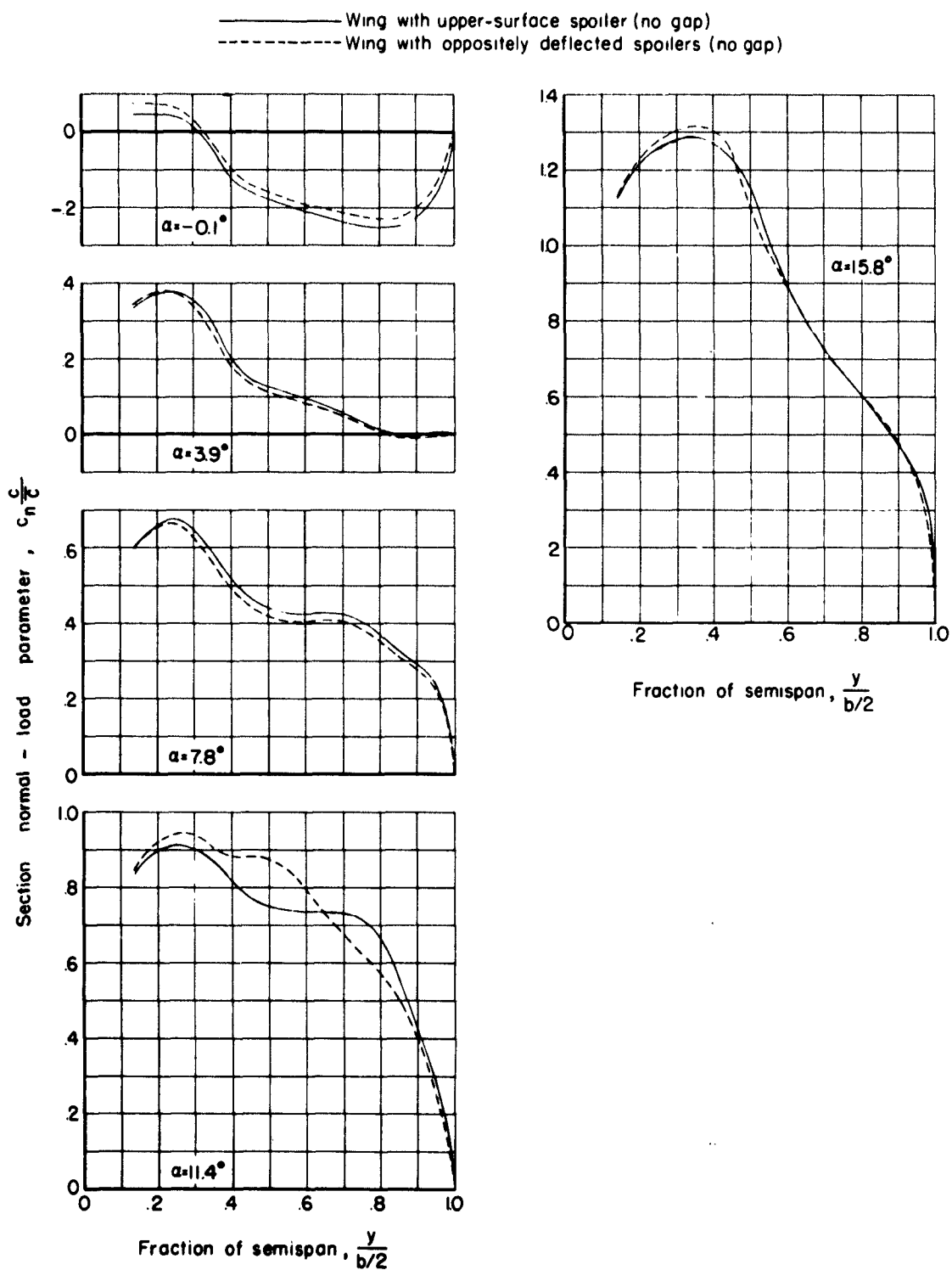
(c) $M = 0.98$.

Figure 15.- Concluded.

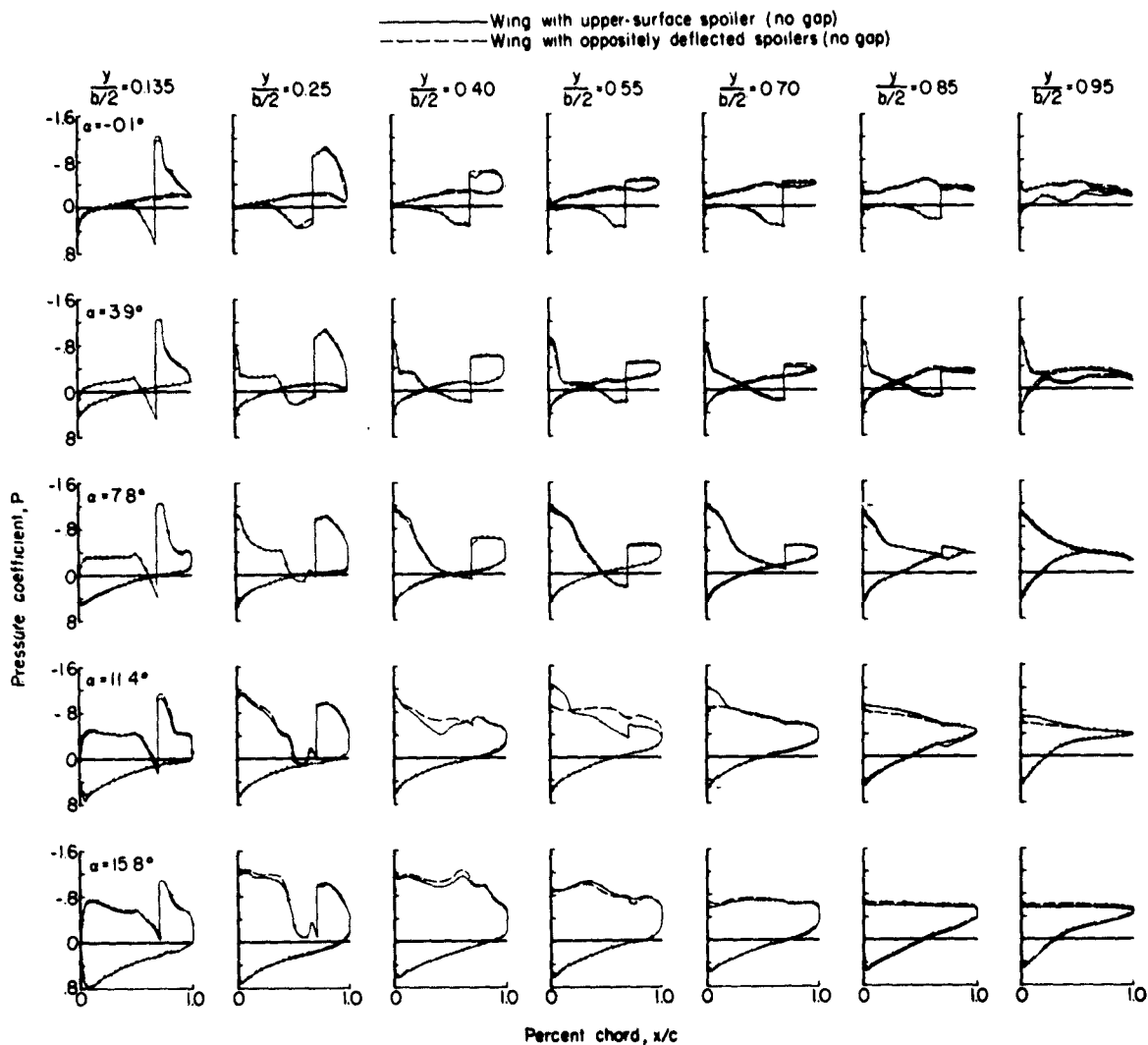
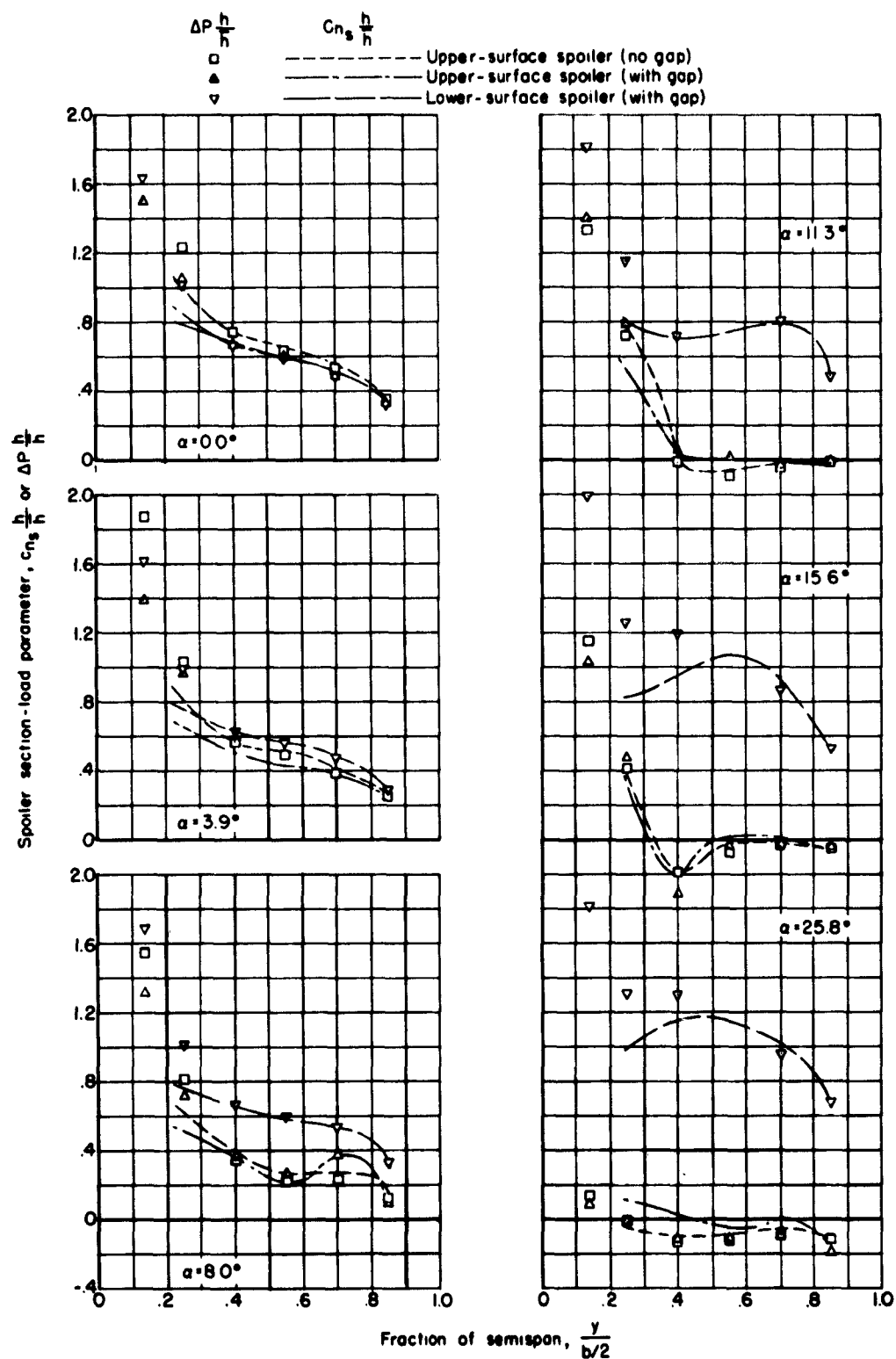


Figure 16.- Effect of a lower-surface spoiler on the opposite wing chord-wise pressure distributions for the upper-surface spoiler (no gap) configuration. $M = 0.98$.



(a) $M = 0.60$.

Figure 17.- Spoiler load distributions.

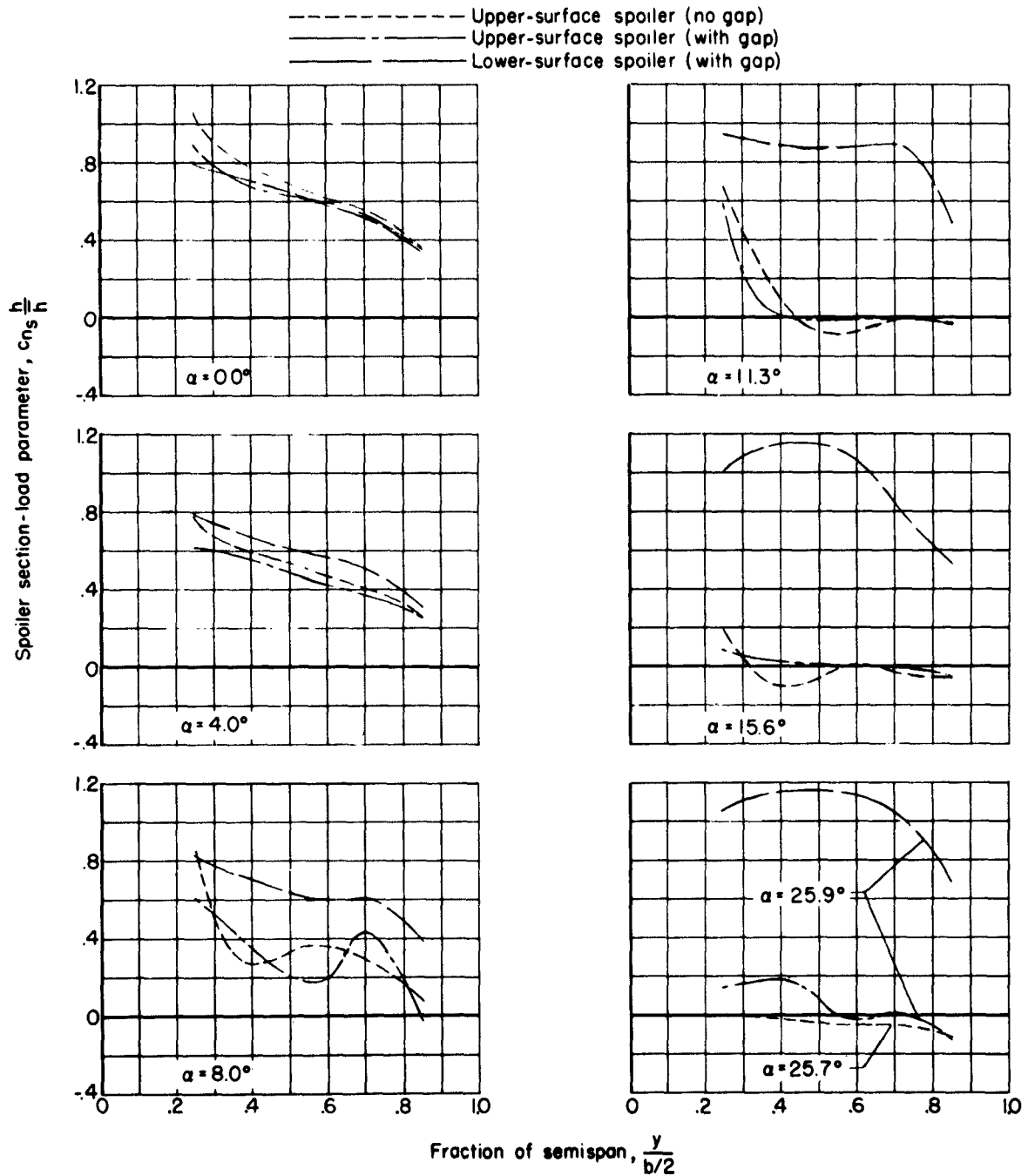
(b) $M = 0.80$.

Figure 17.- Continued.

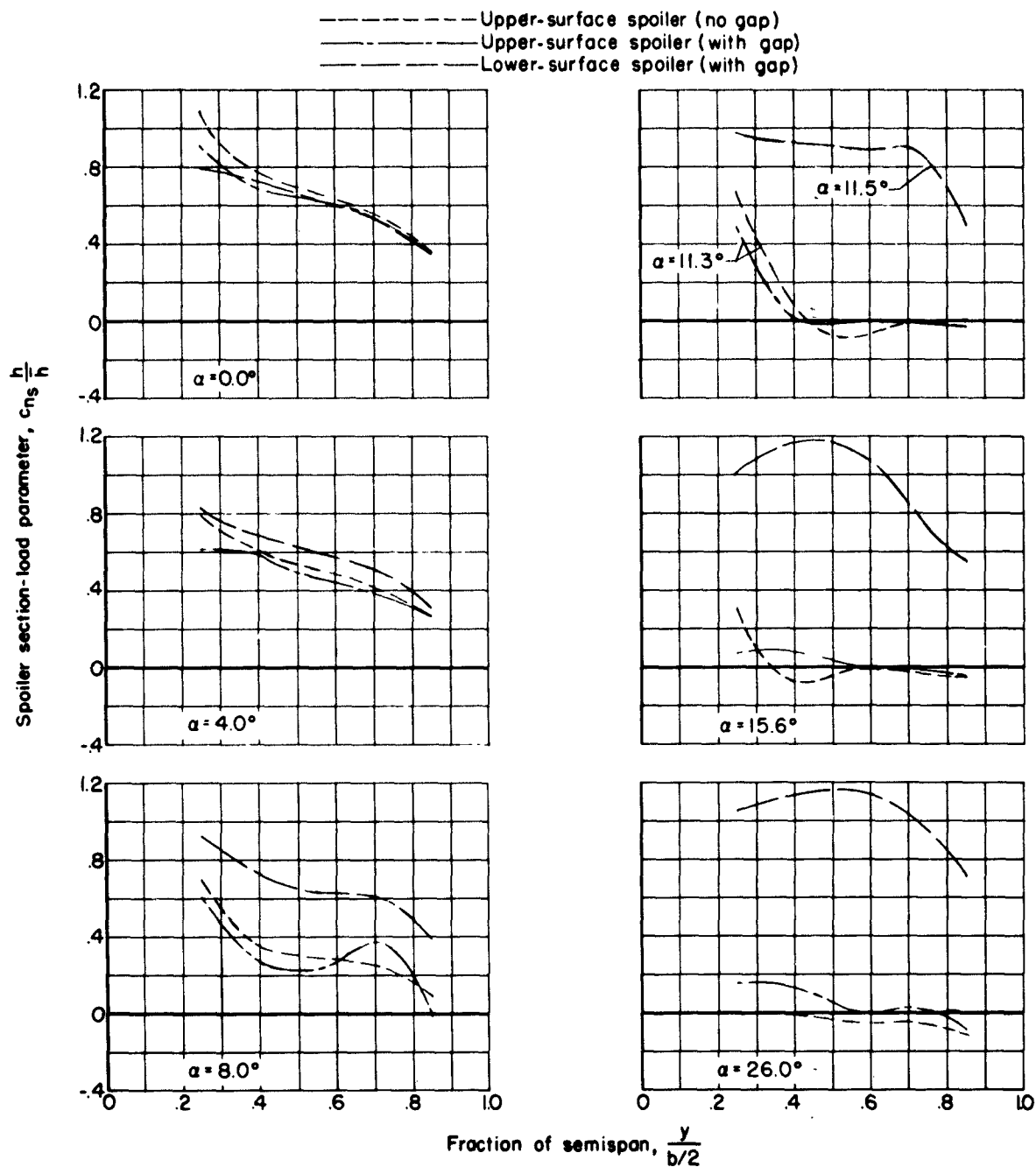
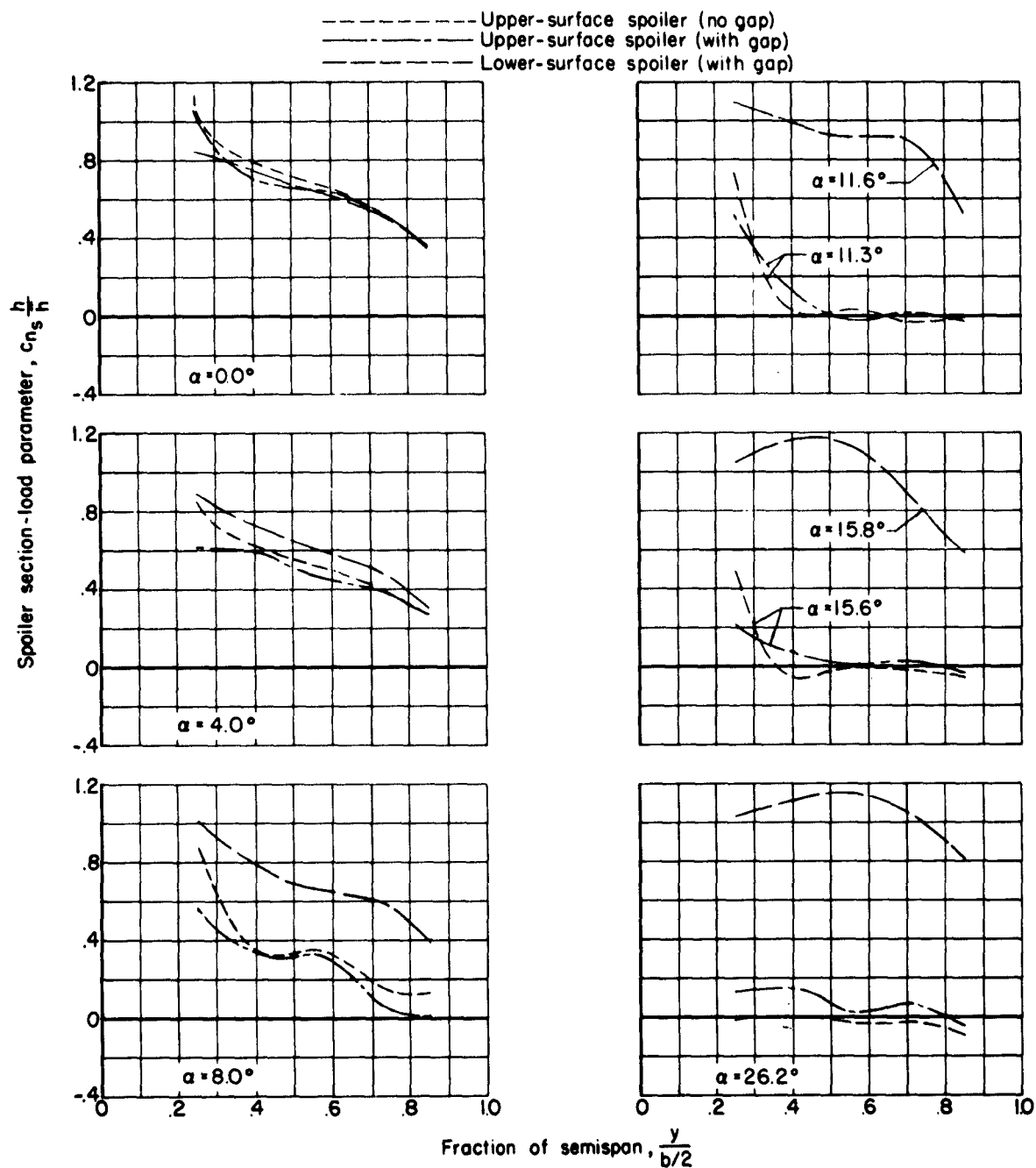
(c) $M = 0.85$.

Figure 17.- Continued.



(d) $M = 0.90$.

Figure 17.- Continued.

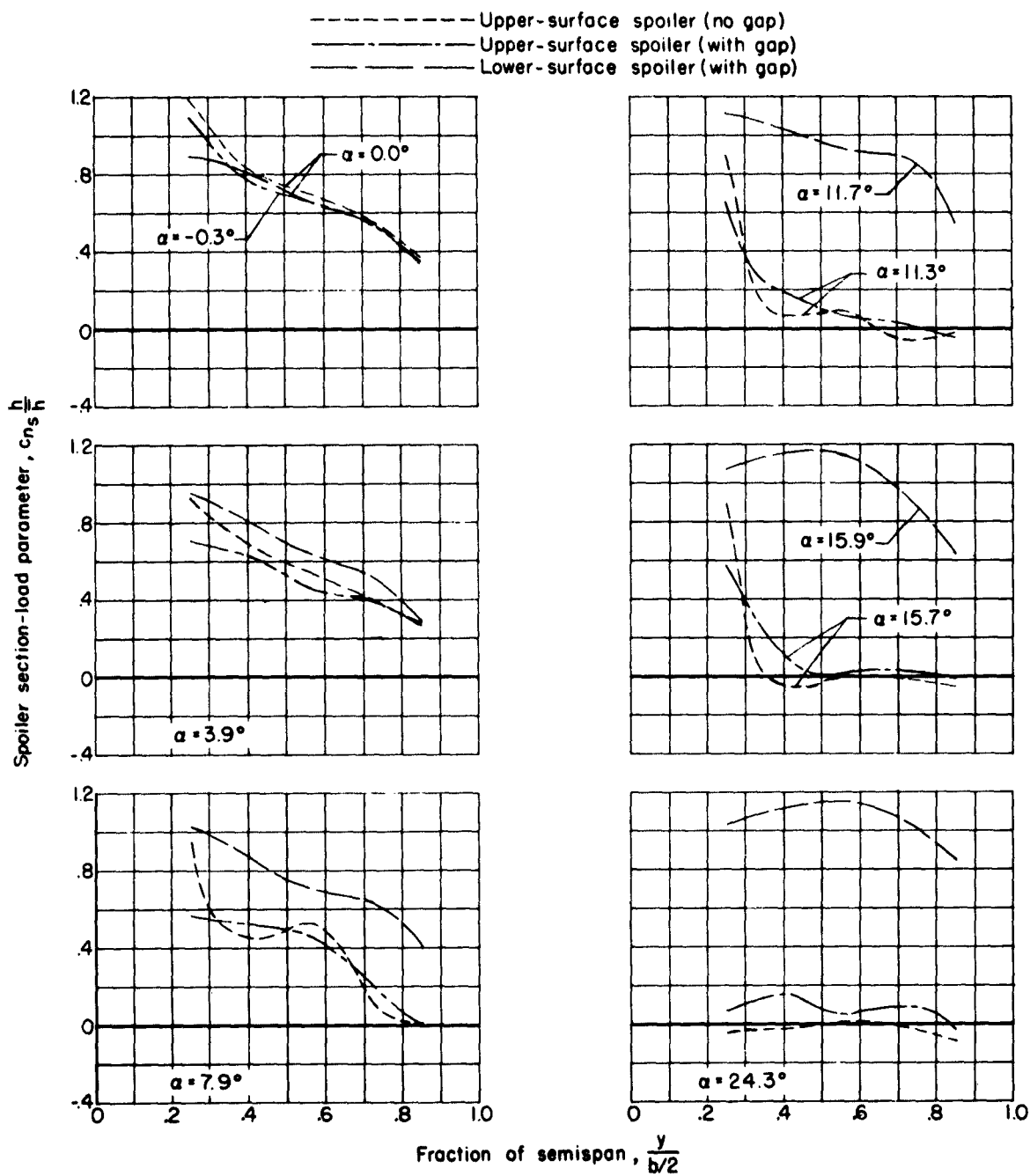


Figure 17.- Continued.

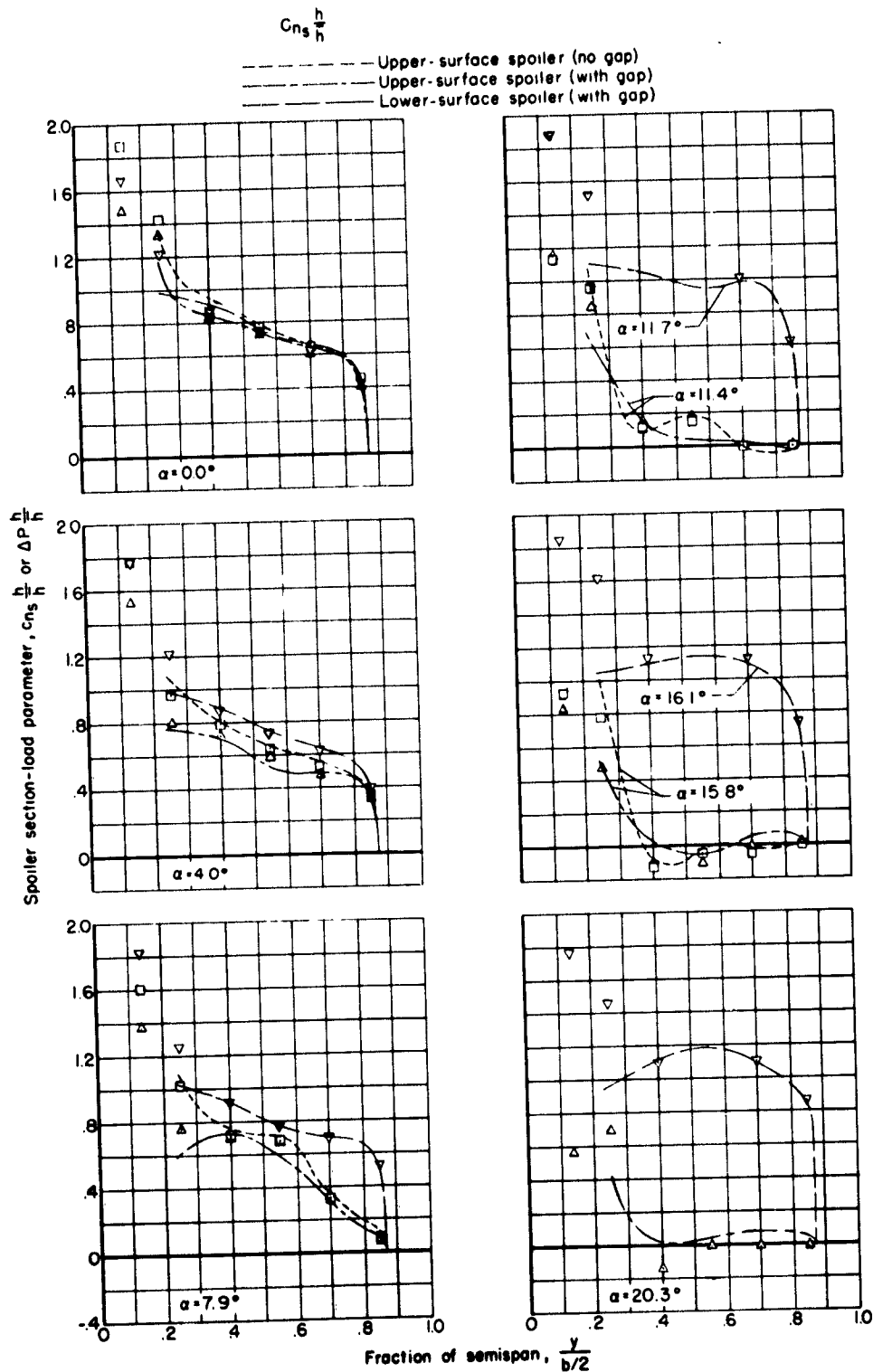
(f) $M = 0.98$.

Figure 17.- Continued.

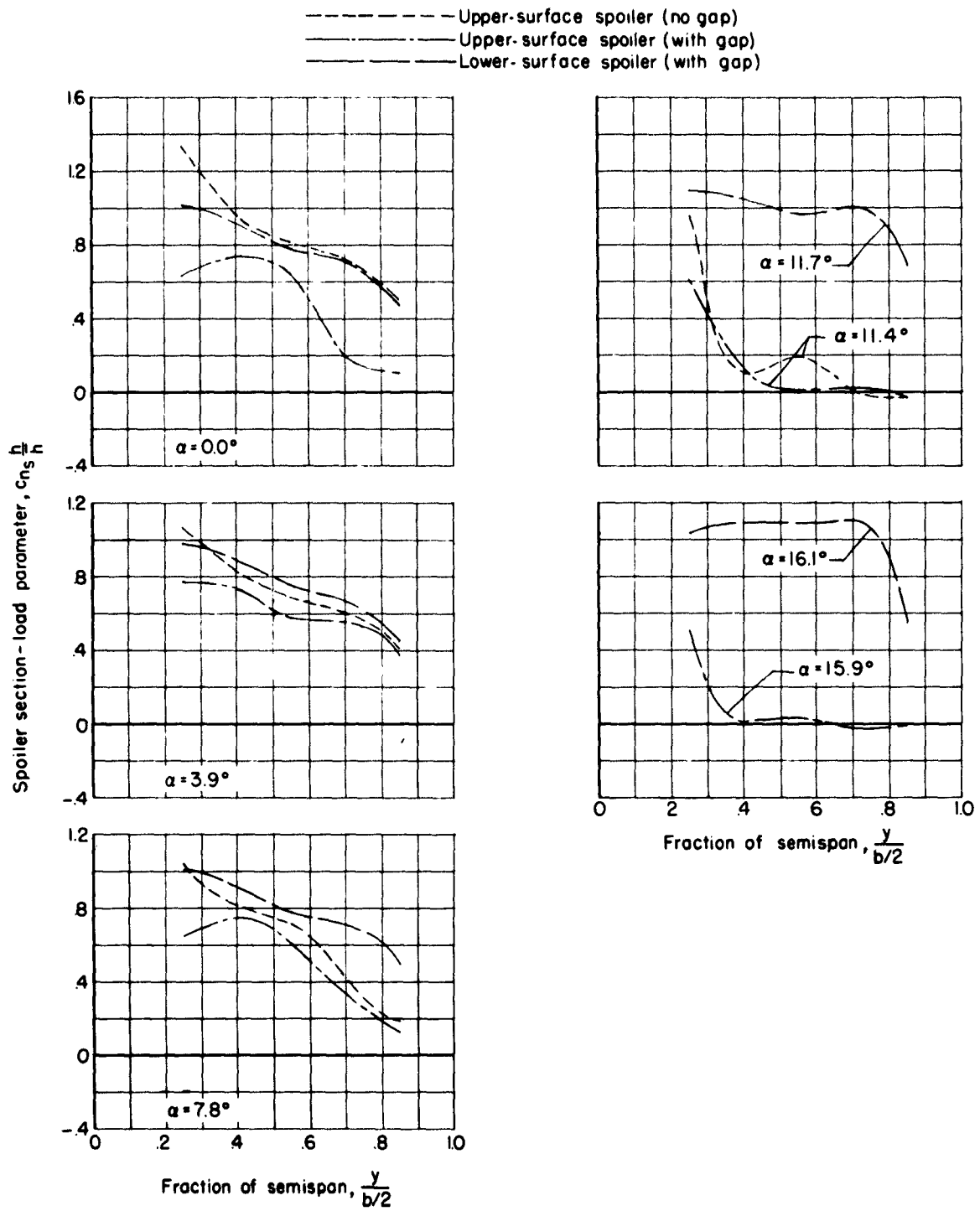
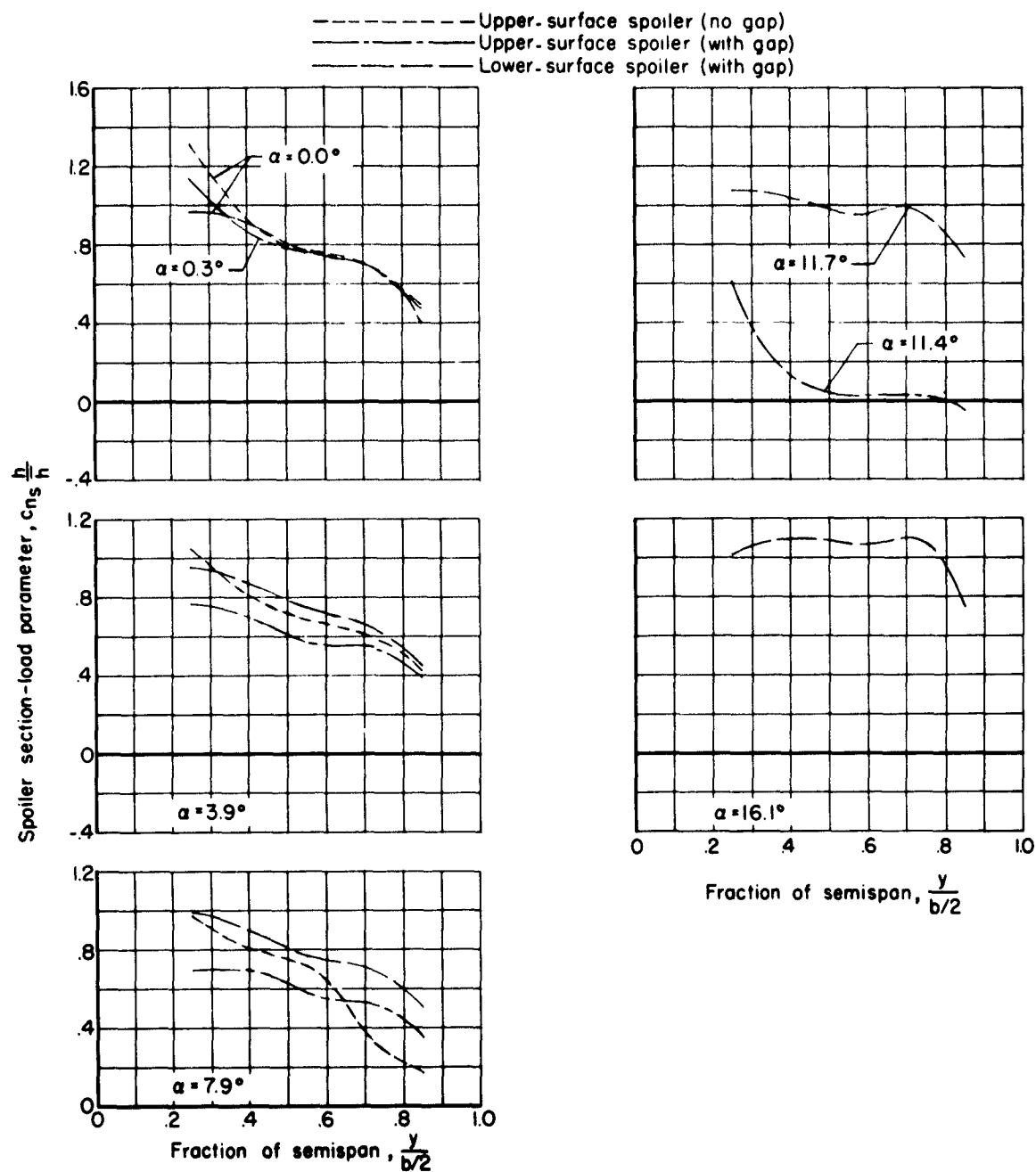
(g) $M = 1.00$.

Figure 17.- Continued.



(h) $M = 1.03$.

Figure 17.- Concluded.

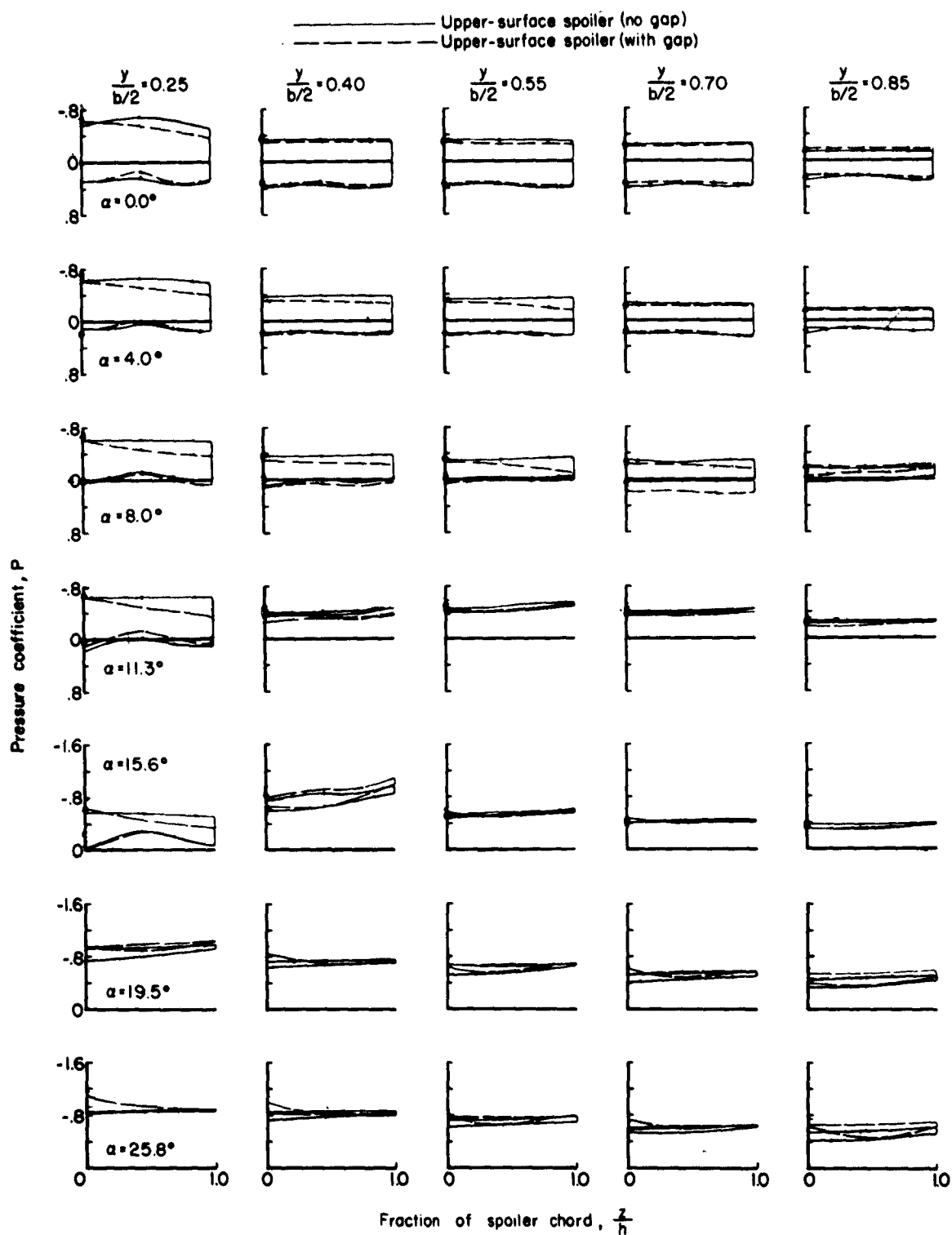
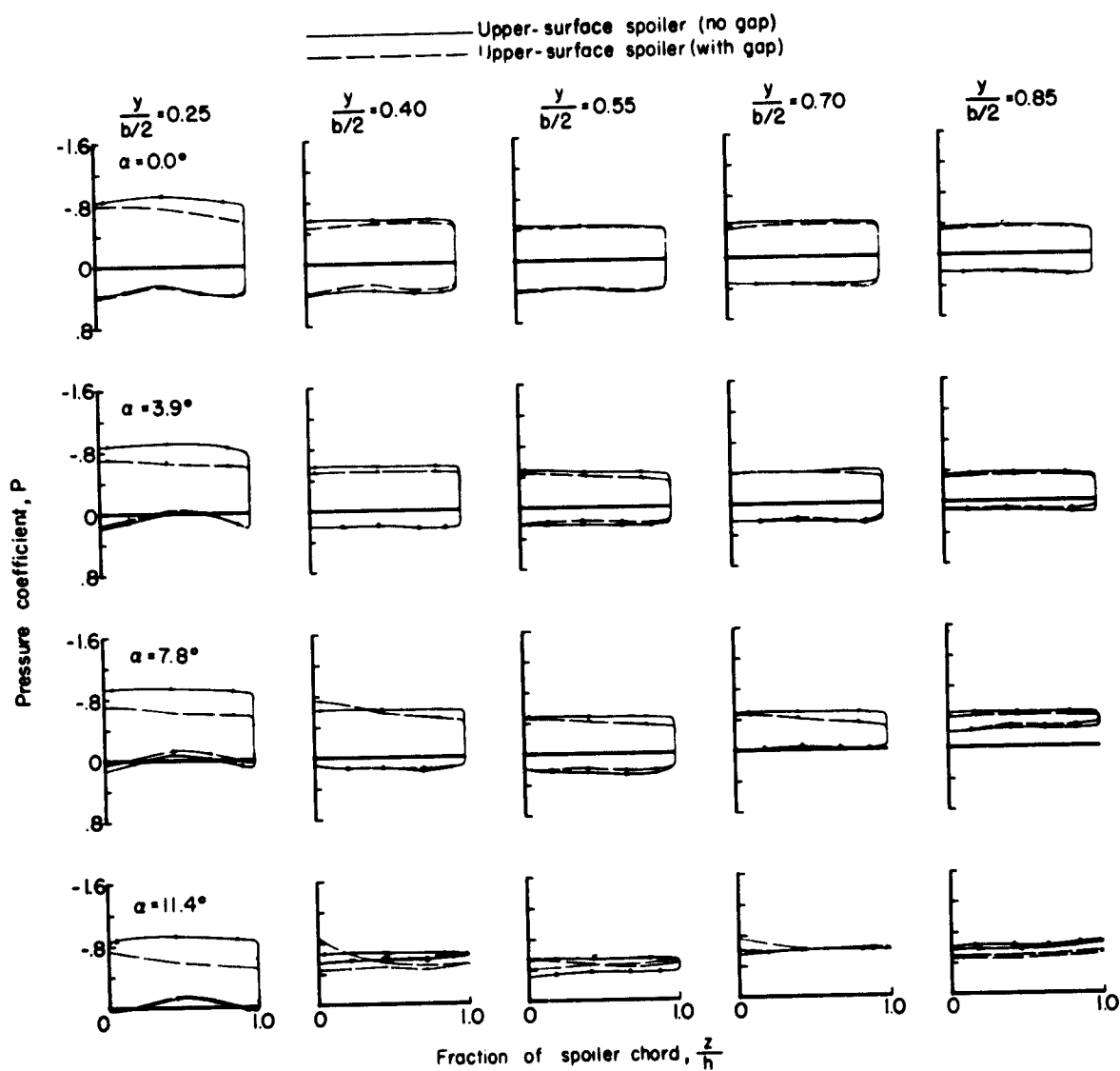
(a) $M = 0.60$.

Figure 18.- Section pressure distributions on the spoiler; upper-surface spoiler (with gap) compared with the upper-surface (no gap) configuration.



(b) $M = 1.00$.

Figure 18.- Concluded.

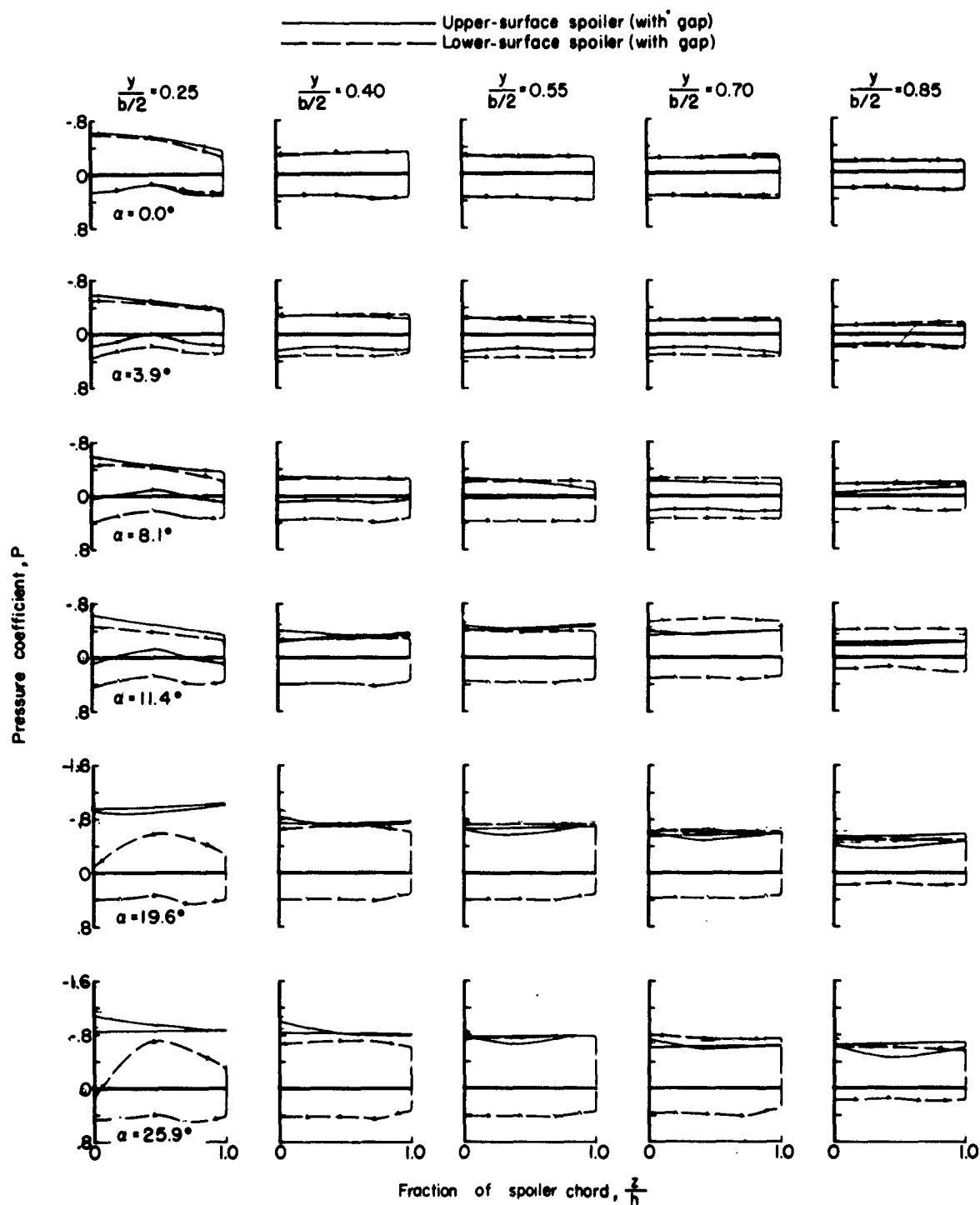


Figure 19.- Section pressure distributions on the spoiler; upper-surface spoiler (with gap) compared with the lower-surface spoiler (with gap) configuration.

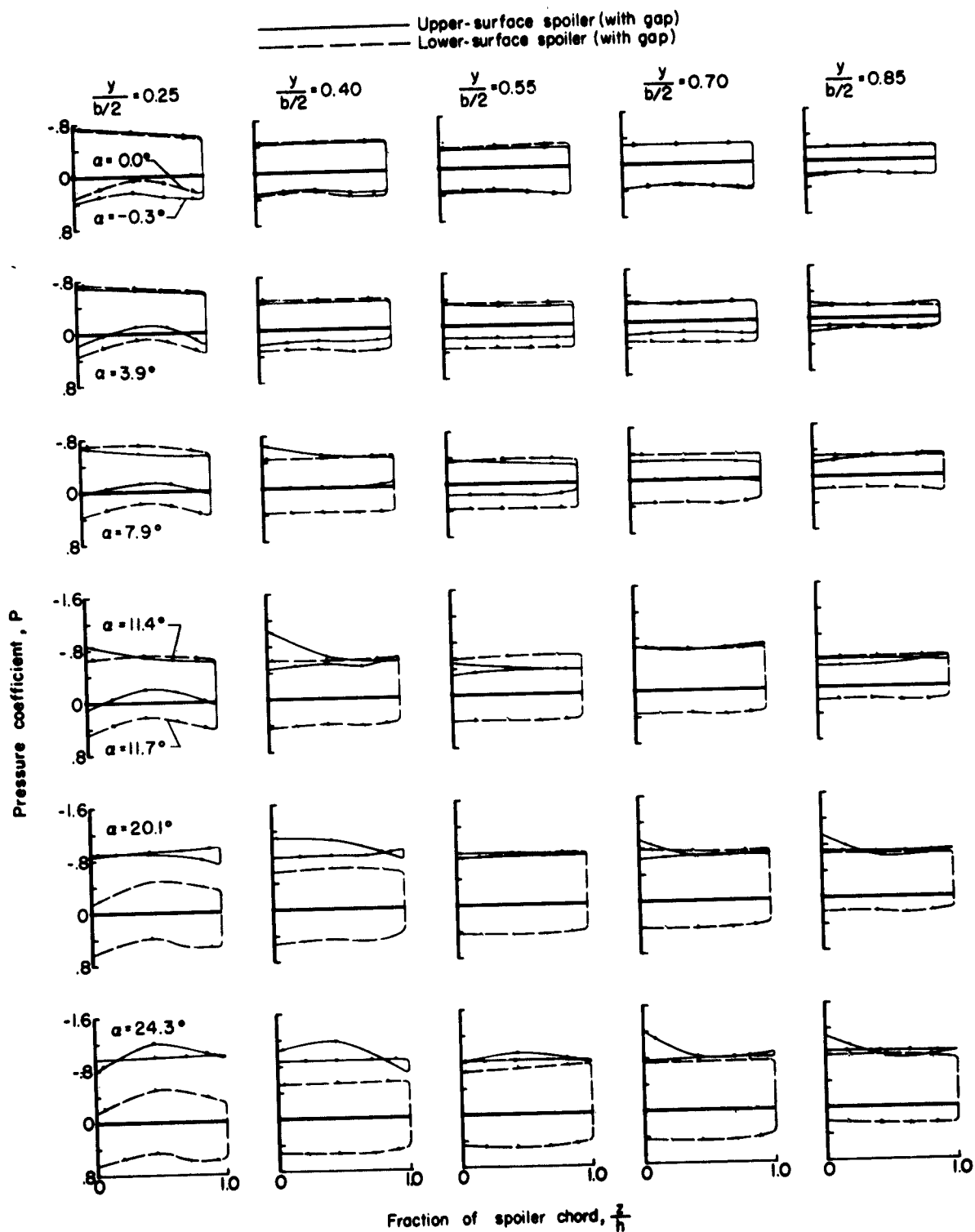
(b) $M = 0.94$.

Figure 19.- Concluded.

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Jr. and George Liner. May 1954. 162p. diagrs.,
4 tabs. (NACA RM L54C17a) CONFIDENTIAL

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airfoil sections parallel to the plane of symmetry.
Test configurations included upper- and lower-
surface spoilers, with and without a gap through the
wing behind the spoiler. Chordwise pressure-
distribution data were obtained at seven spanwise sta-
tions at Mach numbers from 0.60 (Reynolds number,

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1. Wings, Complete - Sweep (1.2.2.2.3)
2. Controls, Spoiler - Complete Wings (1.2.2.4.2)
3. Boundary-Layer Characteristics - Complete Wings (1.2.2.8.1)
4. Wing-Fuselage Combinations - Airplanes (1.7.1.1.1)
5. Control, Lateral (1.8.2.2)
6. Loads, Aerodynamic - Wings (4.1.1.1)
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5.1 x 10⁶) to 1.03 (Reynolds number, 6.2 x 10⁶) for an angle-of-attack range that usually extended to 20° or higher. The effects of upper- and lower-surface spoilers on chordwise pressure distributions, wing normal forces, wing span-load distributions, and positions of center of pressure are shown. Spoiler load distributions are also shown.



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- III. Liner, George
- IV. NACA RM L54C17a

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NACA RM L54C17a
National Advisory Committee for Aeronautics.
**EFFECTS OF SPOILER ALERONS ON THE AERO-
DYNAMIC LOAD DISTRIBUTION OVER A 450
SWEEPBACK WING AT MACH NUMBERS FROM
0.60 TO 1.03.** Joseph M. Hallissy, Jr., F. E. West,
Jr. and George Liner. May 1954. 162p. diagrs.,
4 tabs. (NACA RM L54C17a) **CONFIDENTIAL**

An investigation was conducted with 73-percent-
semispan inboard-spoiler allerons, having heights of
4 percent of the local wing chord and located along
the 70-percent wing-chord line of a 450 sweepback-
wing-fuselage combination. The wing had an aspect
ratio of 3.98, a taper ratio of 0.61, and NACA 65A006
airfoil sections parallel to the plane of symmetry.
Test configurations included upper- and lower-
surface spoilers, with and without a gap through the
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1. Wings, Complete - Sweep (1.2.2.2.3)
2. Controls, Spoiler - Complete Wings (1.2.2.4.2)
3. Boundary-Layer Characteristics - Complete Wings (1.2.2.8.1)
4. Wing-Fuselage Combinations - Airplanes (1.7.1.1.1)
5. Control, Lateral (1.8.2.2)
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5.1 x 10⁶) to 1.03 (Reynolds number, 6.2 x 10⁶) for an angle-of-attack range that usually extended to 20° or higher. The effects of upper- and lower-surface spoilers on chordwise pressure distributions, wing normal forces, wing span-load distributions, and positions of center of pressure are shown. Spoiler load distributions are also shown.



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